

AD624413

CONTRACT REQUIREMENTS	CONTRACT ITEM	MODEL	CONTRACT NO.	DATE
		<i>Model</i>	84454	

REPORT

NO. XAR-A-45

DATE Aug. 26, 1963

EXPERIMENTAL STUDY OF HIGH SPEED HYDROFOILS

VOLUME II

PART IV

CODE 26512

H. R. Wright Jr.
H. R. Wright
PREPARED BY:

G. J. Wennagel
G. J. Wennagel
APPROVED BY:

DDC
RECEIVED
DEC 9 1965
NAVISTV E
T. B. Street
T. B. Street
CHECKED BY:
J. Hubert
J. Hubert

REVISIONS

DATE	REV. BY	REVISIONS & ADDED PAGES	REMARKS

TABLE OF CONTENTS

Page No.

Summary	i
Introduction.	ii
Conclusions	iii
Discussion.	iv
List of Figures	v

SUMMARY

This test program was conducted to measure the hydrodynamic characteristics of a series of transcavitating, and supercavitating hydrofoil models with various planforms and sections to provide a design procedure to predict hydrodynamic characteristics for this type hydrofoil. Twenty hydrofoil models were tested in the Grumman Whirling Tank at three depths and at a sufficient number of speeds and angles of attack to fully define their hydrodynamic characteristics. Lift, drag, pitching moment, and strut base pressure were measured for all models, and cavity pressure and foil pressure were measured for the supercavitating models.

Classic supercavitating hydrofoil theory is shown to predict test results with an accuracy adequate for design if modified value for the cross-flow lift coefficient is accepted.

INTRODUCTION

The work under this contract consisted of testing twenty supercavitating and transcavitating hydrofoil models in the Grumman Whirling Tank at three depths and at a sufficient number of speeds and angles of attack to fully define their hydrodynamic characteristics. The models include aspect ratios of 1.0 to 5.0, sweep angles of 0 to 56.3°, taper ratios of 0 to 1.0, design lift coefficients of 0 to .350, and were tested at cavitation numbers of .029 to .145. All models were mounted on a single 15% thick, blunt base parabolic strut except the aspect ratio = 5 model which was mounted on a twin strut combination

Lift, drag, pitching moment, and strut base pressure were measured for all models, and cavity pressure and foil base pressure were measured for the supercavitating models. The foil cavity pressure was measured by means of a static pressure tap centrally located on the upper surface of one semi-span.

Analysis of the test results was limited to the supercavitating lift and drag data; the transition from fully wetted to supercavitating did not cover significant angle of attack range on any foil tested. Transition from wetted to cavitated flow on these foils was characteristically accompanied by a very high frequency vibration audible as a loud screech. A study of this "singing" phenomenon did not conclusively identify the source but gave some indication that it was associated with a stall flutter type of instability.

The work of this contract has been submitted in three volumes. Volume I presents the measured results in tabular form, the correlation of results, the structural analysis, and an analysis of the "singing" phenomenon encountered during the tests. This volume presents the measured results in graphical form. Volume II summarizes the test results as a procedure for the prediction of the lift and drag for a supercavitating hydrofoil of arbitrary section and planform with a numerical example. Volume III is written as a complete summary report suited to wide distribution.

CONCLUSIONS

Classic supercavitating hydrofoil theory provides adequate correlation for these results if a modified value for the cross-flow lift coefficient is accepted. With the modified cross-flow lift coefficient, the correlation of the fifty-nine measured lift and drag curves with classic supercavitating hydrofoil theory may be summarized statistically as a prediction reliability (+ error indicates measurement higher than predicted):

	<u>Mean Error</u>	<u>Probable Error</u>
Lift Coefficient	-.002	±.012
Drag Coefficient	.0012	±.0028

DISCUSSION

The measured data is presented in graphical form in this section as plots of lift coefficient, drag coefficient, and pitching moment coefficient. The speeds specified are nominal. The actual speeds varied by about 5%, however, all data reduction was based upon the actual measured dynamic pressure.

The specified submergences are in mean geometric chords. The drag coefficients include strut drag. Moment coefficients are referenced to the quarter-chord of the mean geometric chord. The measured ventilated coefficients have not been corrected for cavity pressure and generally correspond to atmospheric cavity pressure.

FIG. C. JULY

Figures

<u>Foil No.</u>	<u>Speed Knots</u>	<u>Fig. No. & Page No.</u>	<u>Foil No.</u>	<u>Speed Knots</u>	<u>Fig. No. & Page No.</u>
1	40	IV.1	7	40	IV.35
	45	IV.2		45	IV.36
	50	IV.3		50	IV.37
	60	IV.4		60	IV.38
	70	IV.5		70	IV.39
	80	IV.6			
2	40	IV.7	8	40	IV.40
	45	IV.8		45	IV.41
	50	IV.9		50	IV.42
	60	IV.10		60	IV.43
	70	IV.11		70	IV.44
				80	IV.45
3	40	IV.12	9	40	IV.46
	45	IV.13		45	IV.47
	50	IV.14		50	IV.48
	60	IV.15		60	IV.49
	70	IV.16		70	IV.50
	80	IV.17		80	IV.51
4	40	IV.18	10	40	IV.52
	45	IV.19		45	IV.53
	50	IV.20	11	40	IV.54
	60	IV.21		45	IV.55
	70	IV.22		50	IV.56
	80	IV.23		60	IV.57
5	40	IV.24		70	IV.58
	45	IV.25		80	IV.59
	50	IV.26	12	40	IV.60
	60	IV.27		45	IV.61
	70	IV.28		50	IV.62
	80	IV.29		60	IV.63
6	40	IV.30		70	IV.64
	45	IV.31		80	IV.65
	50	IV.32			
	60	IV.33			
	70	IV.34			

LIST OF FIGURES (Cont'd.)

<u>Foil No.</u>	<u>Speed Knots</u>	<u>Fig. No. & Page No.</u>	<u>Foil No.</u>	<u>Speed Knots</u>	<u>Fig. No. & Page No.</u>
13	40	IV.66	17	40	IV.90
	45	IV.67		45	IV.91
	50	IV.68		50	IV.92
	60	IV.69		60	IV.93
	70	IV.70		70	IV.94
	80	IV.71		80	IV.95
14	40	IV.72	18	90	IV.96
	45	IV.73		96	IV.97
	50	IV.74		40	IV.98
	60	IV.75		50	IV.99
	70	IV.76		60	IV.100
	80	IV.77		70	IV.101
15	40	IV.78	19	80	IV.102
	45	IV.79		90	IV.103
	50	IV.80		95	IV.104
	60	IV.81		40	IV.105
	70	IV.82		50	IV.106
	80	IV.83		60	IV.107
16	40	IV.84	20	70	IV.108
	45	IV.85		80	IV.109
	50	IV.86		90	IV.110
	60	IV.87		96	IV.111
	70	IV.88		40	IV.112
	80	IV.89		50	IV.113
				60	IV.114
				70	IV.115
				80	IV.116
				90	IV.117
				95	IV.118

LIST OF FIGURES (Cont'd.)Flapped Foil Tests

<u>Foil No.</u>	<u>Speed Knots</u>	<u>d/c</u>	<u>Fig. No. & Page No.</u>
13	40	.75	IV.119
		1.00	IV.120
		1.25	IV.121
	45	.75	IV.122
		1.00	IV.123
		1.25	IV.124
	50	.75	IV.125
		1.00	IV.126
		1.25	IV.127
	60	.75	IV.128
		1.00	IV.129
		1.25	IV.130
	70	.75	IV.131
		1.00	IV.132
		1.25	IV.133
	80	.75	IV.134
		1.00	IV.135
		1.25	IV.136
14	40	1.00	IV.137
	45		IV.138
	50		IV.139
	60		IV.140
	70		IV.141
	80		IV.142
15	40	1.00	IV.143
	45		IV.144
	50		IV.145
	60		IV.146
	70		IV.147
	80		IV.148

K.E. 70 100 120

FROUDE NO.	SYMBOL	L/C	$C_{D,TRUST}$	CAV. NO.
6.40	○	.75	.0032	491
5.56	□	1.00	.0042	505
5.00	◇	1.25	.0056	520

LIFT COEFFICIENT C_L

.02

.04

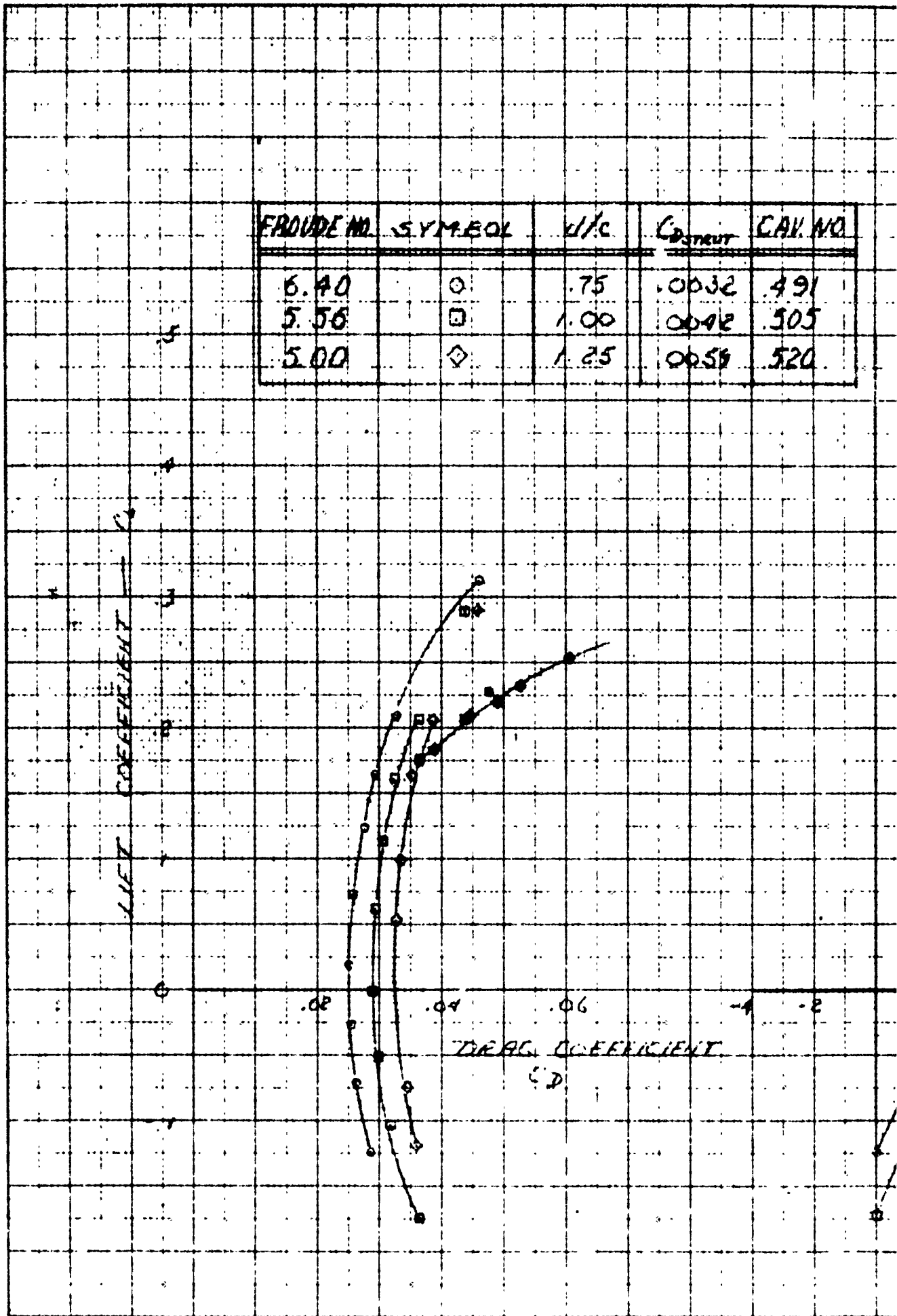
.06

.08

.10

DRAG COEFFICIENT

C_D



WHIRLING TRAIL TEST RESULTS

NO. 13 84454

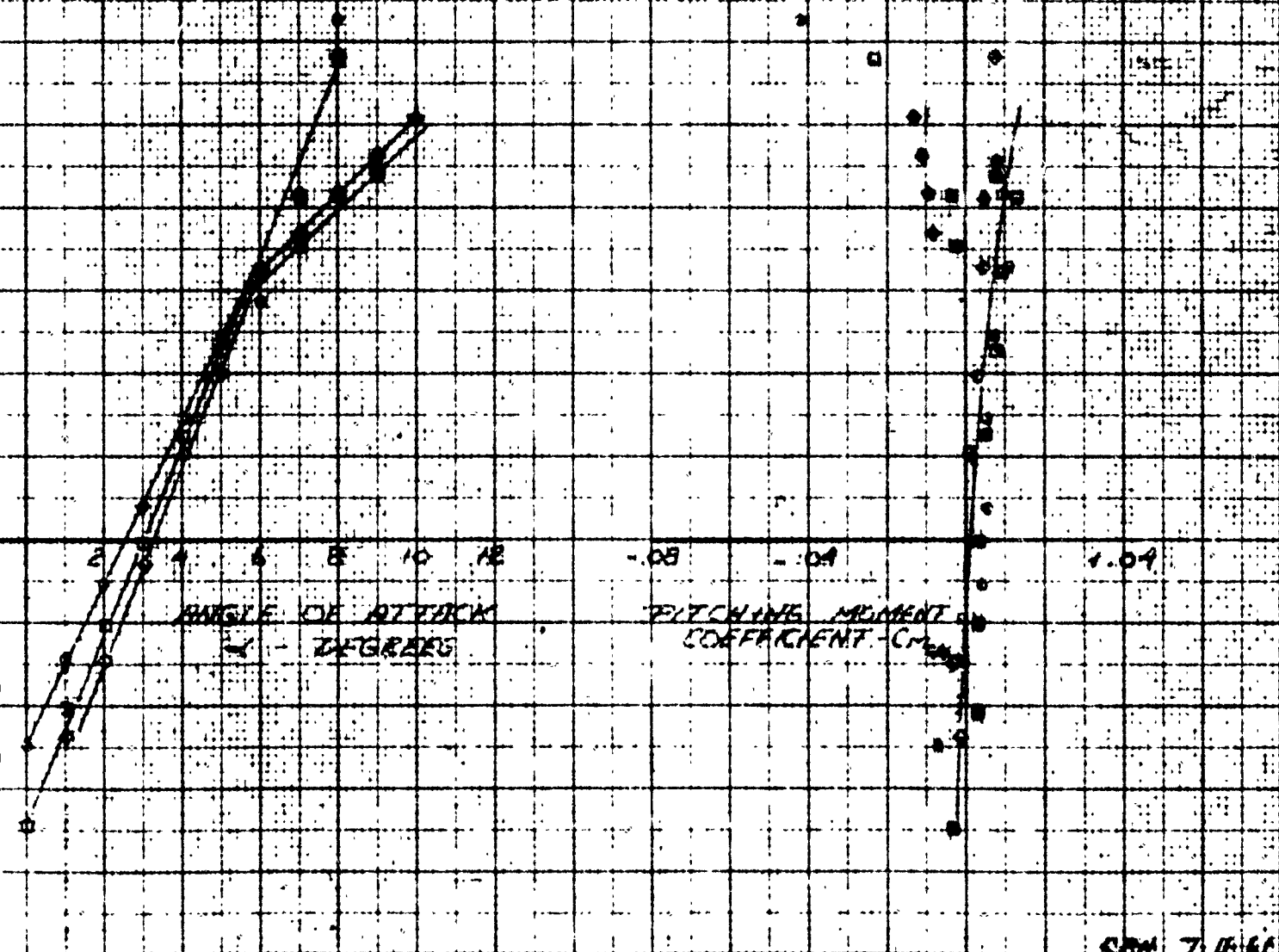
MODEL NO. 1

40 KTS

MAC = 1.39 FT.

NOTE

1. DARKENED SYMBOLS
INDICATE VENTED FLOW



LIFT COEFFICIENT C_L

PROVIDE NO	CAN NO	SYMBOL	C_L/C_D	C_{Lmax}
8.40	401	⊙	.75	.0529
5.56	412	⊠	1.00	.0039
5.00	415	◇	1.25	.0055

DRAG COEFFICIENT C_D

.02 .04 .06 .8 2

WHIRLING TANK TEST NO. 20

NO. 63 94958

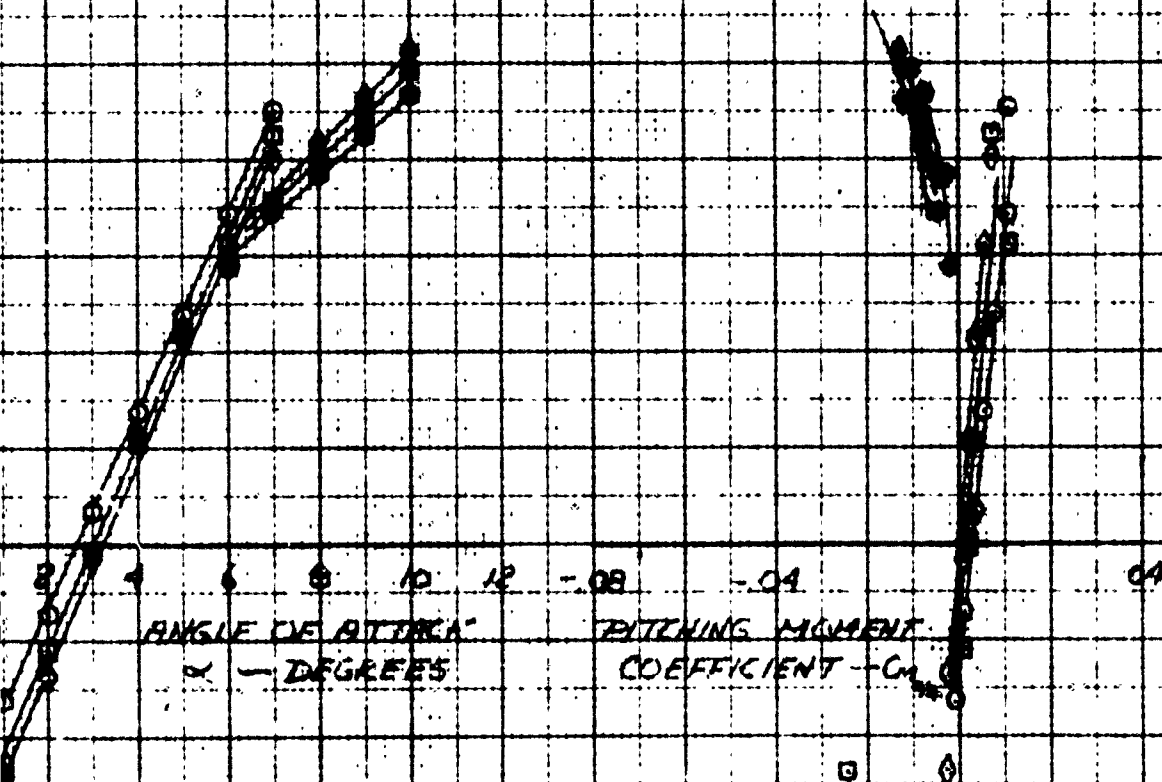
MODEL NO. 1

15 FT

MAR = 139 FT

NOTE:

1. DARKENED SYMBOLS
INDICATE VENTED FLOW



PROUDE NO	CPH NO	SYMBOL	U/C	C_{Dmax}
640	320	○	.75	0028
358	335	□	1.00	0039
500	350	◇	1.25	0055

WET COEFFICIENT

DRAG COEFFICIENT

148 200000 100 200

PAGE IV.3

WHIRLING TANK TEST NO. 2

NO. 63 84454

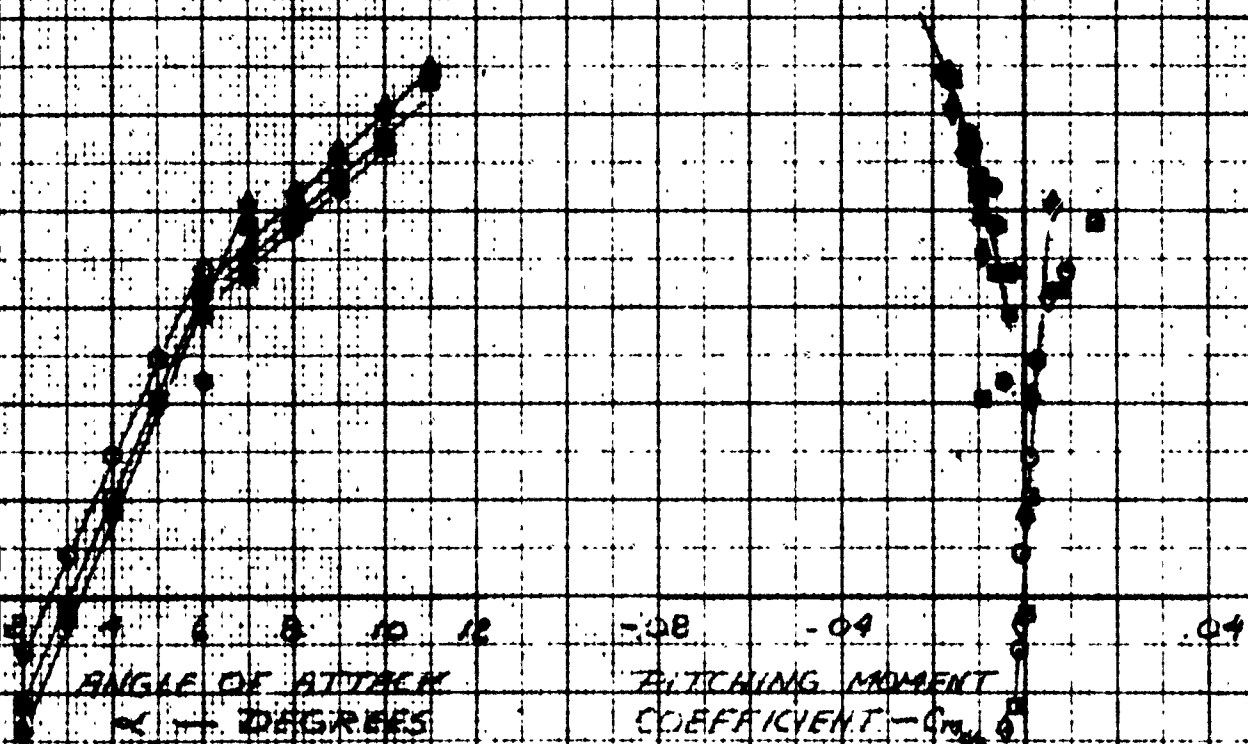
MODEL NO. 1

50 KTS

MAC = 139 FT

NOTE:

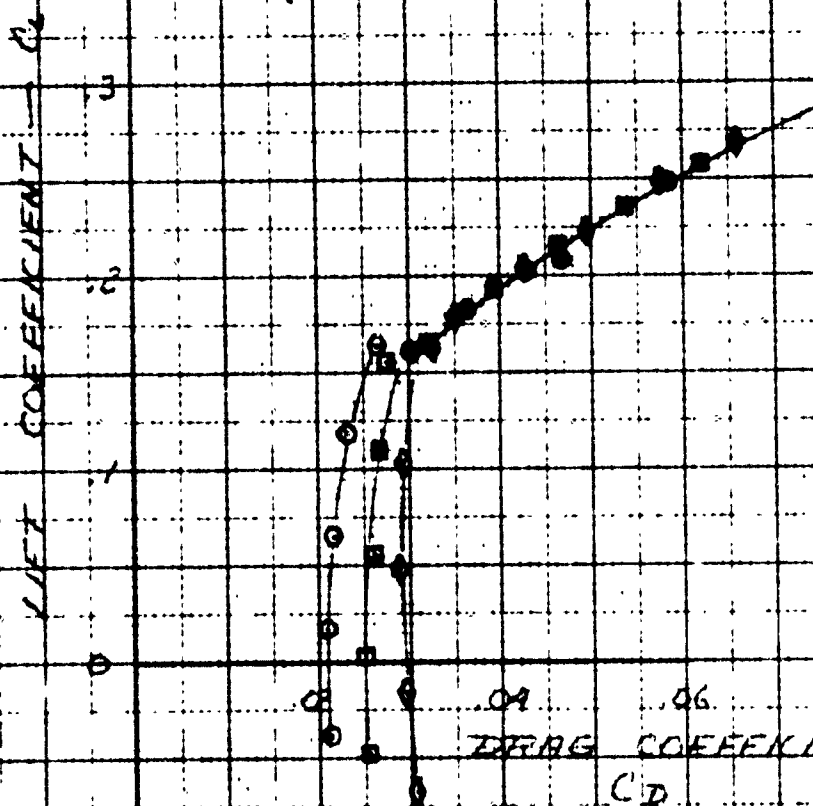
1. DARKENED SYMBOLS
INDICATE VENTED FLOW



CEH 7-11-60

W.E. B. DUBOIS

FRONDE NO.	SYMBOL	d/c	C ₂₀₀₀	CAV NO.
640	○	75	0029	243
556	□	1.00	0039	259
500	◇	1.25	0055	271



PAGE IV. 4

WHIRLING TANK TEST NO. 20

NO. 63 84454

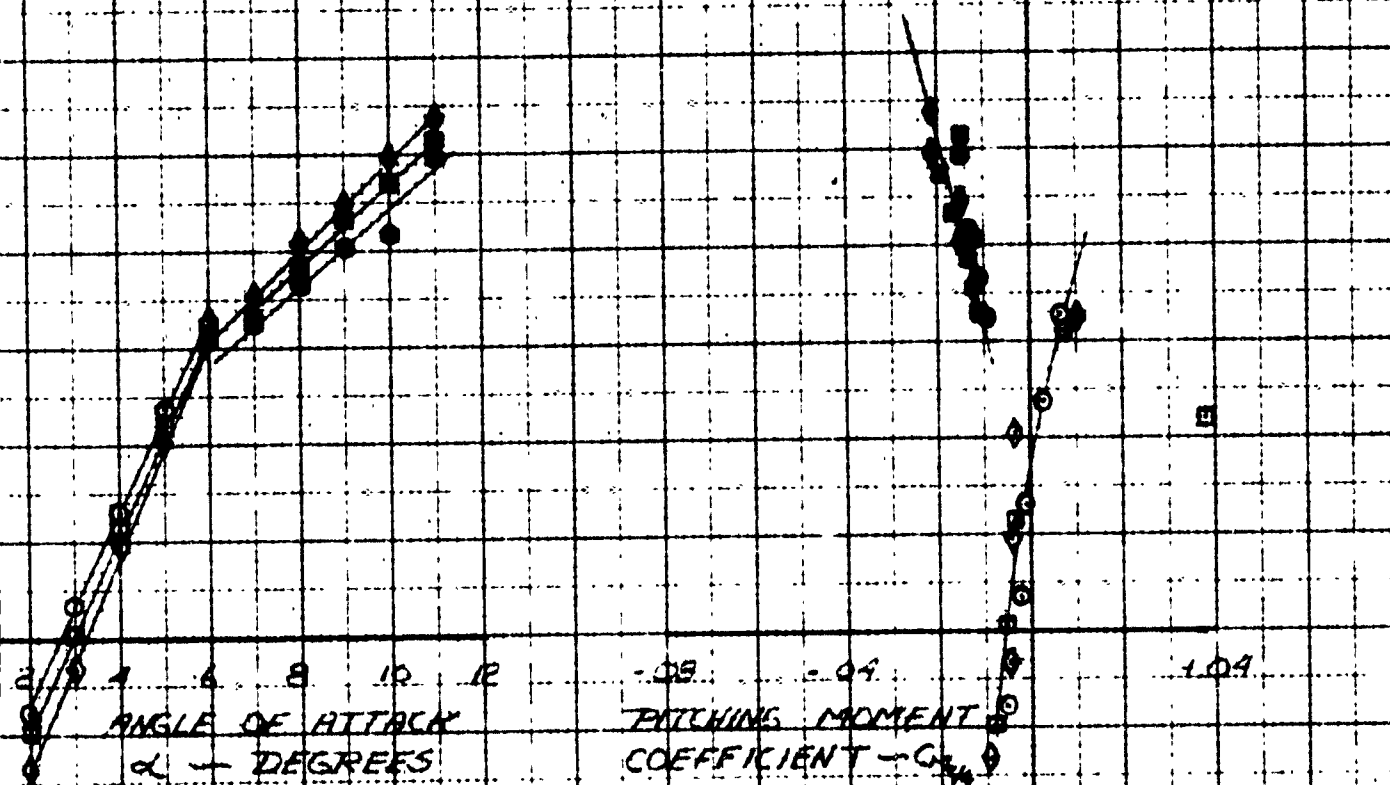
MODEL NO. 1

60 KTS.

M.A.C. = 139.51

NOTE:

1. DASHED SYMBOLS
INDICATE VENTED FLOW



CPH 7-14-61

52959

FRIDGE NO	SYMBOL	d/c	Co. STANT	CAN NO
640	⊙	75	0030	195
556	⊠	1.00	0041	210
500	⬠	1.25	0039	224



PAGE IV 5

WHIRLING TANK TEST NO. 20

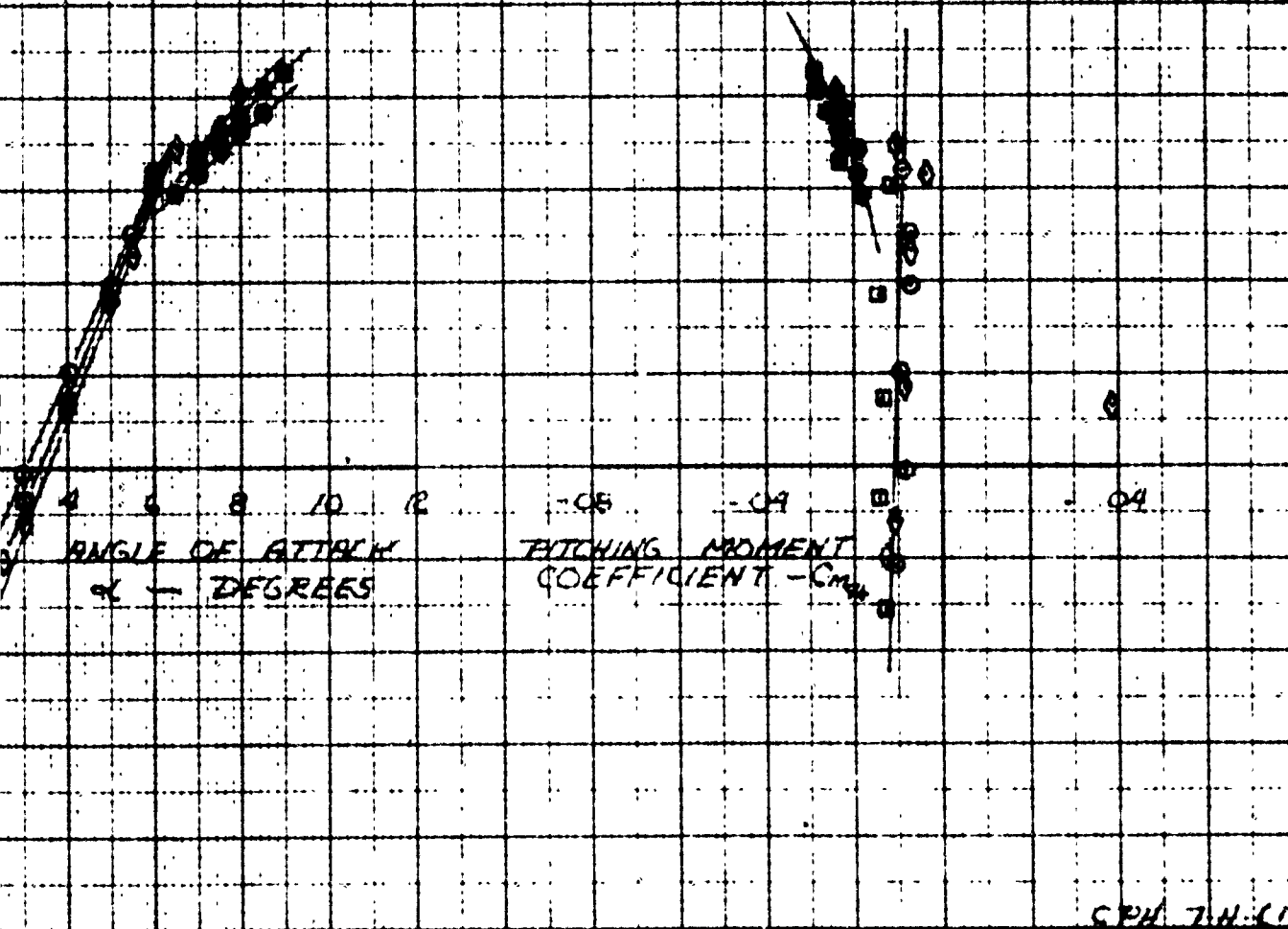
NO. 84454

MODEL NO. 1

TO HTS

MAC = 1.137 FT

NOTE:
1. DARKENED SYMBOLS
INDICATE VENTED FLOW



CPH 7.11.11

MEASUREMENTS
1953

LIFT COEFFICIENT C_L

5

1

3

2

1

0

1

FROUDE NO.	SYMBOL	γ/δ	C_{Dmax}	CAN. NO.
6.40	⊕	.75	.0032	167
5.56	⊞	1.00	.0040	179
5.00	⊙	1.25	.0056	201

.02

.04

.06

-1

-2

2

DRAG COEFFICIENT

C_D

0

1

2

3

4

5

6

7

8

9

PAGE IV. 6

WHIRLING TRAIL TEST N. 20

NO. 65 23-4454

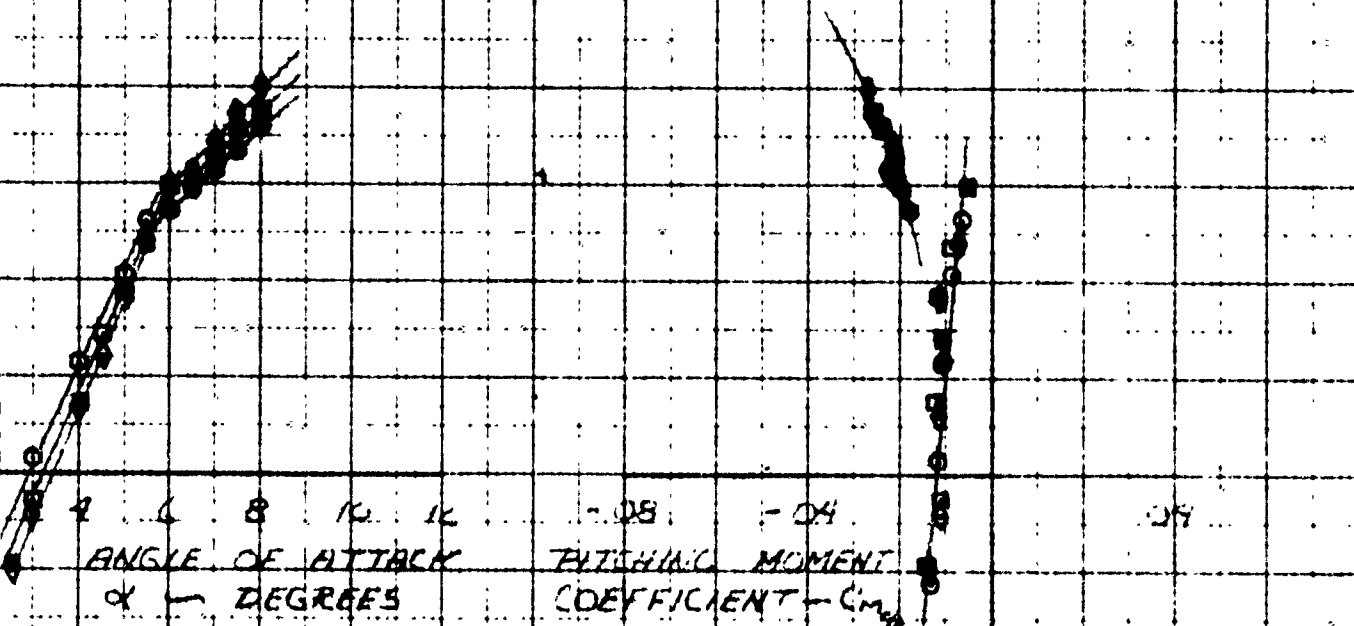
MODEL N. 1

5. 175

MAG. = 139 FT

NOTE:

1. DARKENED SYMBOLS
INDICATE VENTED FLOW

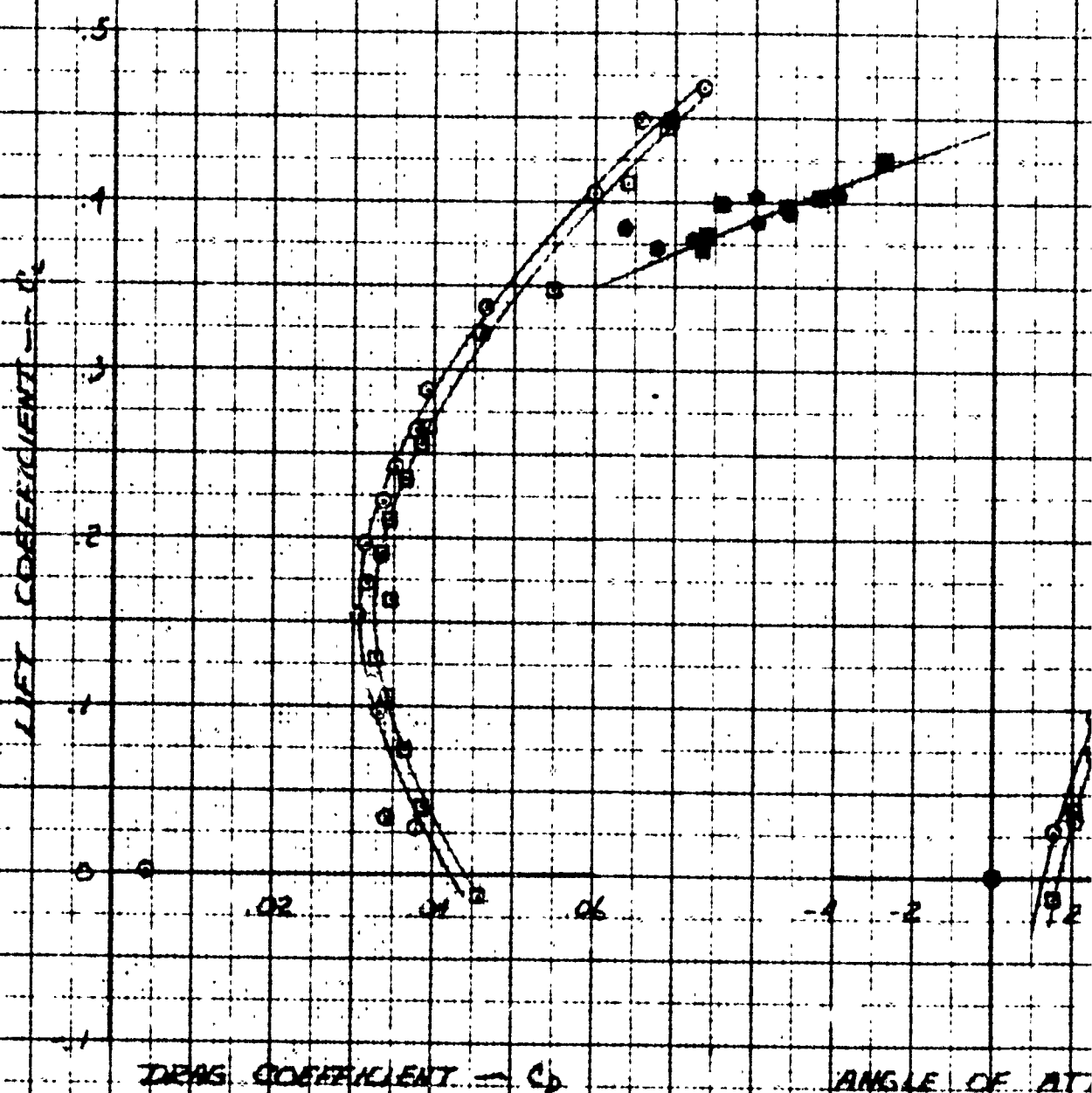


CPH 7-10-61

WE 4000000000 1211

EXPER. NO.	SYMBOL	d/k	$C_{D, \text{STATION}}$	CAN. NO.
556	•	1.00	.0042	500
500	•	1.25	.0057	520

NOTE:
L. DARK
INDIC



DRAG COEFFICIENT - C_D

ANGLE OF AT.

PAGE 12 7

WHIRLING TANK TEST NO. 204

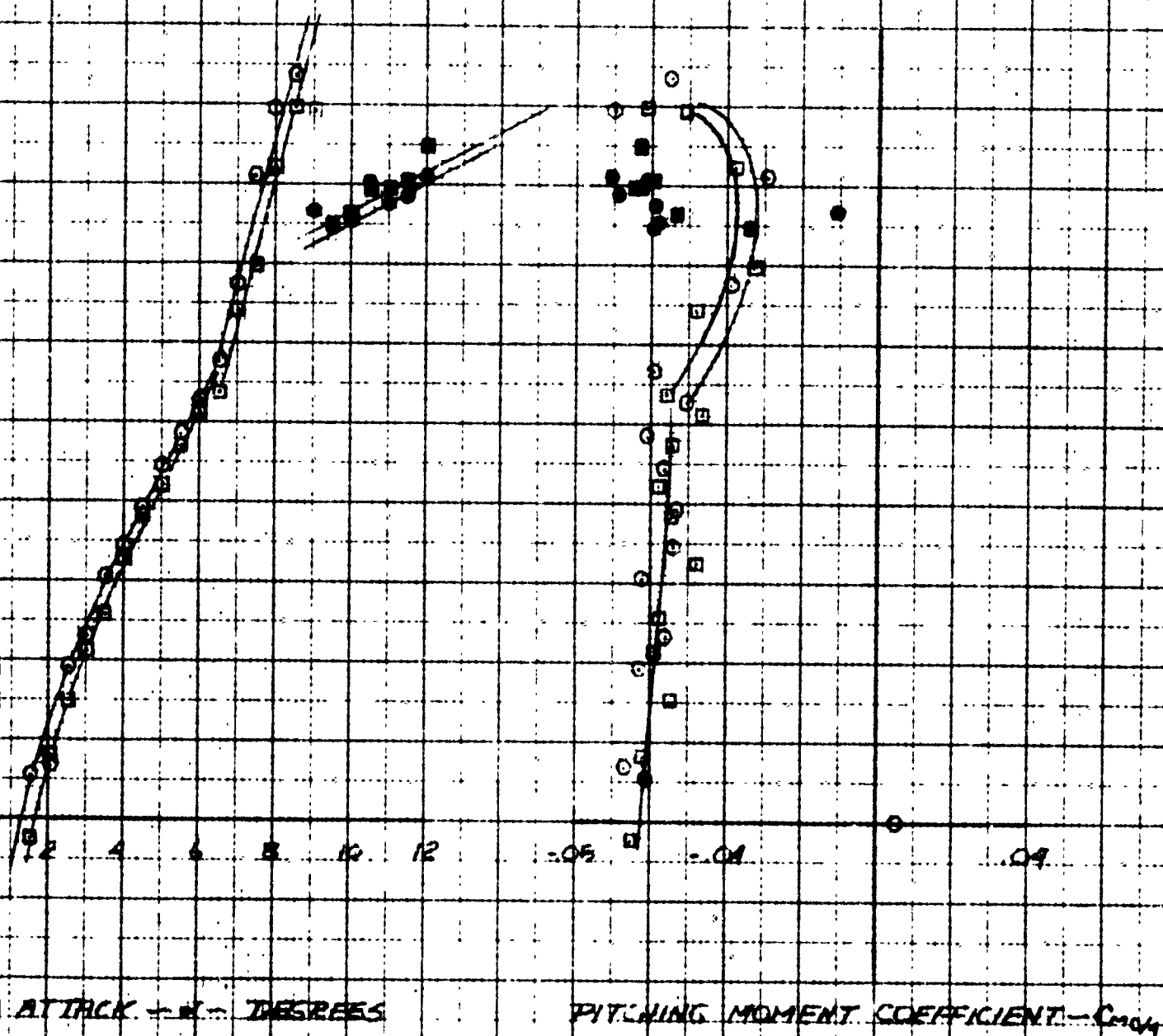
NO. 50. B4454

MODEL NO. 2

40 KTS

MAC = 139 FT

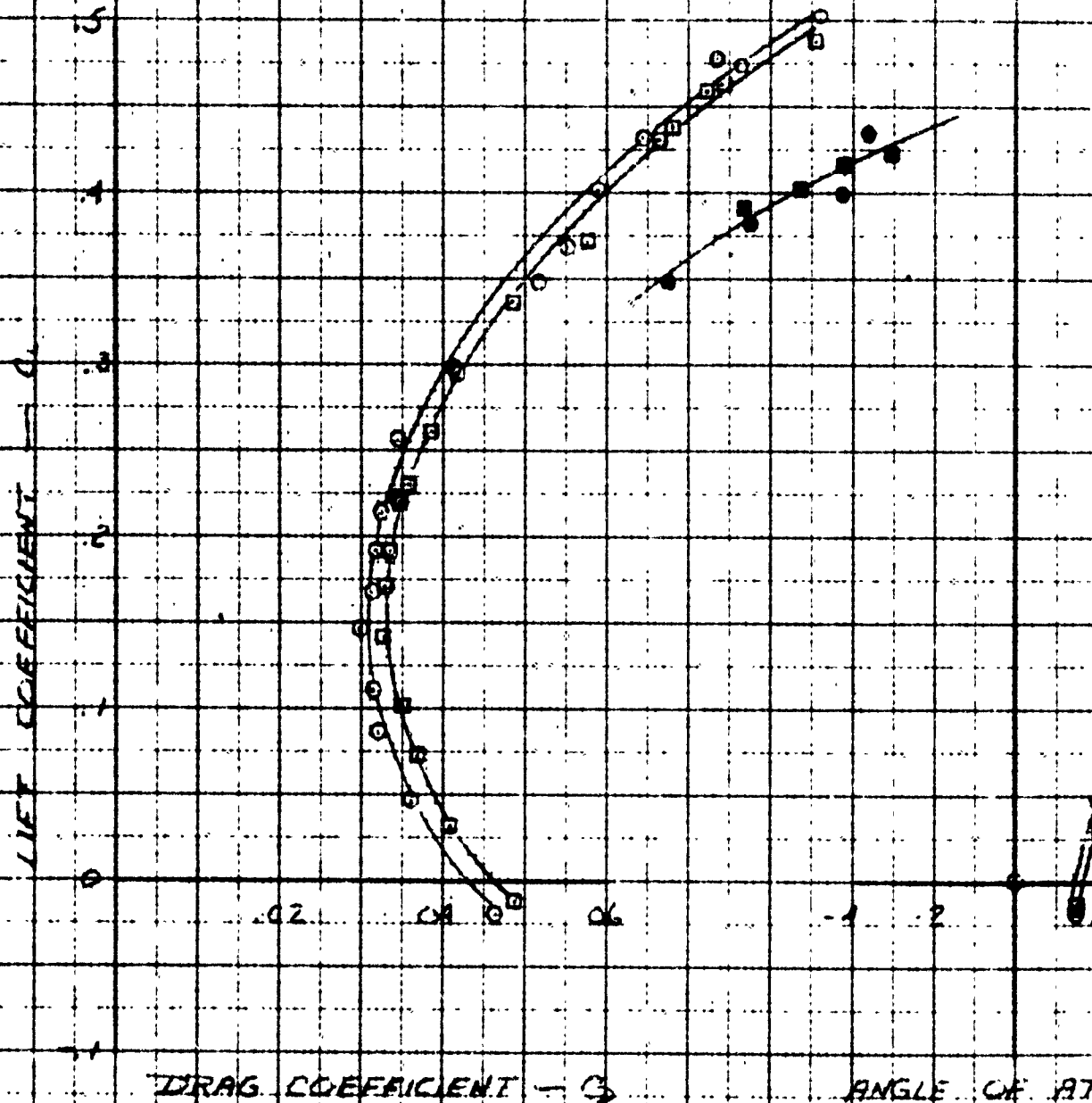
ARMED SYMBOLS
INDICATE VENTED FLOW



W-2 ALBANY, 1957 12773

FROUDE NO	SYMBOL	JA	C_{Dmax}	CAR NO
5.58	•	1.00	0039	407
5.00	•	1.25	0055	440

NOTE:
DARKENED
INDICATE
FLOW.



PAGE IV. 8

WHIRLING TANK TEST NO. 20a

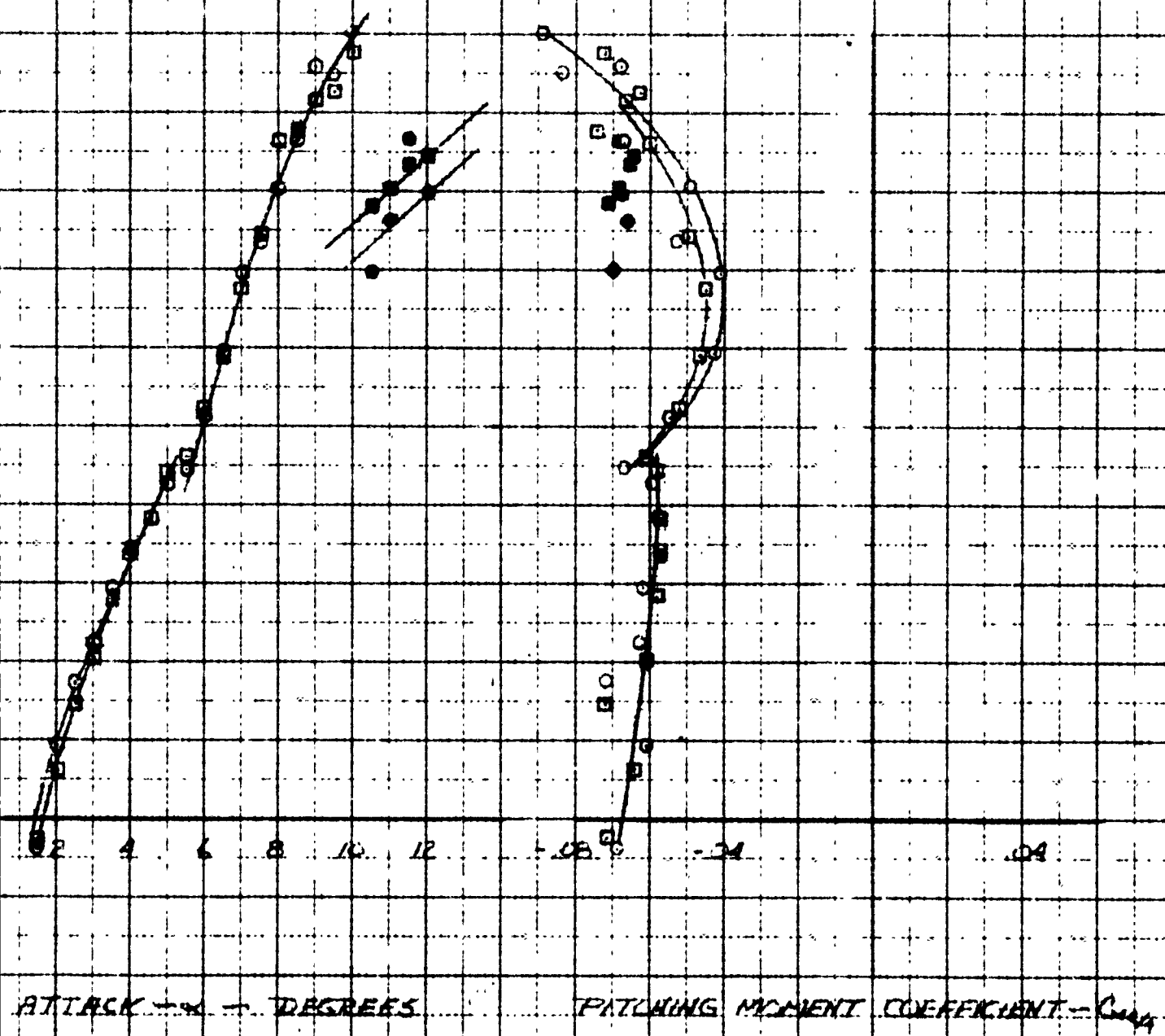
NO 6s 84454

MODEL NO. 2

45 KTS

M.A.C. = .137 FT

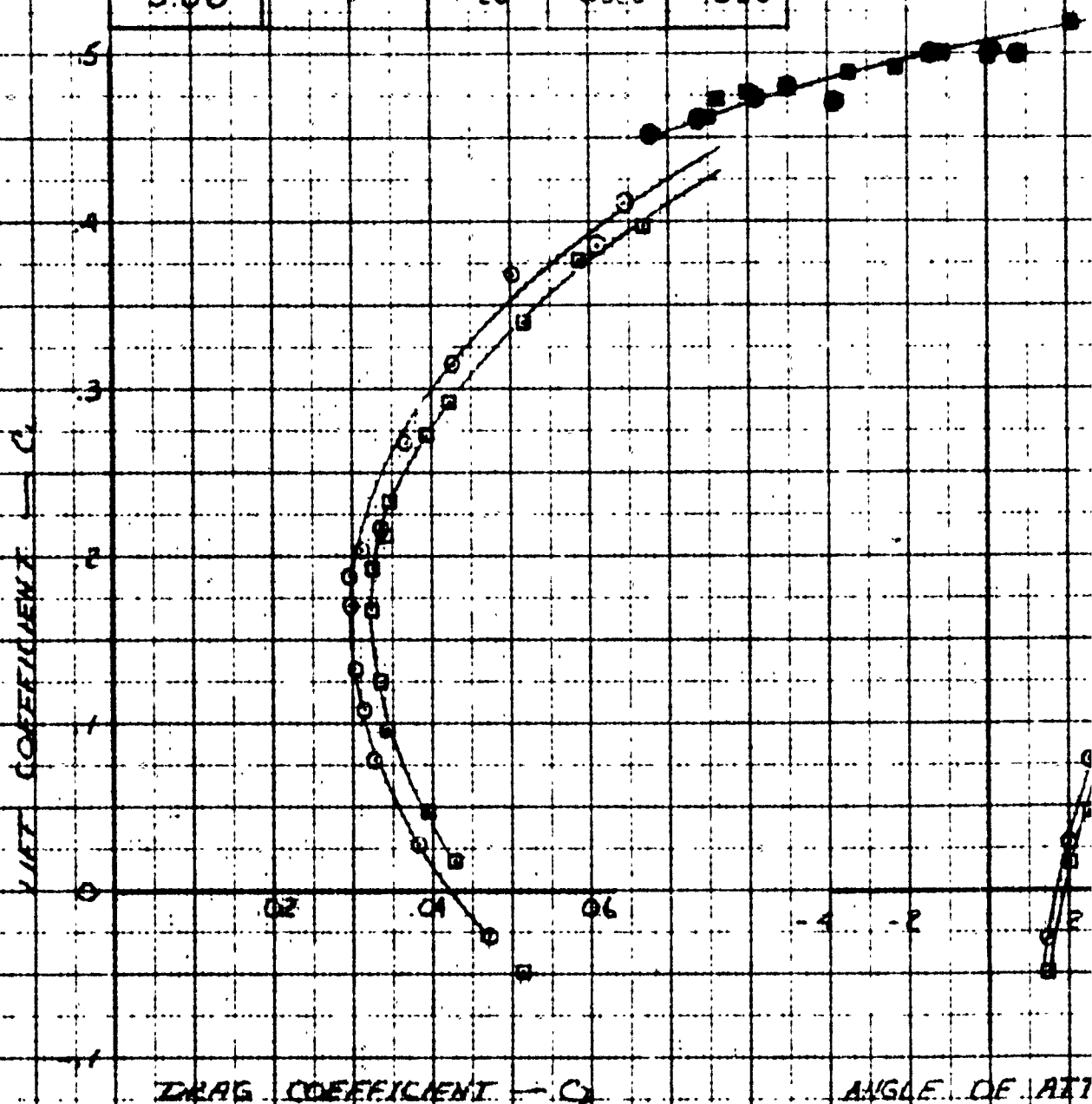
OPENED SYMBOLS
WATE FULLY VENTED.



W-E ALABAMA TEST
2
210

FROUDE NO.	SYMBOL	d/c	$C_{D, max}$	CAV. NO.
5.56	○	1.00	0.65	348
5.00	●	1.25	0.55	360

NOTE
1 DARKENED
FULLY CAV



PAGE IV. 9

WHIRLING TANK TEST NO 204

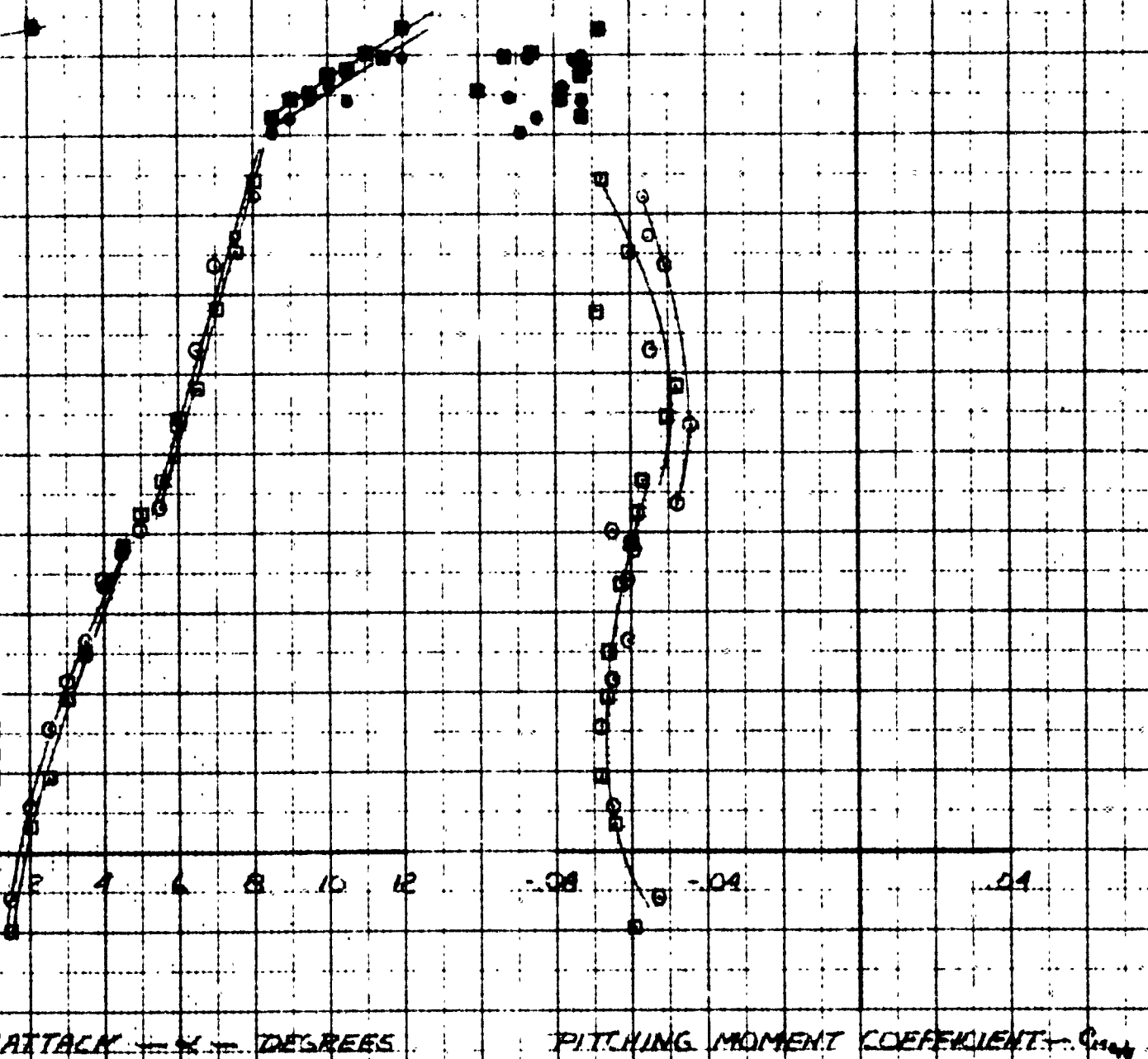
NO 65 84454

MODEL NO 2

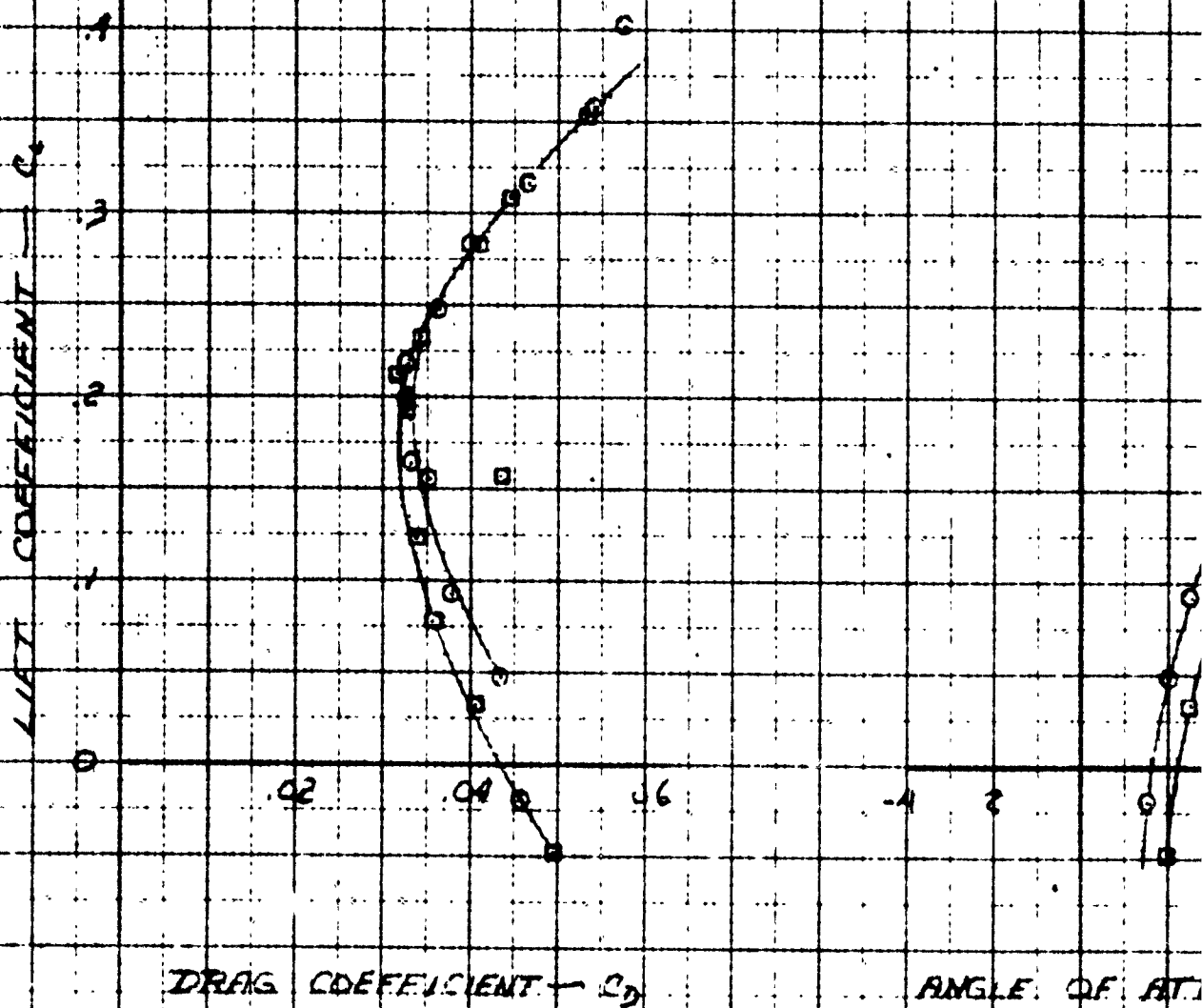
50 KTS

M.A.C. = 139 FT

ED SYMBOLS INDICATE
CAVITATED UPPER SURFACE



PROBE NO	CAN NO	SYMBOL	IN	Count
556	259	•	1.00	0039
500	275	•	1.25	0053



PAGE 12 10

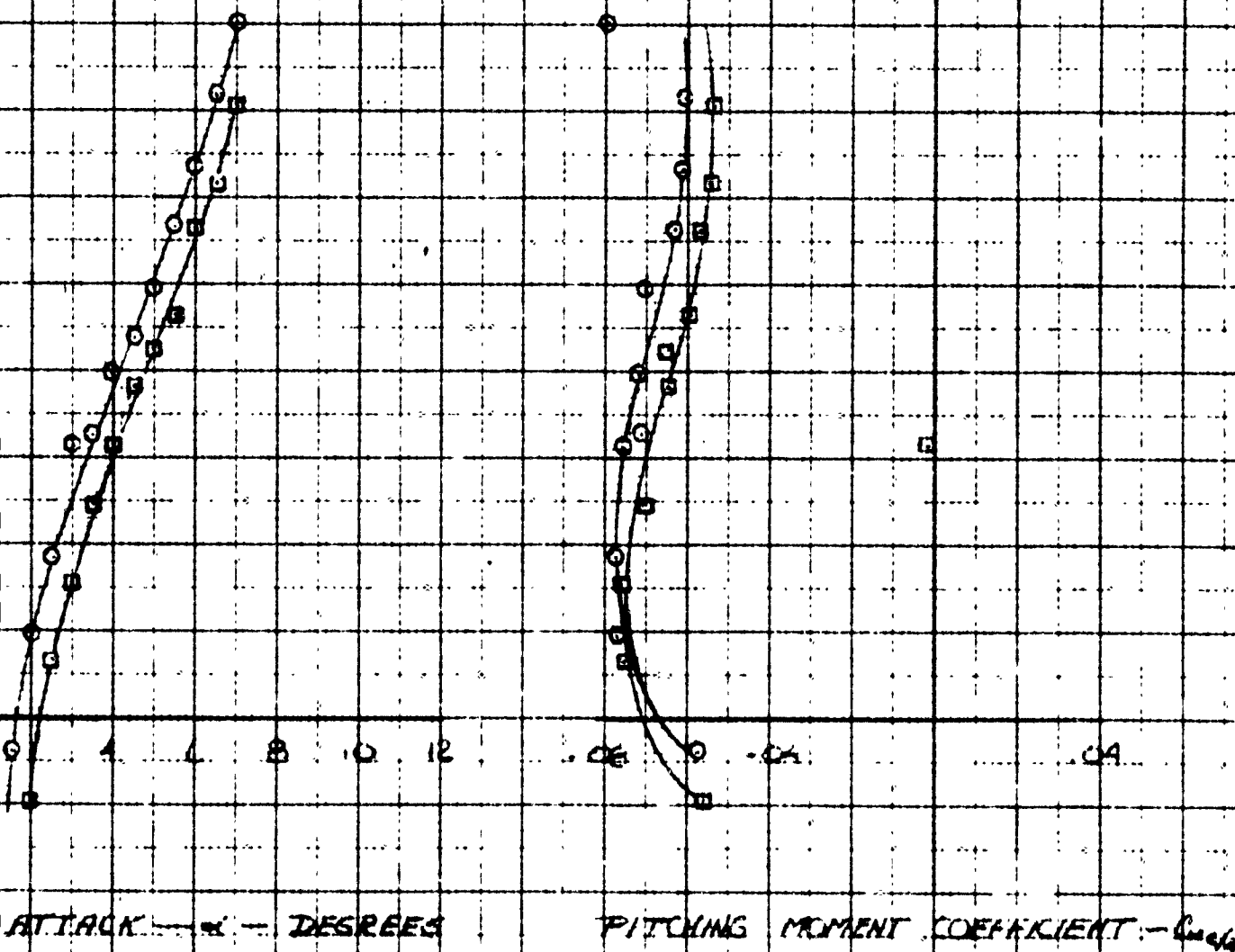
WHIRLING TANK TEST NO 20

NO 6 24454

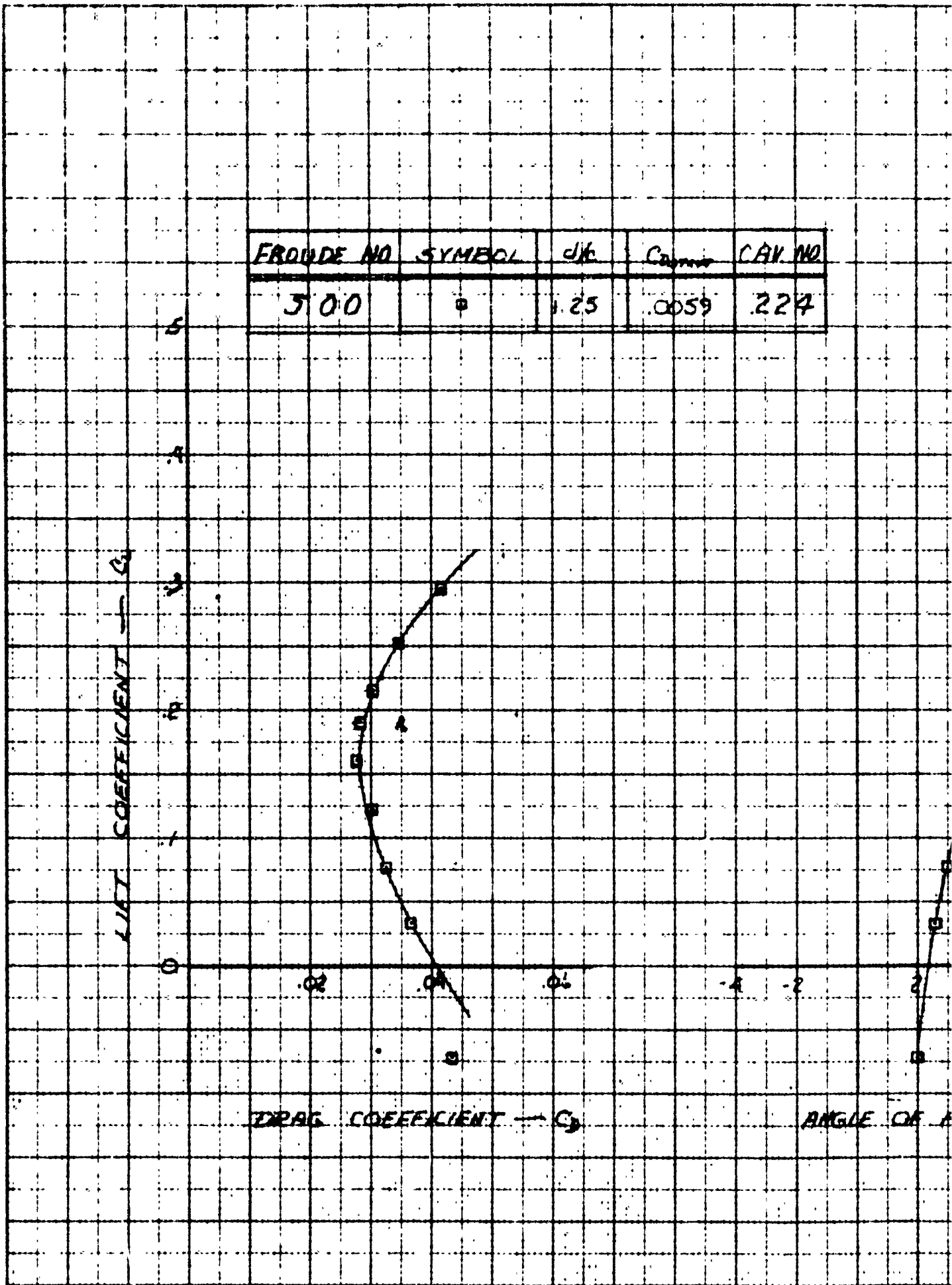
MODEL NO 2

60 KTS.

M.A.C. = 139 FX



ME ALGARD 1957 5275



PAGE III. 11

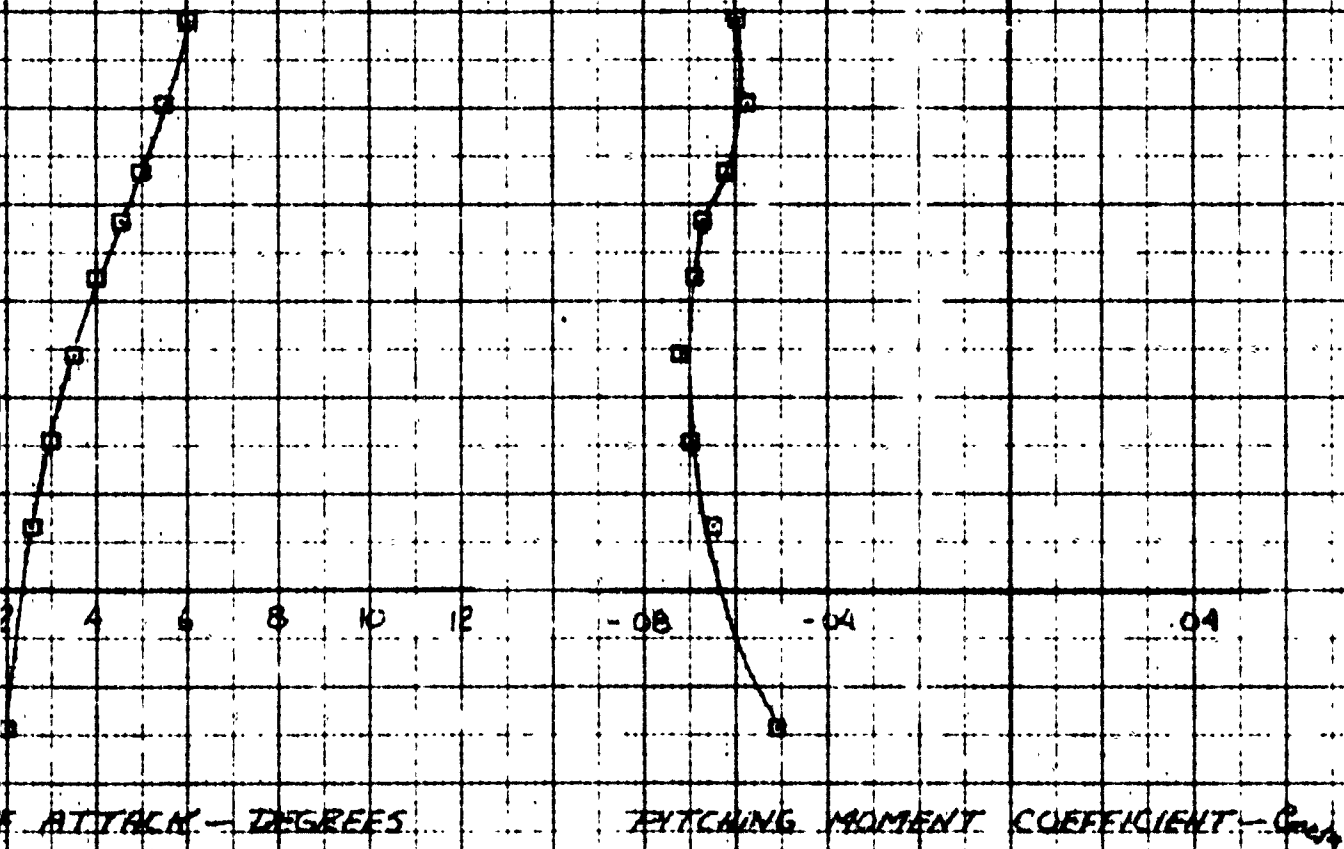
WHIRLING TANK TEST NO. 20

NO. 45. 84454

MODEL NO. 2

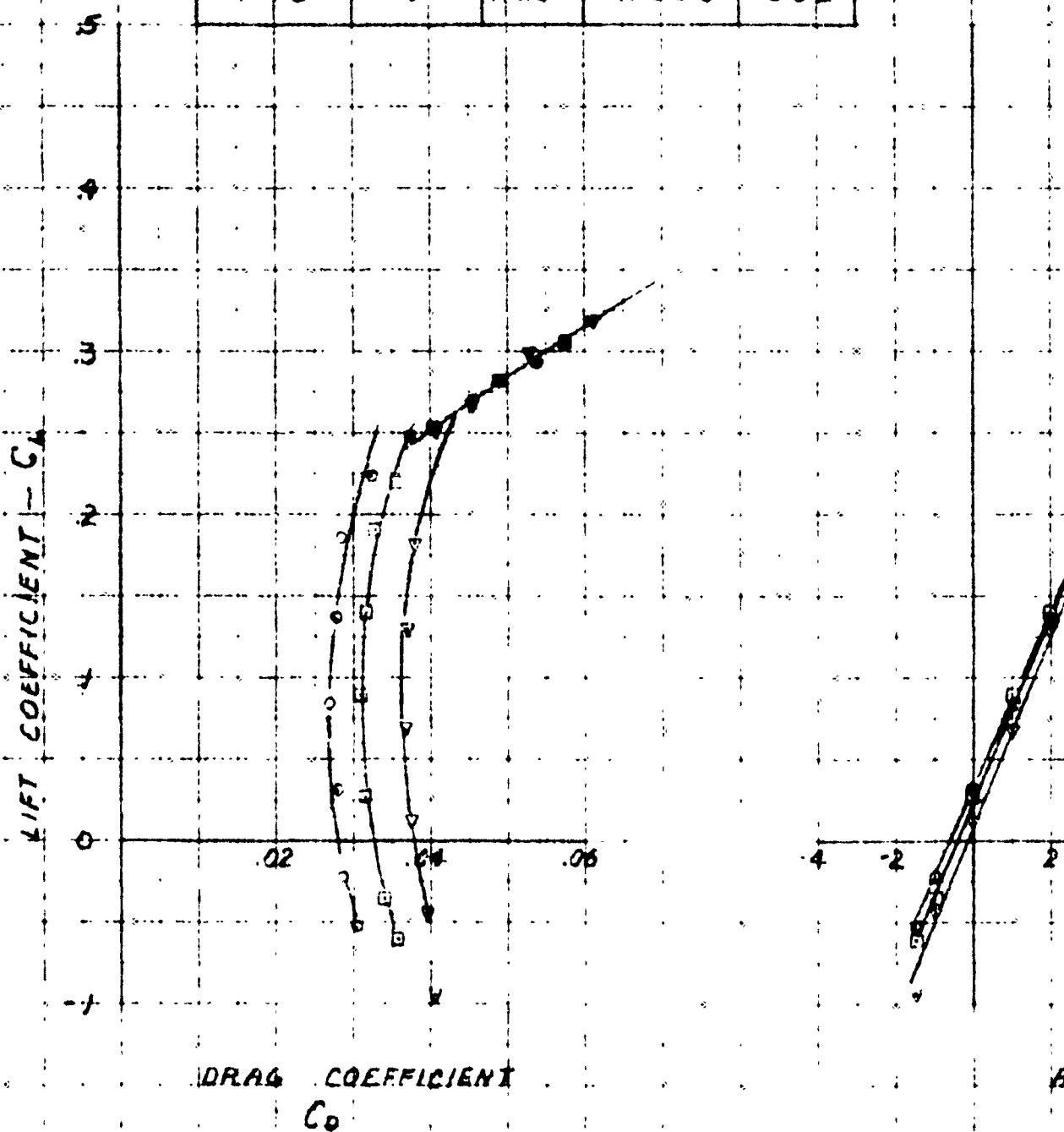
70 KTS.

M.A.C. = 139 FT



FROUDE NO.	SYMBOL	$\frac{h}{c}$	C_D STALL	CHV NO
6.12	○	.75	.0023	4124
5.32	□	1.00	.0034	515
4.79	▽	1.25	.0053	532

NOTE



WHIRLING TANK TEST NO 204

NO 65 84454

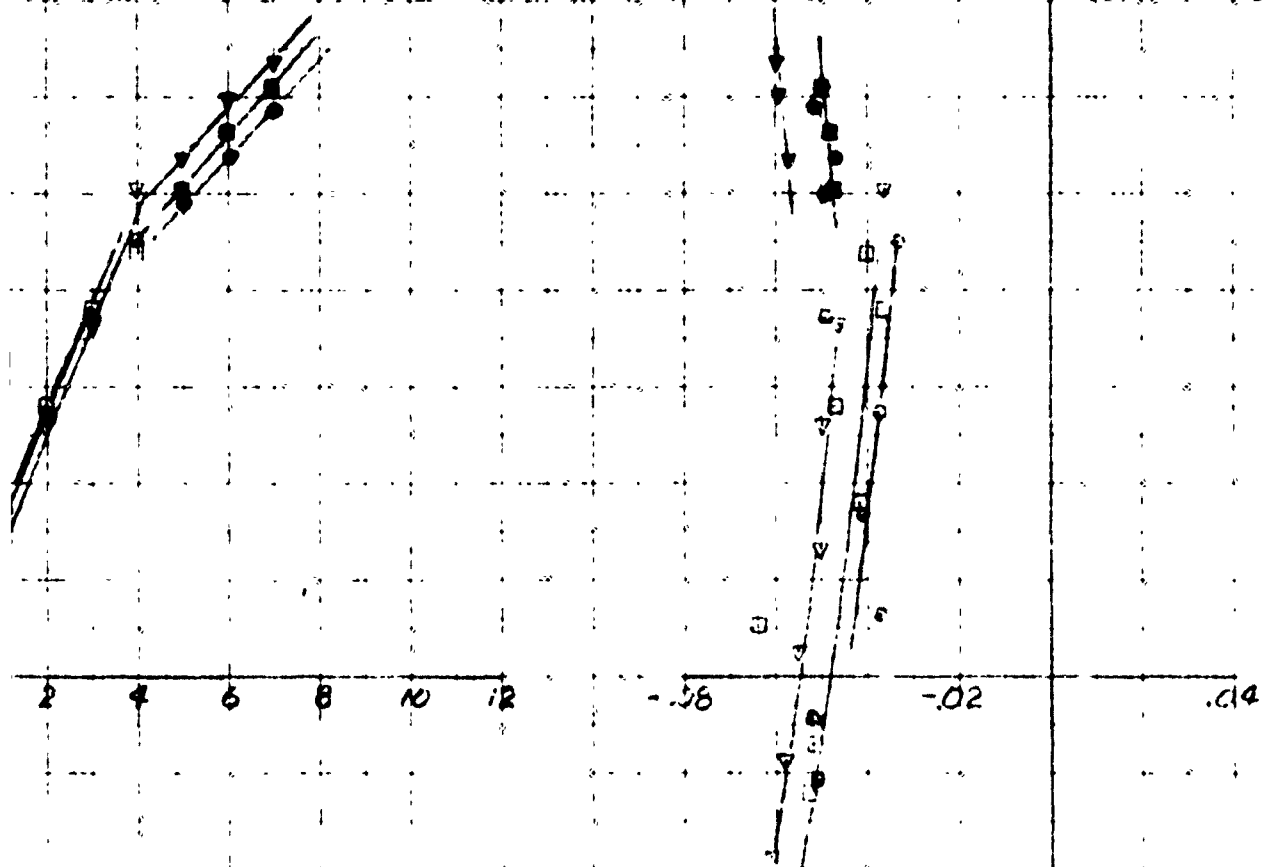
MODEL NO. 3

40 KTS.

MAC = .152 FT.

TE:

1. DARKENED SYMBOLS INDICATE
VENTED FLOW.



ANGLE OF ATTACK
 α - DEGREES

PITCHING MOMENT
COEFFICIENT - C_m

FROUDE NO.	SYMBOL	d/c	C_D SUBT	CAV NO.
5.12	○	.75	.0021	408
5.32	□	1.00	.0036	424
4.79	▽	1.25	.0048	437

LIFT COEFFICIENT C_L

DRAG COEFFICIENT C_D

WHIRLING TANK TEST NO. 204

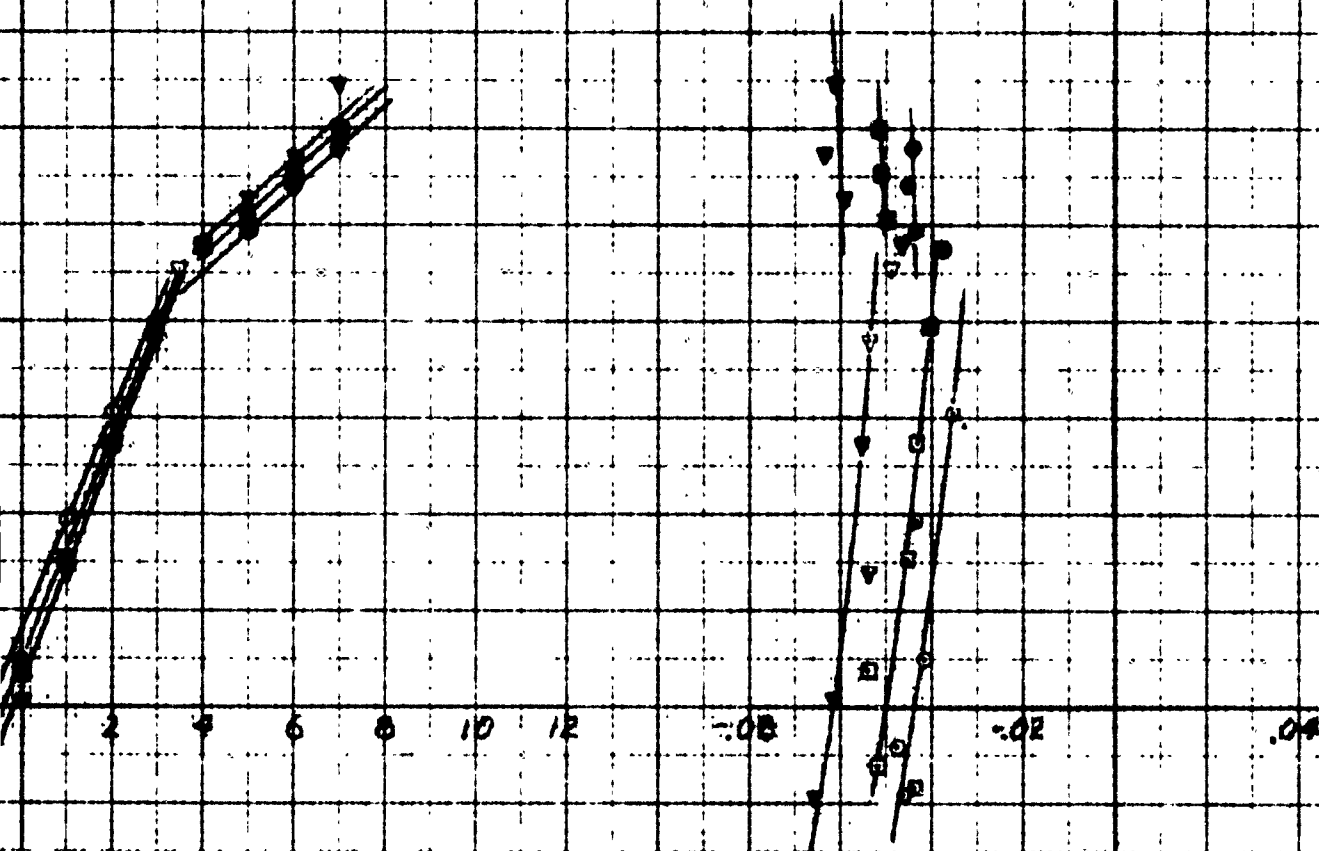
NO. 65 B4454

MODEL NO. 3

45 KTS.

M.A.C. = 152 FT

NOTE:
1. DARKENED SYMBOLS INDICATE
VENTED FLOW.



ANGLE OF ATTACK
 α - DEGREES

PITCHING MOMENT
COEFFICIENT - C_m/C_q

SYMBOL	t/c	C_D WENT	FROUDE NO.	CAV. NO.
⊙	75	.0024	6.12	345
□	100	.0039	5.32	354
▽	125	.0054	4.79	370

LIFT COEFFICIENT - C_L

DRAW COEFFICIENT
 C_D

WHIRLING TANK TEST NO. 204

NO. 65 B4454

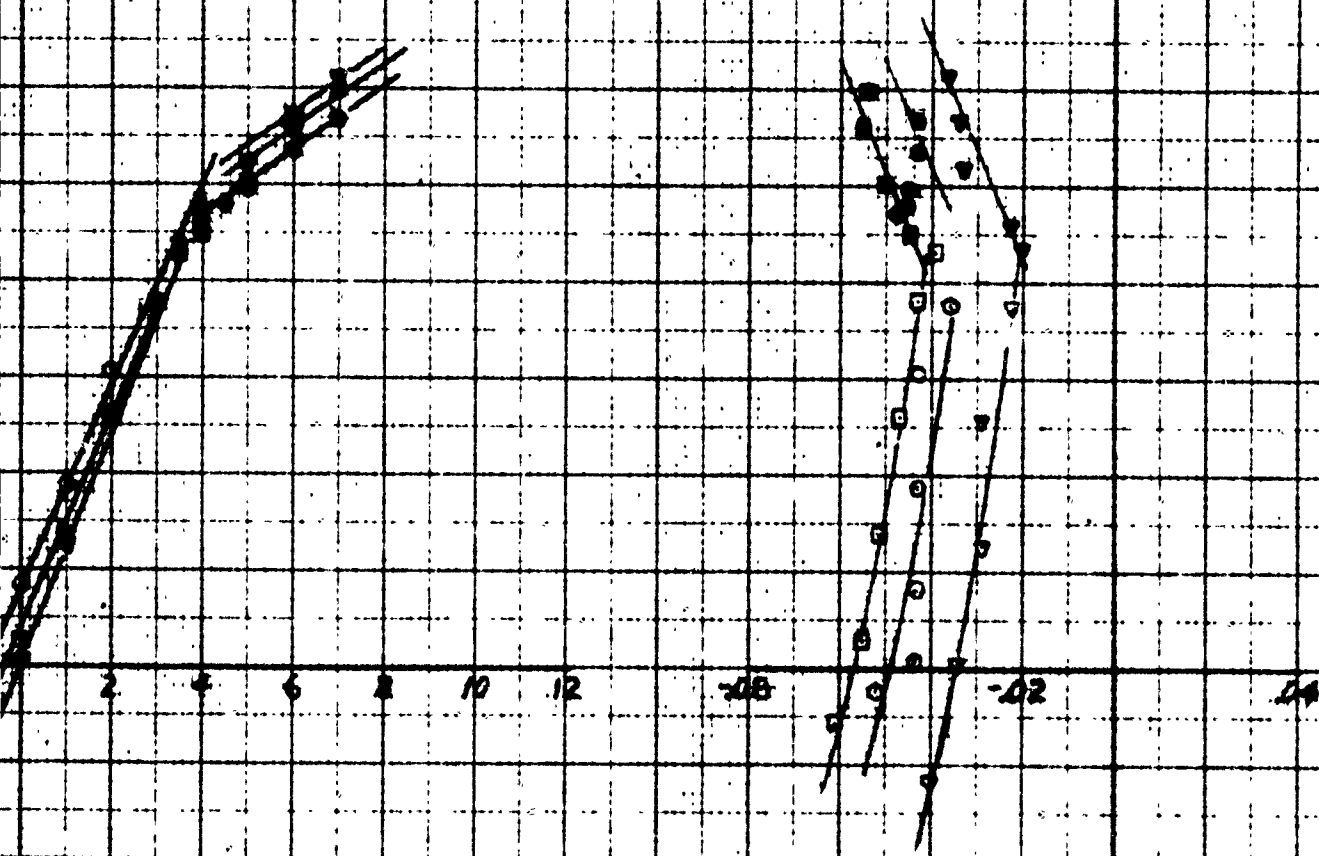
MODEL NO. 3

50 KTS

MAR 2 1952 FT

NOTE:

1. DARKENED SYMBOLS INDICATE
VENTED FLOW.



ANGLE OF ATTACK
 α - DEGREES

PITCHING MOMENT
COEFFICIENT - $C_{m\alpha}$

SYMBOL	s/c	$C_{D \text{ START}}$	FROUDE NO.	CAY. NO.
⊙	75	.0027	6.12	.250
□	100	.0042	5.32	.265
▽	125	.0057	4.79	.280

LIFT COEFFICIENT $\sim C_L$

DRAG COEFFICIENT
 C_D

WHIRLING TANK TEST NO. 204

NO. 65 B4454

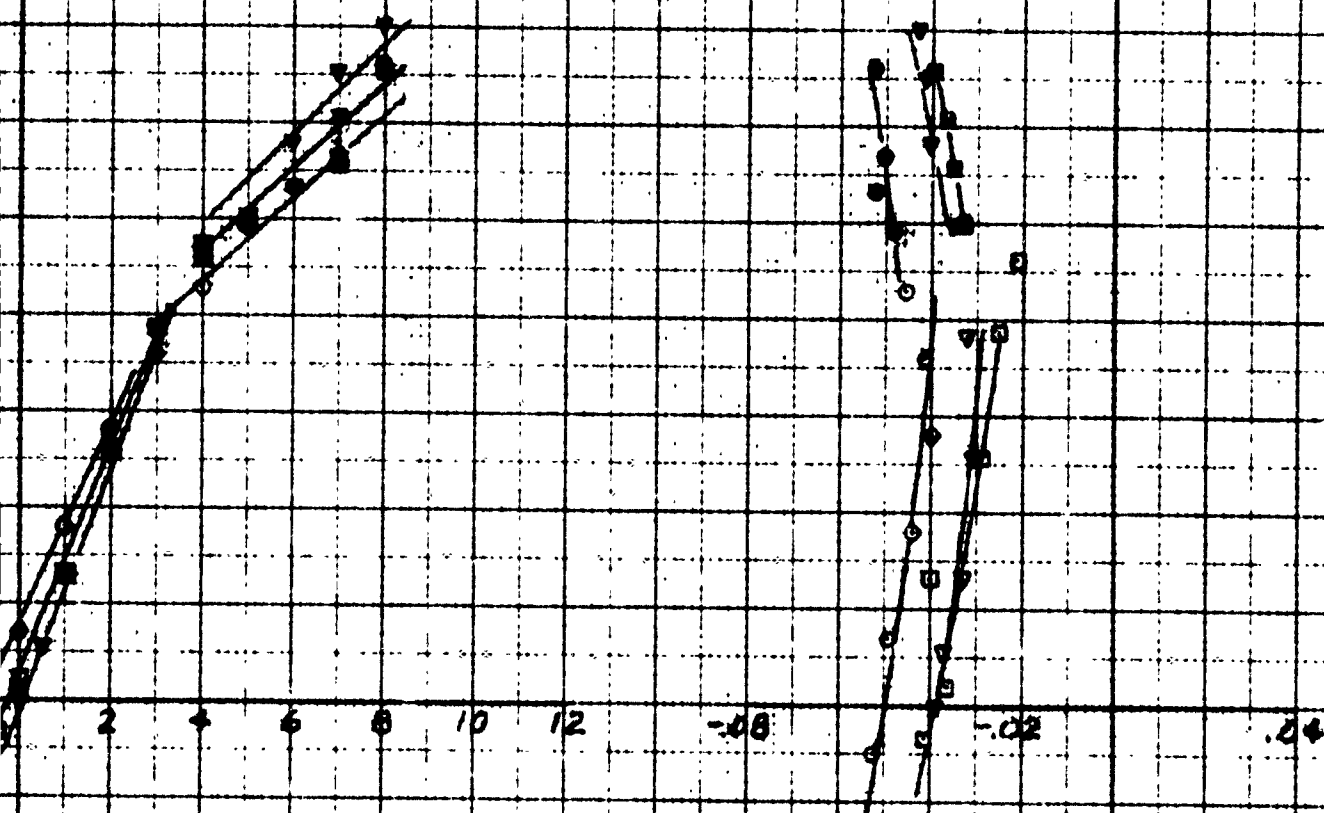
MODEL NO. 3

60 MTS.

M.A.C. = 152 FT

NOTE:

1. DARKENED SYMBOLS INDICATE
VENTED FLOW.



ANGLE OF ATTACK
 α - DEGREES

PITCHING MOMENT
COEFFICIENT - C_m

SYMBOL	q/c	C_D STRUT	FROUDE NO.	CAV. NO.
○	75	.0029	0.12	199
□	100	.0042	5.32	215
▽	125	.0056	4.79	232

NOT

LIFT COEFFICIENT $\sim C_L$

DRAG COEFFICIENT
 C_D

WHIRLING TANK TEST NO. 20H

NO. 65 B4454

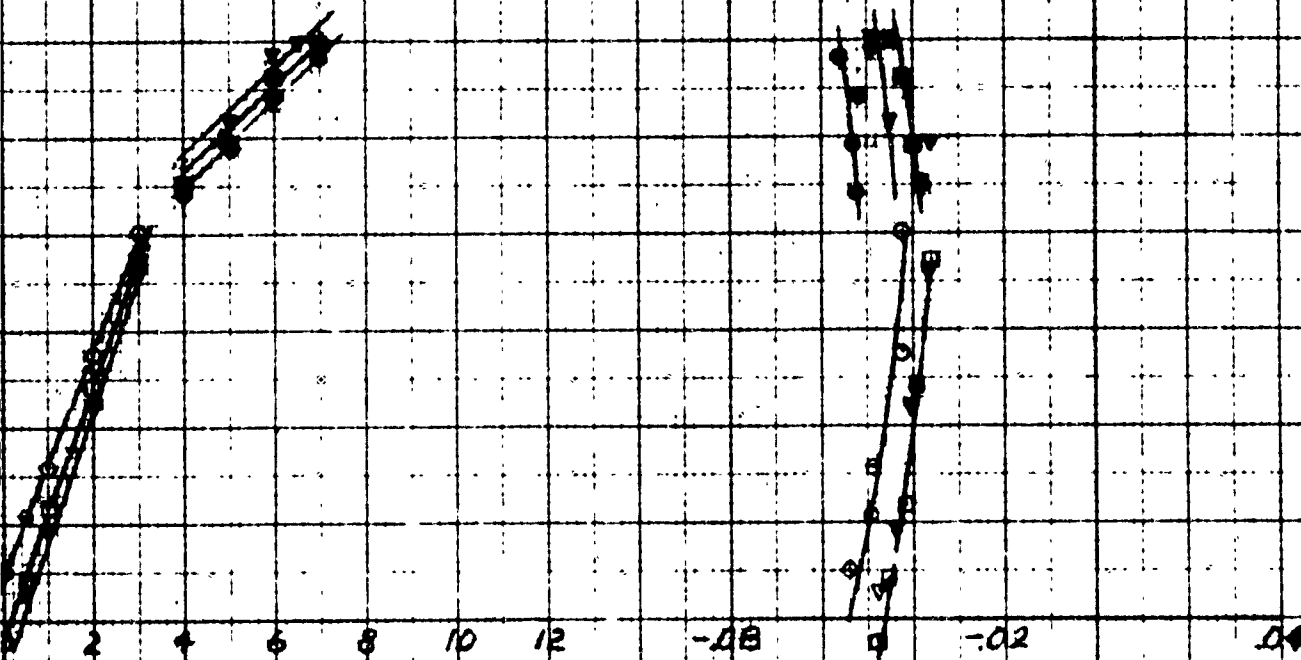
MODEL NO. 3

70 KTS.

M.A.C. 1.152 Ft.

NOTE:

1. DARKENED SYMBOLS INDICATE
VENTED FLOW



ANGLE OF ATTACK
 α - DEGREES

PITCHING MOMENT
COEFFICIENT - C_M

FROUDE NO.	SYMBOL	$\frac{4}{10}$	C_D STRUT	CAV. NO.	NO.
6.12	○	75	0027	177	
5.32	□	100	0042	192	
4.79	▽	125	0050	210	

LIFT COEFFICIENT C_L

0.2 0.4 0.6

DRAG COEFFICIENT C_D

WHIRLING TANK TEST NO. 20H

NO. 65 84454

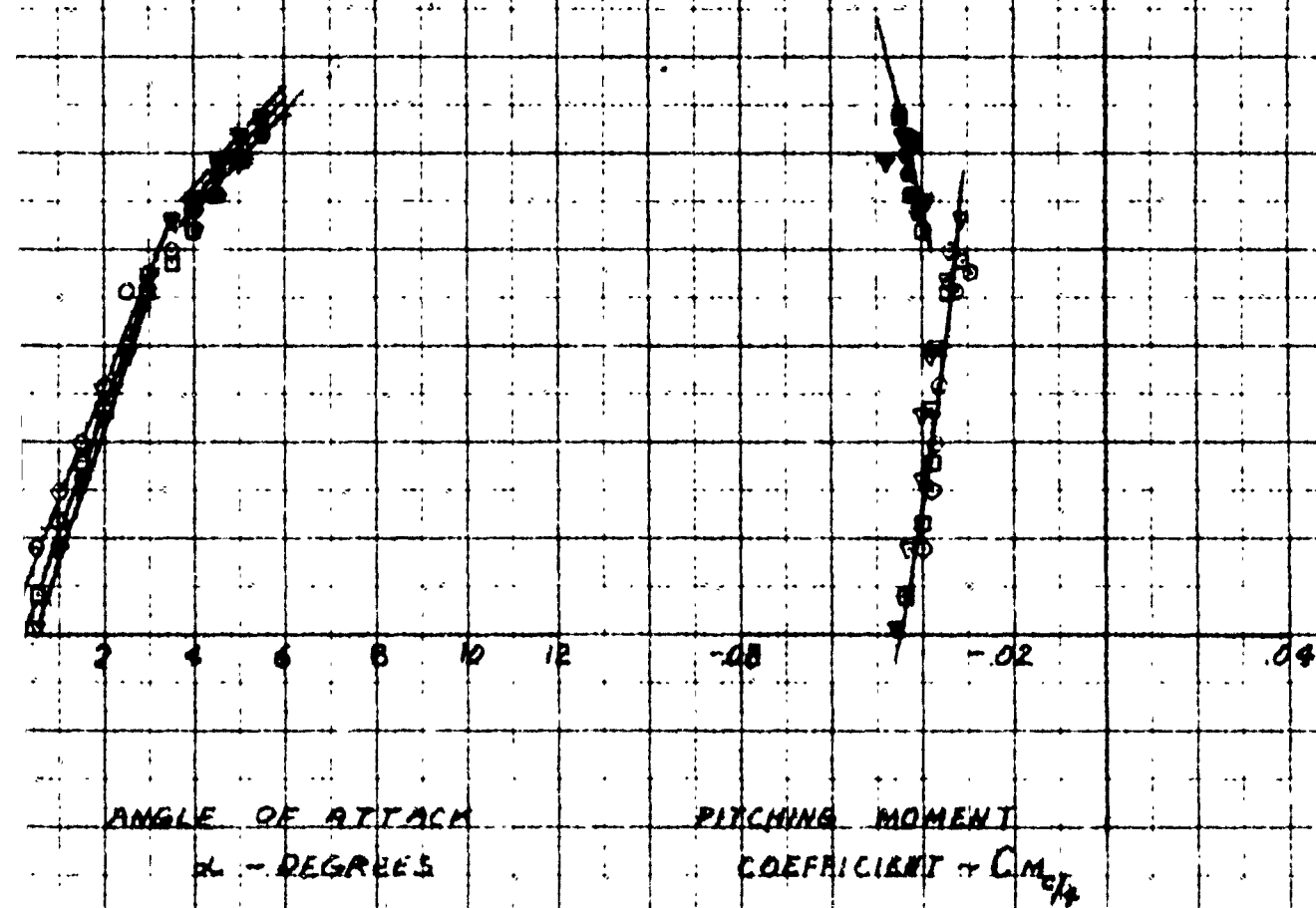
MODEL NO. 3

80 KTS.

M.A.C. 152 FT

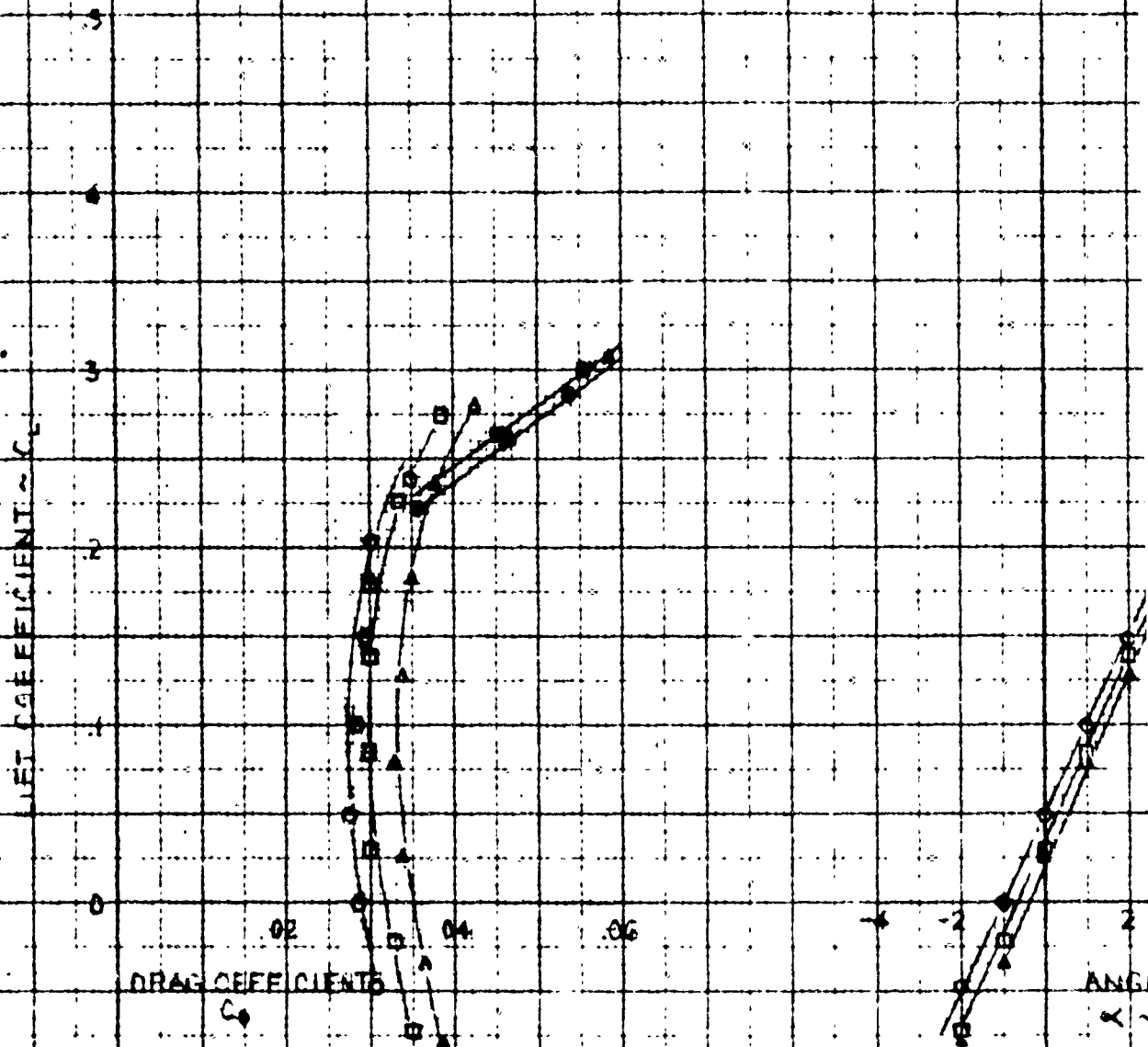
NOTE:

DARKENED SYMBOLS INDICATE
VENTED FLOW.



FROUDE NO.	SYMBOL	d/c	$C_{D \& RL}$	CAV No.
6.2913	○	.75	0019	493
5.4682	□	1.00	0027	511
4.9141	△	1.25	0042	517

NOTES
1. CA

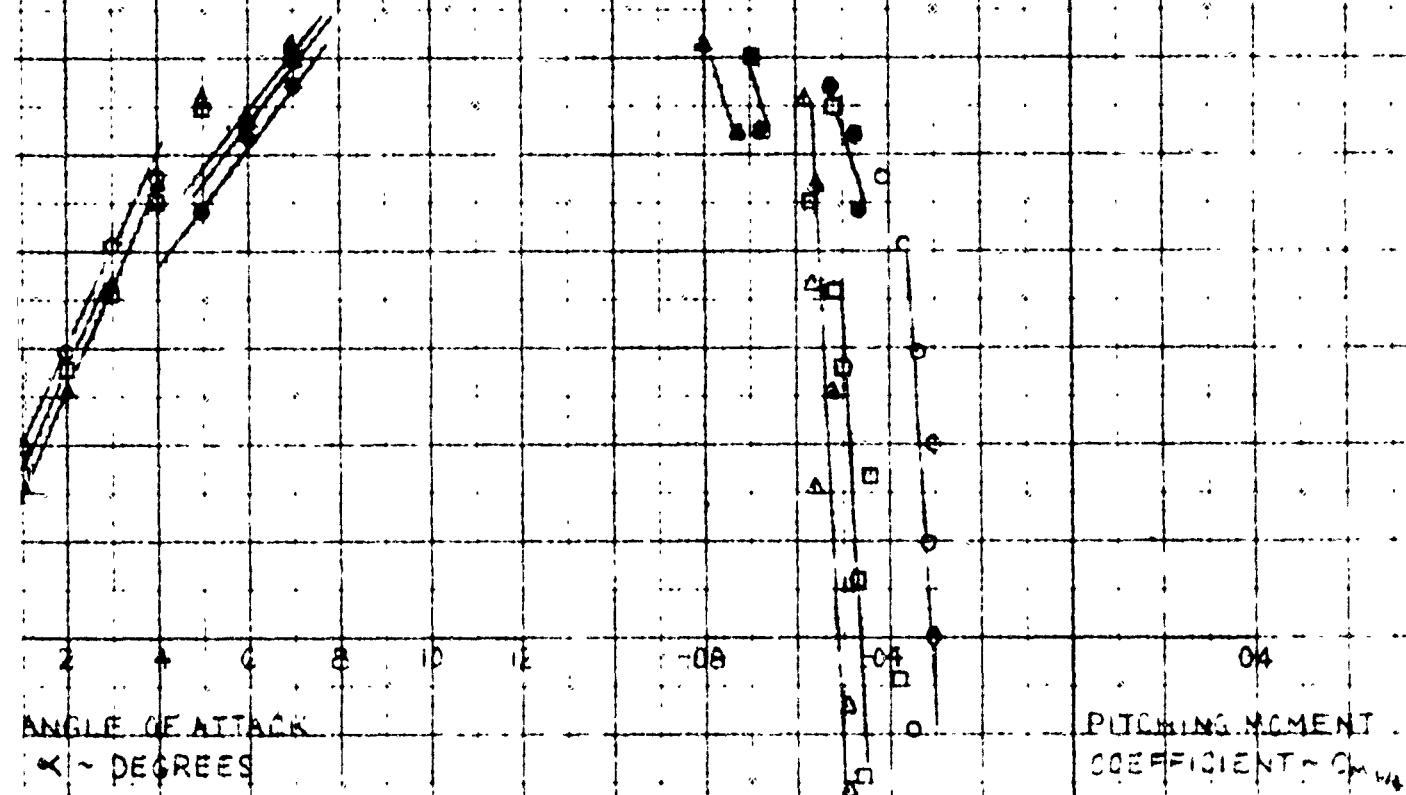


GRUMMAN AIRCRAFT ENGINE

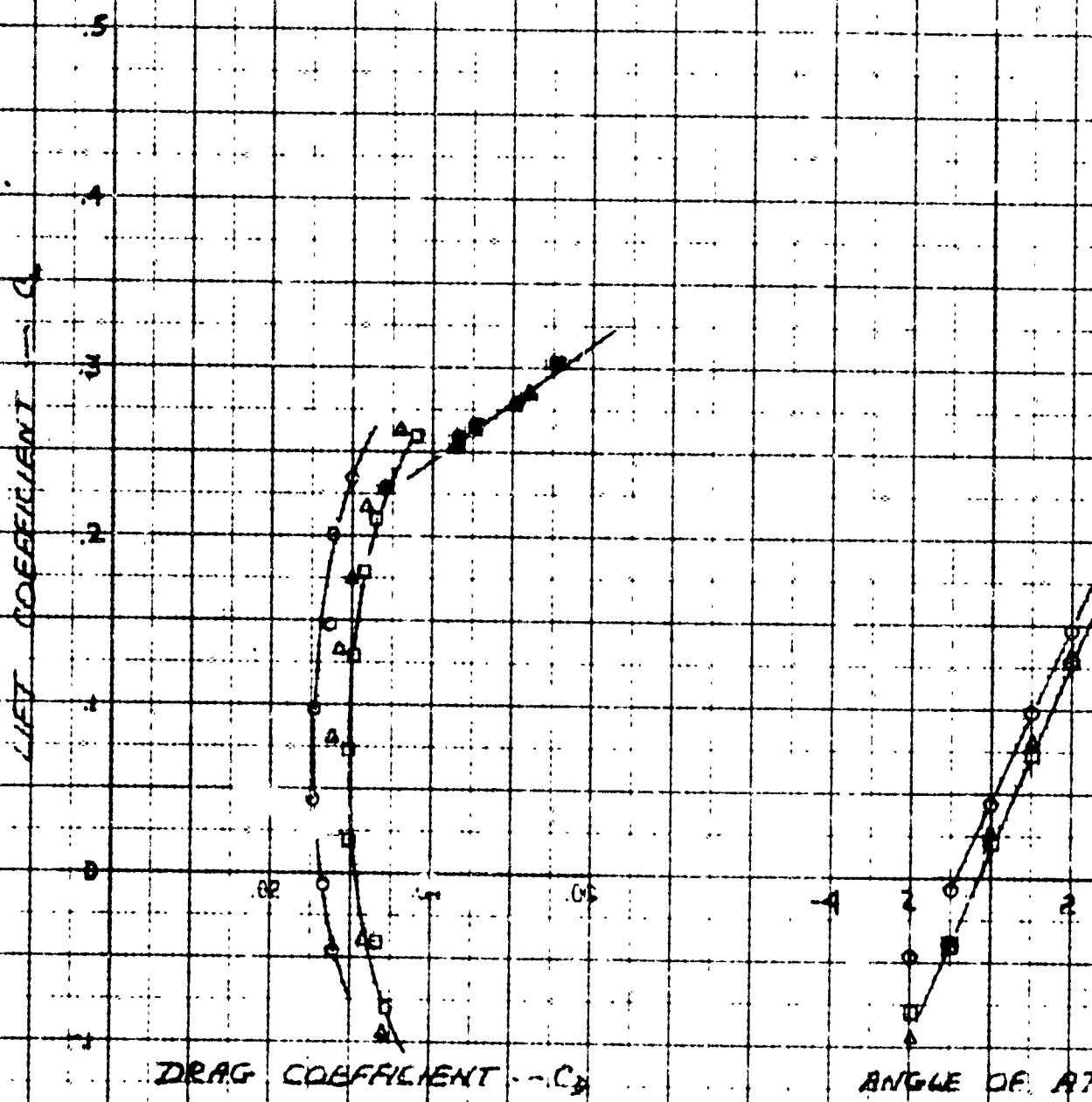
1. DARKENED SYMBOLS INDICATE
VENTED FLOW

WHIRLING TANK TEST NO 236
BU SHIPS NO 4
40 KTS

M.A.L. = 144 FT



FROUDE No.	SYMBOL	d/k	C _{pm}	C _{pv} No.	NO
62913	+	.75	.0021	.402	1.
54002	o	1.00	.0033	.414	
49141	+	1.25	.0042	.433	2.



NOTE

1. DARKENED SYMBOLS INDICATE
VENTED FLOW

2. DATA @ $\alpha = 125$ IS OPEN
TO QUESTION.

WHIRLING TANK TEST NO 208

BUSHIPS NO 4
45 KTS

M.A.C. = 144 FT.



ATTACK — α — DEGREES

PITCHING MOMENT COEFF — $C_{m\alpha}$

FRONDE NO.	SYMBOL	d/c	CD STAT	CAV. NO.
6.2818	○	.75	0022	337
5.4698	○	1.00	0034	348
4.9191	▲	1.25	0046	366

LIFT COEFFICIENT - C_L

DRAG COEFFICIENT - C_D

ANGLE OF A

GRUMMAN AIRCRAFT EN

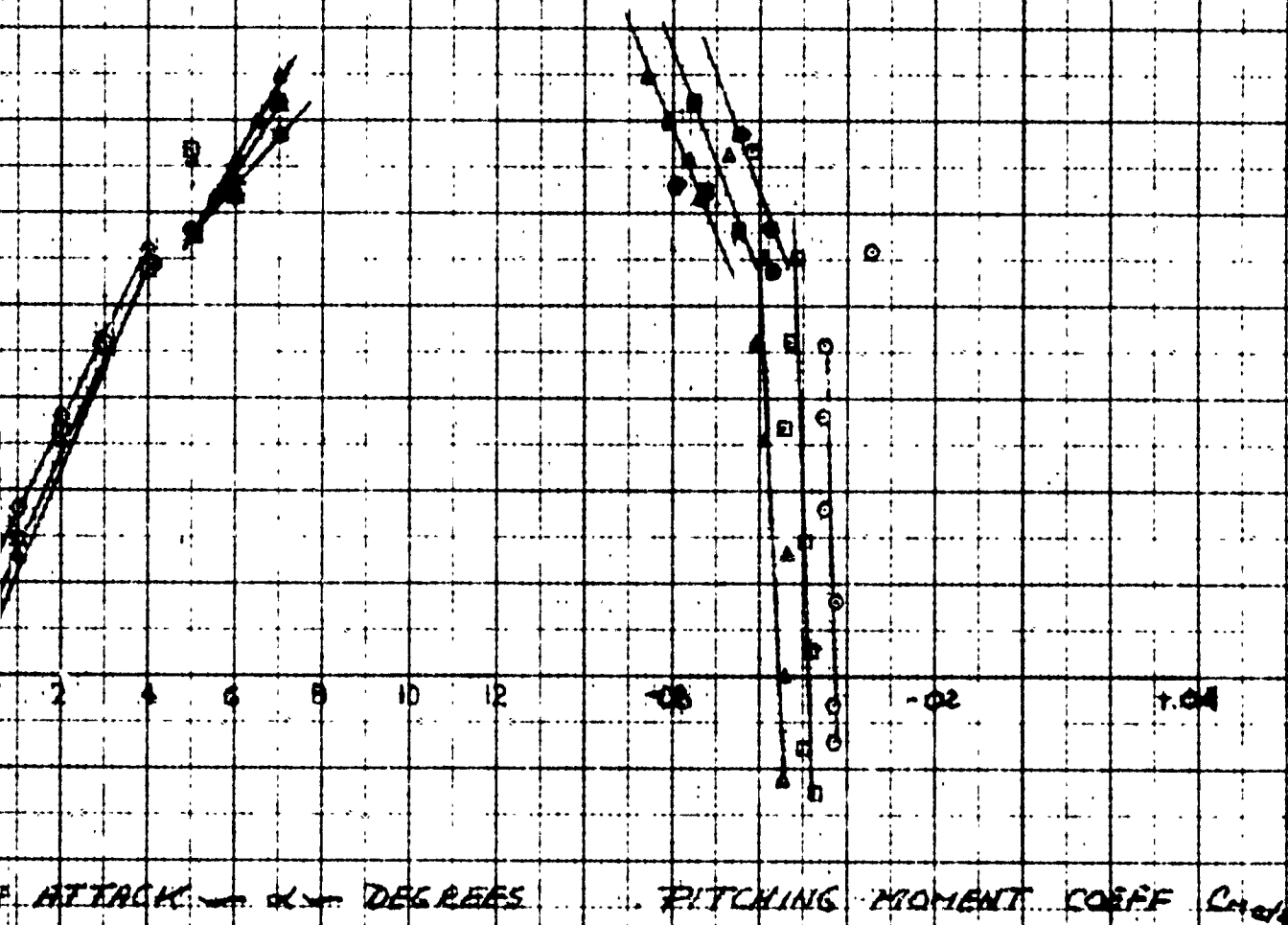
NOTE:
DARKENED SYMBOLS INDICATE
VENTED FLOW

WHIRLING TANK TEST NO. 206

PUSHIES NO. 4

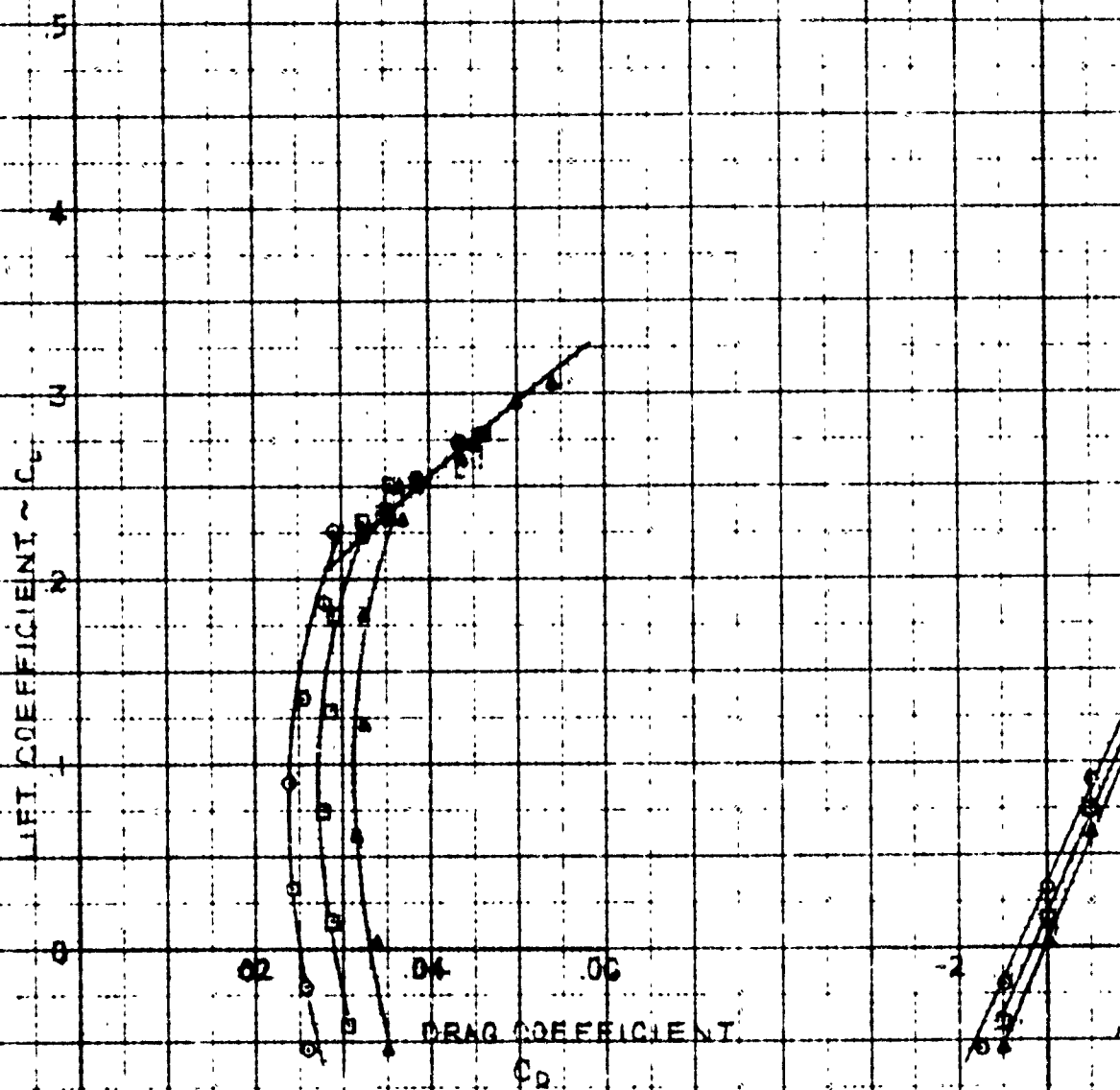
50 KTS

M.A.S. 144.51



FROJDE NE	SYMBOL	dist	Construct	CAV. No.
6.2713	•	.75	0025	251
5.9682	•	1.00	0039	264
4.9141	•	1.25	0051	279

NOTE



GRUMMAN AIRCRAFT EN

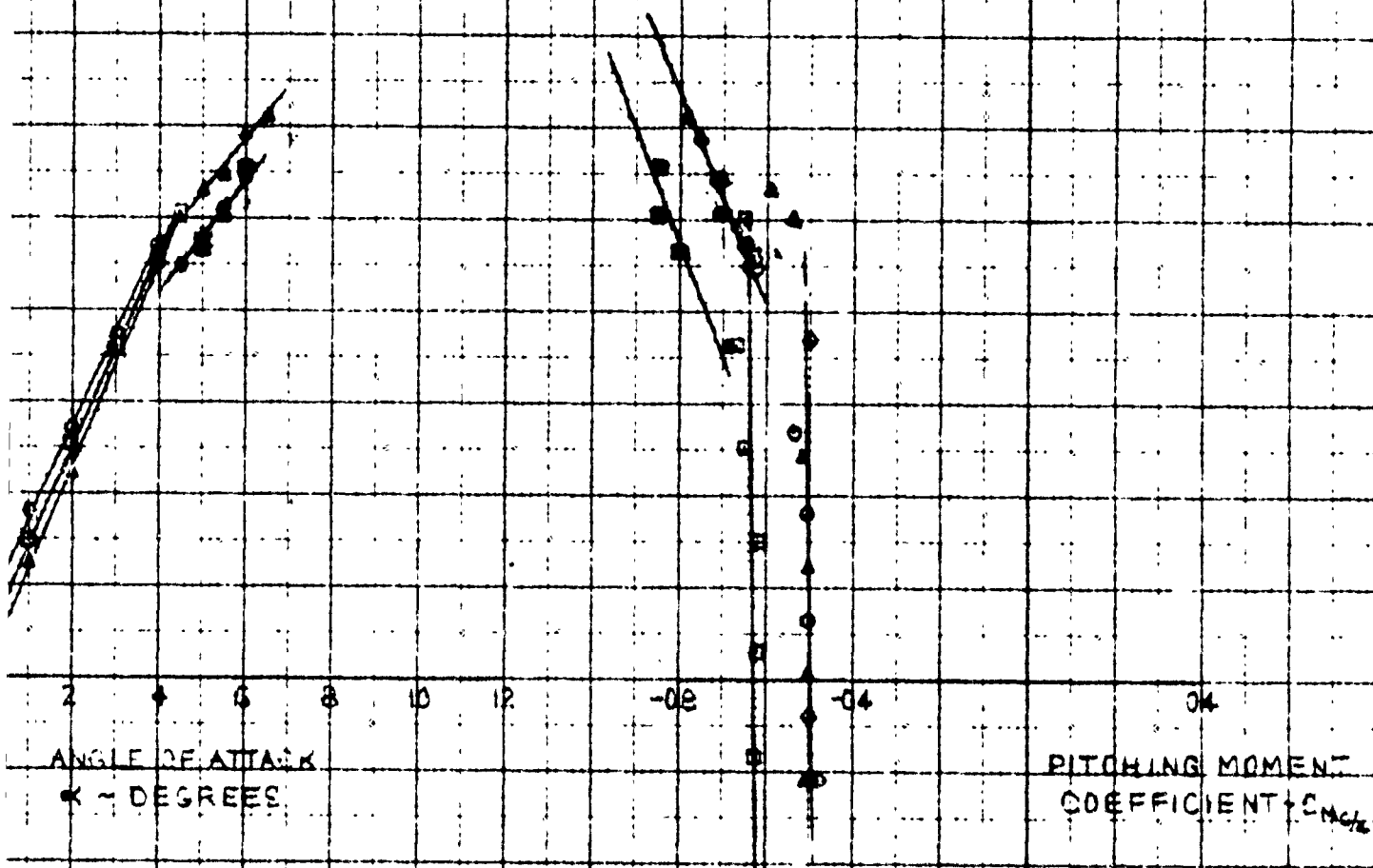
WHIRLING TANK TEST NO. 206

BUSHIPS NO. 4

30 KTS

M.A.C. \pm .144 FT.

NOTE:
1. DARKENED SYMBOLS INDICATE
VENTED FLOW



FROUDE No.	SYMBOL	d/c	C_D STRUT	CAY. No.
6.2818	○	75	.0026	.196
5.4682	□	100	.0037	.211
4.9141	▽	125	.0053	.228

LIFT COEFFICIENT $\sim C_L$

DRAO COEFFICIENT
 C_D

WHIRLING TANK TEST NO. 206

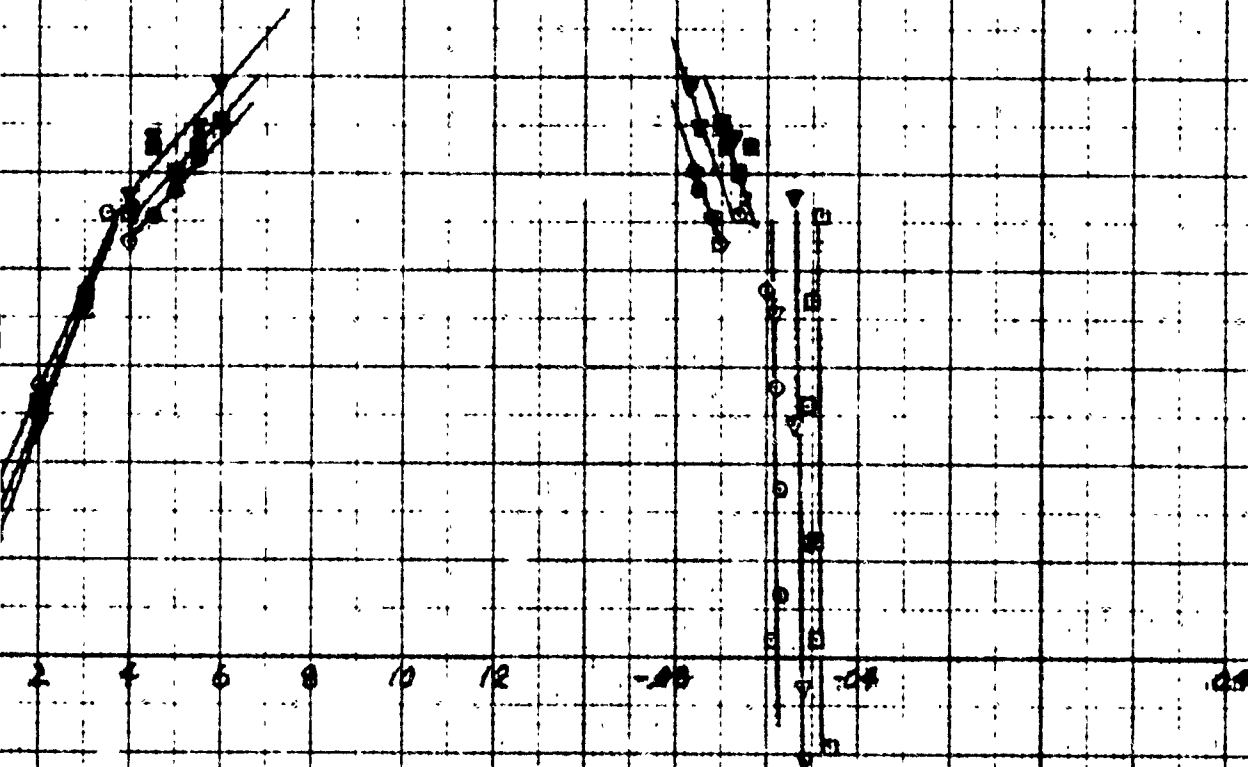
BU SHIPS NO. 4

70 MTS.

M.A.C. = 144 FT

NOTE:

1. DARKENED SYMBOLS INDICATE
VENTED FLOW.



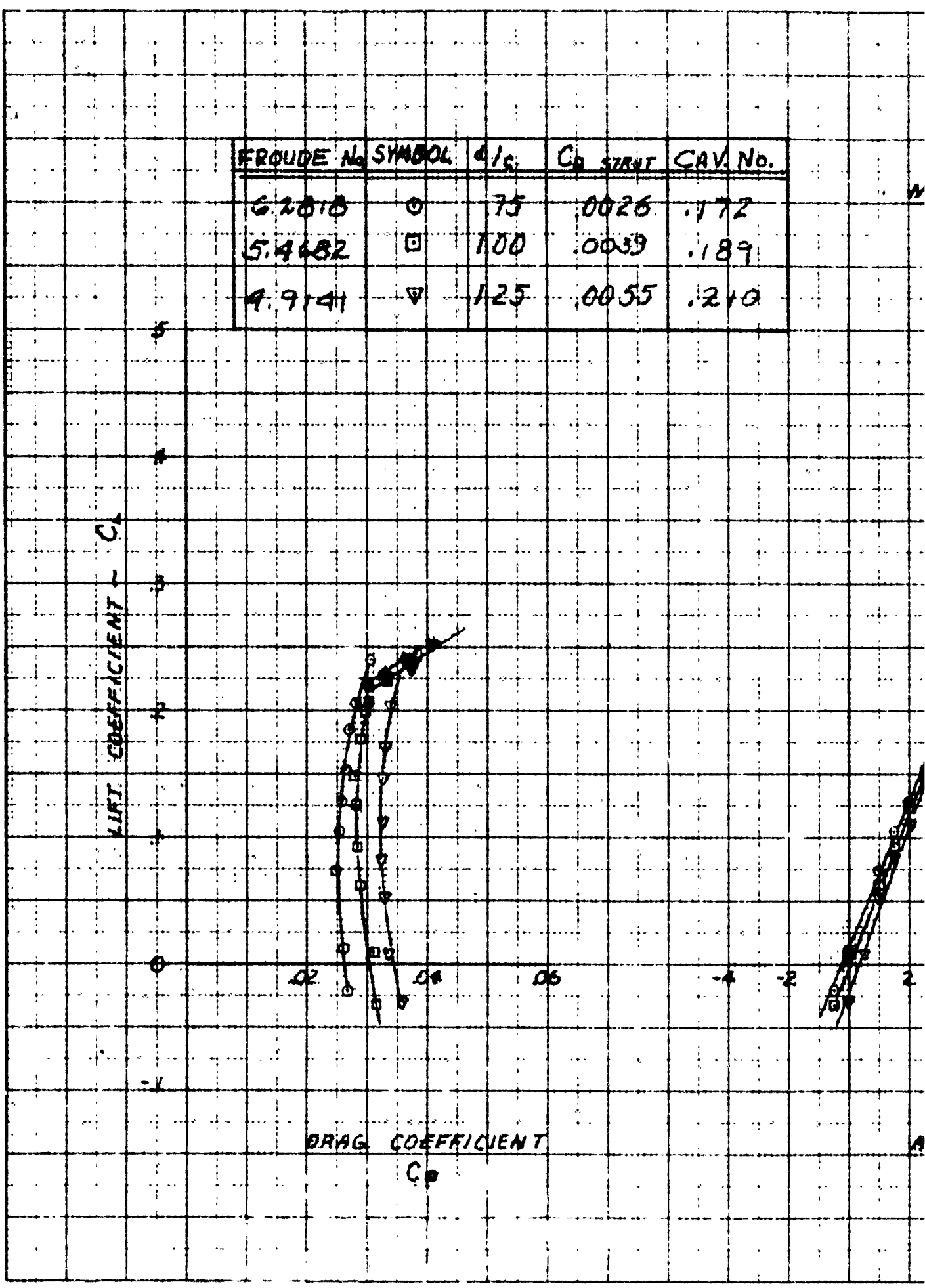
ANGLE OF ATTACK
 α - DEGREES

PITCHING MOMENT
COEFFICIENT C_m

FROUDE No	SYMBOL	4/5	C _D STRUT	CAV. No.
6.2818	○	75	.0026	.172
5.4482	□	100	.0039	.189
4.9141	▽	125	.0055	.210

LIFT COEFFICIENT - C_L

DRAG COEFFICIENT
C_D



WHIRLING TANK TEST NO. 206

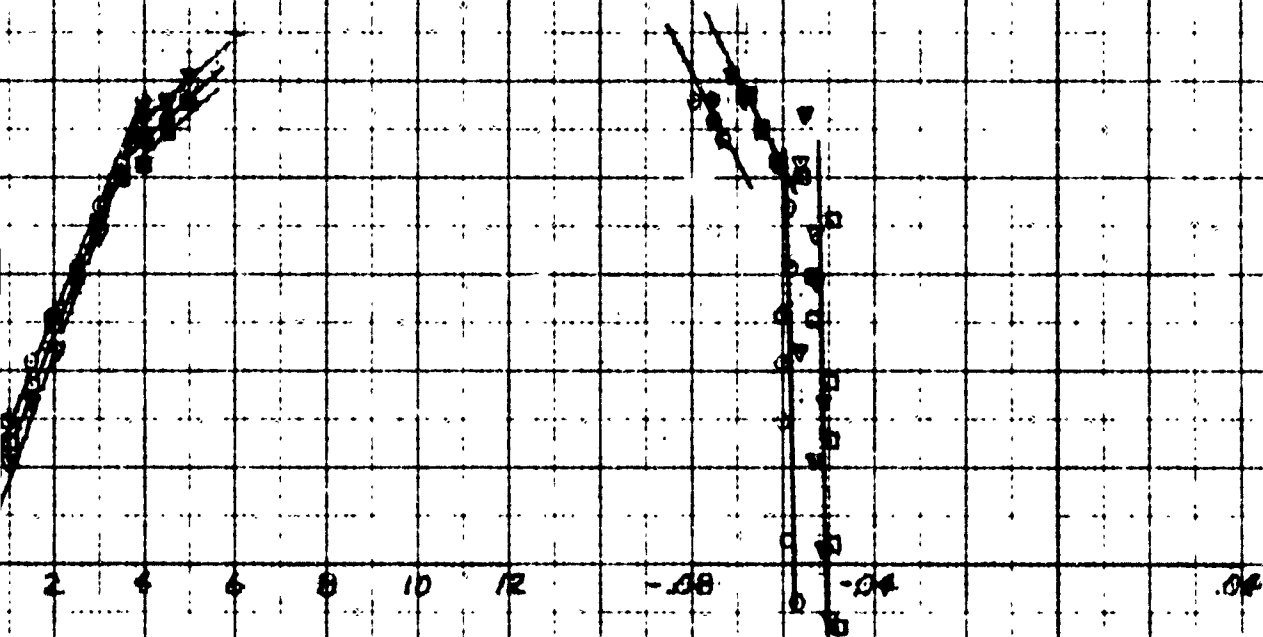
BU SHIPS NO. 4

80 KTS

MAC = 144 FT

NOTE:

1. DARKENED SYMBOLS INDICATE
VENTED FLOW.



ANGLE OF ATTACK
 α - ATTACK

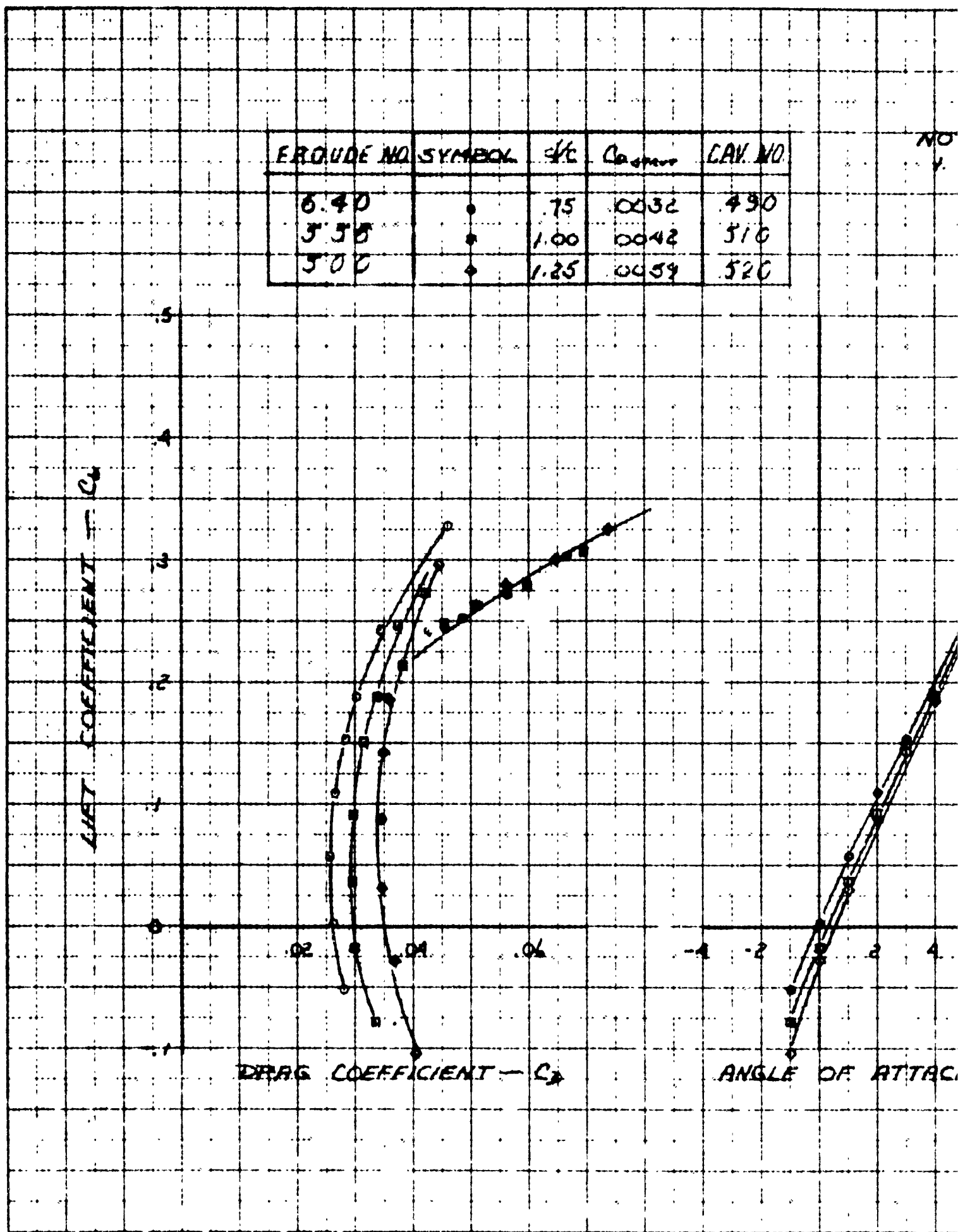
PITCHING MOMENT
COEFFICIENT - C_m

EROUDE NO	SYMBOL	$\frac{V}{V_c}$	$C_{L_{max}}$	CAY. NO.	NO.
8.40	•	.75	.0032	490	
5.55	•	1.00	.0042	510	
5.00	•	1.25	.0059	520	

LIFT COEFFICIENT - C_L

DRAG COEFFICIENT - C_D

ANGLE OF ATTACK



PAGE IV 24

WHIRLING TANK TEST NO 208

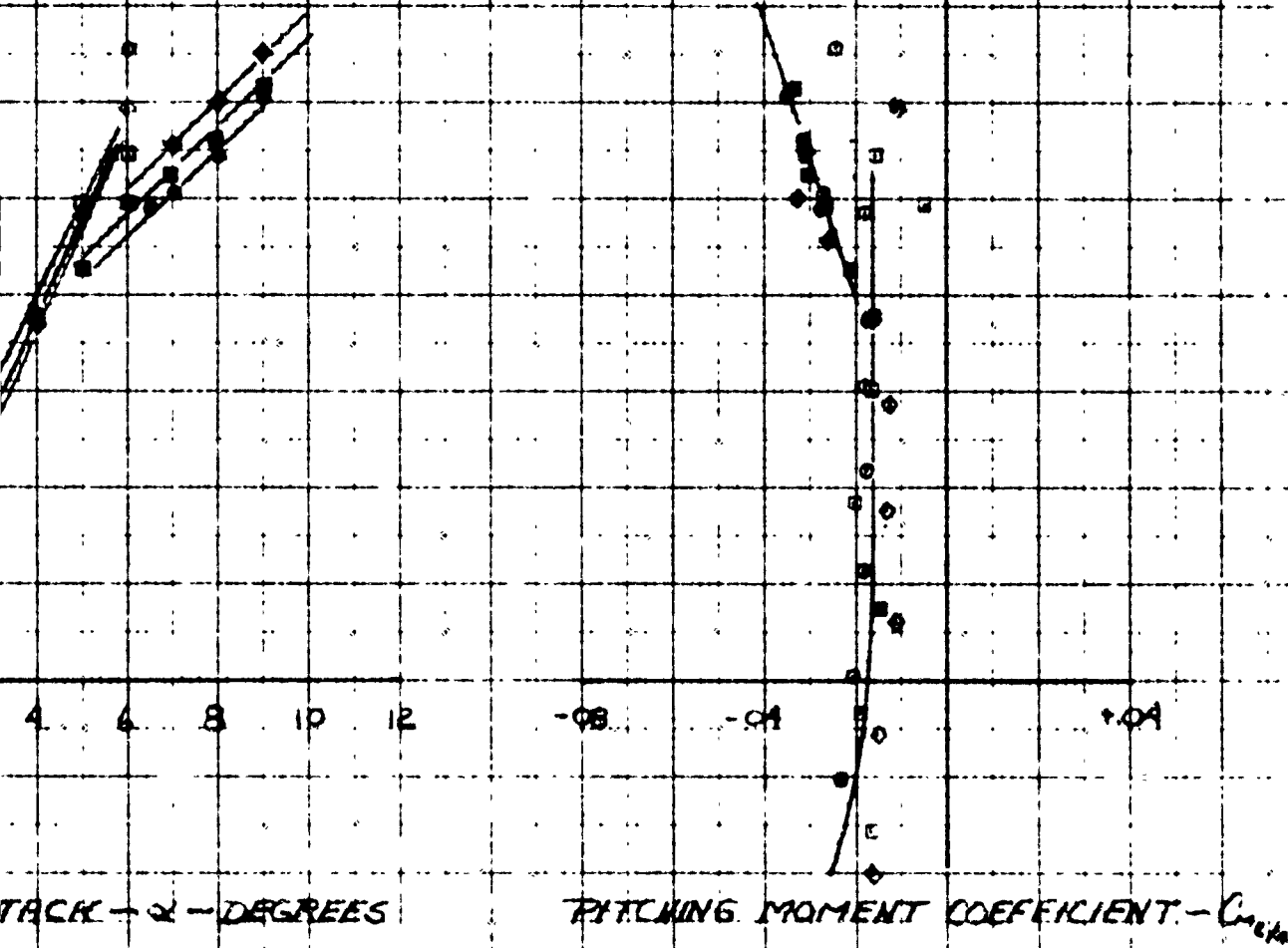
NO 4. 84959

MODEL NO 5

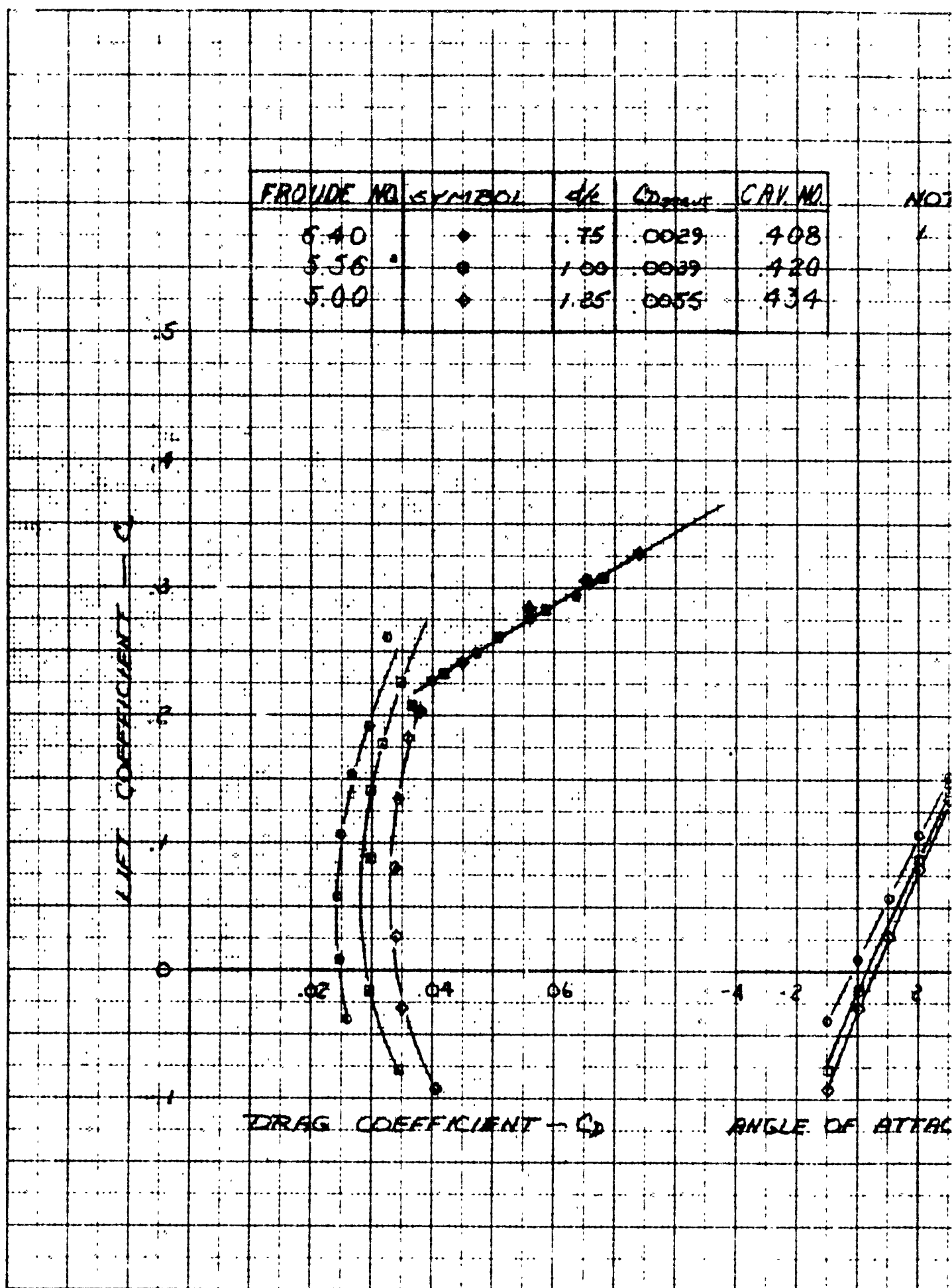
40 KTS

M.R.C. = 1.25 FT.

NOTE
1. DARKENED SYMBOLS
INDICATE VENTED FLOW



22.0



WHIRLING TANK TEST NO. 208

NO 4s 84454

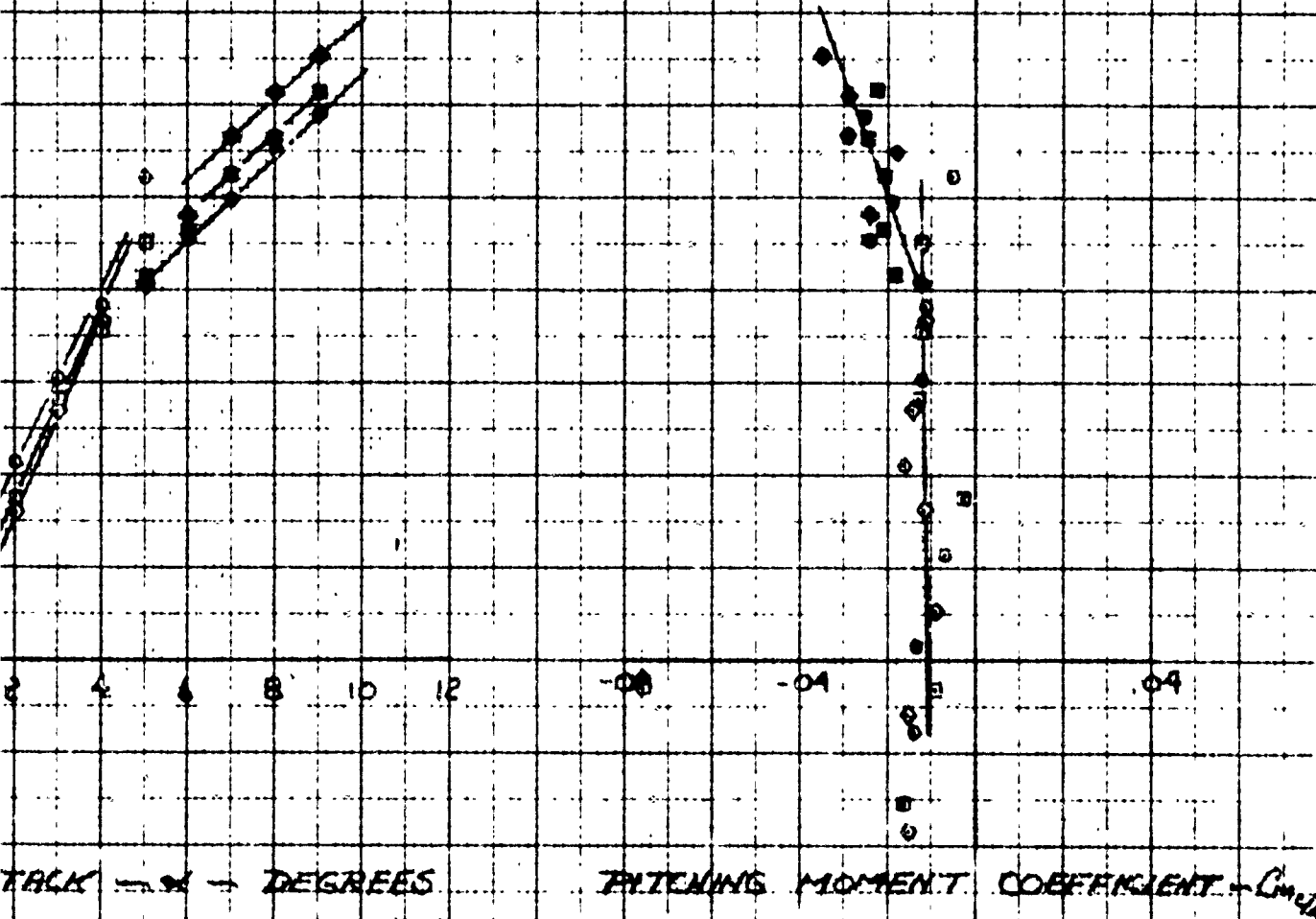
MODEL NO 5

45 KTS

M A C - 137 FT

NOTE:

1. DARKENED SYMBOLS
INDICATE VENTED FLOW



ME ALBANY 1947 5200

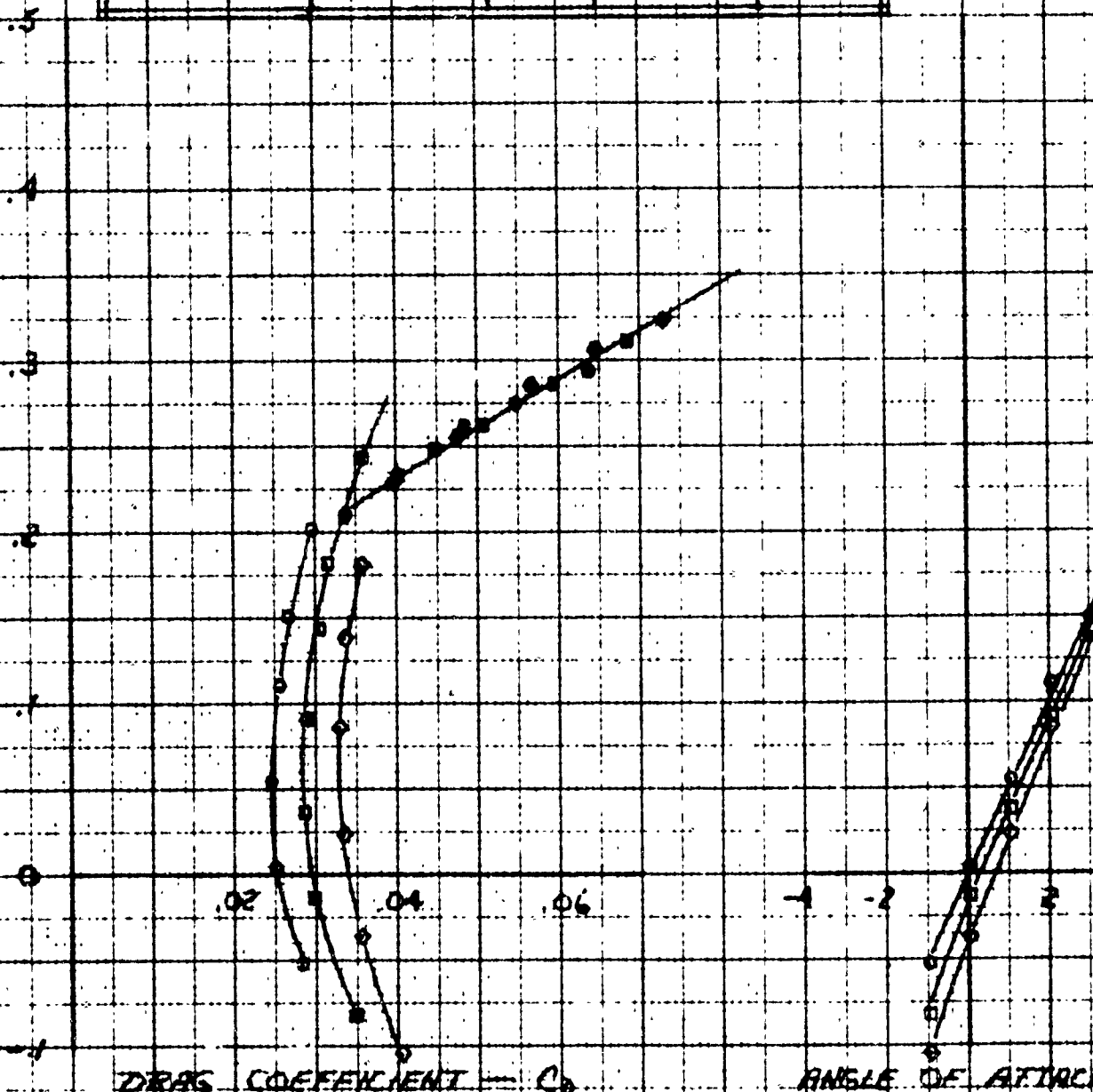
FRONTS NO.	SYMBOL	d/A	C _D (max)	CAV. NO.
6.40	◆	.75	.0028	340
5.56	●	1.00	.0039	350
5.00	◆	1.25	.0055	370

NOTE:
1. DAPK
INDIC

LIFT COEFFICIENT - C_L

DRAG COEFFICIENT - C_D

ANGLE OF ATTACK



WHITLING TANK TEST, NO 202

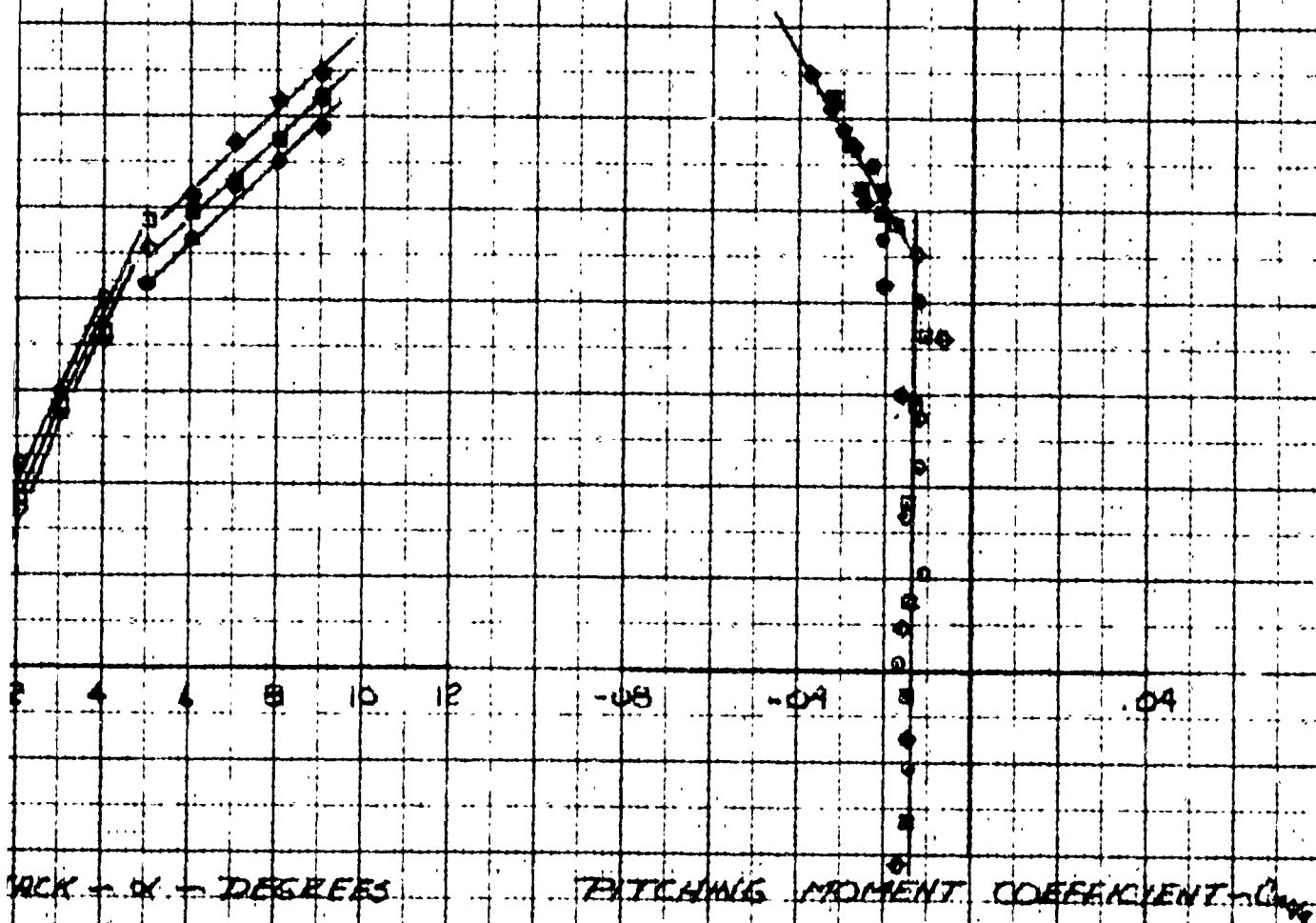
NO 63 84454

MODEL NO 5

50 KTS

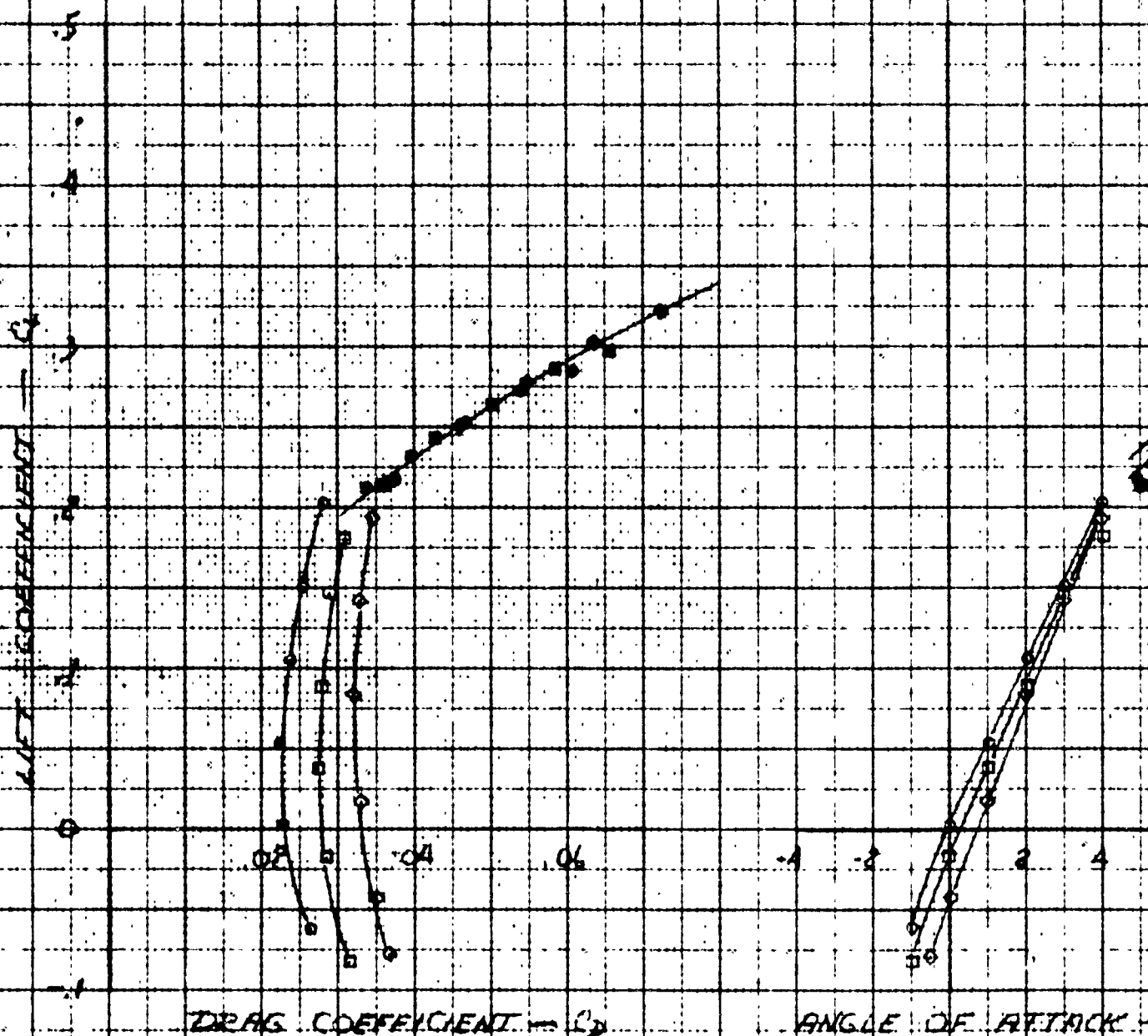
W.B.S. = 1.37 FT

OPENED POINTS
INDICATE VENTED FLOW



FROUDE NO.	SYMBOL	CA	CD AREA	CAR. NO.
6.40	◆	.75	0029	250
5.56	●	1.00	0039	260
5.00	◆	1.25	0055	275

NOTE:
1. DARKER
INDICATE



WHIRLING TANK TEST NO. 20 B.

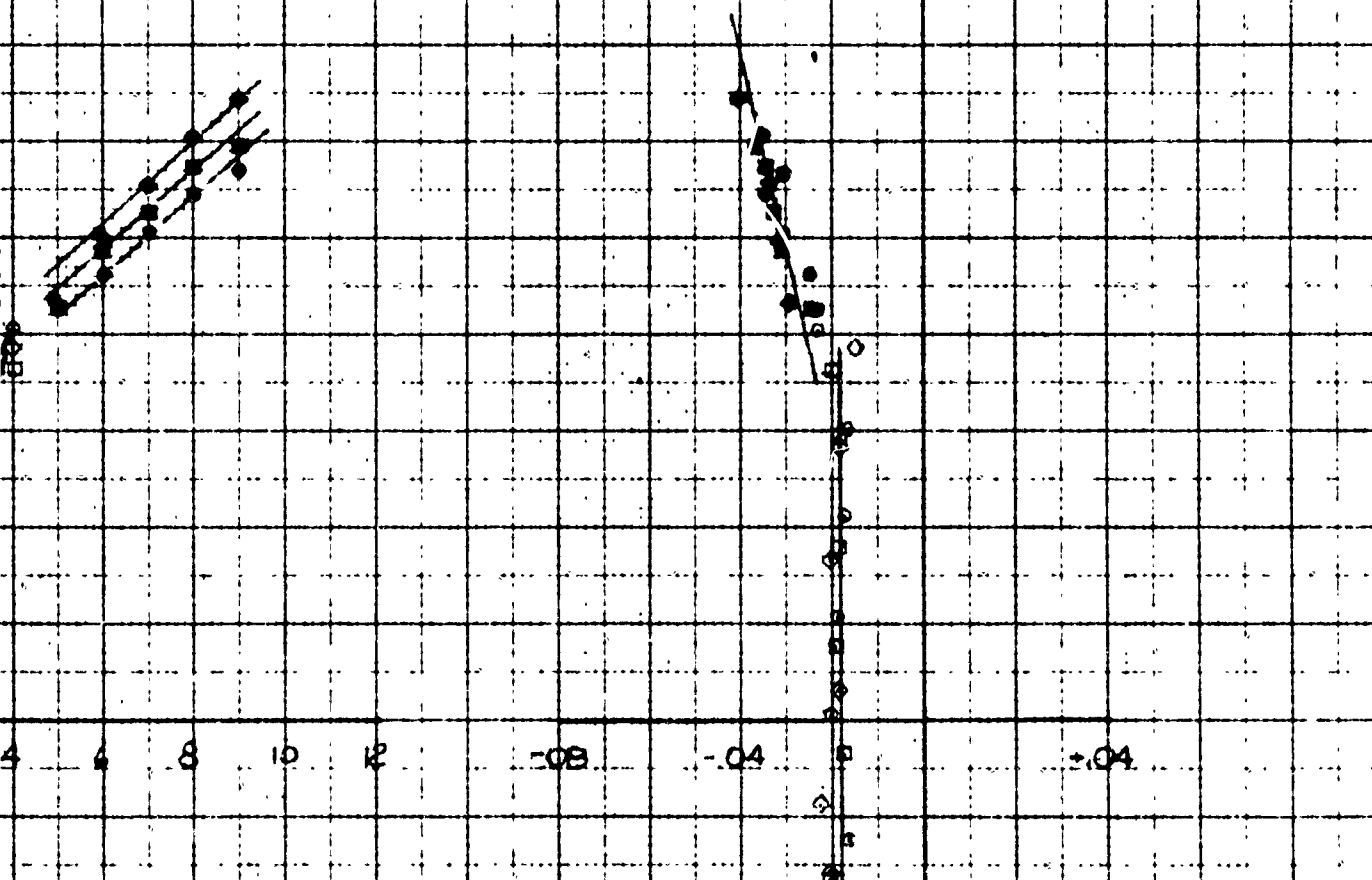
NO. 43 84454

MODEL NO. 5

60 KTS

M.H.C. = 137 FT.

OPENED SYMBOLS
CREATE VENTED FLOW

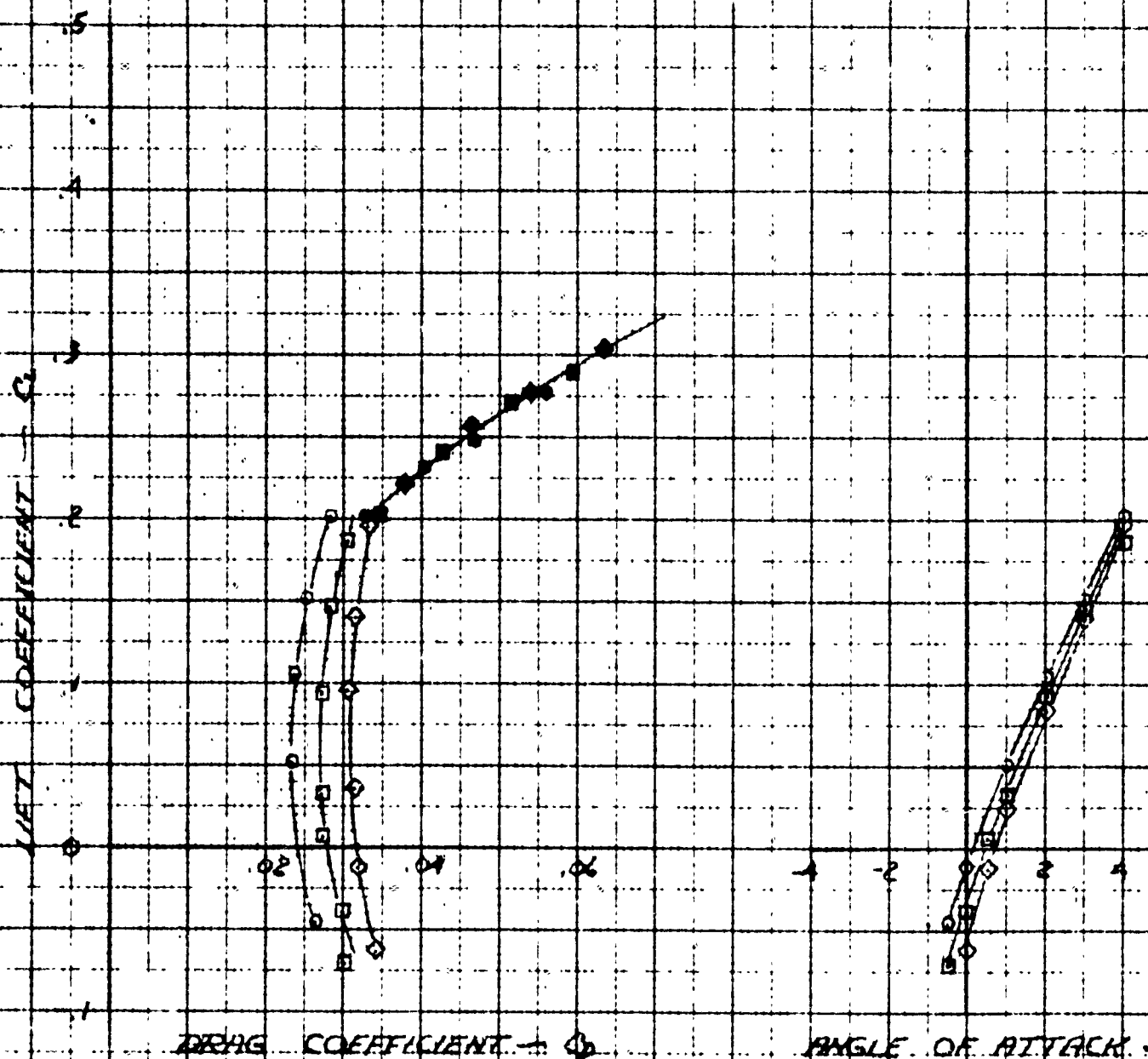


ANGLE - α - DEGREES

PITCHING MOMENT COEFFICIENT - C_m

EXPOUSE NO.	CAR. NO.	SYMBOL	d/c	C_{Lmax}
640	196	◆	75	0030
556	210	■	1.00	0041
500	225	◇	1.25	0059

NGTE:
1. 72
IN



WHIRLING TANK TEST NO. 208

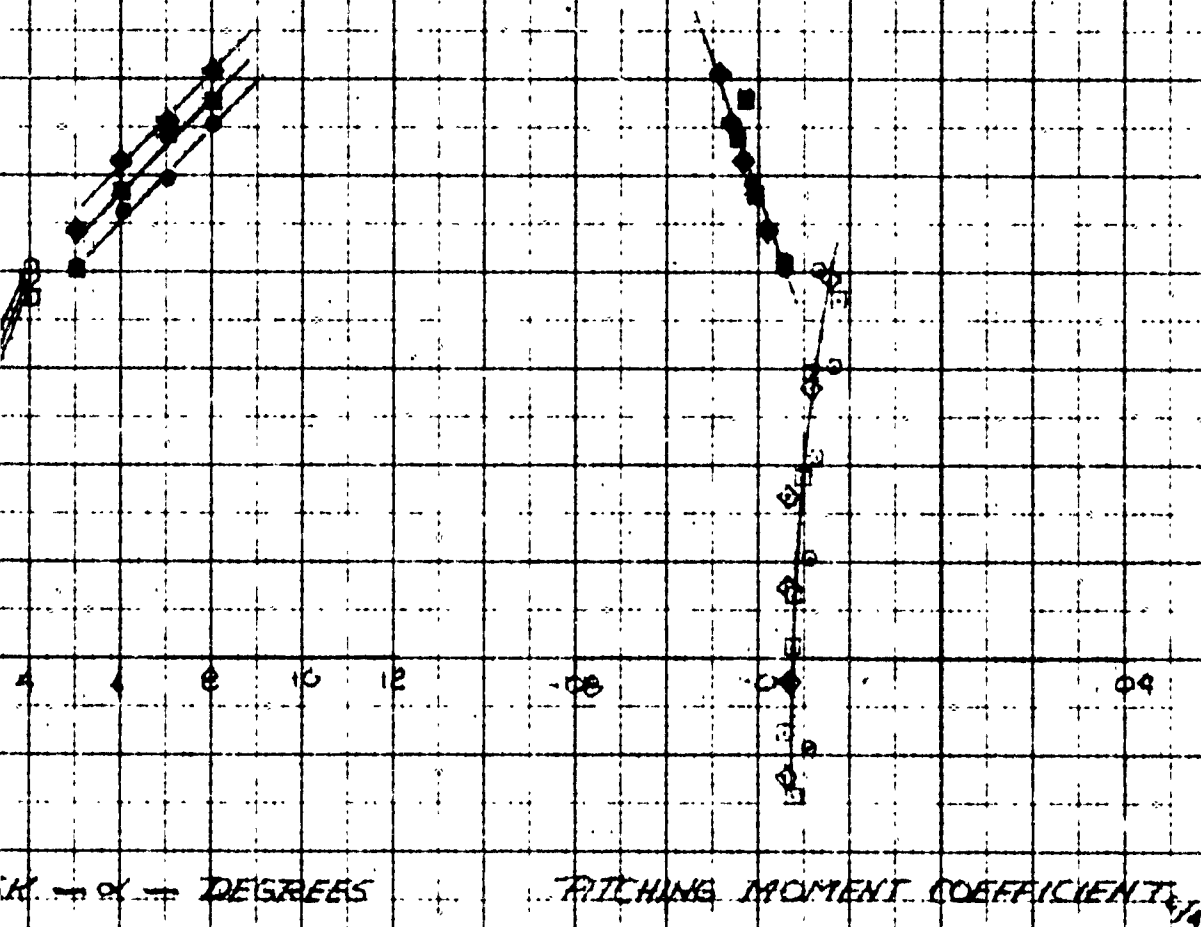
NO. 6: 84954

MODEL NO. 5

7.0 KTS

W.H.C. 139 FT

TE:
DARKENED SYMBOLS
INDICATE VENTED FLOW

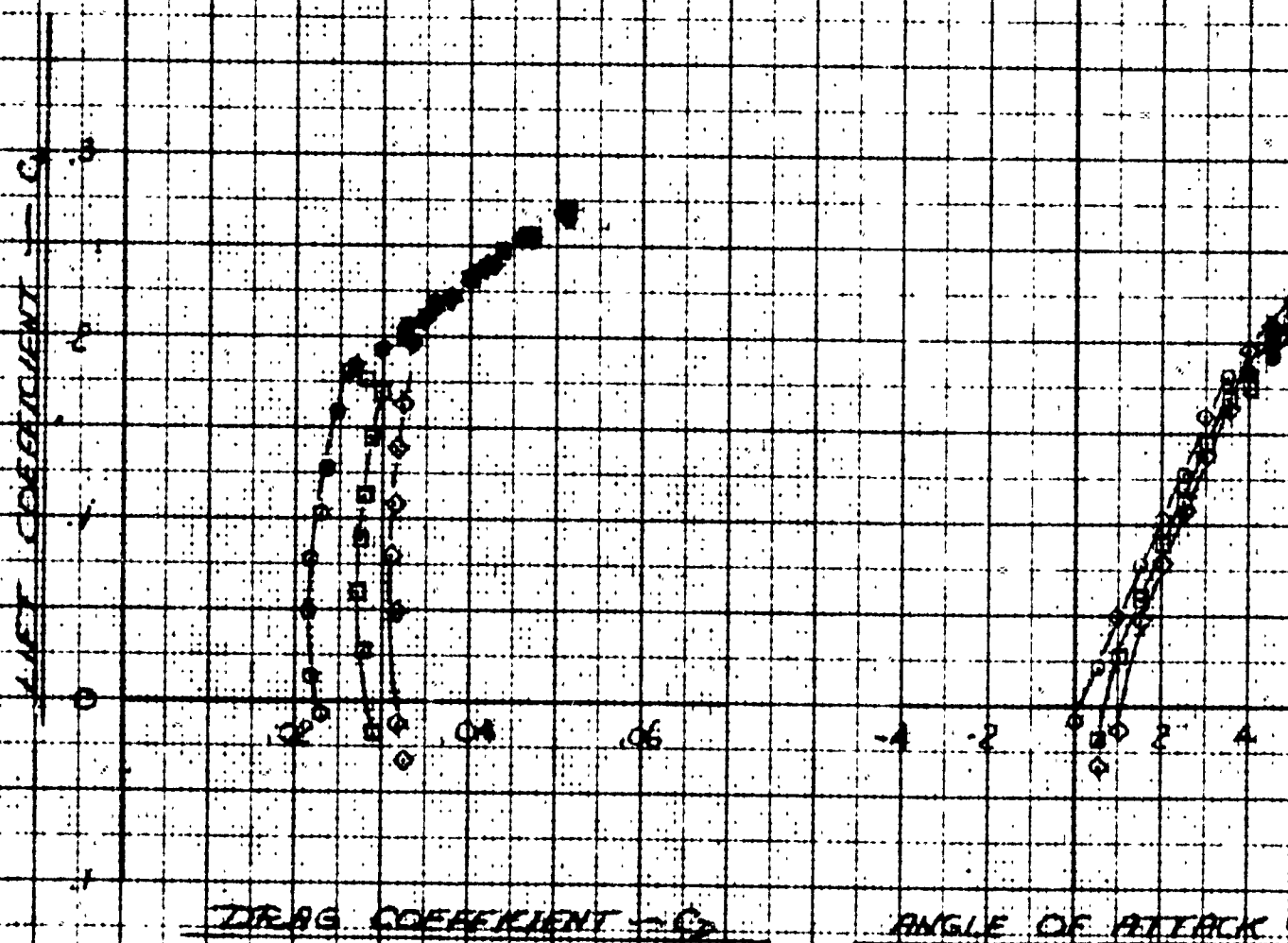


α - DEGREES

PITCHING MOMENT COEFFICIENT $C_{m, \frac{1}{4}}$

FRONDE NO.	SYMBOL	d/c	C _{drag}	CAV. NO.
5.40	◆	.75	0032	175
5.56	●	1.00	0040	189
5.00	◆	1.25	0056	203

NOTE:
1. DARK
INDICA



WHIRLING TANK TEST NO. 208

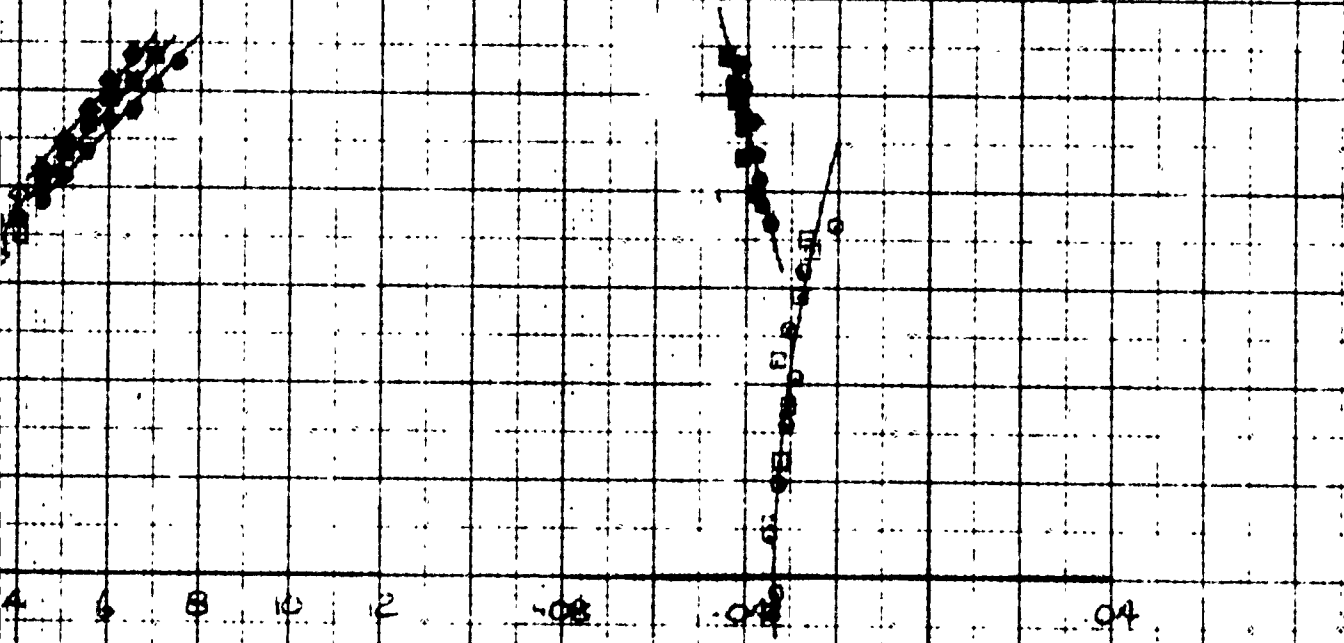
NO. 42 B4454

MODEL NO. 5

80 KTS

W.T.S. = 13.7 FT.

OPENED SYMBOLS
INDICATE VENTED FLOW

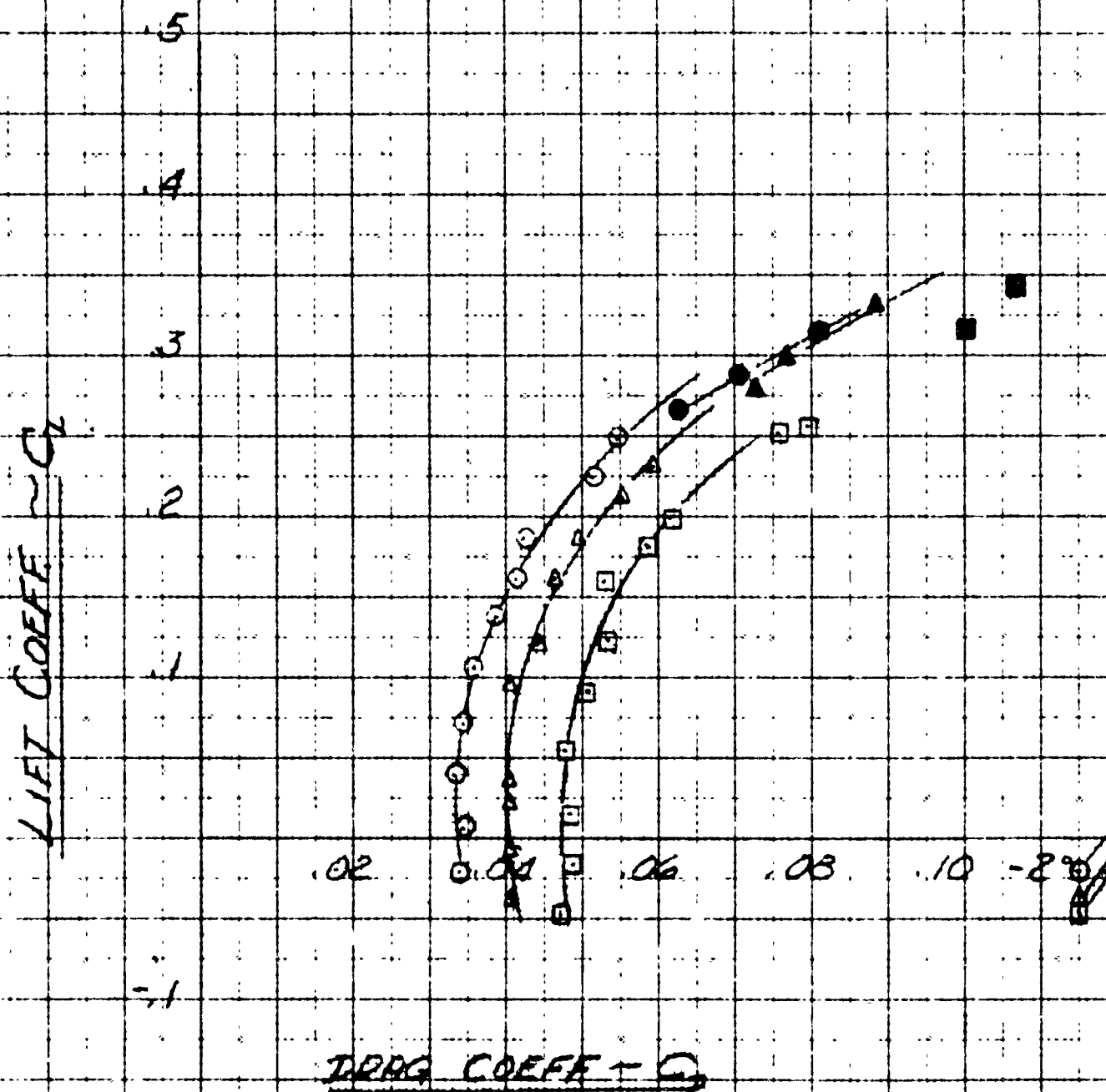


ANGLE - DEGREES

PITCHING MOMENT COEFFICIENT - $C_{m\alpha}$

EROUDE No	CAV. No	SYMBOL	%	C _D STOUT
4.8189	.528	○	0.75	.0038
4.2032	.553	▲	1.00	.0069
3.7859	.585	□	1.25	.0101

NOTE:
1. DARI
IN



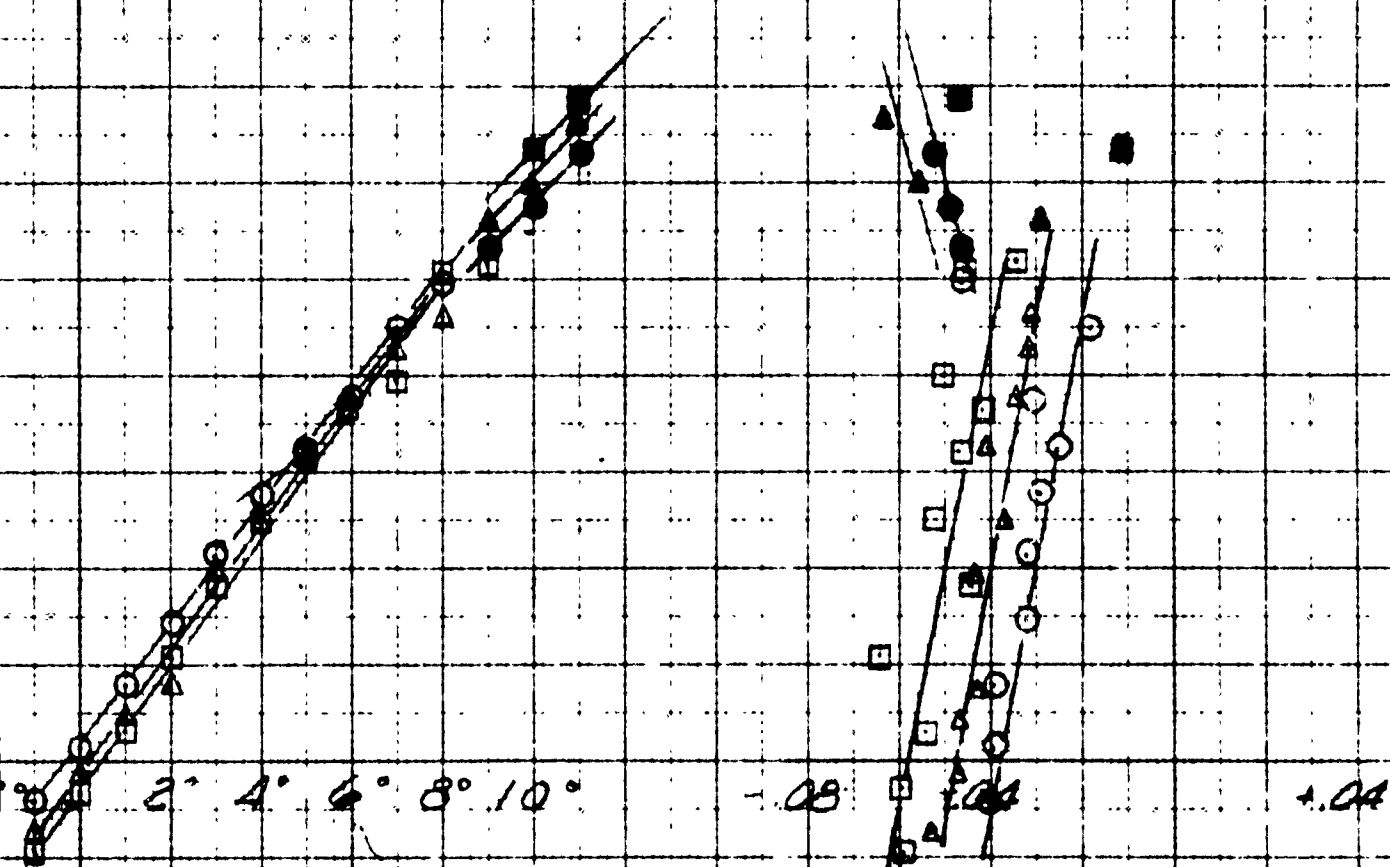
WHIRLING TANK TEST NO 20 F

BUSHIPS NO 6

40 KTS

MBC = 250 FT

ENLARGED SYMBOLS
INDICATE VENTED
FLOW.

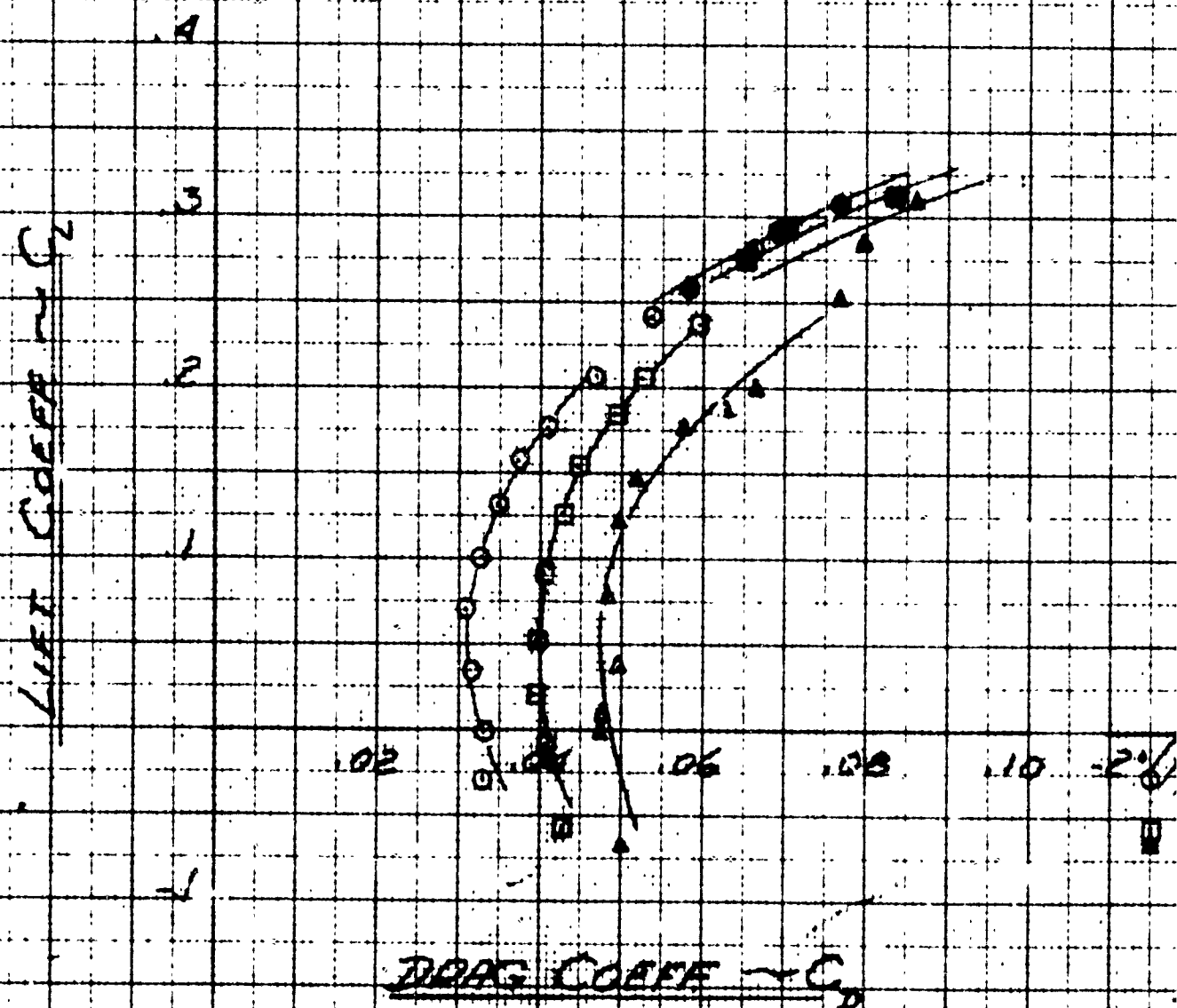


ANGLE ° ATTACK
X - DEGREES

PITCHING MOM.
COEFF. C_m

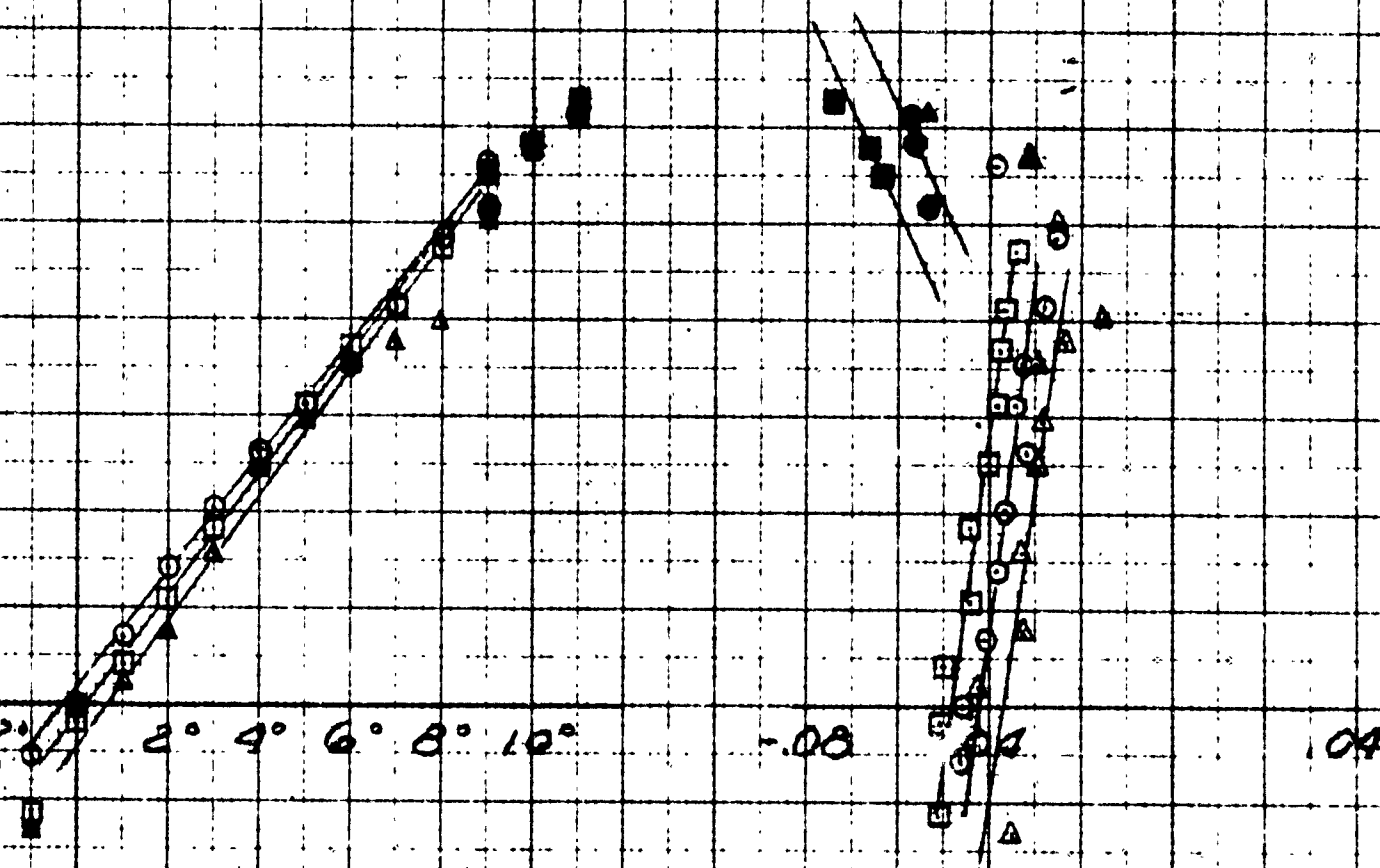
FROUDE No	CAN. No	SYMBOL	α/c	C_D STRUT
4.8189	435	○	0.75	.0046
4.2032	458	□	1.00	.0072
3.7859	485	▲	1.25	.0103

NOTE
I.D.



WHIRLING TANK TEST NO. 20 F
BUSHIPS NO. 6
15 KTS

NOTE:
DARKENED SYMBOLS
INDICATE VENTED
FLOW.

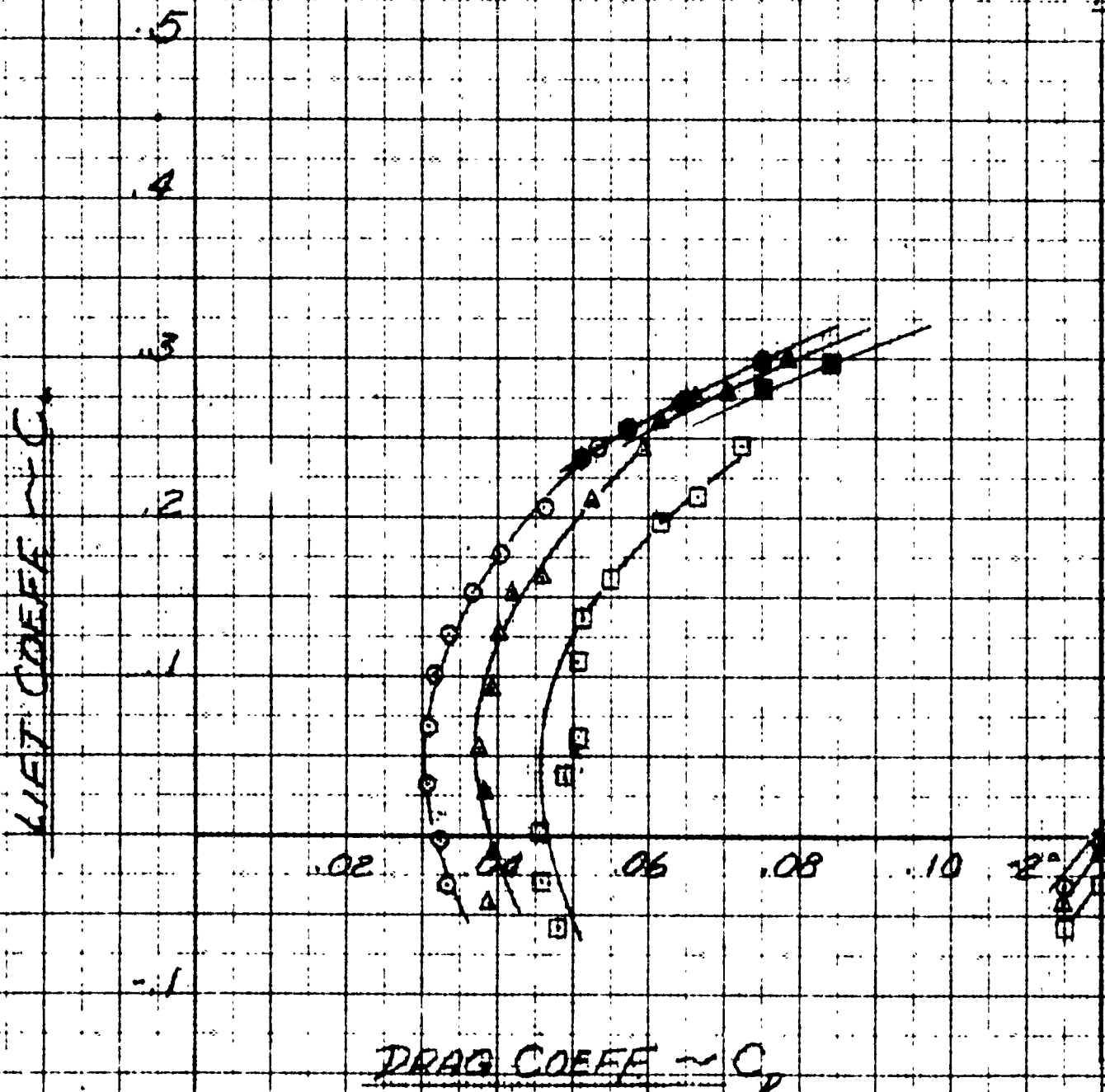


ANGLE OF ATTACK
X ~ DEGREES

PITCHING MOM.
COEFF ~ C_m^1/4

FRONTS No	CAV No	SYMBOL	%	C _D STRUT
4.8787	.365	○	0.75	0049
4.2032	.395	△	1.00	0074
3.7857	.420	□	1.25	0108

NOTE:
L. PARKER
INDIC



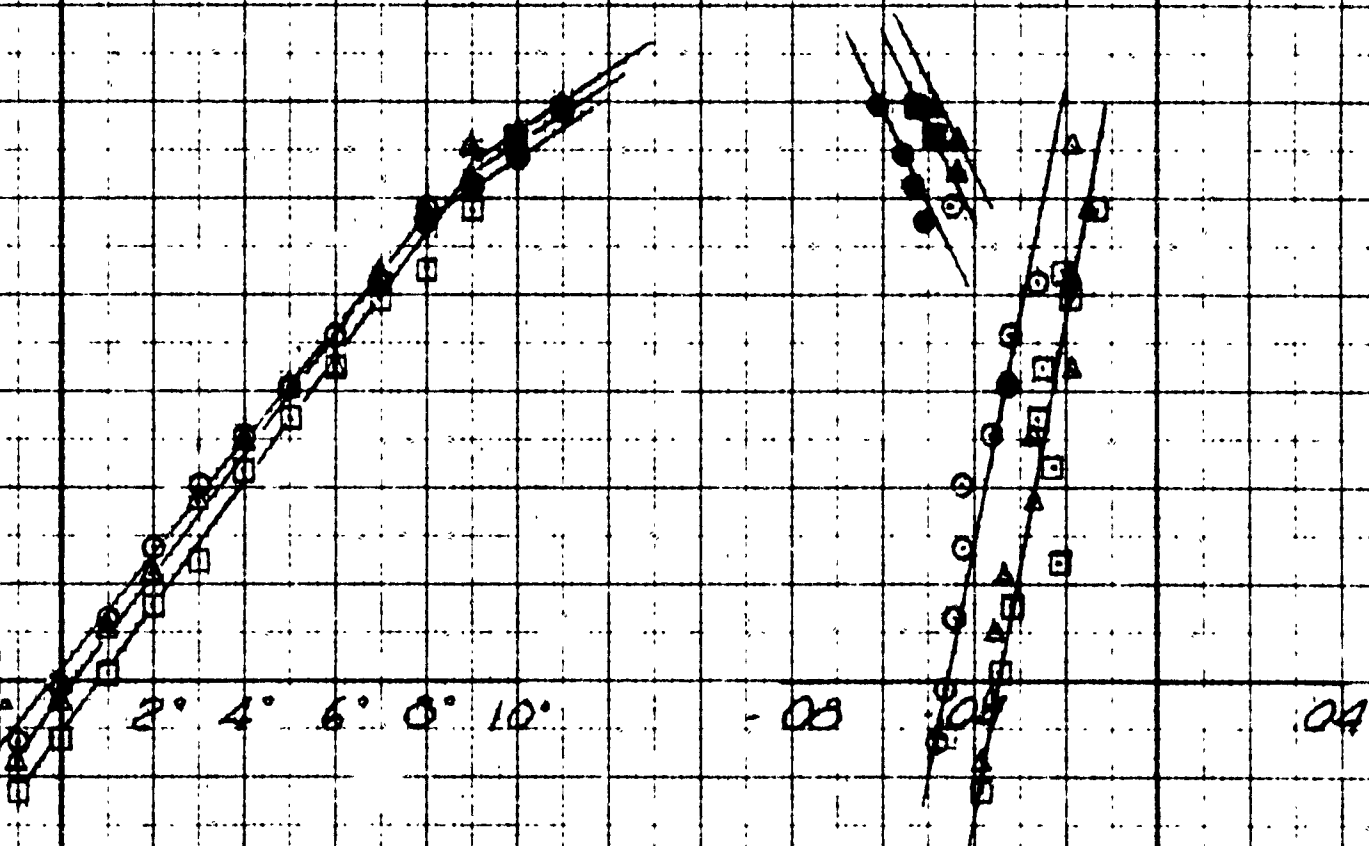
WHIRLING TANK TEST NO. 20 F

BUSHIPS NO. 6

50 KTS.

MAL = 250 FT

BROKEN SYMBOLS
INDICATE VENTED
FLOW.



ANGLE ° ATTACK
& ~ DEGREES

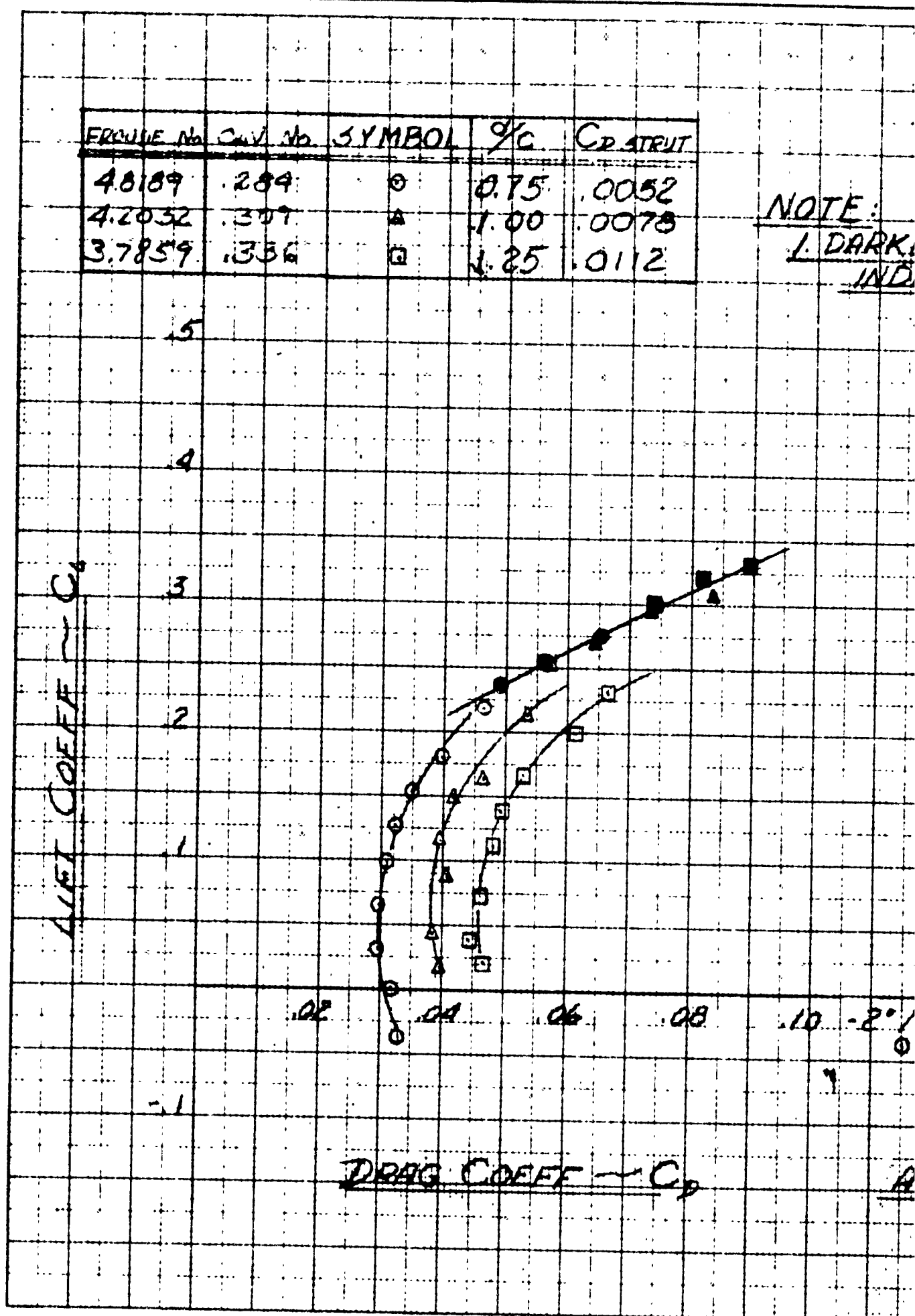
PITCHING MOM
COEFF ~ C_M

FRIDGE No	CAV No	SYMBOL	%C	C _D STRUT
4.8189	.284	○	0.75	.0052
4.2032	.307	△	1.00	.0078
3.7859	.336	□	1.25	.0112

NOTE:
1. DARK
IND

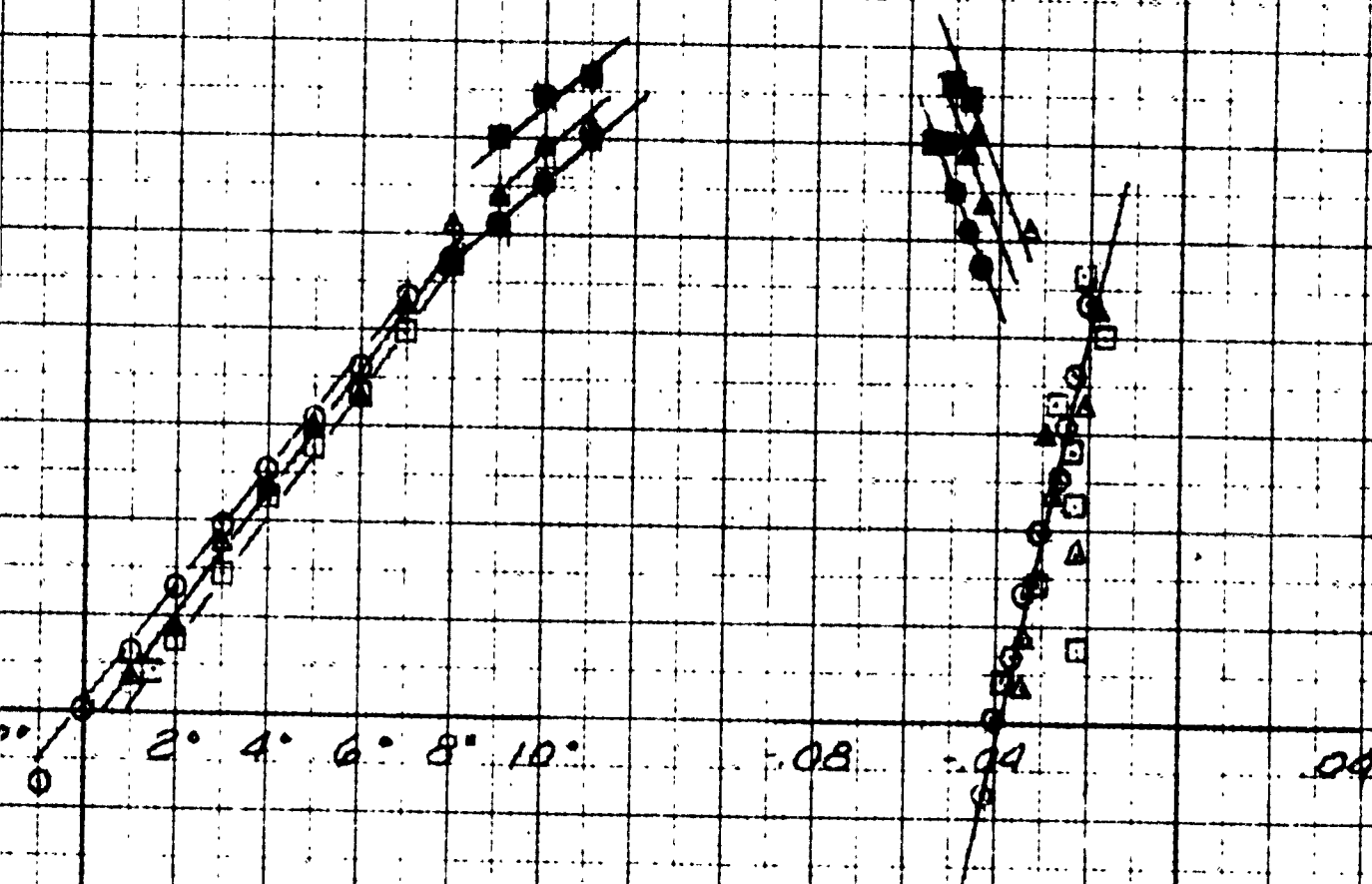
LIFT COEFF - C_L

DRAG COEFF - C_D



WHIRLING TANK TEST NO 20F
BUSHIPS NO. 6
60 KTS.

BROKEN SYMBOLS
INDICATE VENTED
FLOW.



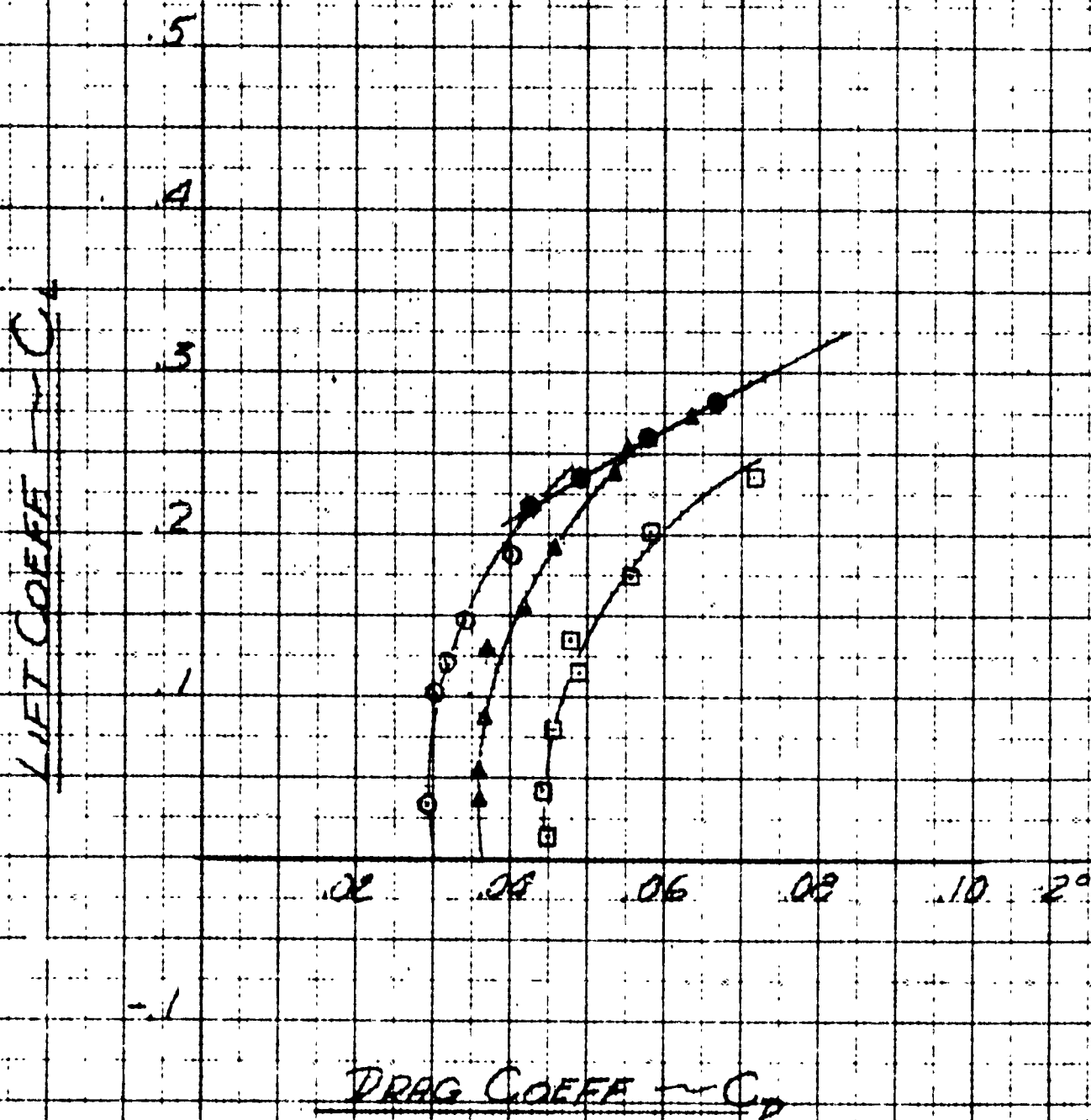
ANGLE OF ATTACK
X ~ DEGREES

PITCHING MOM
COEFF ~ C_m

PROBE NO.	CAL. NO.	SYMBOL	%C	C _D START
48189	231	○	0.75	.0051
42032	252	▲	1.00	.0076
37859	279	□	1.25	.0113

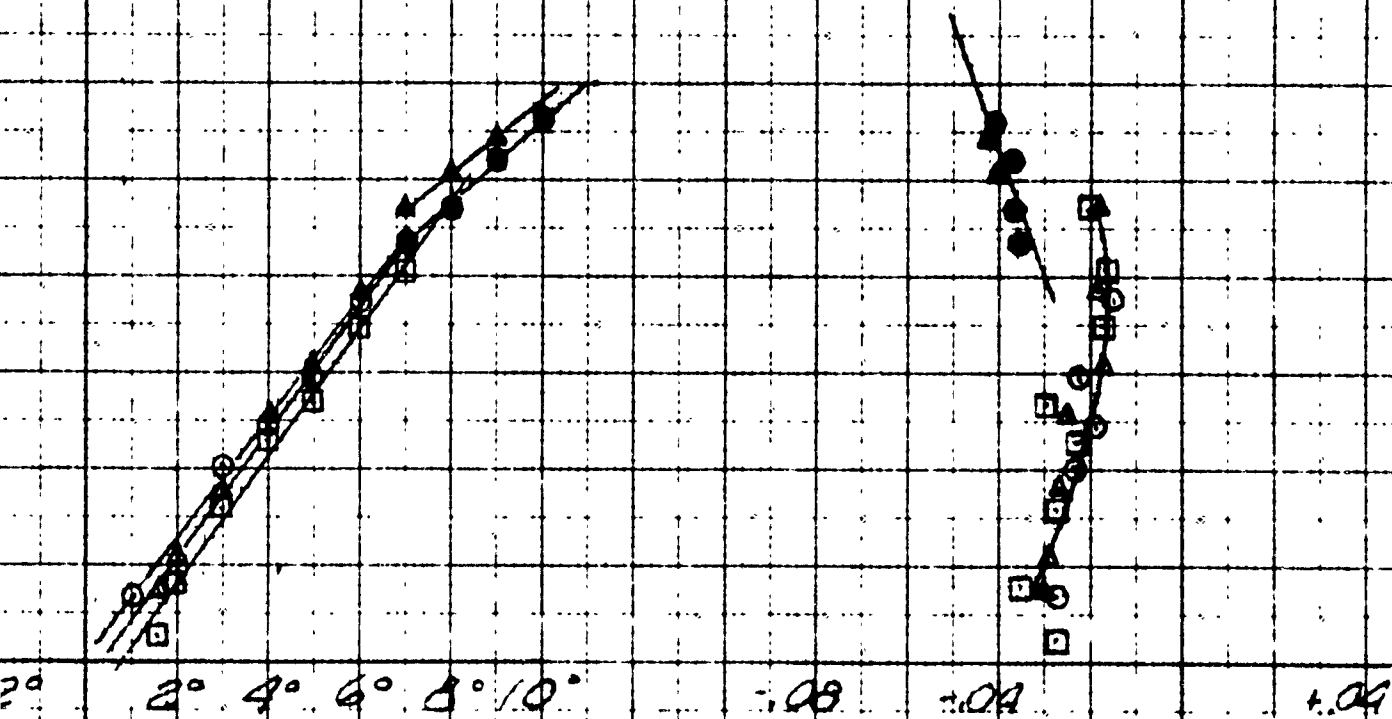
NOTE

LDA



WHIRLING TANK TEST NO. 20F
 BUSHIPS NO. 6
 70 KTS

NOTE:
 DARKENED SYMBOLS
 INDICATE VENTED
 FLOW.



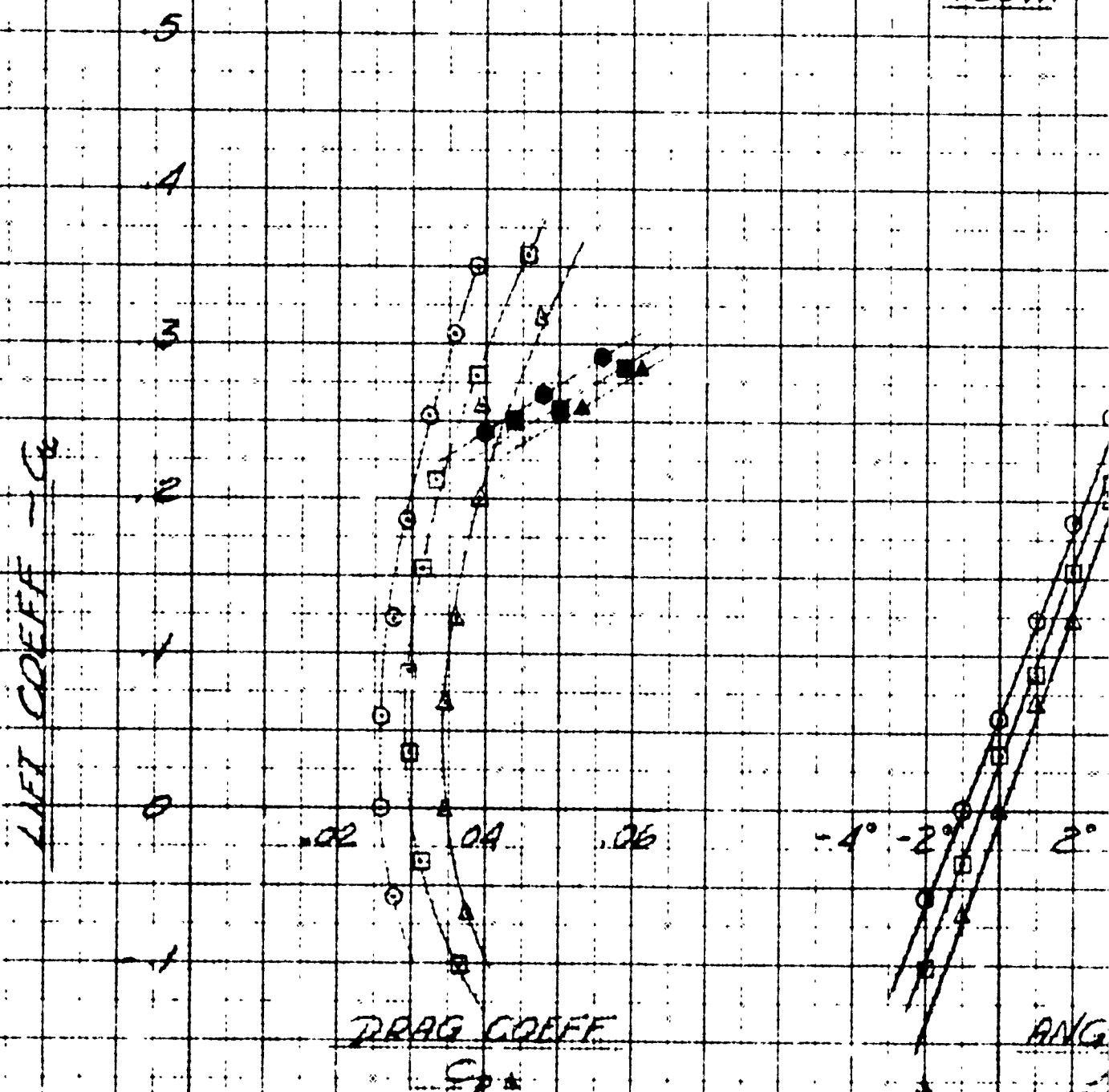
ANGLE OF ATTACK
 IN DEGREES

PITCHING MOM
 COEFF - $C_{M \frac{1}{4}}$

FRONDE No.	CLV. No.	SYMBOL	d/c
82260	478	○	.75
71414	483	□	1.00
64531	489	▲	1.25

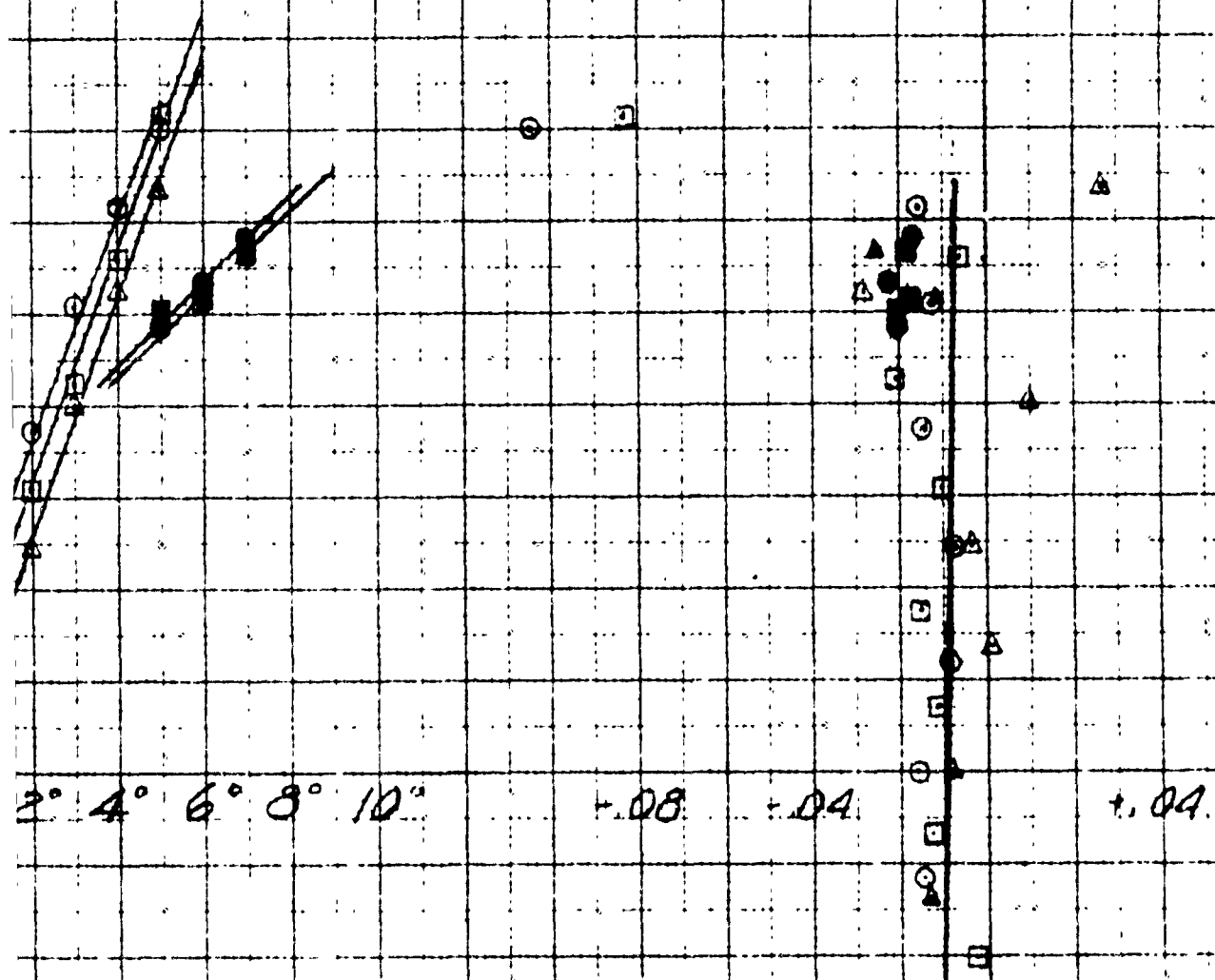
NOTE:

DARKENED \circ
INDICATE RE
FLOW.



WHIRLING TRANS TEST NO. 20D
 BUSHIPS NO 7
 40 KTS
 M.A.C. = .033 FE

SYMBOLS
 VENTED



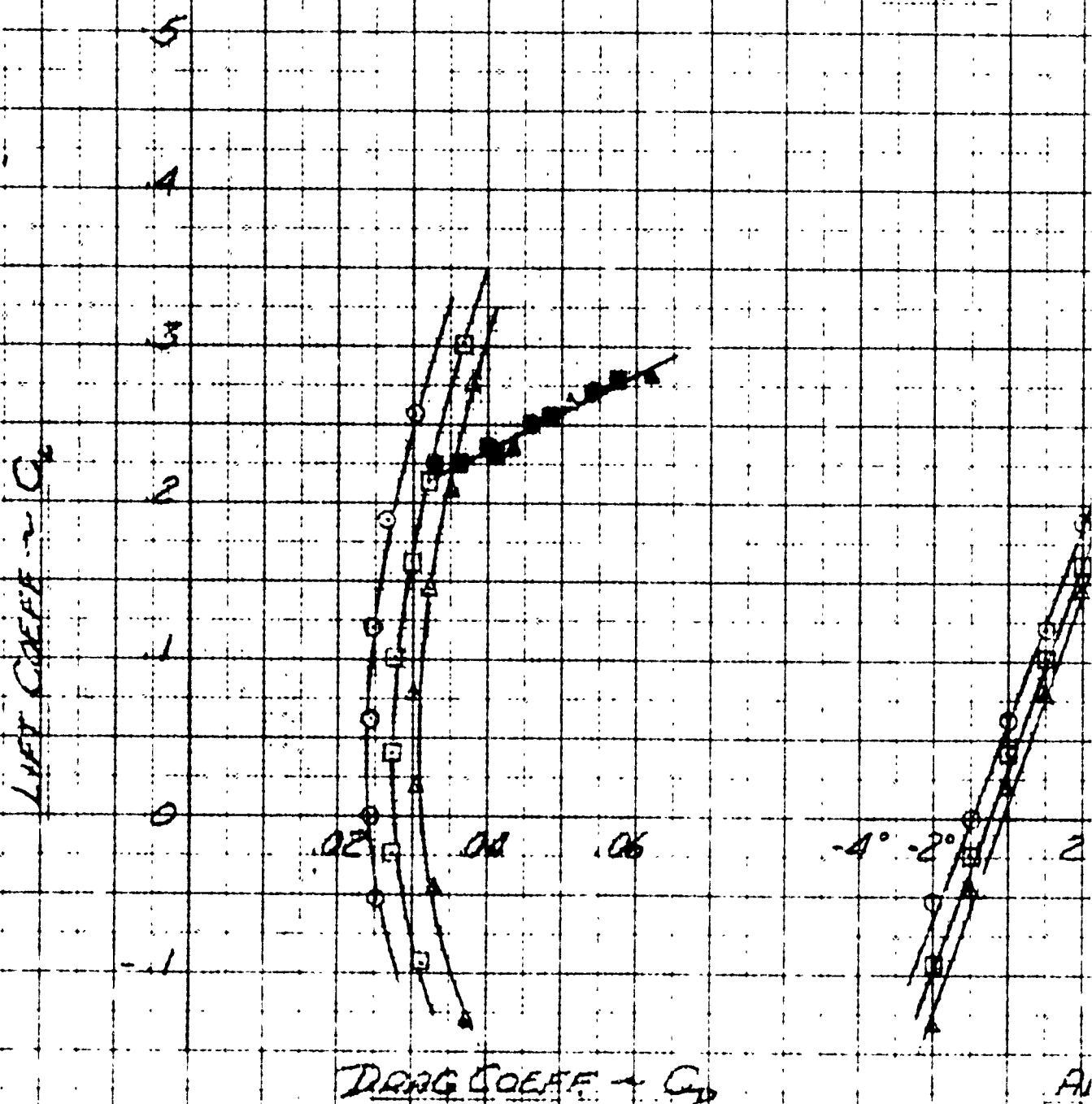
ANGLE OF ATTACK
 α - DEGREES

PITCHING MOM.
 COEFF $\sim C_{M\frac{1}{4}}$

FRONDE NO.	CAV. NO.	SYMBOL	d/c
8.226	.387	○	.75
7.1414	.392	□	1.00
6.4031	.377	▲	1.25

NOTE

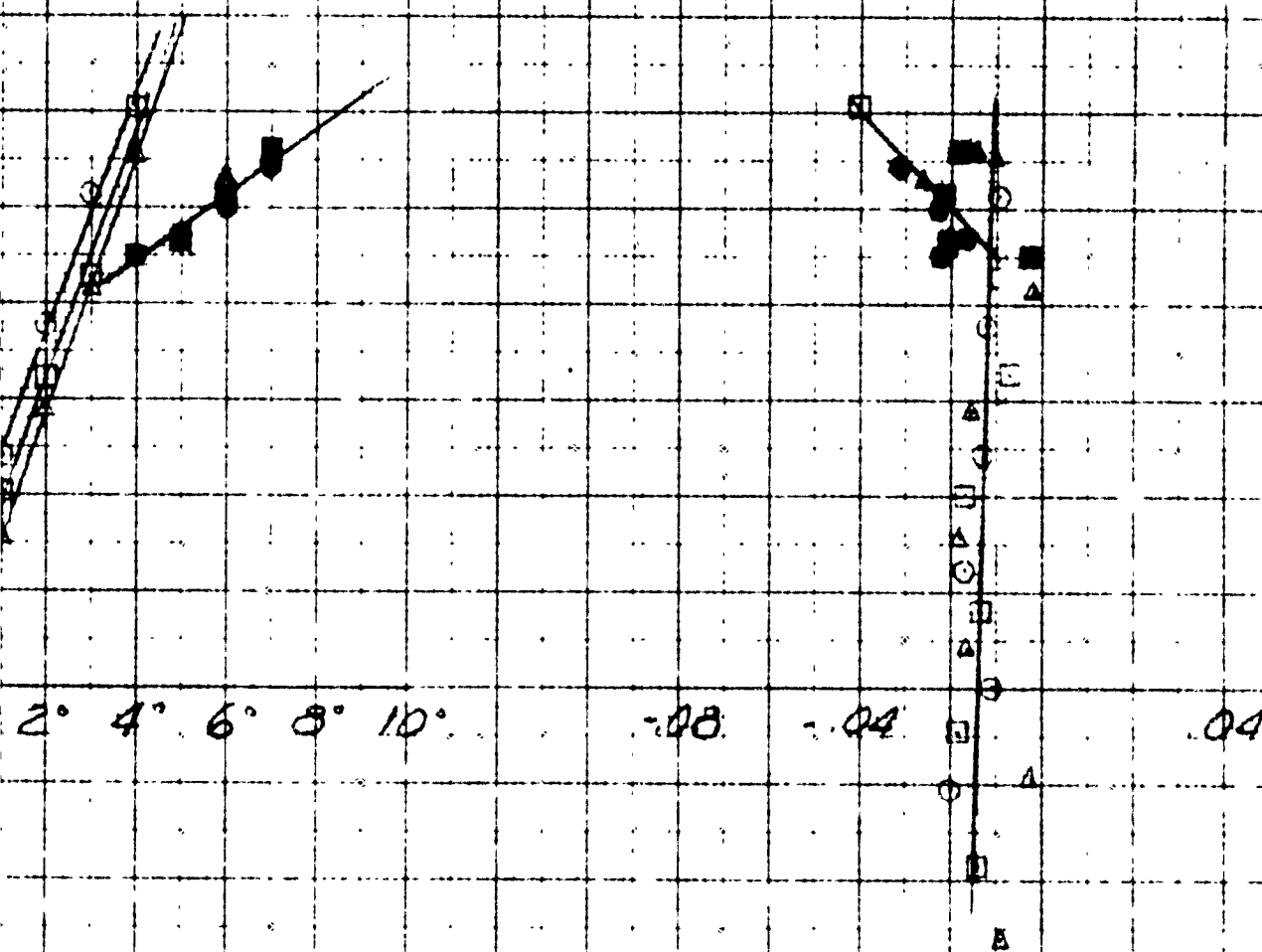
1. DARKENED SYMBOLS
INDICATE VENT
FLOW.



WHIRLING TANK TEST NO 20D
 BUSHIPS NO. 7
 45 KTS.

M.A.C. = 08.3 FT

NEOL
 NTED

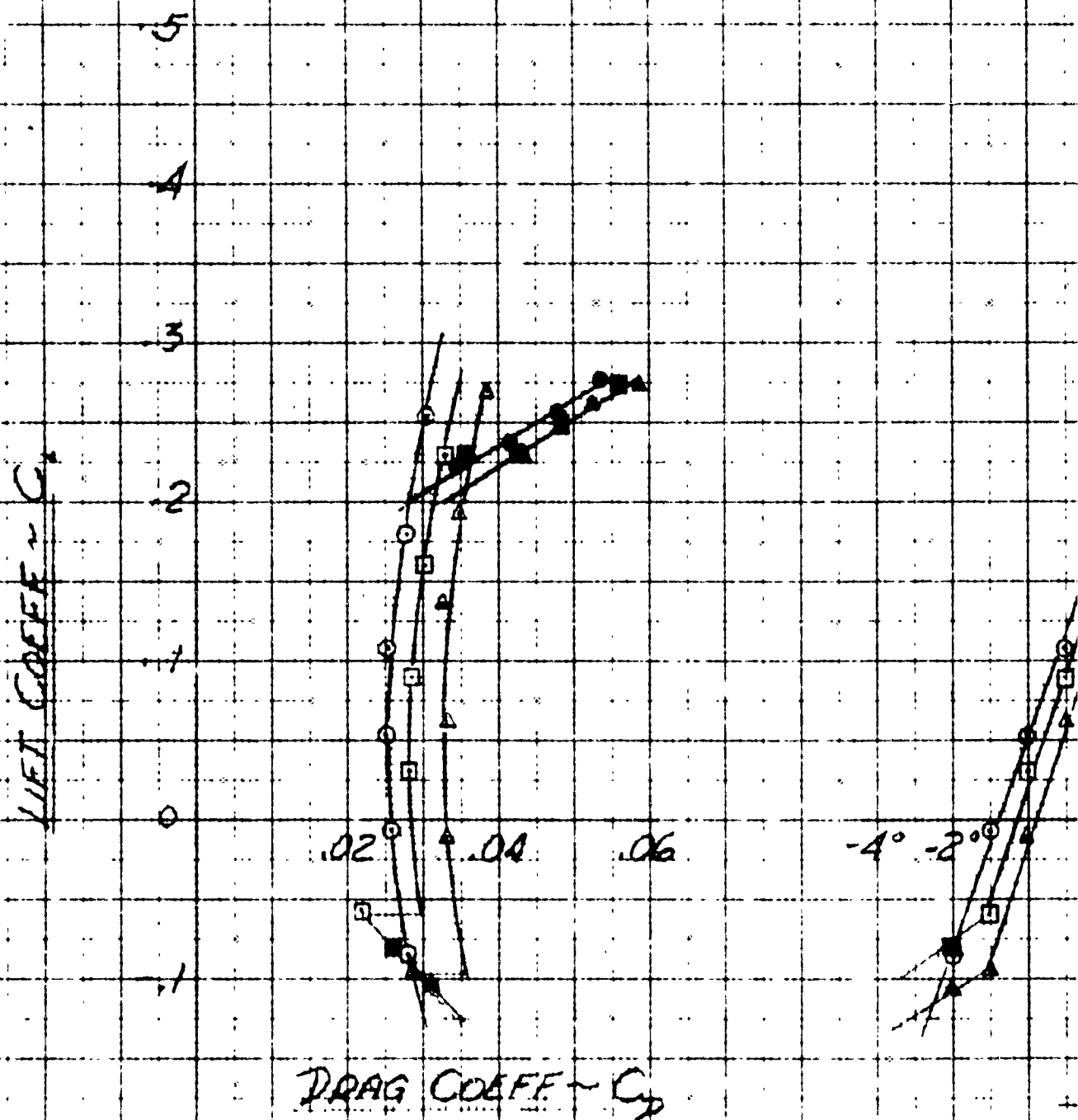


ANGLE OF ATTACK
 α DEGREES

PITCHING MOM.
 COEFF. C_m

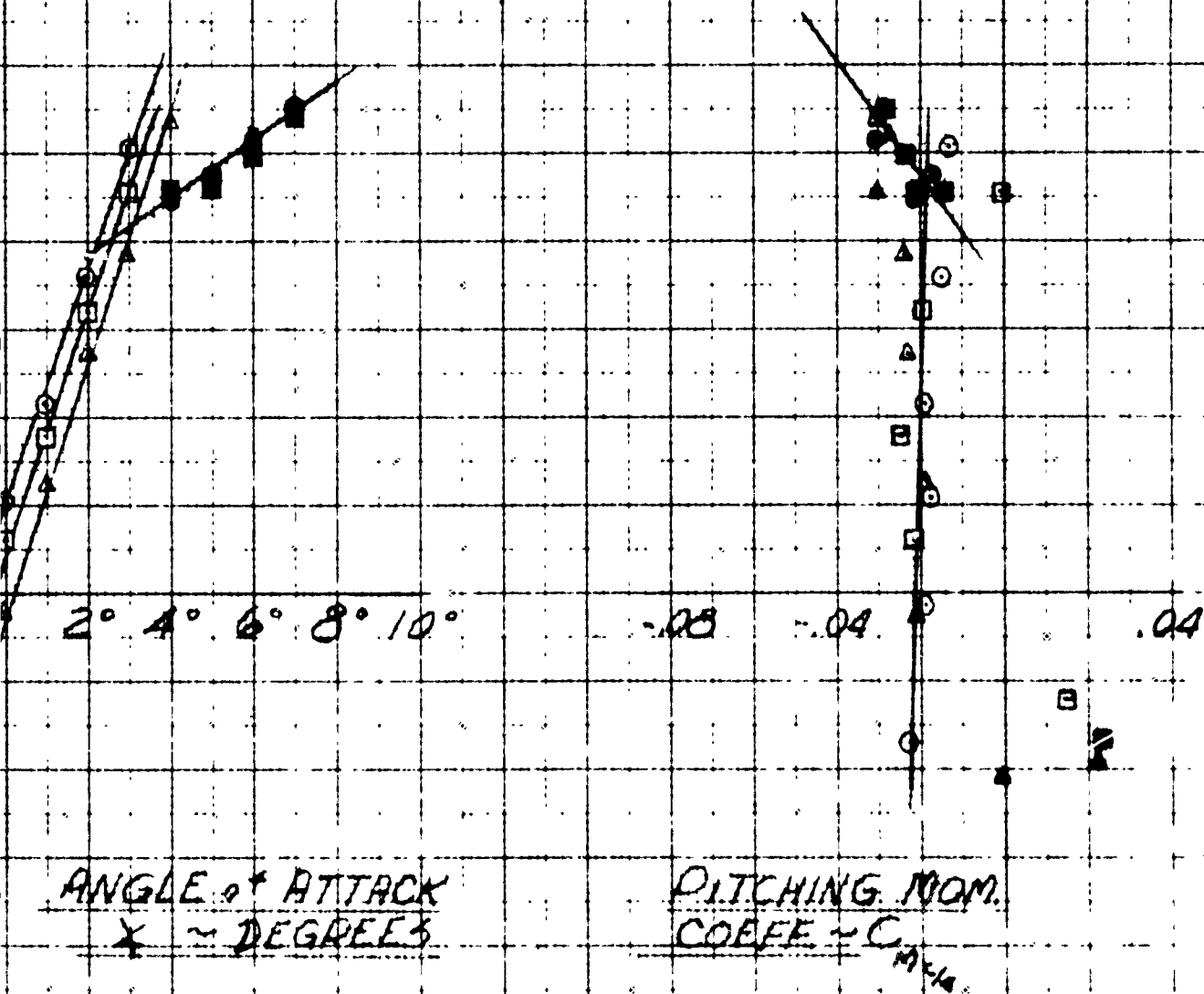
FRONDE No.	CAY. NO.	SYMBOL	q/c
82250	319	⊙	.75
71414	324	⊠	1.00
64031	329	▲	1.25

NOTE:
1. DARKENED
INDICATE
FLO



WHIRLING TANK TEST NO 20D
 BUSHIPS NO. 7
 50 KTS
 M.A.C. = 283 FT

VED SYMBOLS
 TE VENTED
 LOW.



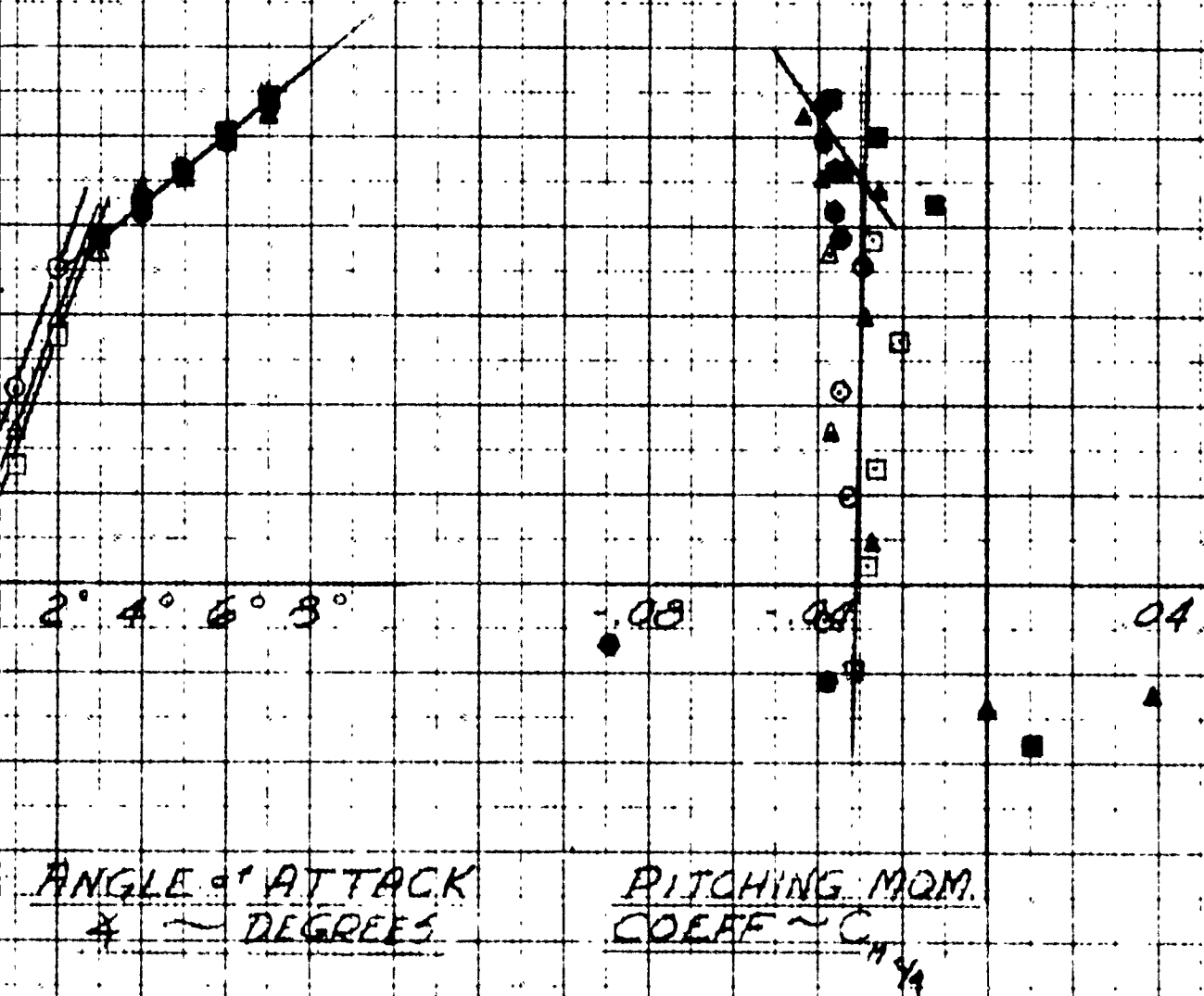
WHIRLING TANK TEST NO 20D

BUSHIPS NO 7

60 KTS

MAC. = 0.83 FT

OPENED SYMBOLS
INDICATE VENTED
FLOW.



ANGLE OF ATTACK
IN DEGREES

PITCHING MOM.
COEFF ~ C_M

PROUDE NO	CAV. NO	SYMBOL	q/c
8.2260	.178	○	.75
7.1414	.185	□	1.00
6.4031	.192	△	1.25

NOTE:
I. PARKER
INDICA
FL

LIFT COEFF C_L

0.02 0.04 0.06

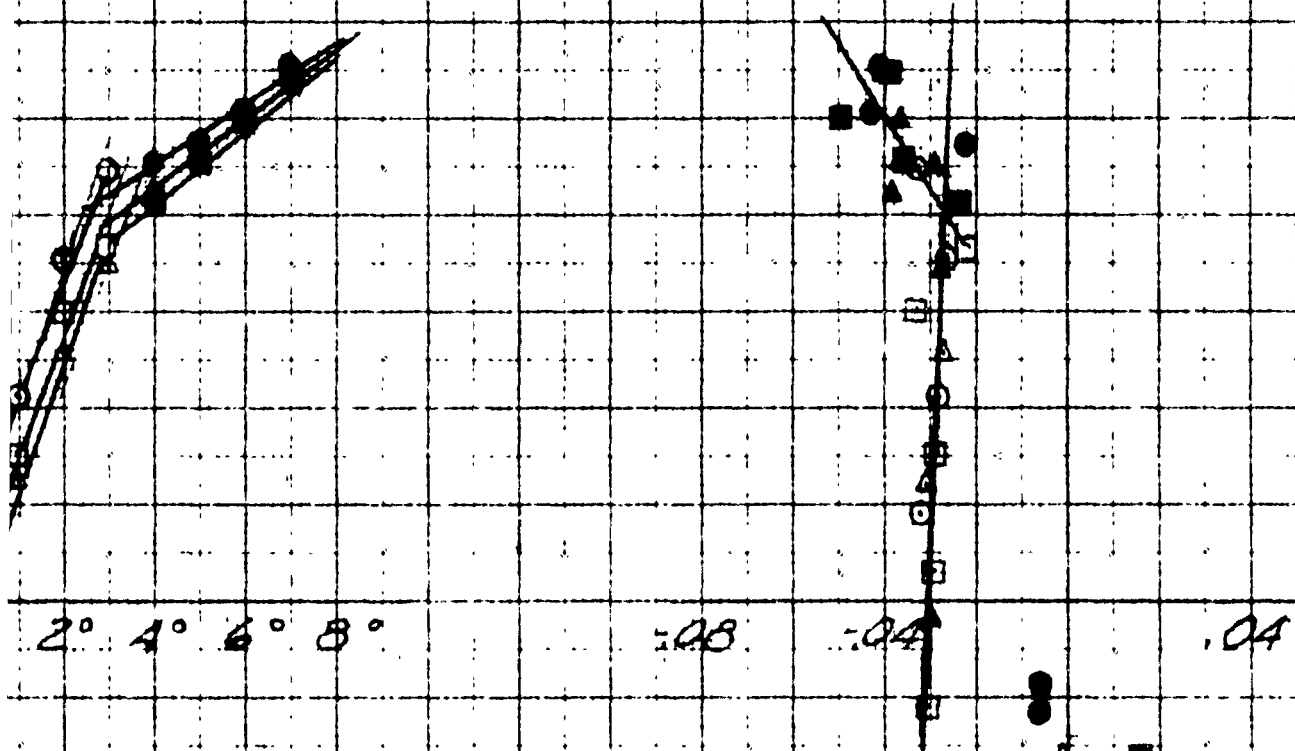
-4° -2° 2

DRAG COEFF C_D

AN

WHIRLING TANK TEST NO 20 D
 BUSHIPS NO 7
 70 KTS
 M.A.C. = .083 FT

OPENED SYMBOLS
 INDICATE VENTED
 FLOW



ANGLE OF ATTACK
 IN DEGREES

PITCHING MOM.
 COEFF C_m

SYMBOL	α	C_{Dmax}	CAL NO.	FROUDE NO.
o	.75	.0035	.500	6.12
□	1.00	.0048	.520	5.32
+	1.25	.0069	.530	4.79

LIFT COEFFICIENT - C_L

DRAG COEFFICIENT - C_D

ANGLE OF α

W.E. ...

NOTE.

1. DARKENED POINTS INDICATE
VENTED FLOW.

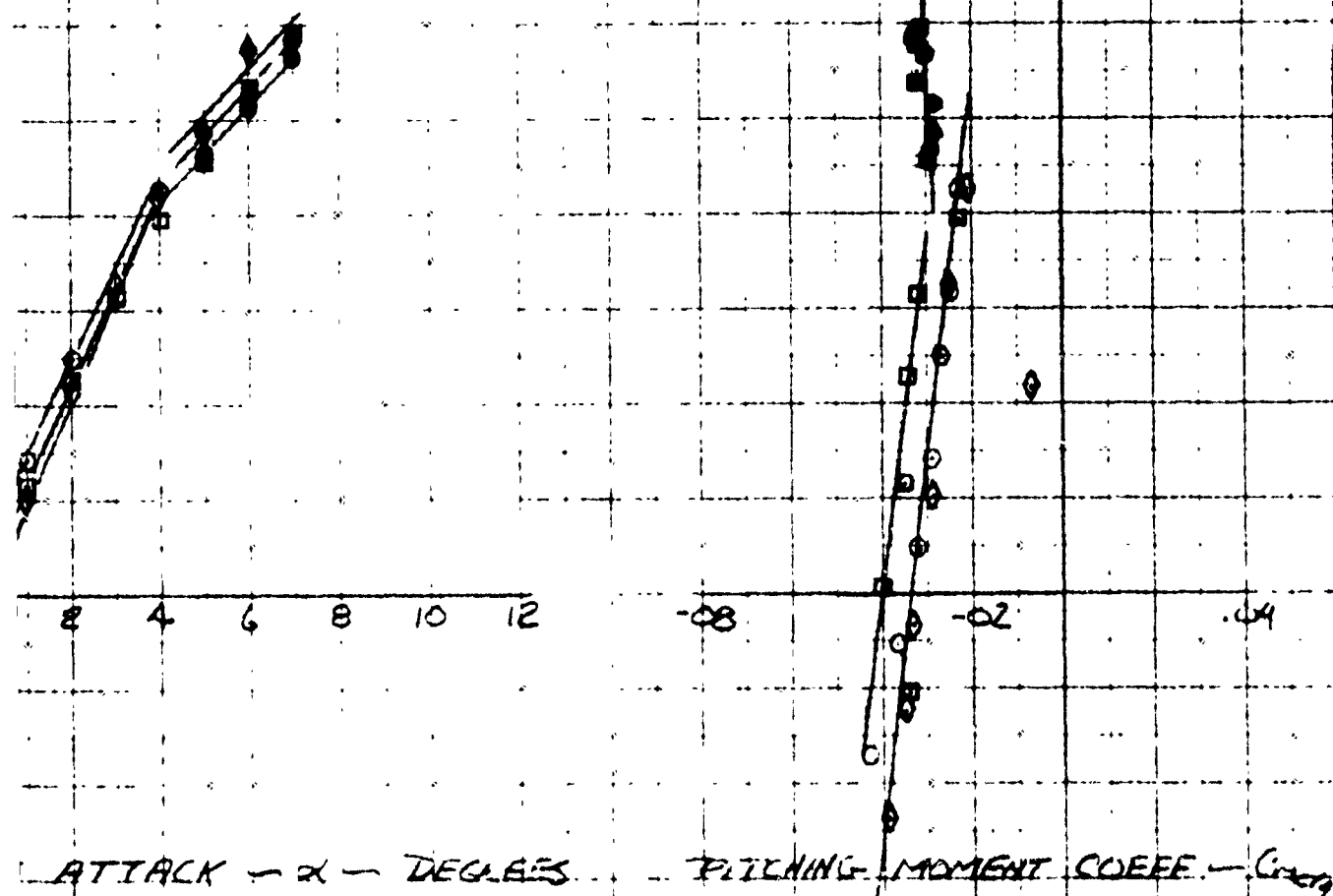
WHIRLING TANK TEST NO. 200.

NO. 63 B4954

MODEL NO. 3

40 KTS

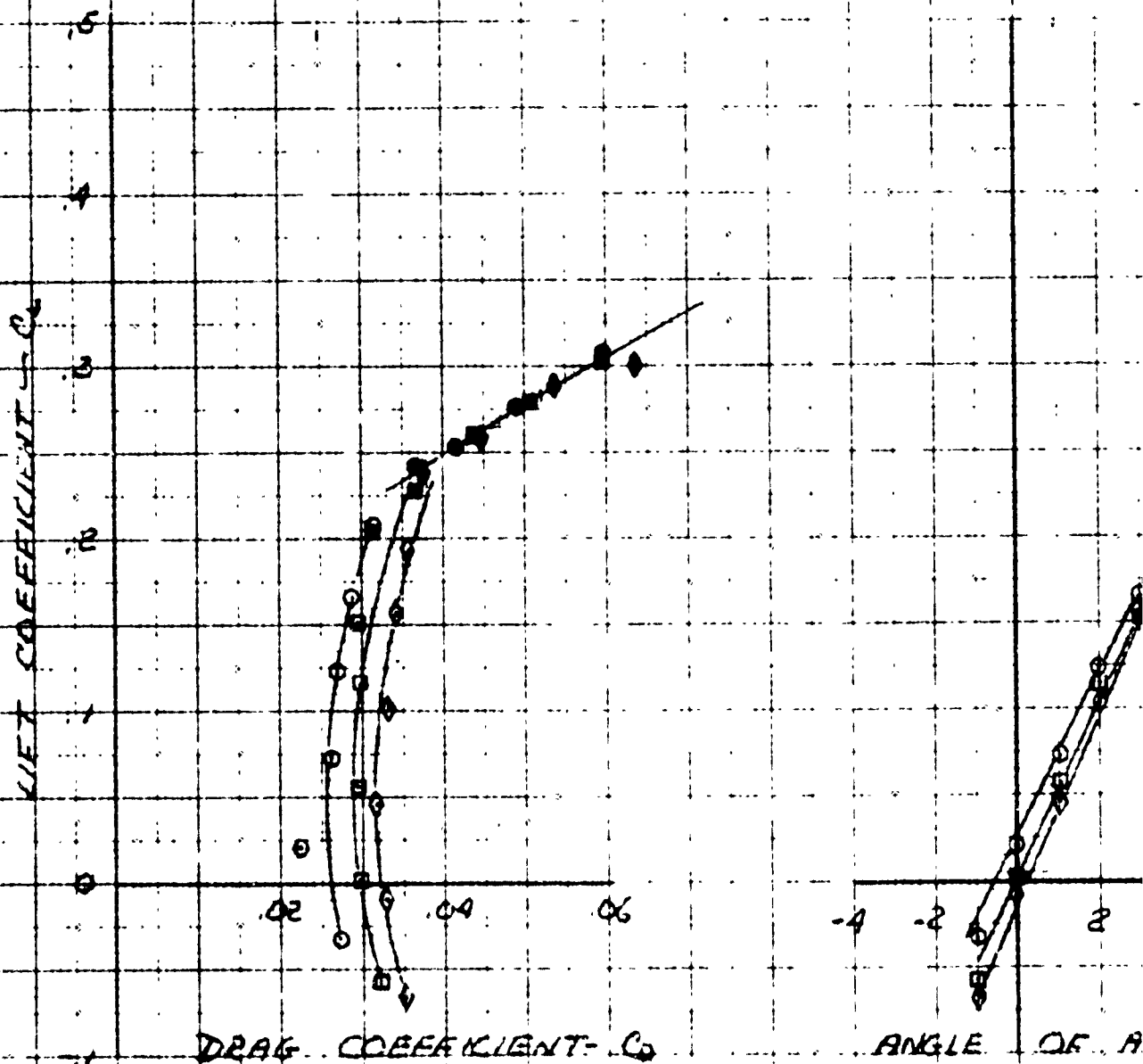
M.A.L. = 152 FT.



ME
3-4

FROUDE NO.	SYMBOL	d/E	C _{pm}	CAV. NO.
5.12	♦	.75	.0031	410
5.32	•	1.00	.0046	422
4.79	♦	1.25	.0061	440

NOTE:
1. DATA
VELOCITY



WHIRLING TANK TEST - AL. CWC

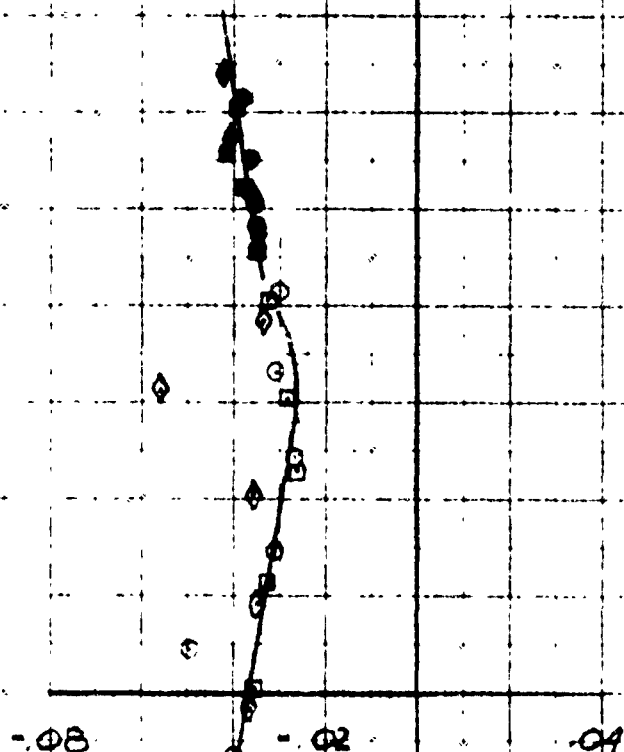
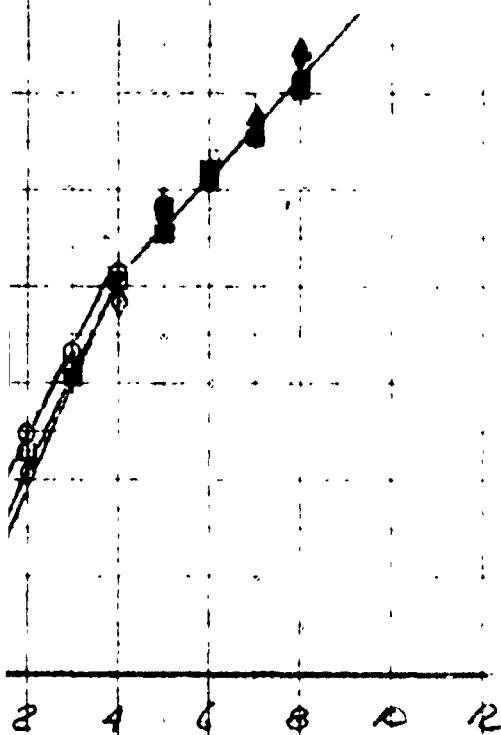
NO. 53 89454

MODEL - NO. 5

45 KTS

W. H. L. = 152 FT.

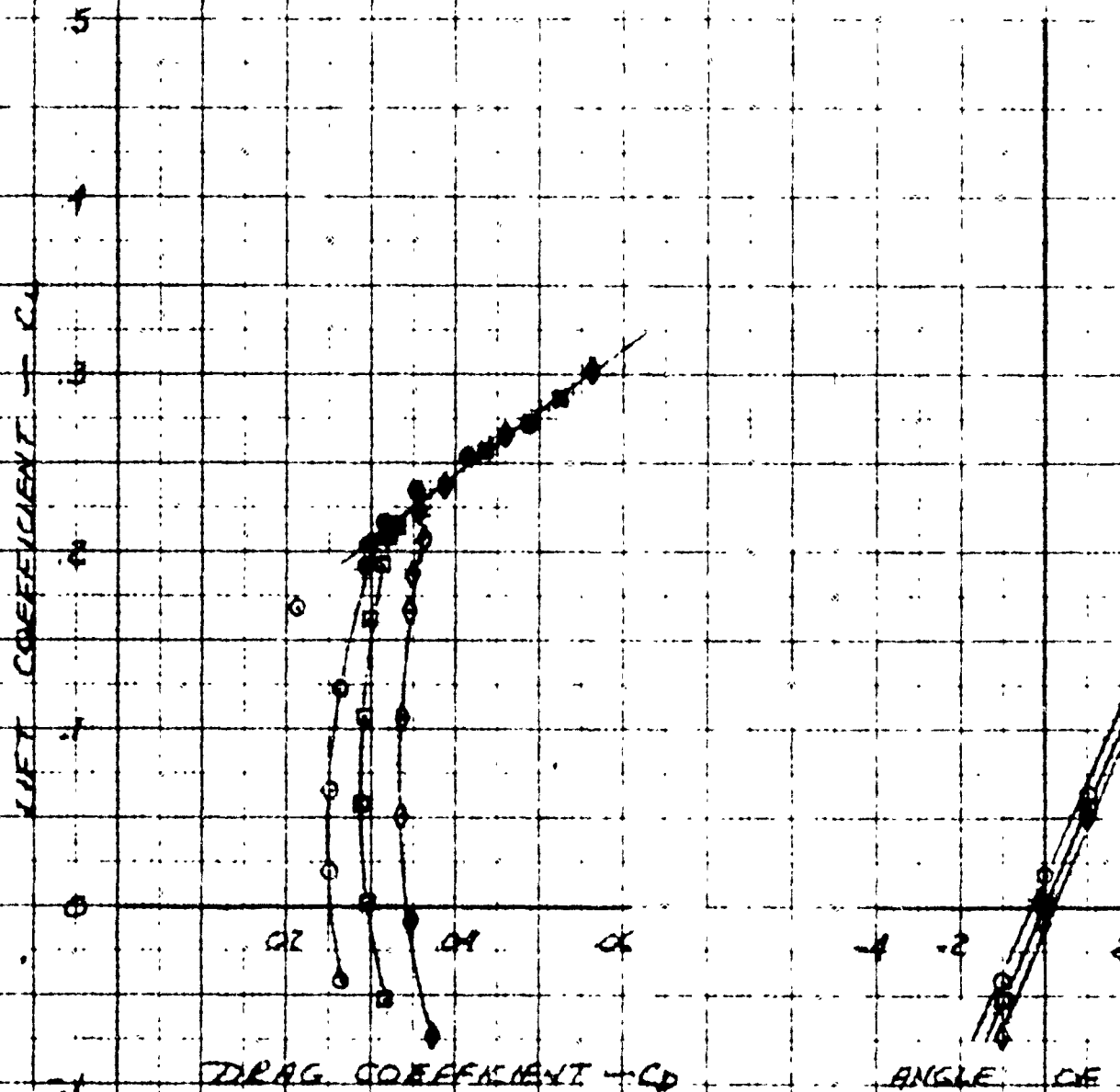
TE:
DARKENED POINTS INDICATE
VENTED FLOW



ATTACK - 2 - DEGREES

ROLLING MOMENT COEFF - C_{RM}

FRONDE NO.	SYMBOL	d/c	C _D max	CAV. NO.
6.12	♦	.75	.0030	338
5.32	•	1.00	.0046	355
4.79	♦	1.25	.0050	375



NOTE:

1. DARKENED SYMBOLS INDICATE
DISTURBED FLOW.

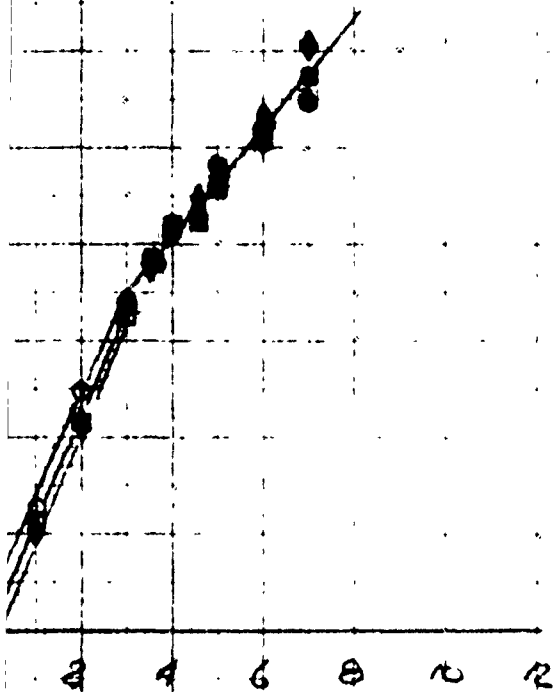
WHIRLING TANK TEST NO. 200

NO. 1-3 34454

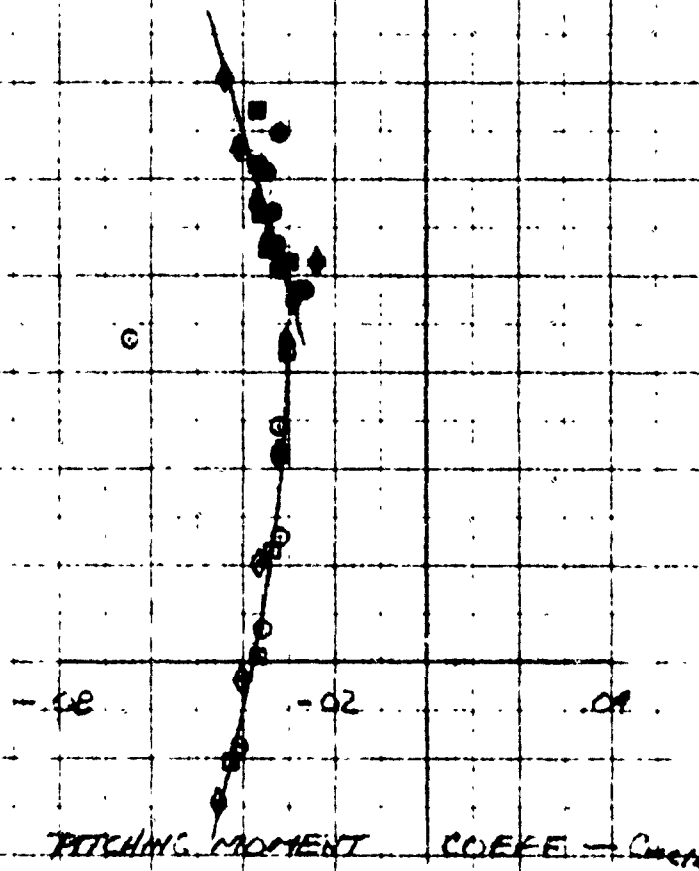
MODEL NO. 8

50 KTS

M.H. = 152 FT



ANGLE OF ATTACK - α - DEGREES



PITCHING MOMENT COEFF - Cm

TRADE NO.	SYMBOL	$\frac{1}{A}$	$C_{D_{\text{min}}}$	CAK. NO.
6.12	♦	.75	.0036	250
5.32	•	1.00	.0046	268
4.79	♦	1.25	.0060	282

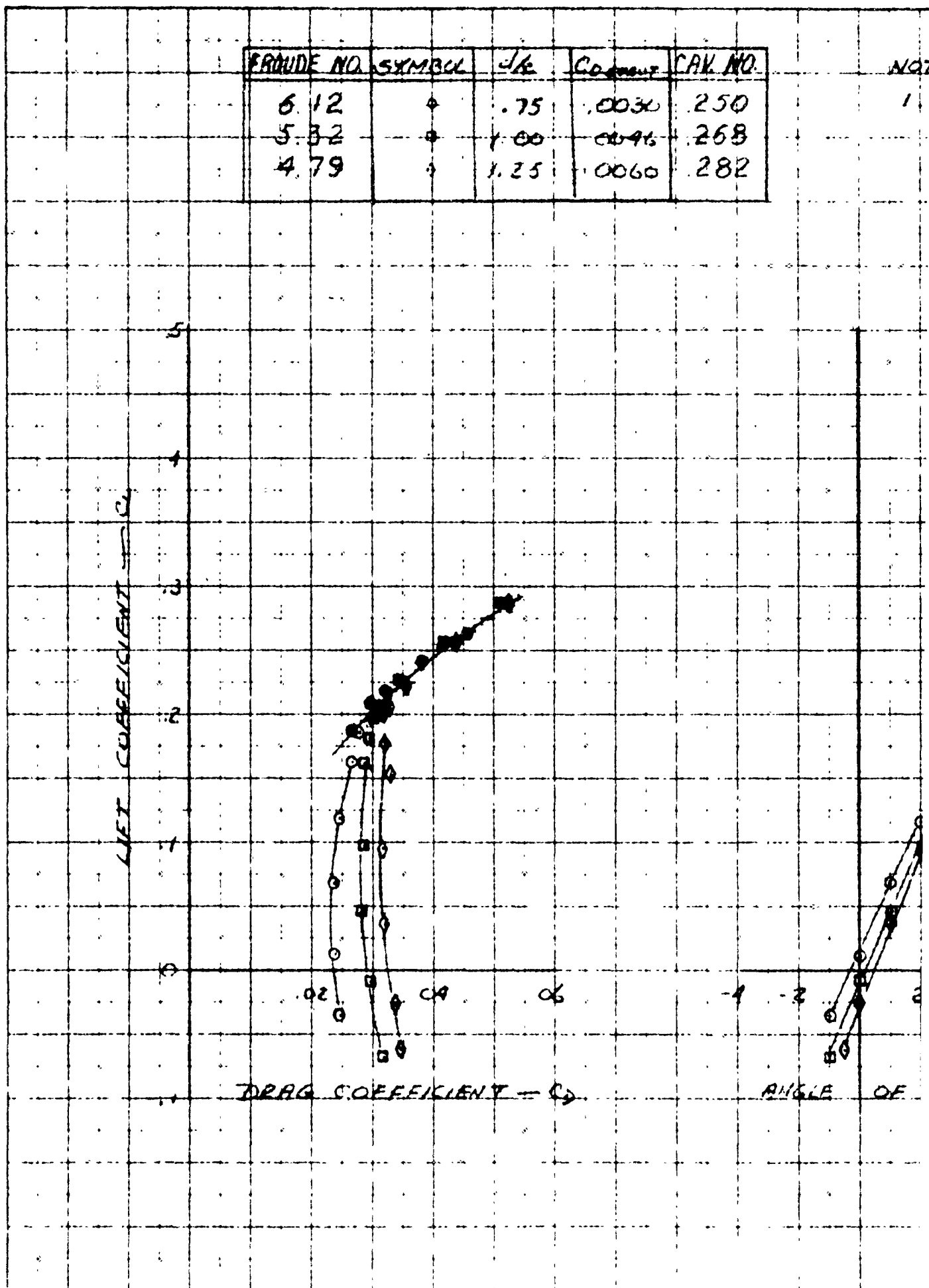
NO. 1

NO. 1

LIFT COEFFICIENT - C_L

DRAG COEFFICIENT - C_D

ANGLE OF



NOTE

1. DARKENED SYMBOLS INDICATE
VENTED FLOW

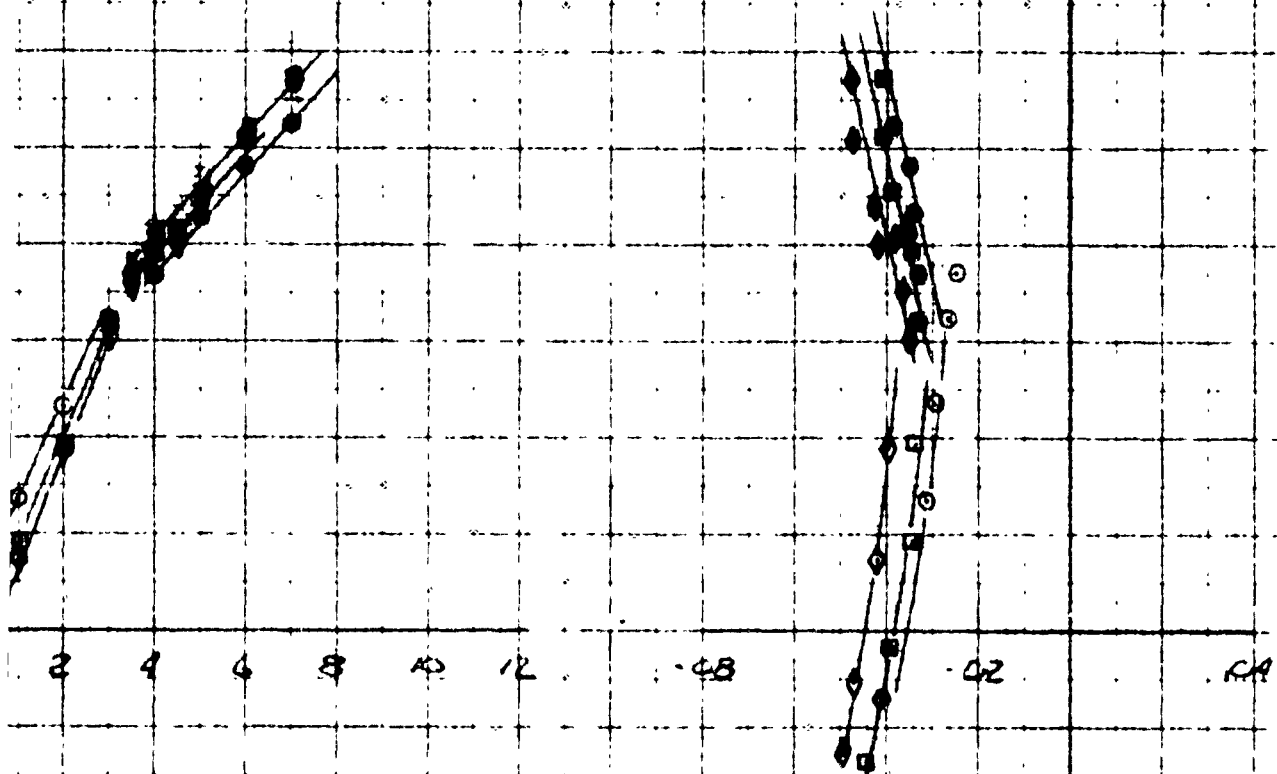
WHIRLING TANK TEST NO. 200

NO. 63 84454

MODEL NO. B

10 KTS

M.H.C. = 154 F.

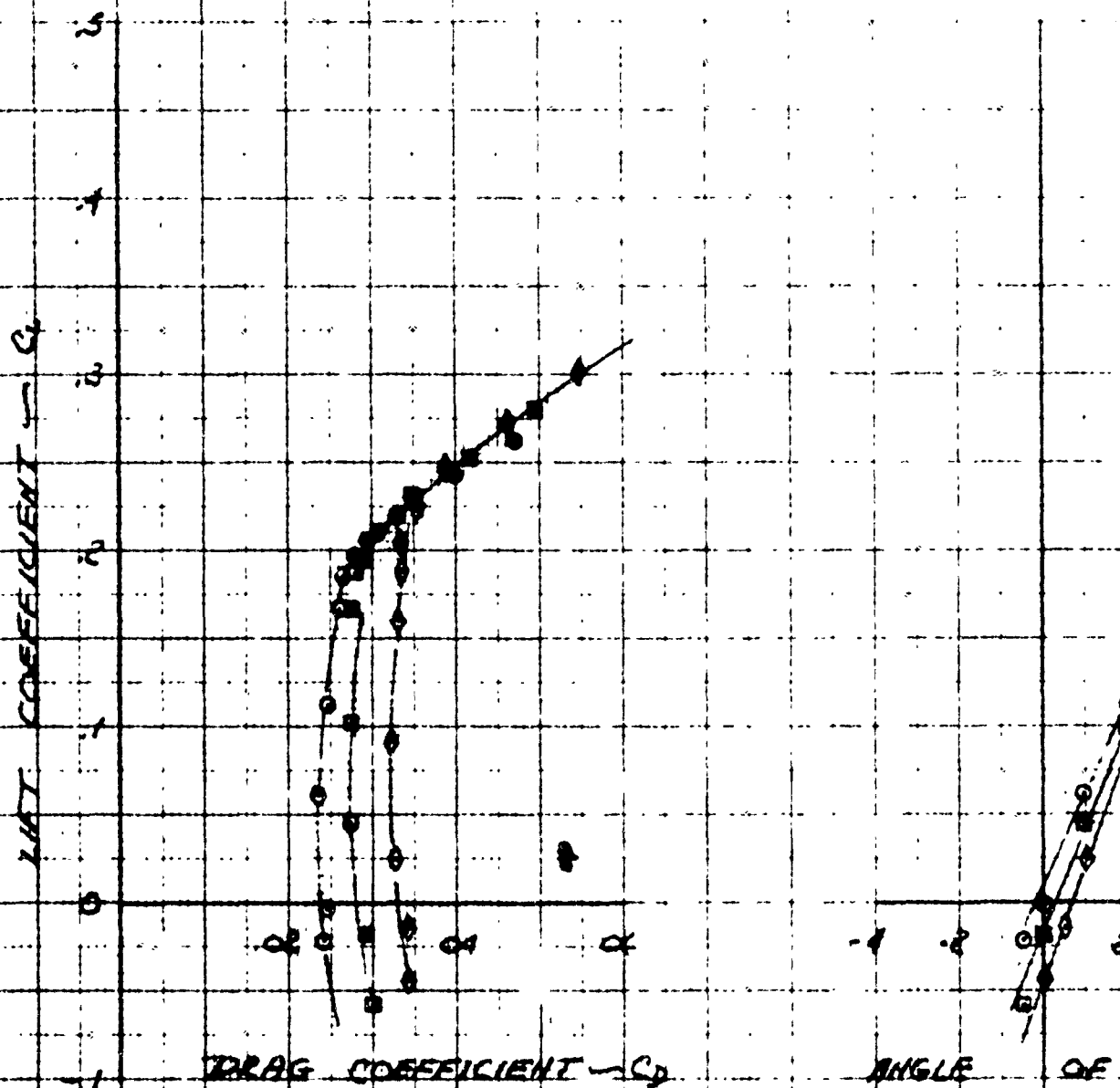


DE ATTACK — 2 — DEG

PITCH VE. MOMENT COEFF — $C_{m\dot{\alpha}}$

FRAUDE NO.	SYMBOL	d/k	$C_{D_{max}}$	CAV NO.
6.12	+	.75	.0042	198
5.32	o	1.00	.0060	215
4.79	+	1.25	.0077	230

NOTE:
1. DARI
VENT.



WHIRLING TANK TEST NO. 20C

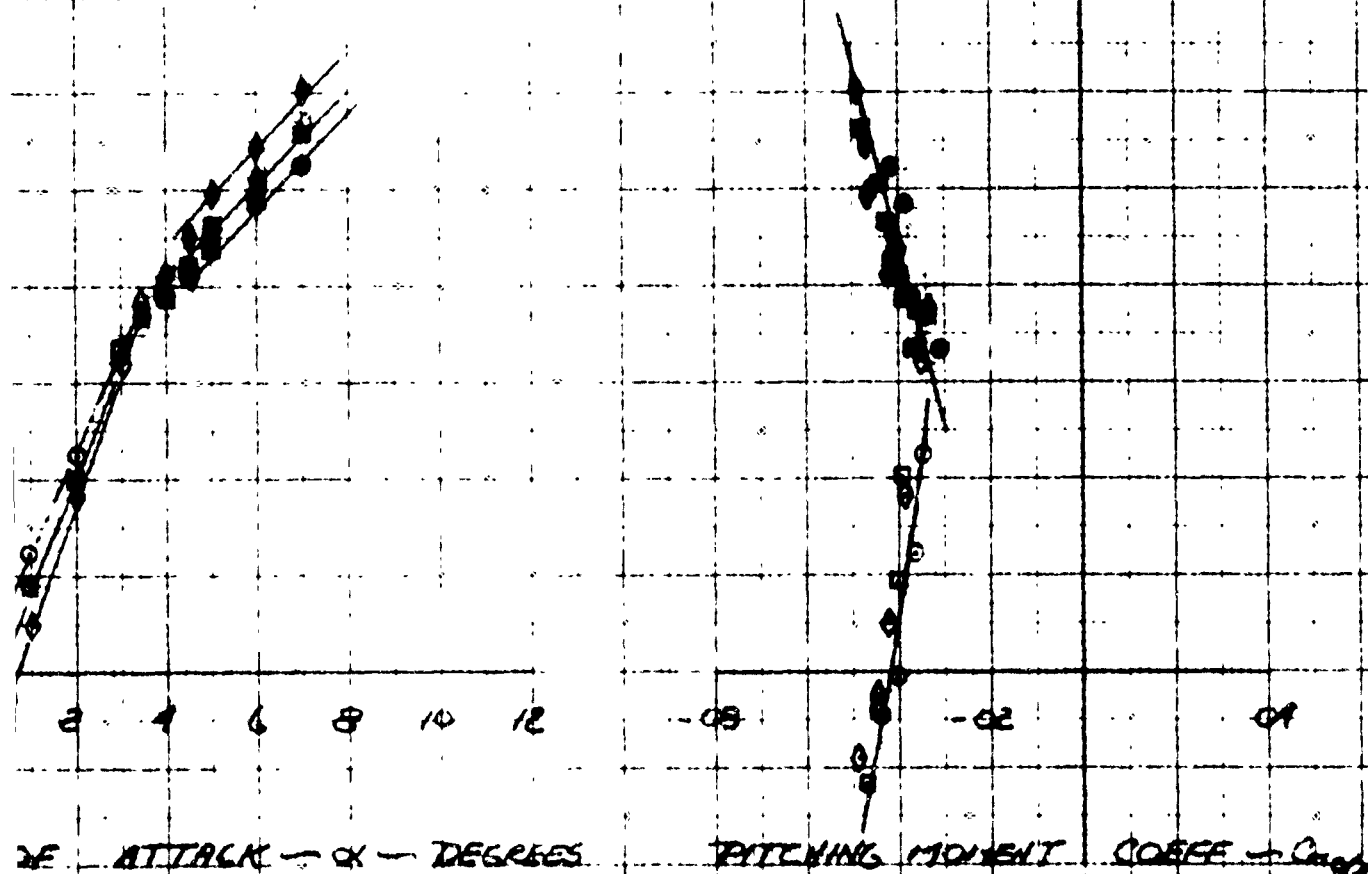
NO. 6-8 84454

MODEL NO. 8

70 KTS

M.H.C. = 152 FT.

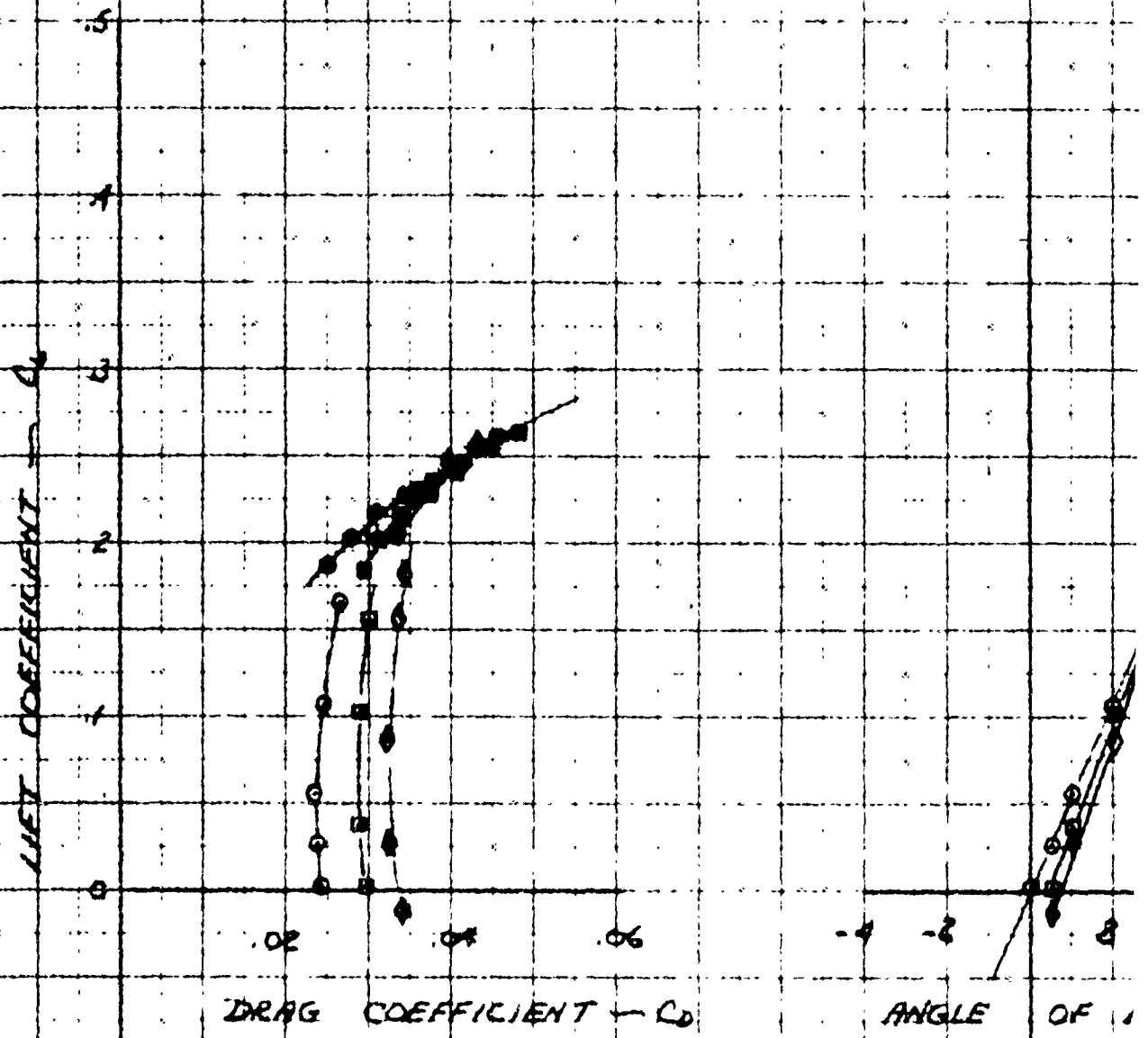
MARKED SYMBOLS INDICATE
REVERSED FLOW.



100-10100-1-304

FROUDE NO.	SYMBOL	d/E	C_{Dmax}	CAV NO.
6.12	◆	.75	.0028	178
5.32	●	1.00	.0051	194
4.79	♦	1.25	.0057	210

NOTE:
1. DATA
VEN



WHIRLING TANK TEST NO 200

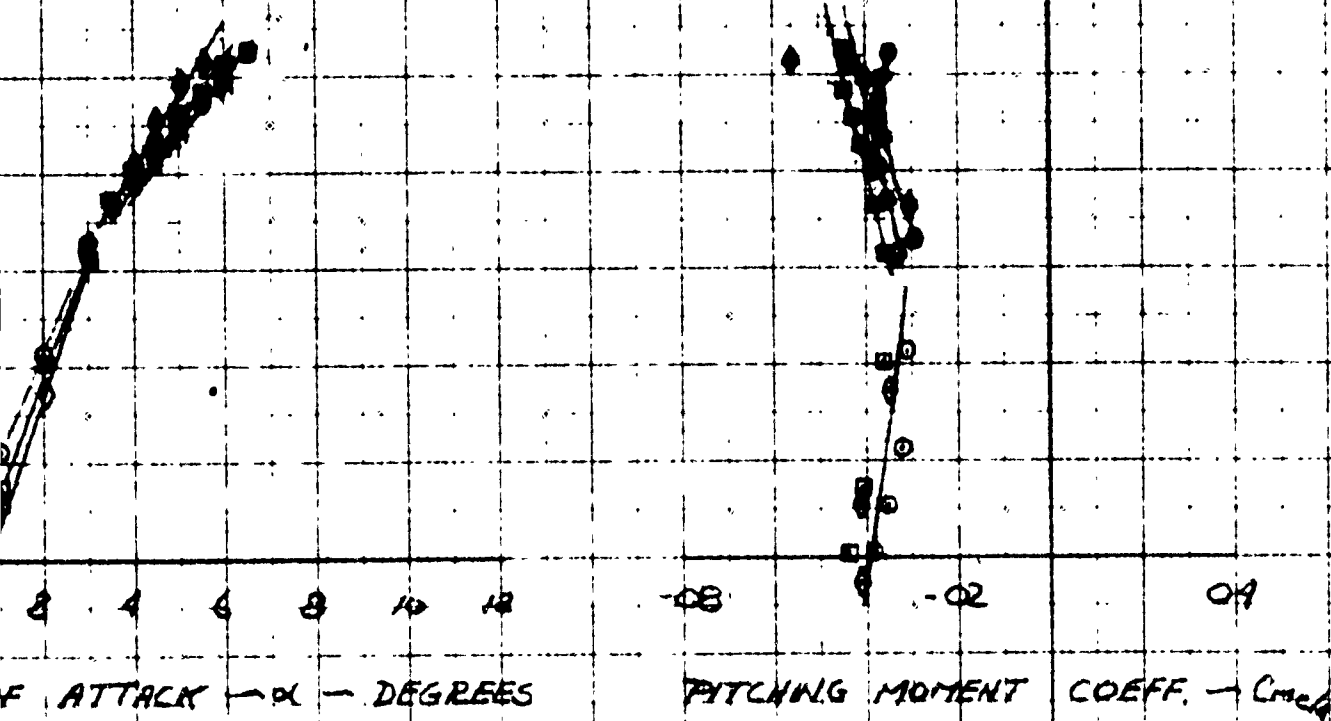
NO. 89454

MODEL NUB

80 KTS

12.25 FT

TE:
DARKENED SYMBOLS INDICATE
VENTED FLOW



SYMBOL	d/c	C _D STRUT	CAR. NO.	AROUDE NO.
○	.75	.0035	.500	6.12
□	1.00	.0049	.519	5.32
▲	1.25	.0065	.532	4.79

NOTES

1. DARK
VE

LIFT COEFFICIENT C_L

DRAG COEFFICIENT
 C_D

GRUMMAN AIRCRAFT ENGINEER

WHIRLING TANK TEST NO. 26E

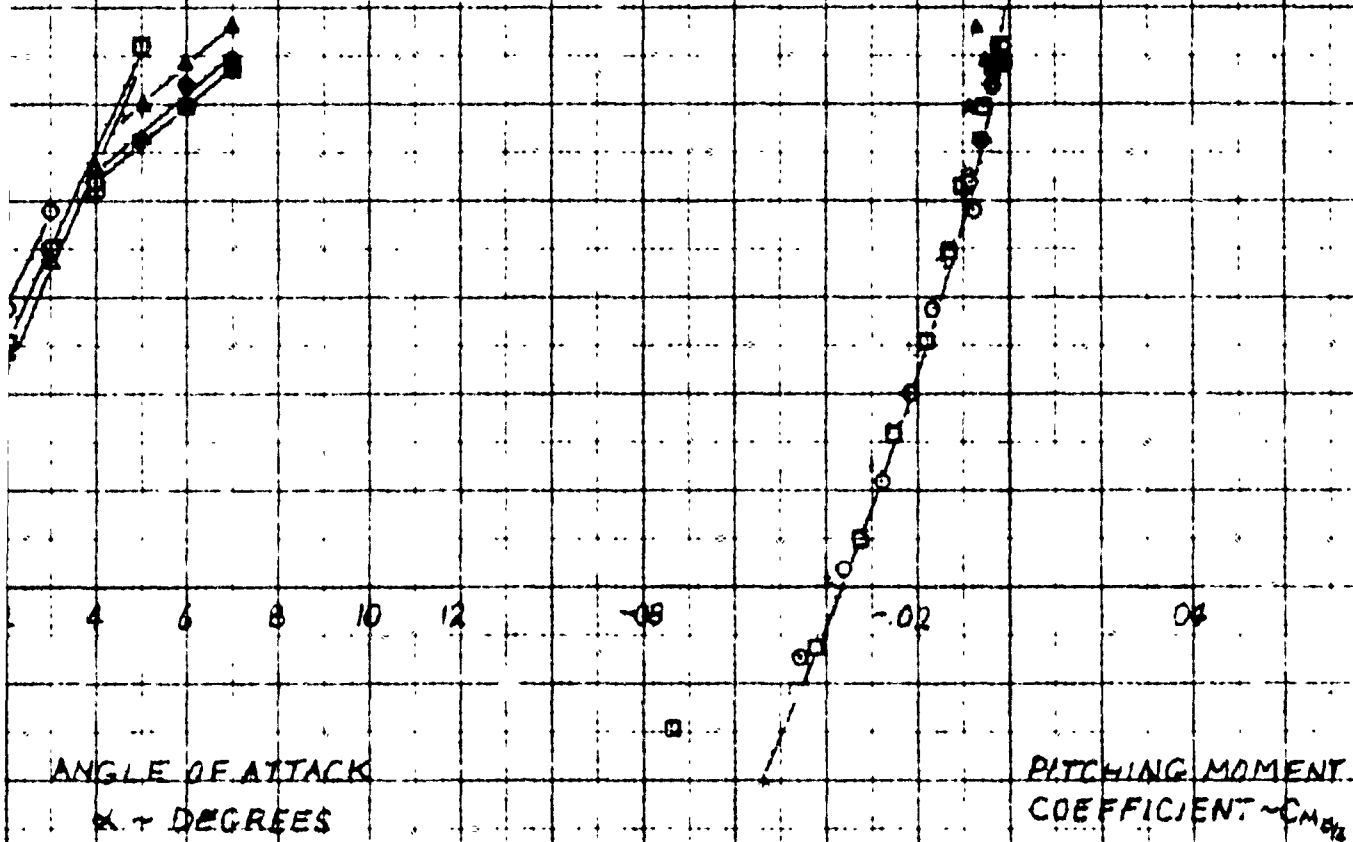
NO. 65-89454

MODEL NO. 9

40 KTS

M.H.C. = 126 FT.

ARKENED SYMBOLS INDICATE
VENTED FLOW



SYMBOL	M/C	CONSTANT	CAY NO.	PROUDE NO.
○	.75	0031	410	5.12
□	1.00	0049	422	5.32
▲	1.25	0061	438	4.79

NOTES
1.

LIFT COEFFICIENT - C_L

DRAG COEFFICIENT

ANGLE
 $\alpha - D$

GRUMMAN AIRCRAFT ENGINEER

WHIRLING TANK TEST NO 22E

NO. 01 84454

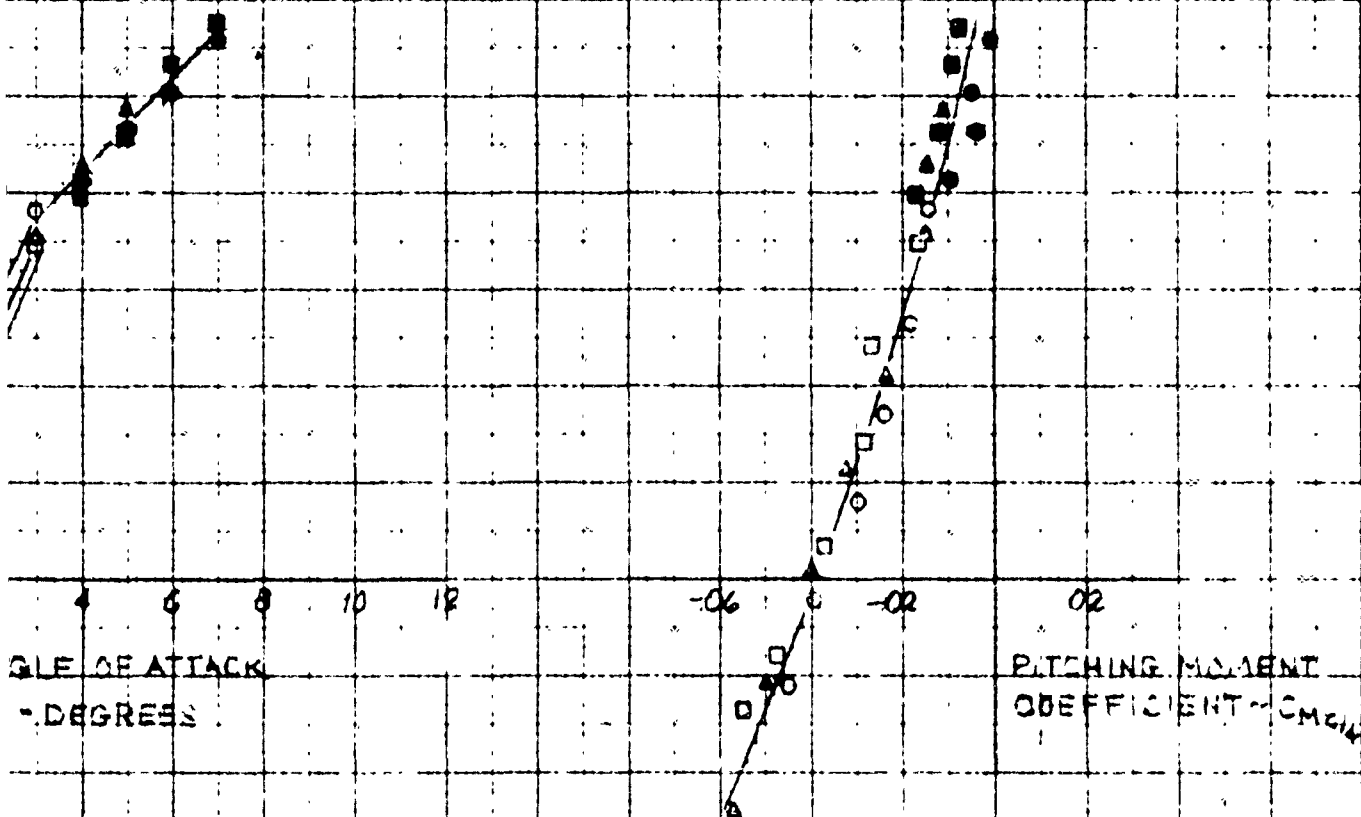
MODEL NO 9

45 KTS

MA. C. 3. 172 57

NOTE:

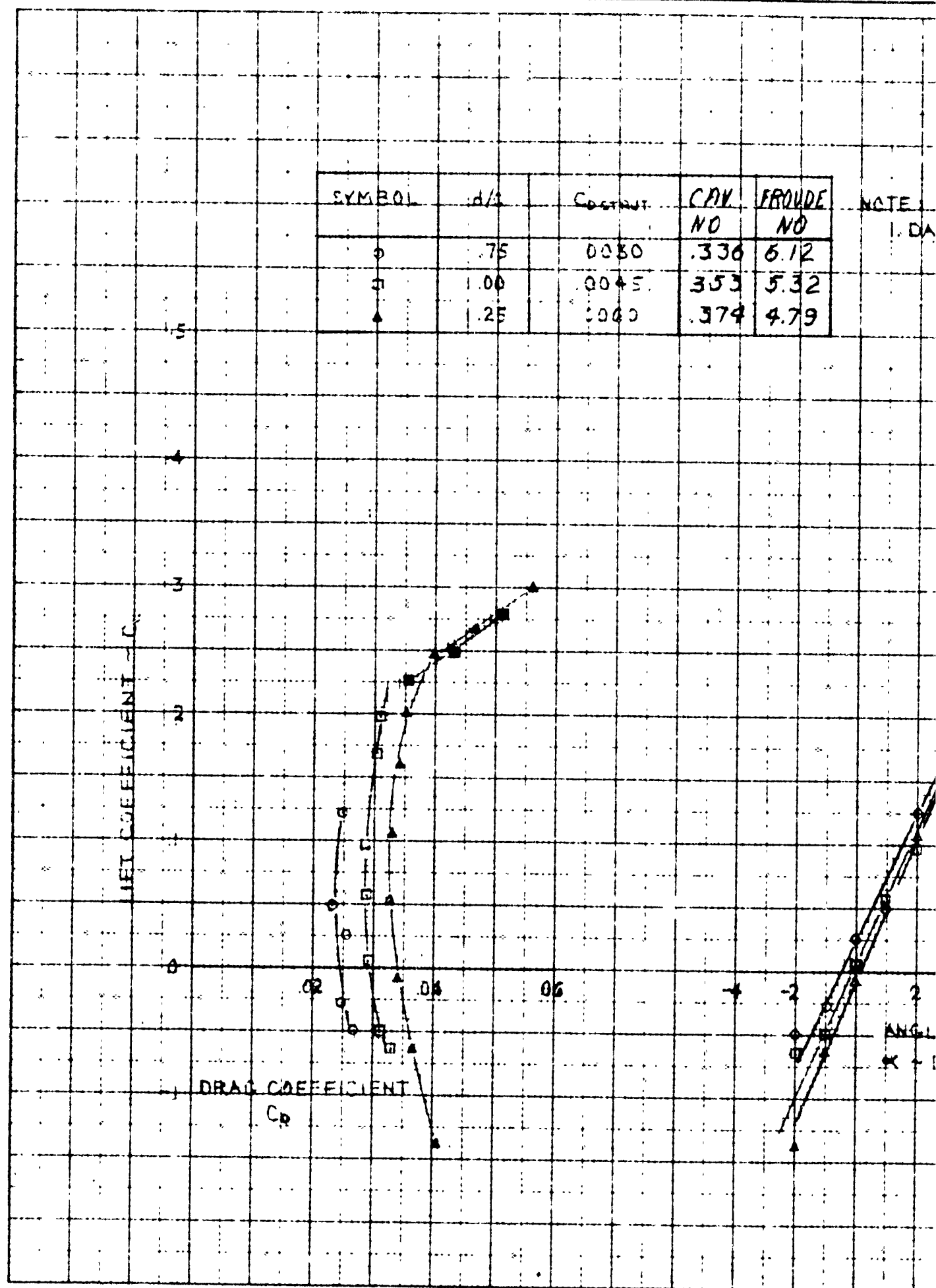
1. DARKENED SYMBOLS INDICATE
VENTED FLOW



ANGLE OF ATTACK
- DEGREE

PITCHING MOMENT
COEFFICIENT - C_m

SYMBOL	d/s	CONSTANT	CAN NO	FROUDE NO
o	.75	0030	336	6.12
□	1.00	0045	353	5.32
▲	1.25	0060	374	4.79



WHIRLING TANK TEST NO 23E

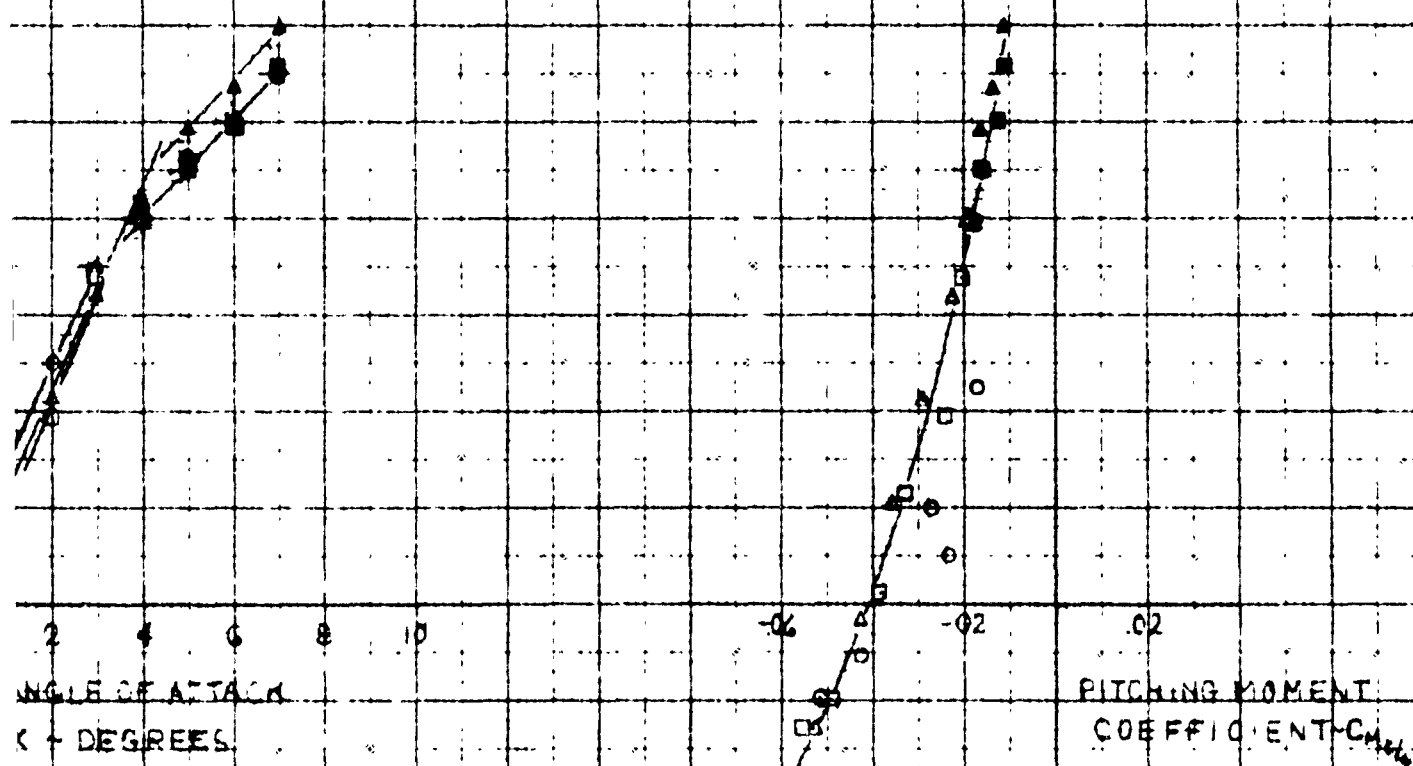
NO 65 B44E4

MODEL NO 9

SURTS

M.F.S. 1152 H7

DARKENED SYMBOLS INDICATE
VENTED FLDW



FROUDE NO.	SYMBOL	$\frac{S}{C}$	Q ST. NO.	CAY NO.
6.12	○	.75	.0032	249
5.32	□	1.00	.0048	261
4.79	▽	1.25	.0060	285

NOTE:

LIFT COEFFICIENT C_L

DRAG COEFFICIENT C_D

WHIRLING TANK TEST NO. 20A

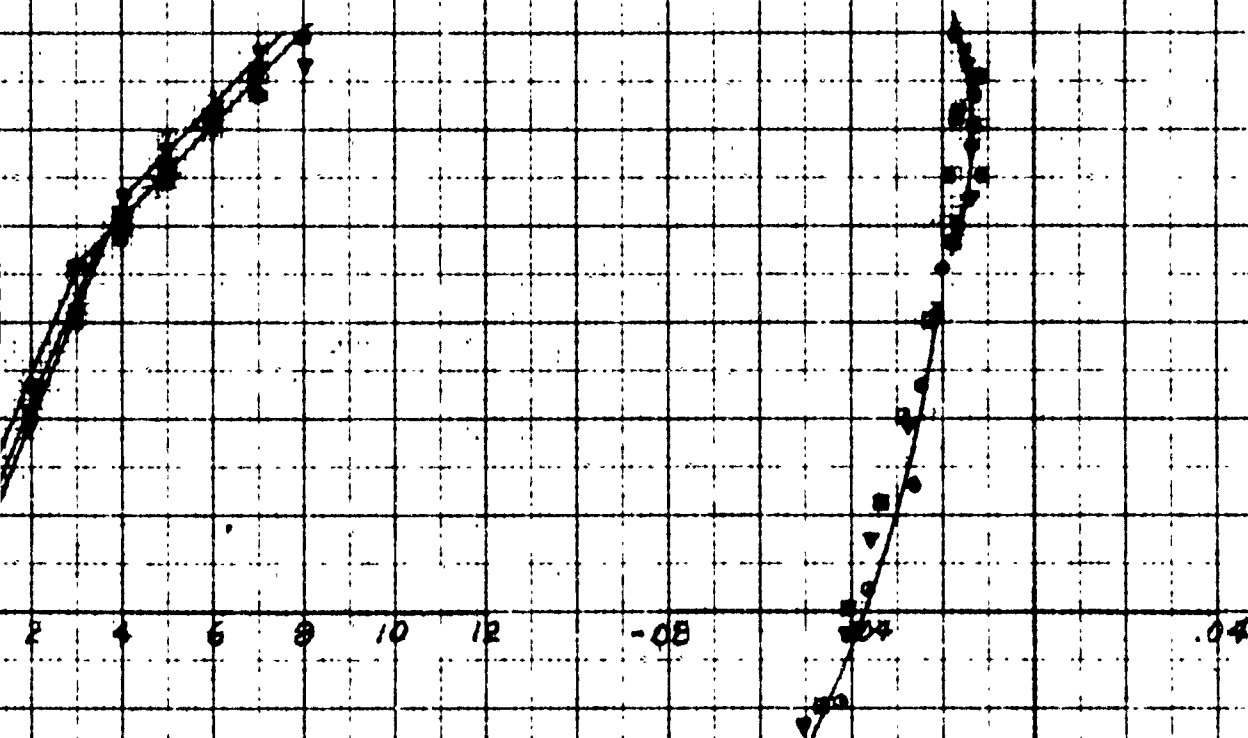
NO. 63 B49.54

MODEL NO. 9

60 MTS.

MACH 1.54

IN DARKENED SYMBOLS INDICATE
VENTED FLOW.



ANGLE OF ATTACK
α - DEGREES

PITCHING MOMENT
COEFFICIENT - $C_{m_{\alpha}}$

FROUDE NO.	SYMBOL	$\frac{4}{c}$	$C_{D\text{ STAY}}$	CAN NO.
6.12	⊙	.75	.0034	.197
5.32	⊠	1.00	.0050	.215
4.79	▽	1.25	.0065	.230

NOTE:

LIFT COEFFICIENT C_L

.02

.04

.06

DRAE COEFFICIENT

C_D

DRAG

C_D

WHIRLING TANK TEST NO. 20E

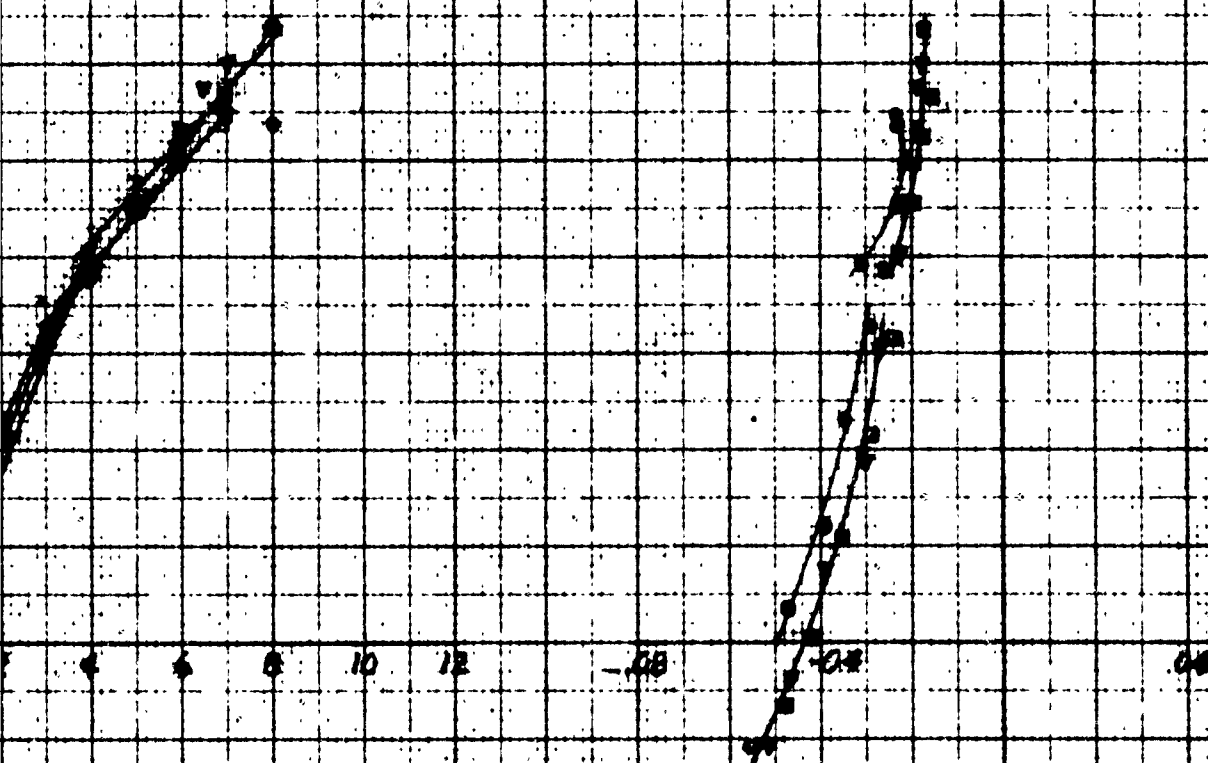
NO. 6-39454

MODEL NO. 9

70 H.T.S.

M.P.C. 2152 F1

DARKENED SYMBOLS INDICATE
VENTED FLOW



ANGLE OF ATTACK
DEGREES

PITCHING MOMENT
COEFFICIENT

WING NO.	SYMBOL	$2/c$	C_{Lmax}	WING NO.
5.12	⊙	75	.0054	177
5.32	⊠	100	.0051	197
4.79	⊗	125	.0067	210

LIFT COEFFICIENT - C_L

.02

.04

.06

.1

.2

.3

DRAG COEFFICIENT

C_D

WING

GRUMMAN AIRCRAFT ENGINEER

WHIRLING TANK TEST NO 208

NO 05 84454

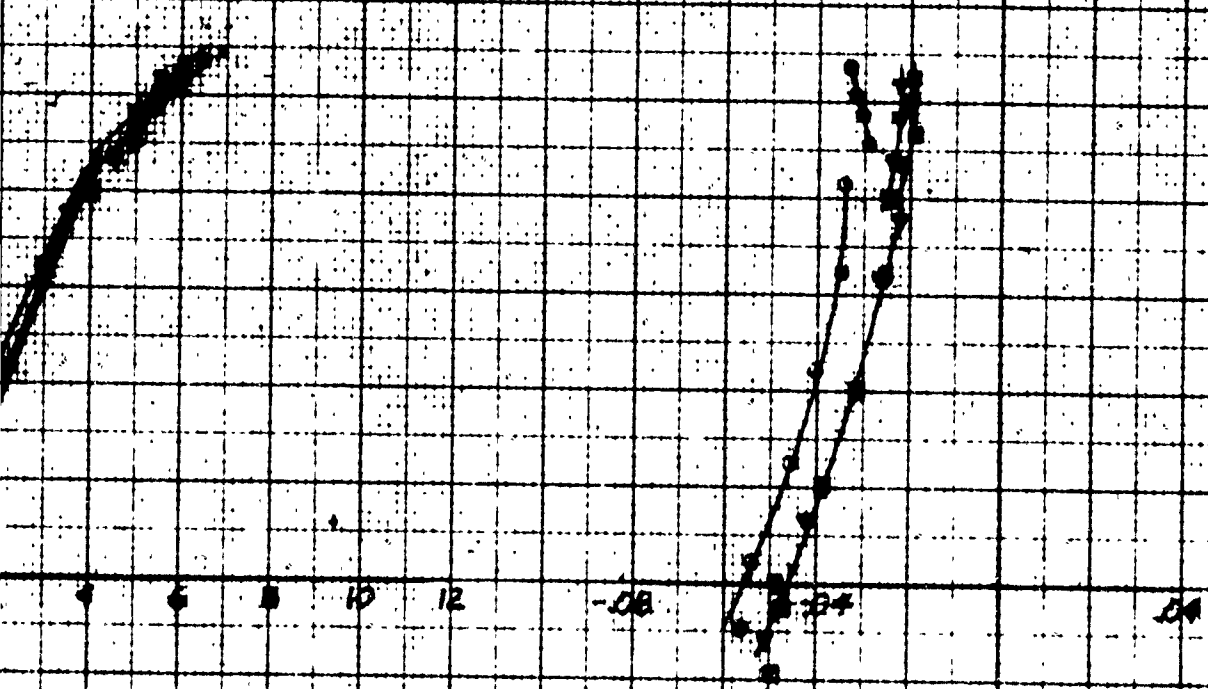
MODEL NO. 9

80 MTS

MAC = 152 FT

NOTE:

DARKENED SYMBOLS INDICATE
VENTED FLOW

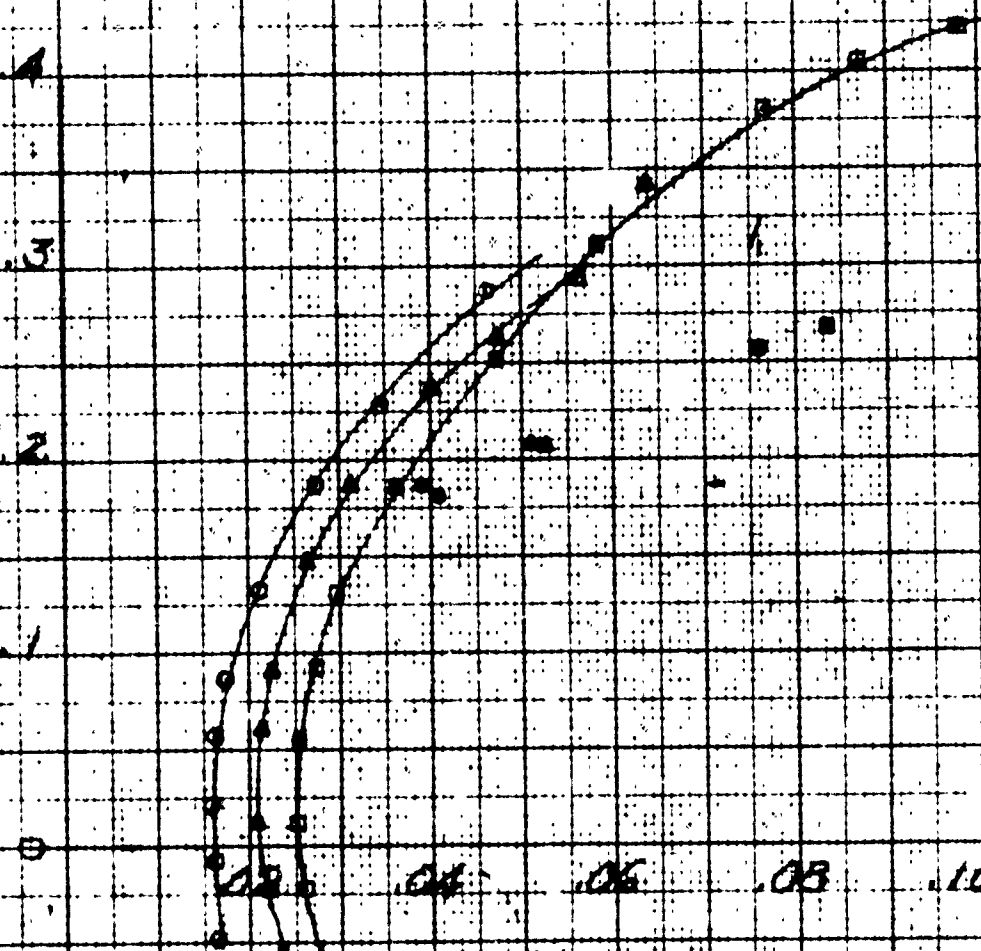


ANGLE OF ATTACK
α - DEGREES

PITCHING MOMENT
COEFFICIENT - C_m

SYMBOL	d/c	FROUDE NO.	CAY. NO.
•	.75	5.0990	52695
▲	1.00	4.4438	54879
○	1.25	4.0002	57041

LIFT COEFF - C_L



DRAW COEFF - C_D

ANGLE
 α

NO.
1.D
1A

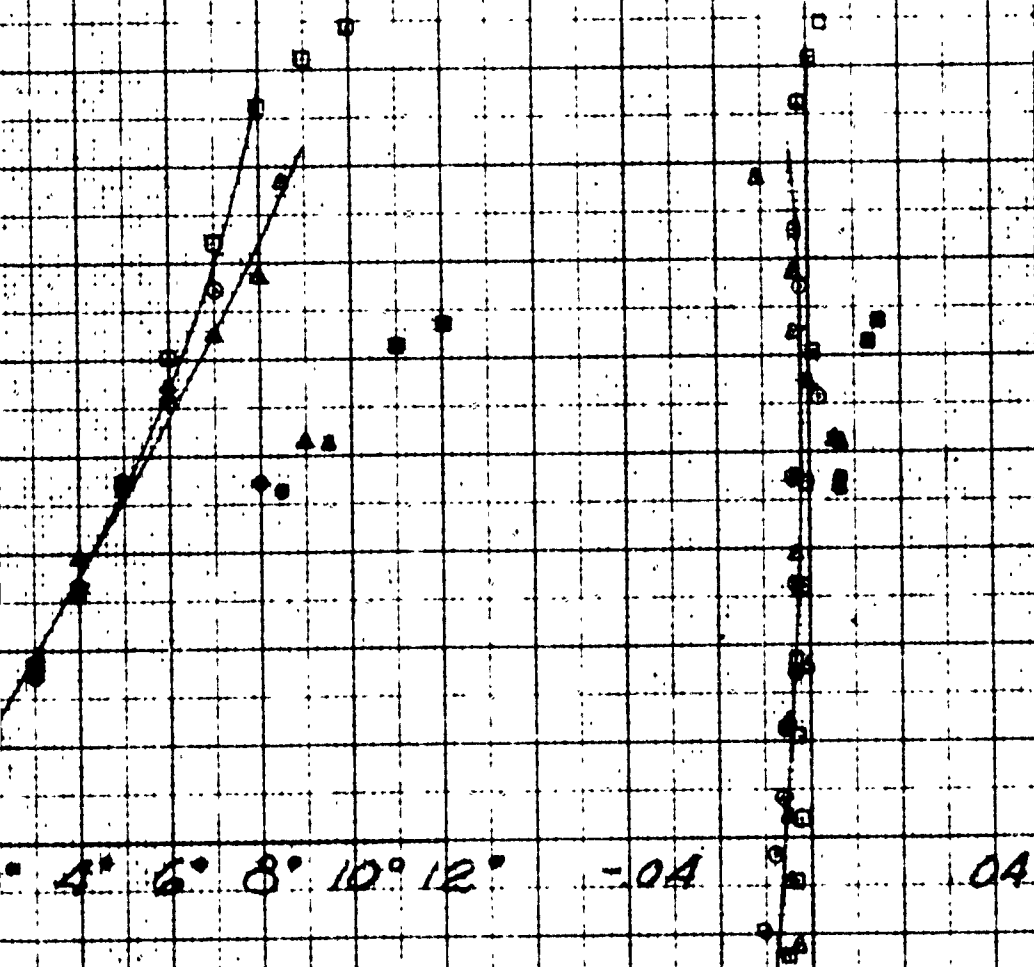
WHIRLING TANK TEST NO. 39

BUSHIPS FOIL NO. 10

M.A.C. = 222 FT - AREA = 8.4"

V = 40 KNOTS

NOTE:
1. DARKENED SYMBOLS
INDICATE VENTED FLOW.



ANGLE OF ATTACK
DEGREES

PITCHING MOMENT
COEFF - $C_{m/4}$

SYMBOL	$\frac{2}{\pi}$	FROUDE No.	CAV No.
○	.75	5.0990	A 3223
▲	1.00	4.4438	A 5866
●	1.25	4.0002	A 8058

NOTE

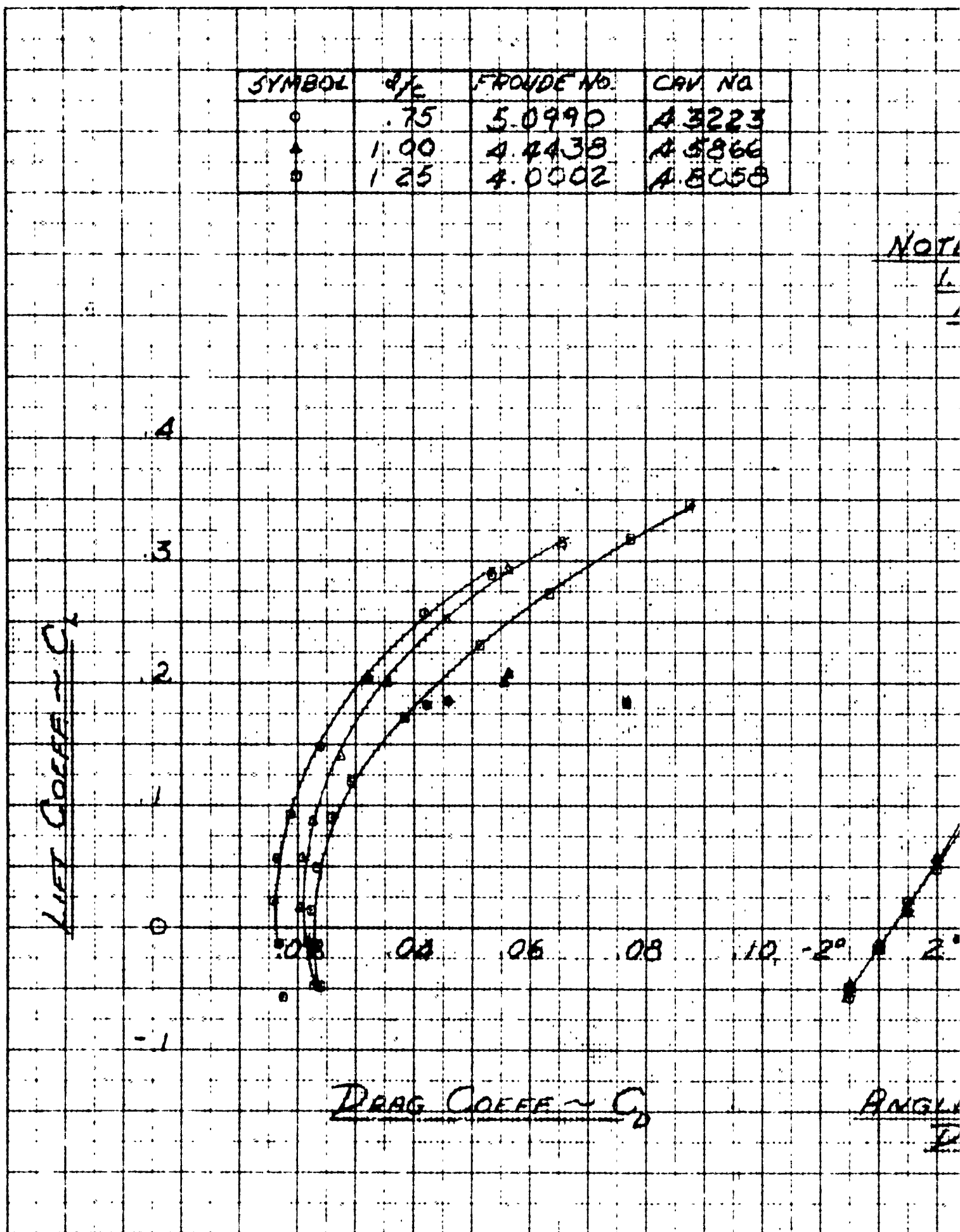
LIFT COEFF - C_L

4
3
2
1
0
-1

.03 .04 .05 .06 .08 .10 -2° 2°

DRAG COEFF - C_D

ANGLE
IN
DEGREES



WHIRLING TANK TEST NO. 39

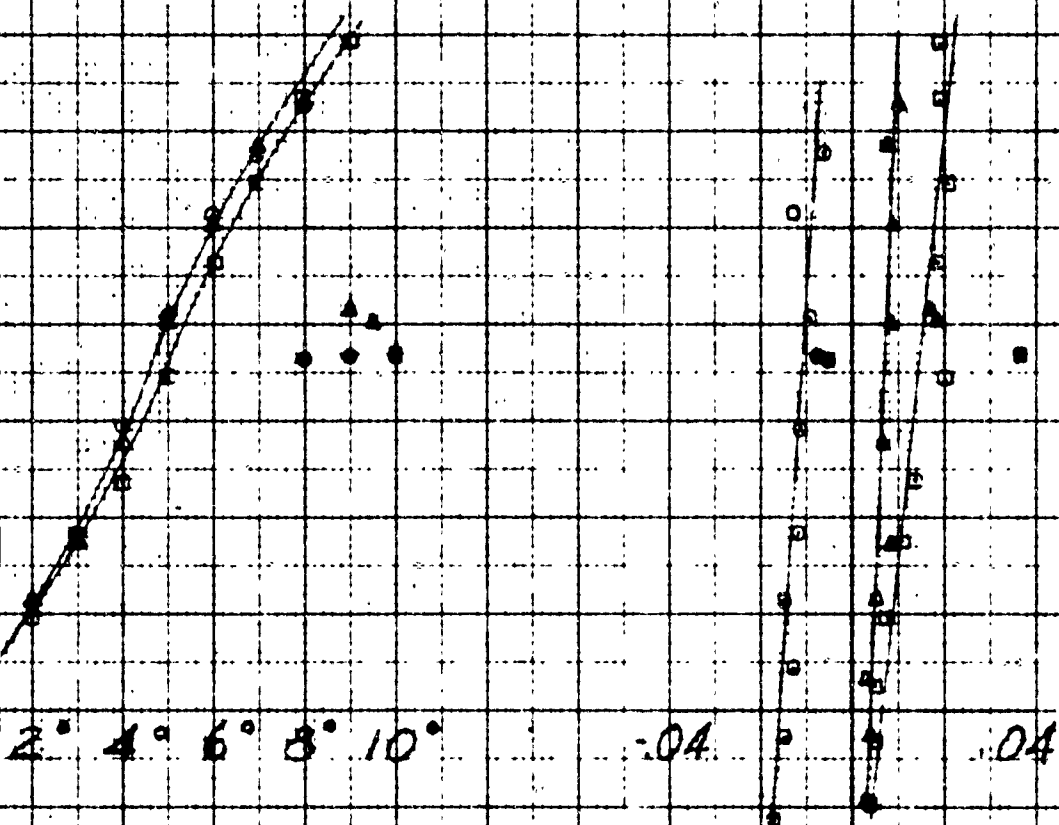
BUSHIPS FOIL NO. 10

M.A.C. = 222 FT ~ AREA = 8.9 "

V = 45 KNOTS

NOTE:

1. DARKENED SYMBOLS
INDICATE VENTED FLOW.



ANGLE OF ATTACK
DEGREES

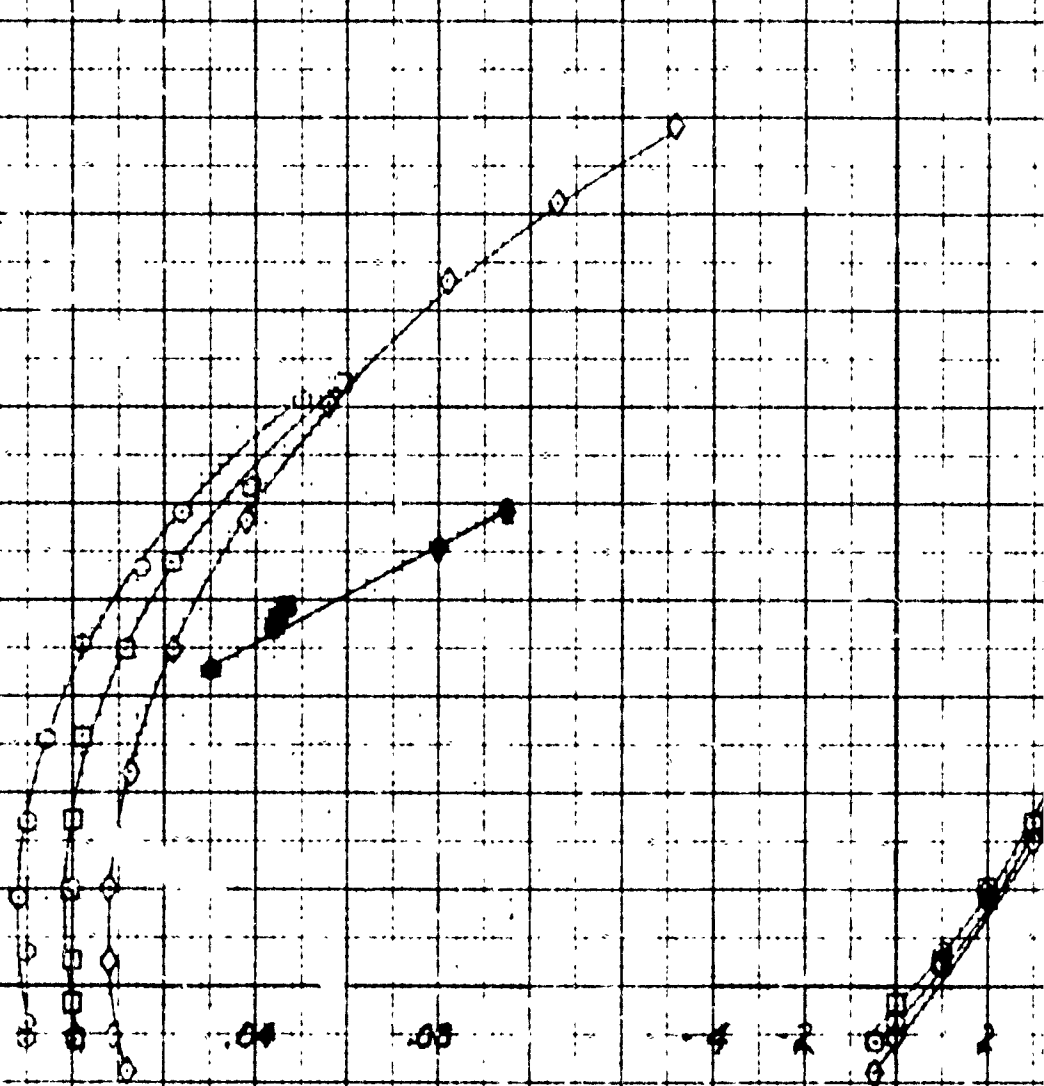
PITCHING MOMENT
COEFF. ~ CMc/4

ME 34
 25.00

CAR NO.	PROVIDE NO.	SYMBOL	d/MAC
5.22	5.0271	○	75
5.46	4.3197	□	1.00
5.65	3.9414	◇	1.25

LIFT COEFFICIENT ~ CL

DRAG COEFFICIENT ~ CD



WHIRLING TANK TEST NO 40

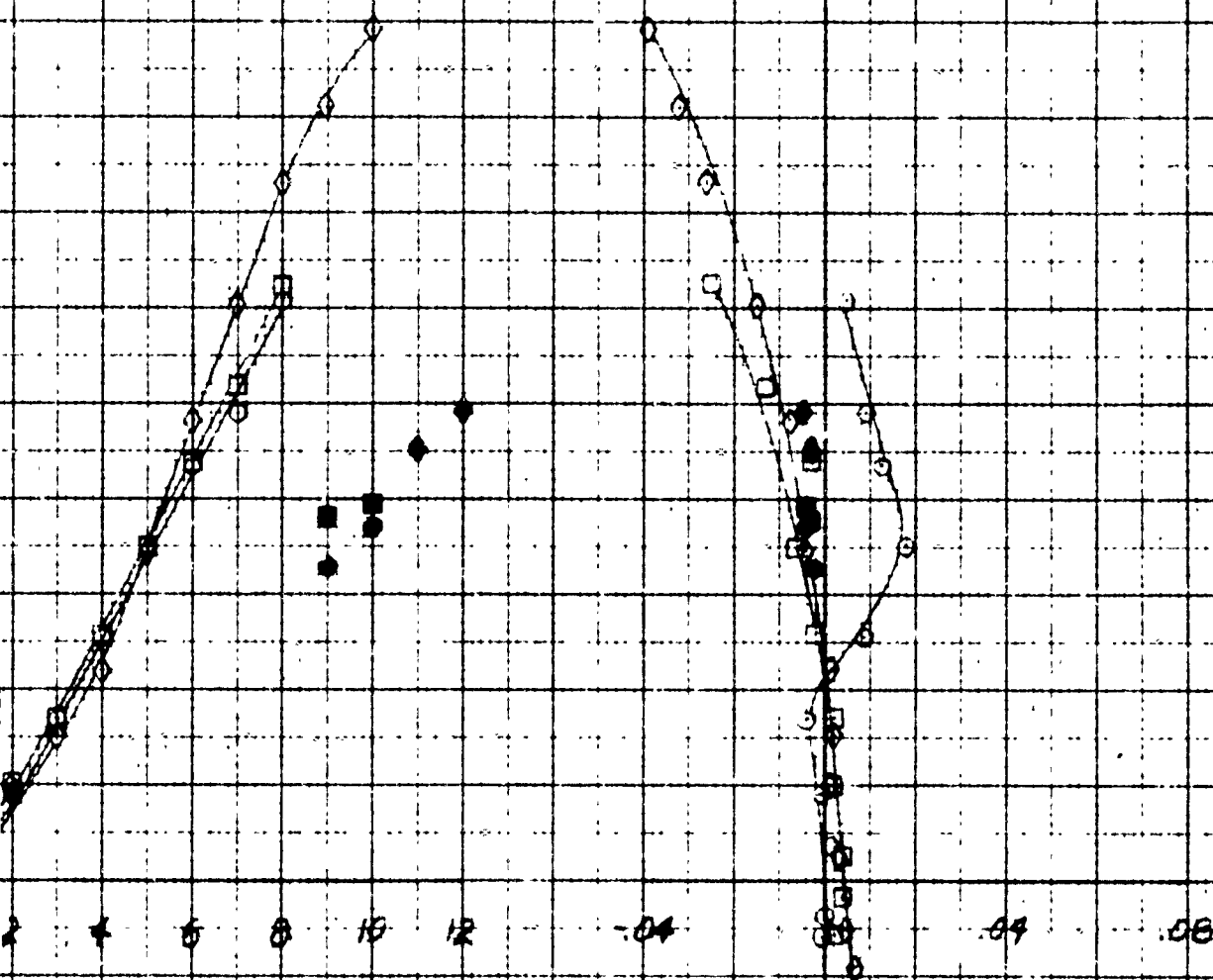
FOIL NO. 11

MAC. = 0.229 FT.

AREA = 9.60 SQ. IN.

 $V = 40$ M.T.S.NOTE:

1. DASHED SYMBOLS
INDICATE VENTED FLOW

ANGLE OF ATTACK $\sim \alpha$ DEGREESPITCHING MOMENT
COEFFICIENT $\sim C_m$

CAR NO	FRONDE NO	SYMBOL	d/MAC
429	50271	○	75
452	43797	□	100
479	39414	◇	125

LIFT COEFFICIENT - C_L

DRAG COEFFICIENT - C_D

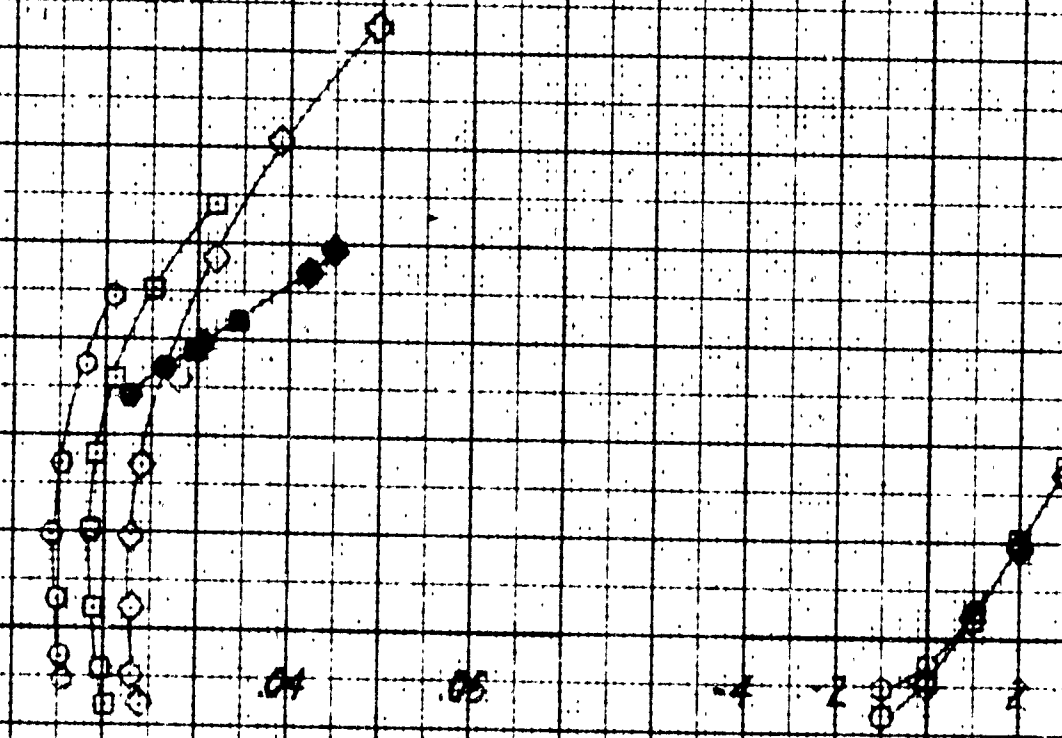
ANGLE

5
4
3
2
1
0

0.1 0.2 0.3 0.4 0.5 0.6 0.7 0.8 0.9 1.0

0.1 0.2 0.3 0.4 0.5 0.6 0.7 0.8 0.9 1.0

0 1 2 3 4 5 6 7 8 9 10

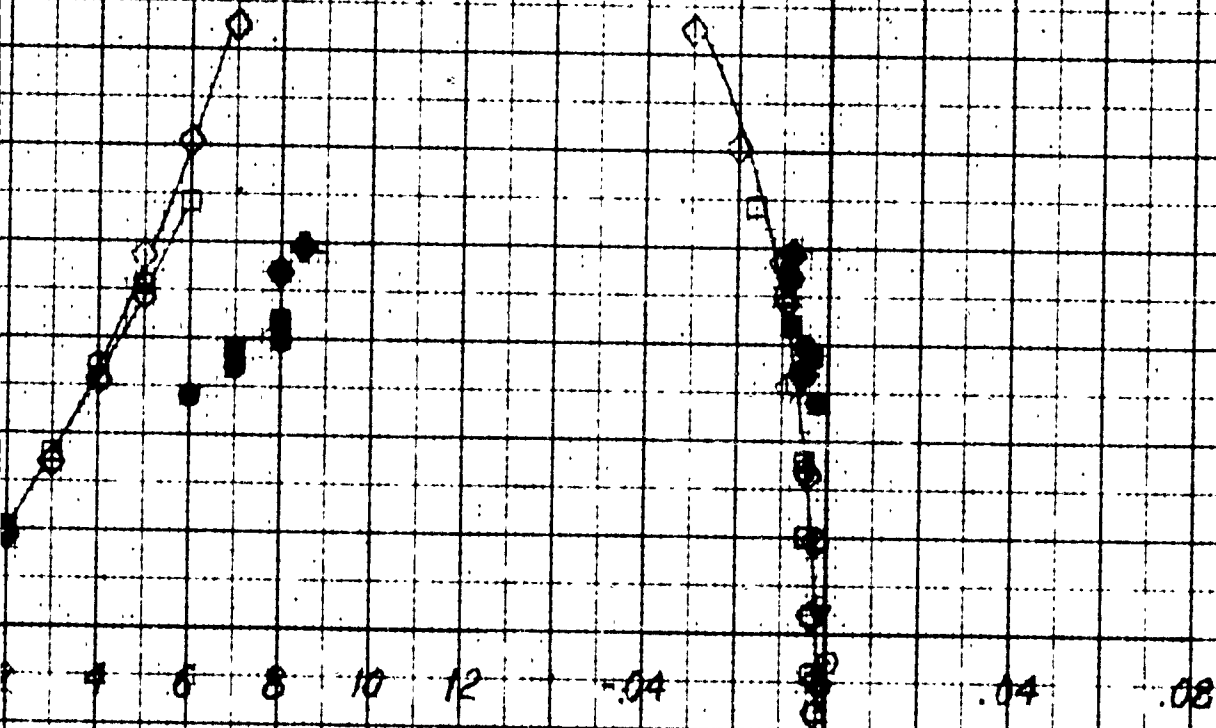


WHIRLING TANK TEST NO. 40.
FOIL NO. 11 MAC = 0.229 FT.
AREA = 9.60 SQ. IN.

$V = 45$ MTS.

NOTE:

1. DARKENED SYMBOLS
INDICATE VENTED FLOW.



ANGLE OF ATTACK - DEGREES

PITCHING MOMENT
COEFFICIENT - $CM_{M_{1/4}}$

NO. 14.55
P. 14.55
1955

WHIRLING TANK TEST NO. 40

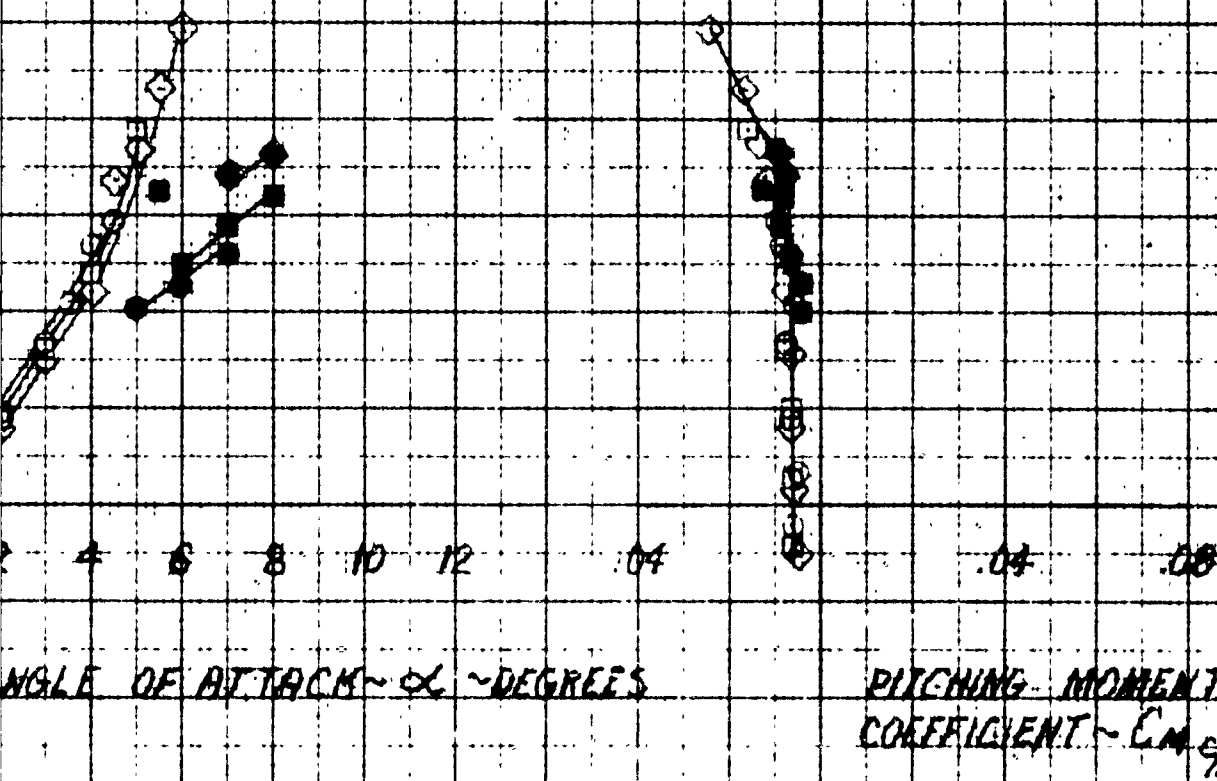
FOIL NO. 11 MAC = 0.229 FT.

AREA = 9.60 SQ. IN.

 $V = 50$ MTS.

NOTE:

1. DARKENED SYMBOLS
INDICATE VENTED FLOW.



14-3
 17-000000-100
 2510

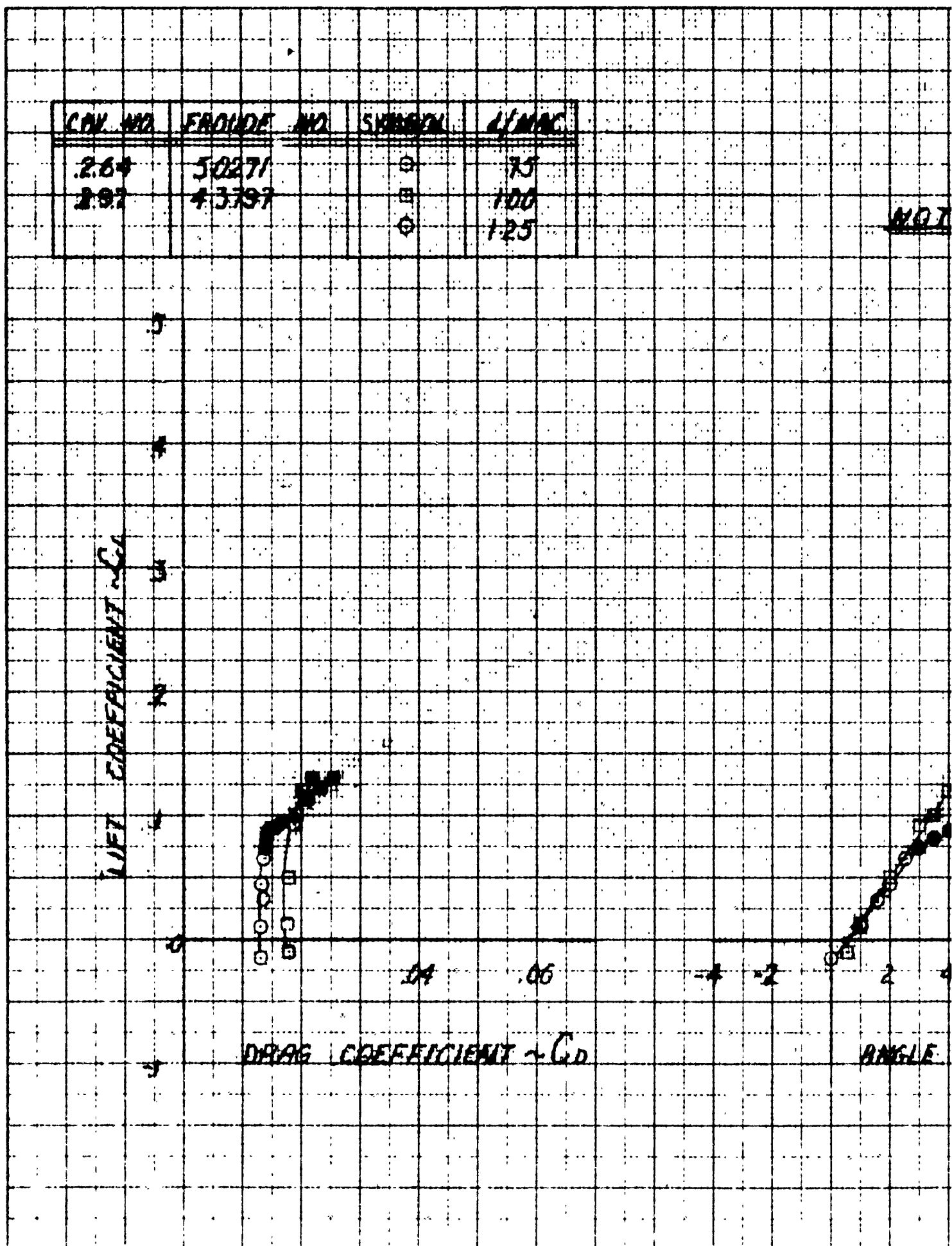
CON. NO.	FRONDE NO.	SKINNO.	L/MAC
2.64	50271	⊙	75
2.97	43797	⊠	100
		⊙	125

NOT

LIFT COEFFICIENT $\sim C_L$

DRAG COEFFICIENT $\sim C_D$

ANGLE



PAGE 157

WHIRLING TANK TEST NO. 40

FOIL NO. 11 MAC = 0.229 FT

AREA = 9.50 SQ IN

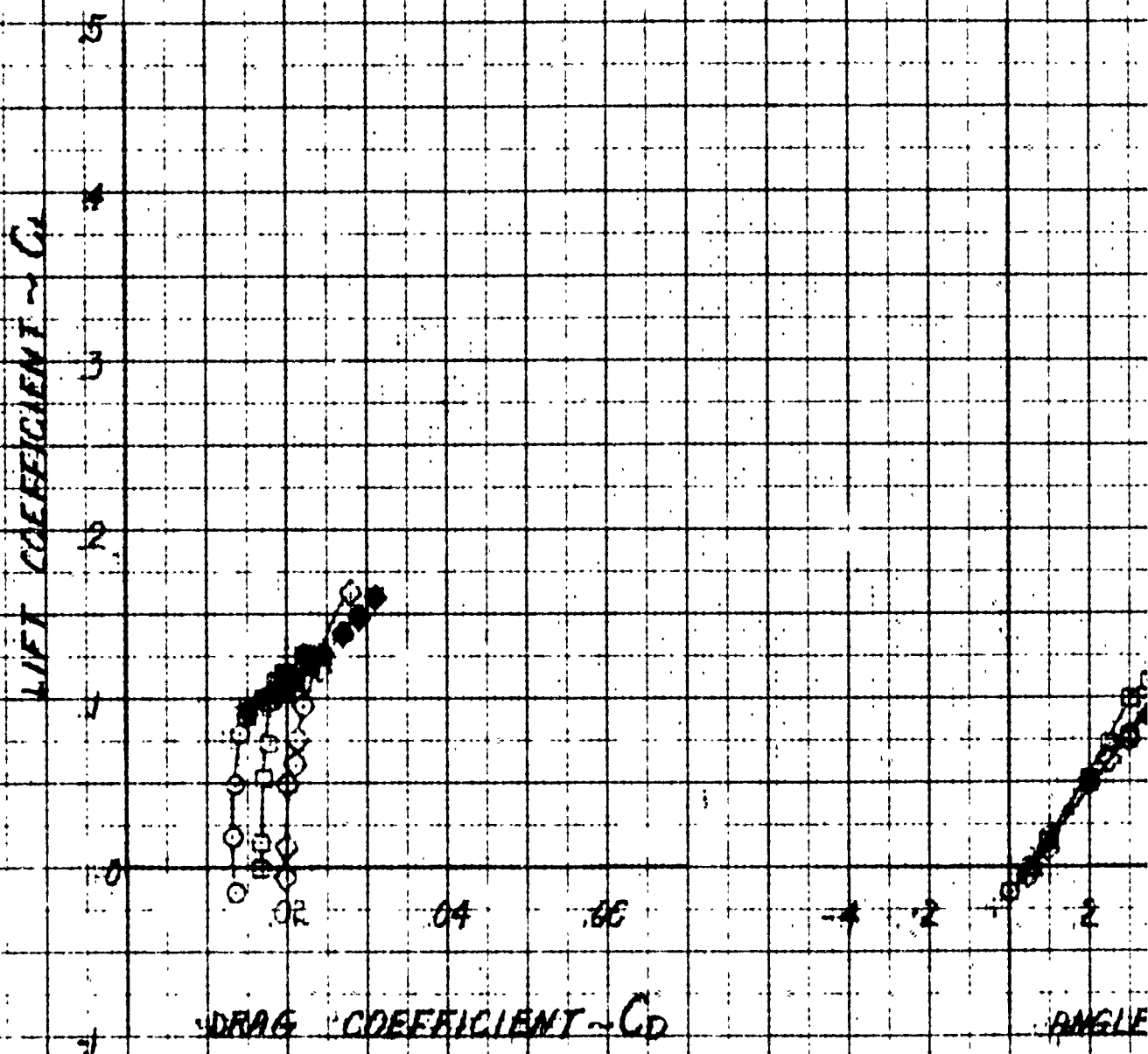
$V = 60$ KTS

NOTE:

1. DARKENED SYMBOLS
INDICATE VENTED FLOW.



CAN. NO.	FACONDE NO.	SYMBOL	L/MHC.
224	50271	⊕	.75
248	43797	⊗	1.00
271	39414	⊙	1.25



WHIRLING TRAIL TEST NO. 40

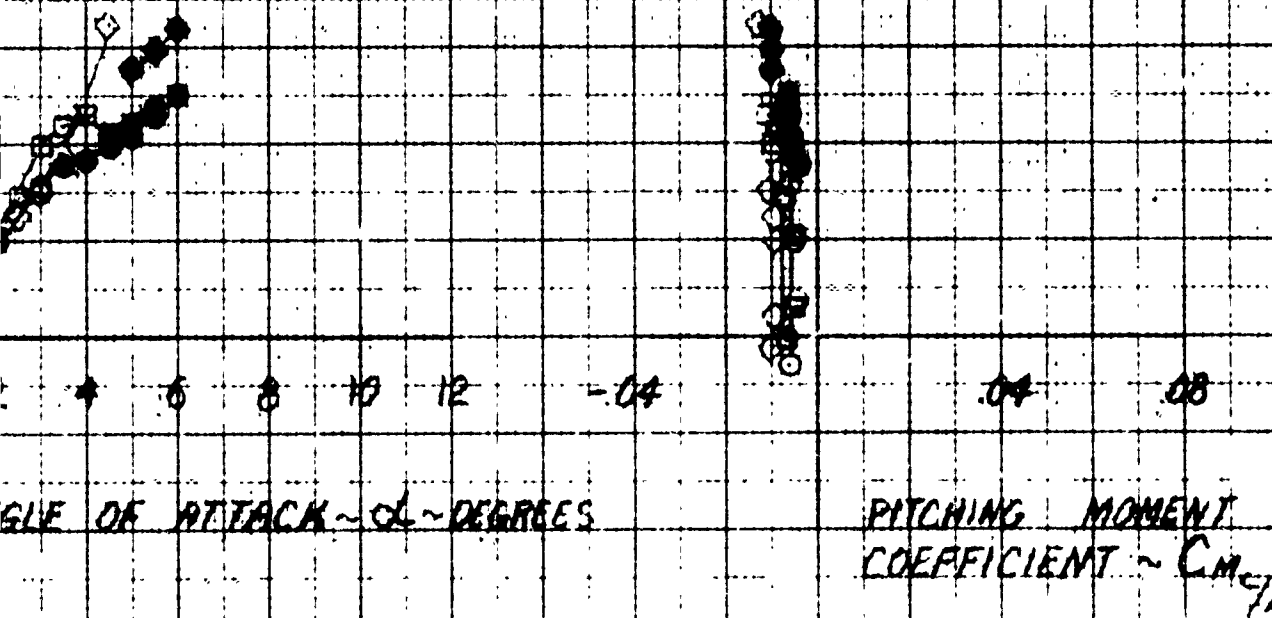
FOIL NO. 11 MAC = 0.289 FT.

AREA = 9.60 SQ. IN.

V = 70 KTS.

NOTE:

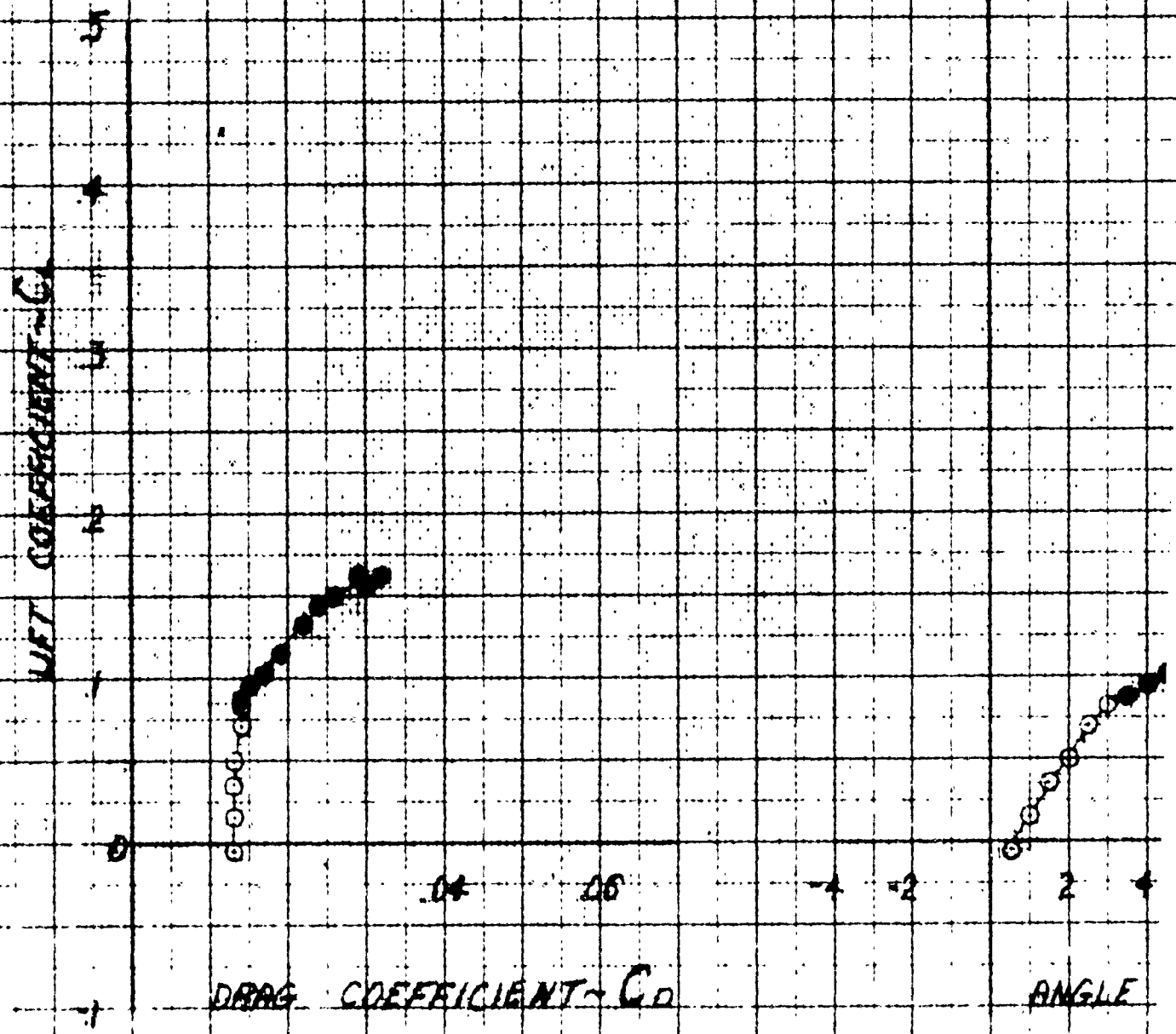
1. DRAINED SYMBOLS
INDICATE VENTED FLOW



ME 9
 1701 200000 3-4
 2-12

CAV NO	FRUIDE NO	SYMBOL	LINE
130	50271	•	75

NOTE:

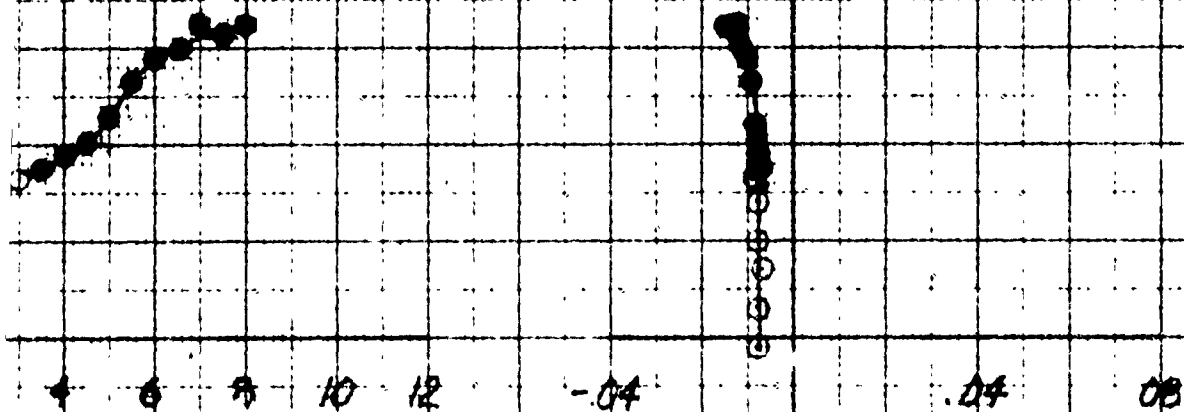


WHIRLING TANK TEST NO. 40
 FOIL NO. 11 MAC. = 0.229 FT.
 AREA = 9.60 SQ. IN.

V = 80 MTS.

1:
 DARKENED SYMBOLS
 INDICATE VENTED FLOW.

2. MODEL FAILED AT $\alpha = 8^\circ$



ANGLE OF ATTACK $\sim \alpha$ - DEGREES

PITCHING MOMENT
 COEFFICIENT $\sim C_{m, \alpha}$

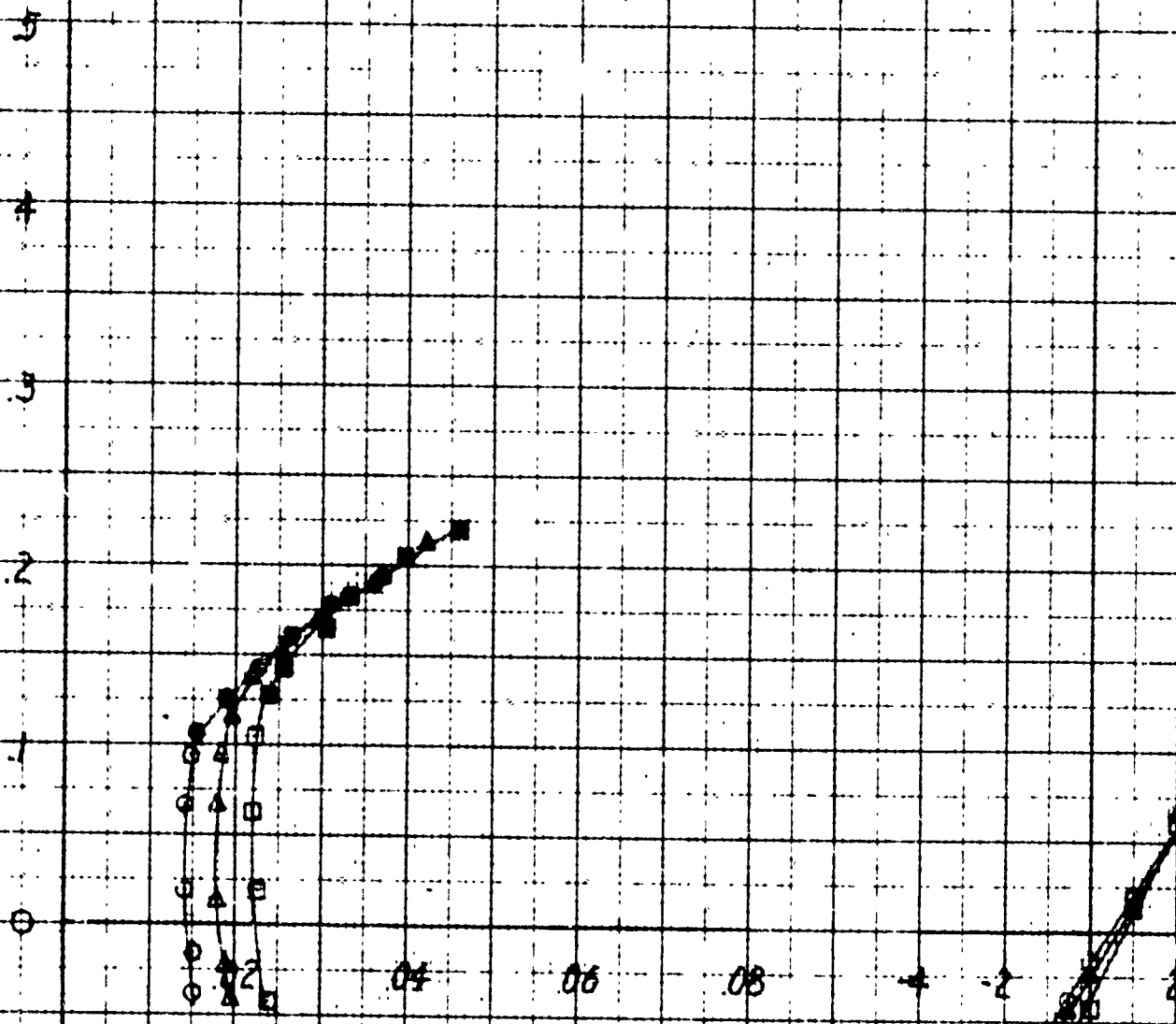
SYMBOL	d/c	FROUDE NO.	CAH NO.
○	.75	5.63	.505
△	1.00	4.89	.522
□	1.25	4.40	.542

NOTE:

NOE ALBANY, 1957

LIFT COEFFICIENT $\sim C_L$

DRAG COEFFICIENT $\sim C_D$

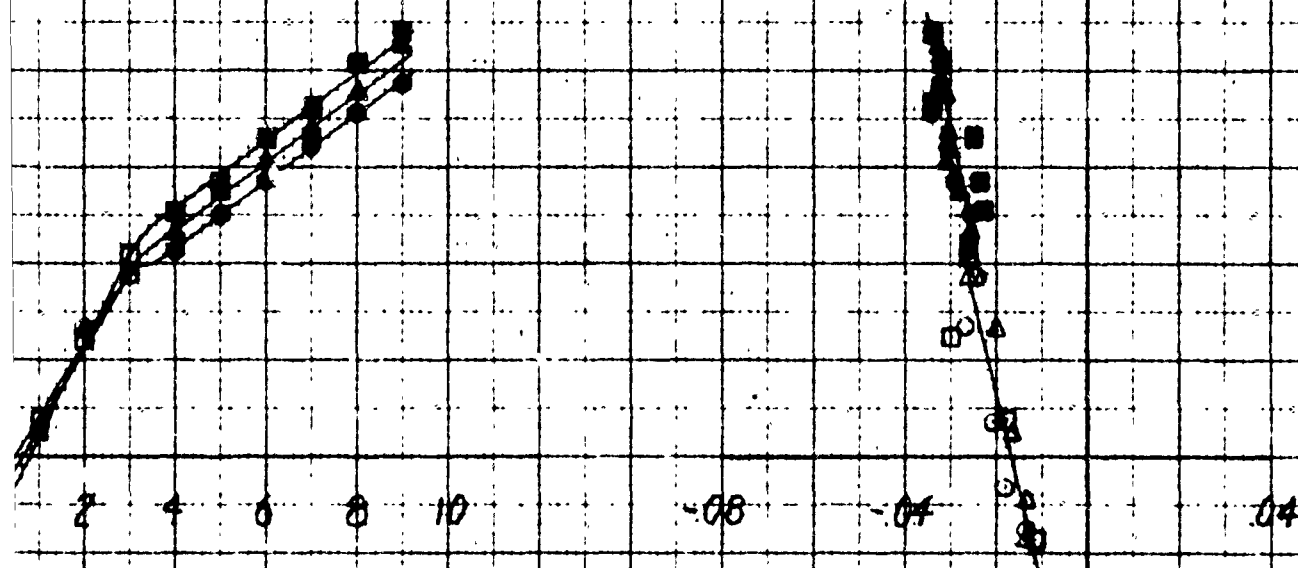


WHIRLING TANK TEST NO. 51

BU SHIPS FOIL NO. 12

MAC. = 182 FT ; $S = 8.00 \text{ m}^2$ $V = 40 \text{ KTS.}$

DARKENED POINTS
INDICATE REVERSED FLOW



ANGLE OF ATTACK ~ X-DEG

PITCHING MOMENT
COEFFICIENT ~ $C_{M_e/4}$

SYMBOL	d/c	FROUDE NO.	CAV. NO.
○	.75	5.6280	4.18
△	1.00	4.8924	4.29
□	1.25	4.4025	4.49

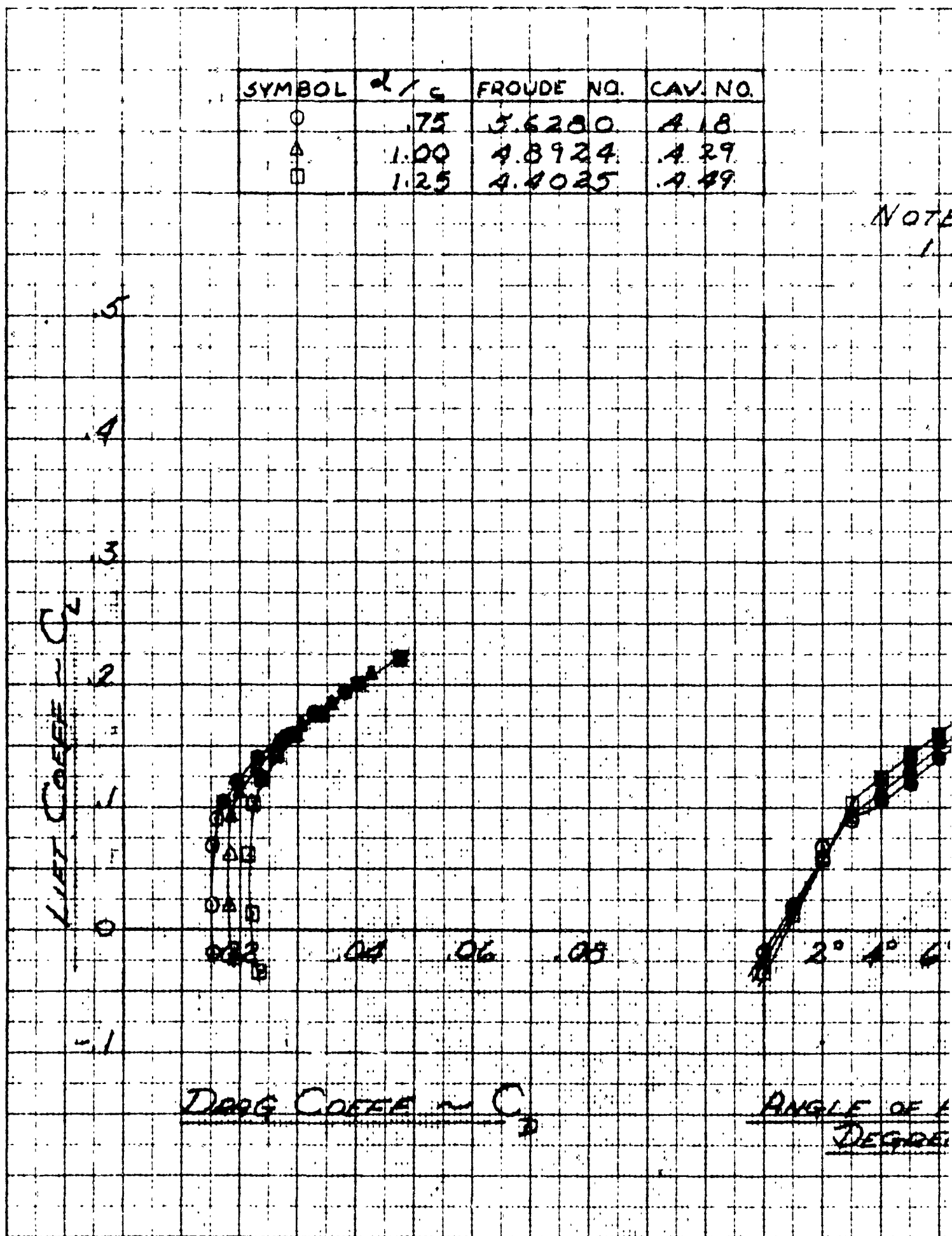
NOTE

WSE PLANNING 1007

LIFT COEFF C_L

DRAW COEFF C_D

ANGLE OF α
DEGREE

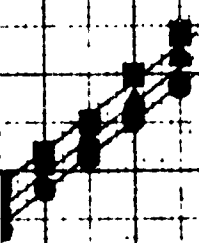


W.T.T. NO. 51 BUSHIPS FOIL NO. 12

M.A.C. = .182 FT. AREA = 8.0 in^2

VEL = 45 KTS

NOTE:
1. DARKENED SYMBOLS
INDICATE VENTED FLOW.

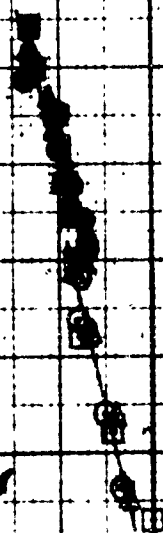


4° 8° 10°

ATTACK
ANGLES

-.08 -.04

PITCHING MOMENT
COEFF ~ C_m



WHIRLING TANK TEST NO. 51
 BU SHIPS FOIL NO 12
 W.L.C. = .182 FT S = 8.00 SQ
 VELOCITY = 50 KTS

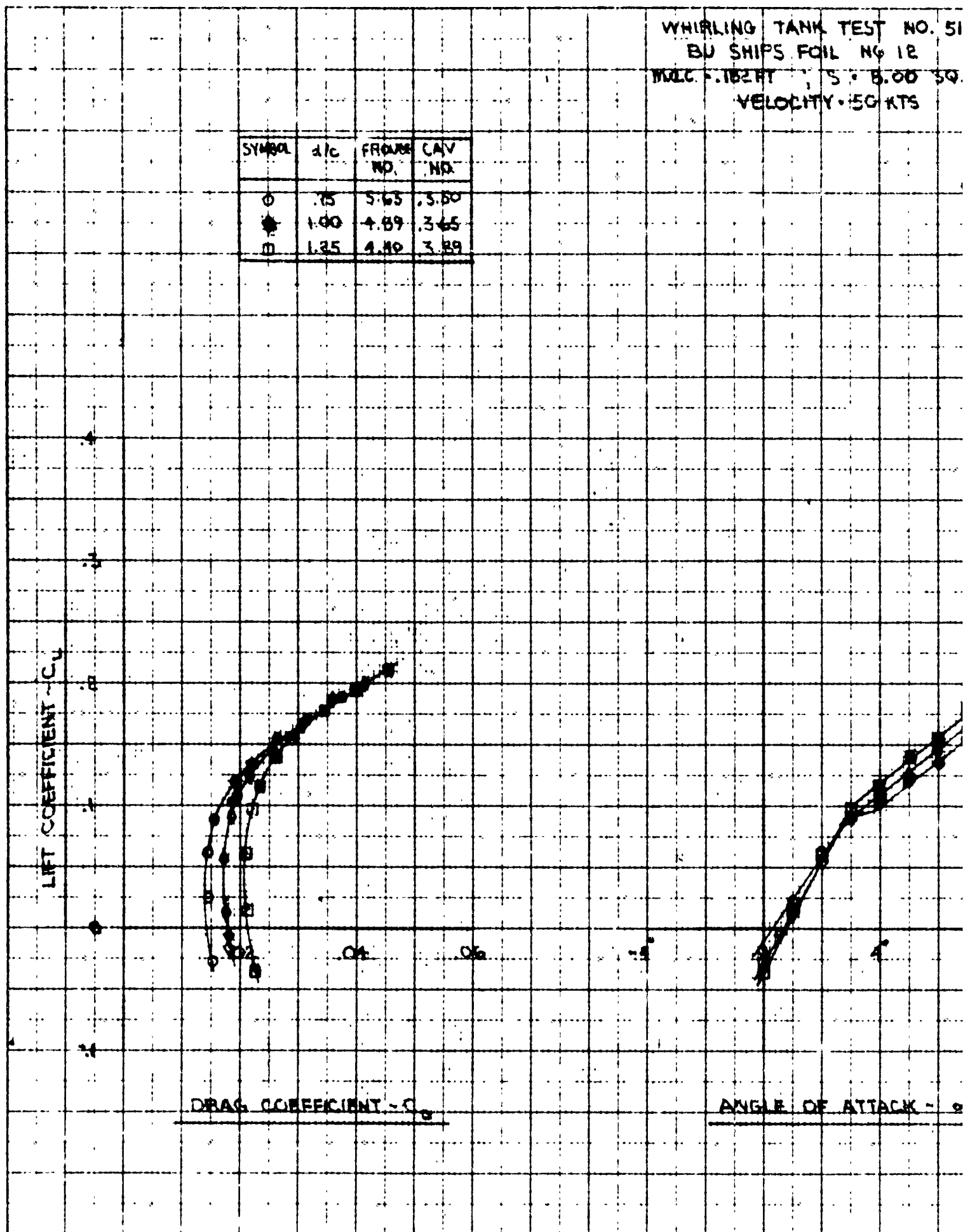
SYMBOL	d/c	FRAME NO.	CAV NO.
○	.75	5.65	3.50
●	1.00	4.89	3.65
□	1.25	4.10	3.89

NO. 2 ALUMINUM FOIL

LIFT COEFFICIENT - C_L

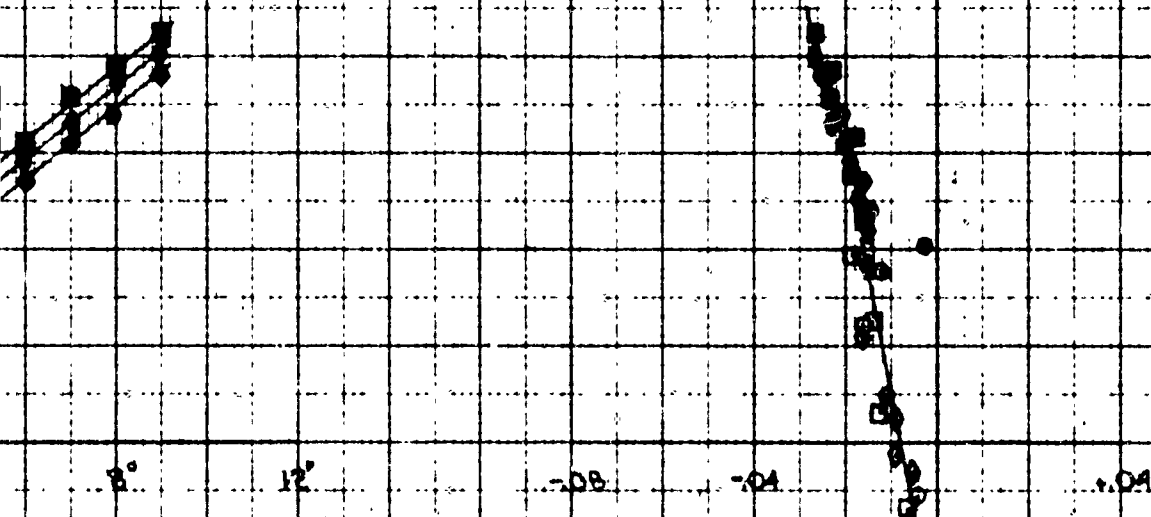
DRAG COEFFICIENT - C_D

ANGLE OF ATTACK - α



50 IN.

NOTE: DARKENED POINTS INDICATE VENTED FLOW

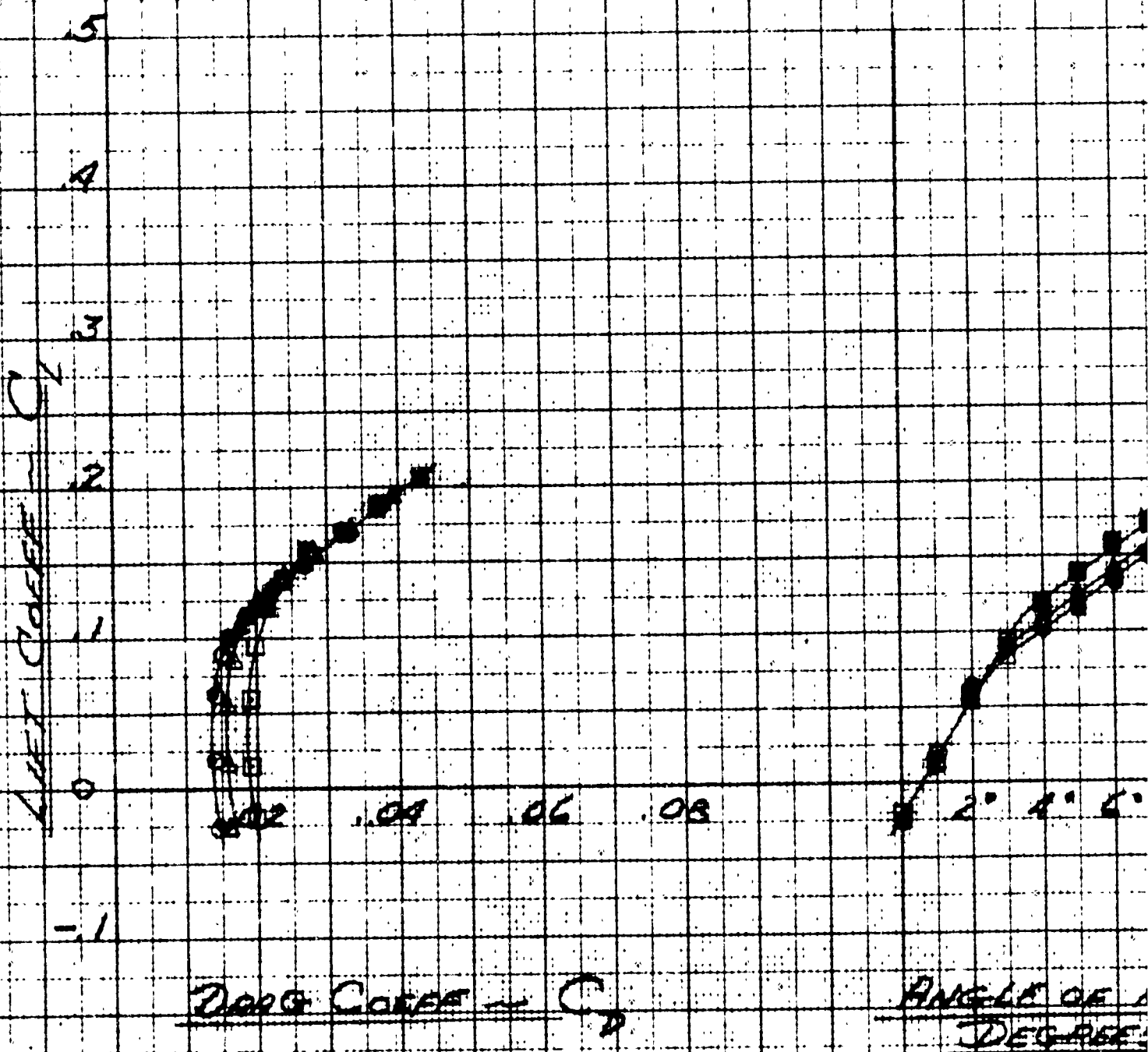


α - DEGREES

PITCHING MOMENT COEFFICIENT - C_m

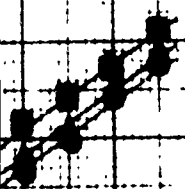
SYMBOL	d / x_c	FROUDE NO.	CAV. NO.
○	.75	5.6280	.261
△	1.00	4.8924	.280
□	1.25	4.4025	.298

NOTE:
1. DATA
INDEX



W.I.T. NO. 51. BUSHIPS FOL. NO. 12
M.A.C. = 182 FT. AREA = 8.000"
VEL = 60 KTS

DARKENED SYMBOLS
INDICATE VENTED FLOW



6° 8° 10°

-08 108 50

E ATTACK
SES

DITCHING MOMENT
COEFF = $C_{M \frac{1}{4}}$



PAGE 1264

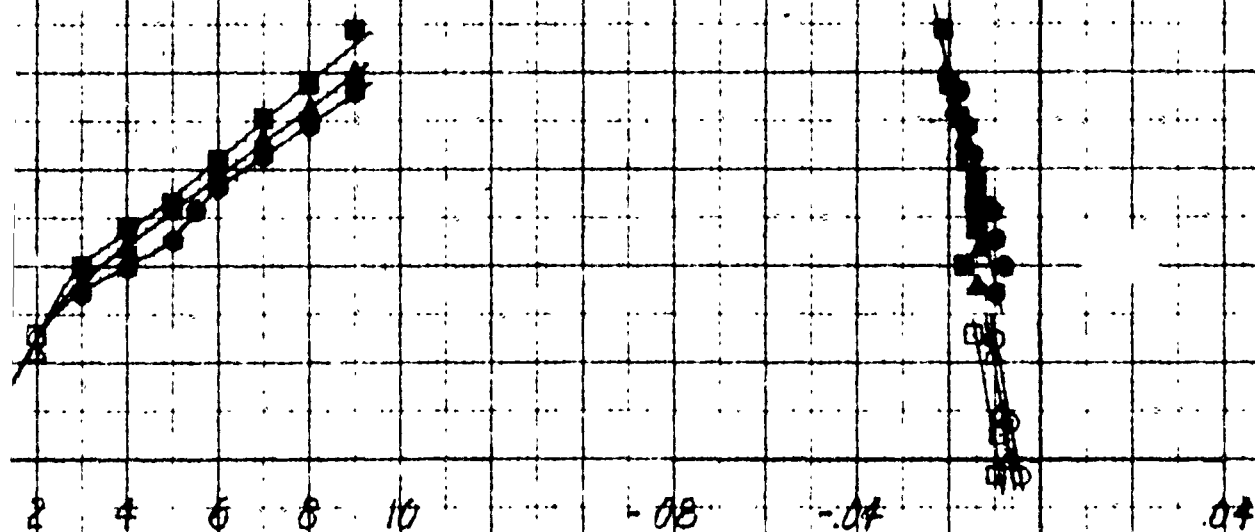
WHIRLING TANK TEST NO. 51

BU SHIPS FOIL NO. 12

MAC = 182 FT; S = 8.06 IN²

V = 70 KTS.

DARKENED POINTS
INDICATE VENTED FLOW.



ANGLE OF ATTACK - α - DEG

PITCHING MOMENT
COEFFICIENT - C_m

SYMBOL	θ / ϕ	FROUDE NO.	CAV. NO.
○	.75	5.6280	.172
△	1.00	4.8924	.192
□	1.25	4.4025	.211

NOTE

1.0

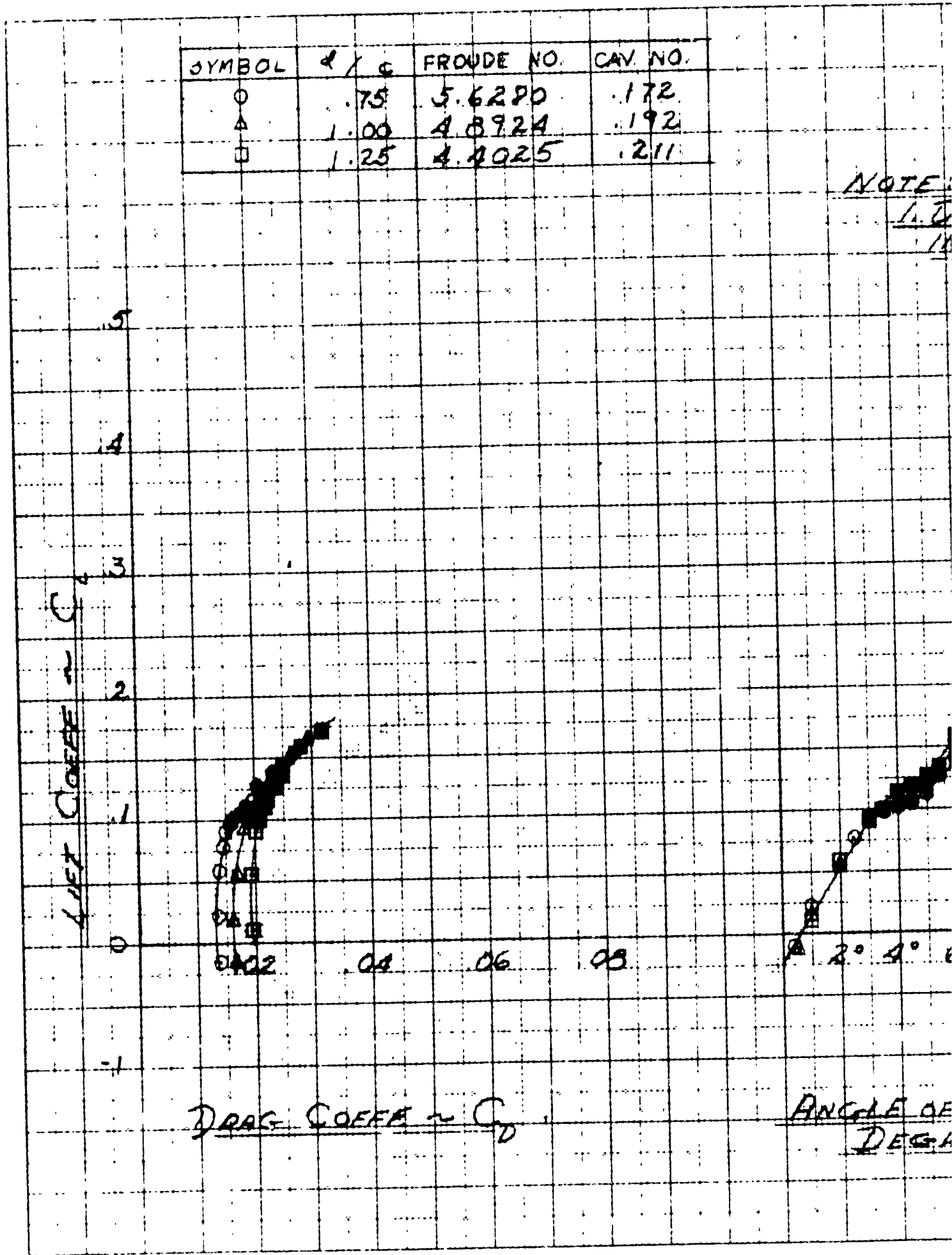
1.0

NOE ALABAMA 1967 2-5

LIFT COEFF C_L

DRAG COEFF C_D

ANGLE OF
DEGREE



W.T.T. NO. 51 BUSHIPS FOIL NO. 12

M.A.C. = 182 FT. AREA = 8.004 "

VBL = 80 KTS

DARKENED SYMBOLS
INDICATE VENTED FLOW.



° 6° 8° 10°

-.08

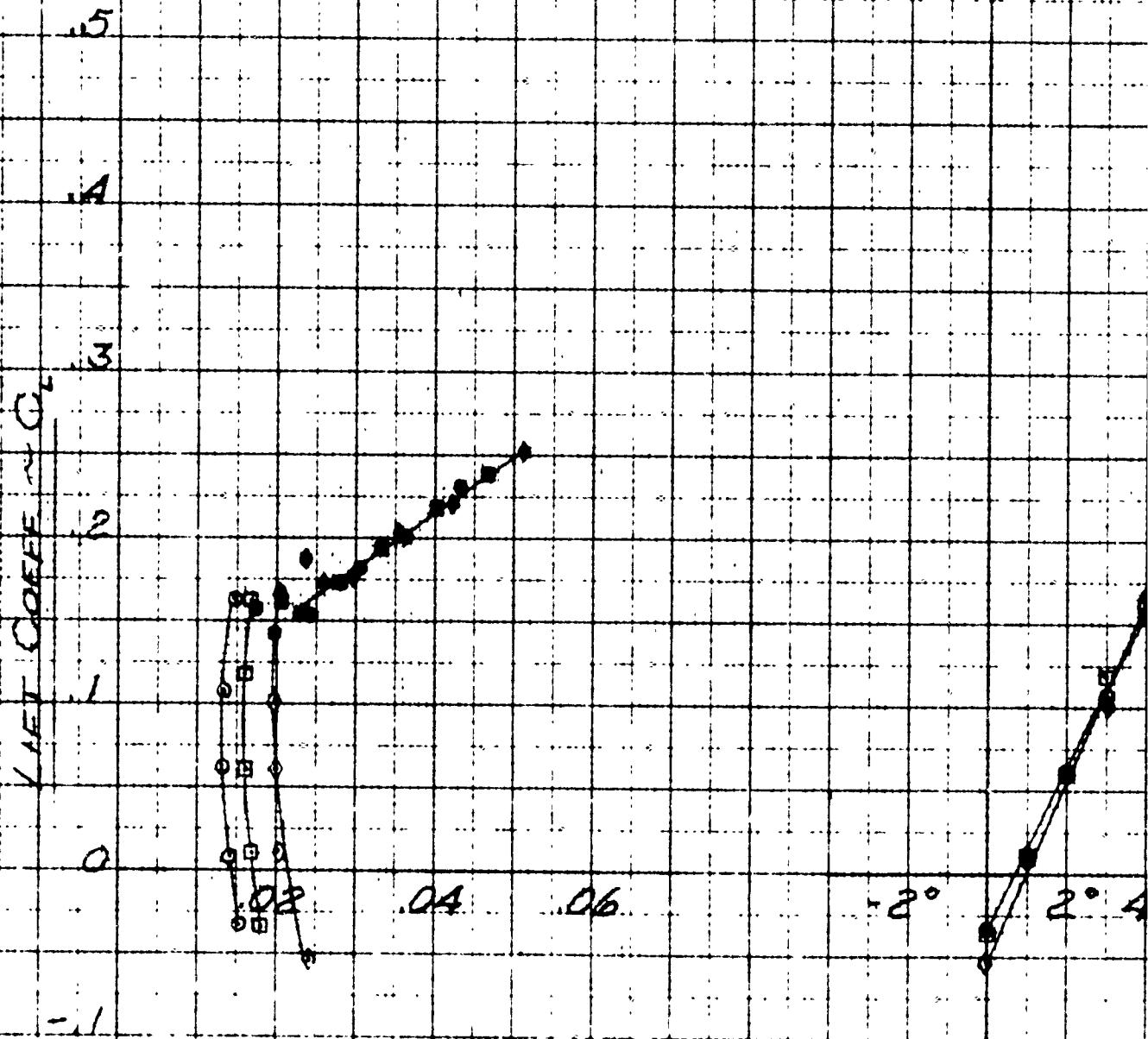
-.04

DE ATTACK
GREES

PITCHING MOMENT
COEFF - $C_{M/A}$

FROUDE NO.	SYMBOL	$R/\mu AC$	CARV. NO.
6.1234	●	.75	A 97
5.3219	□	1.00	515
4.7785	◆	1.25	546

NOTE:
1. DARKENED
INDICATE

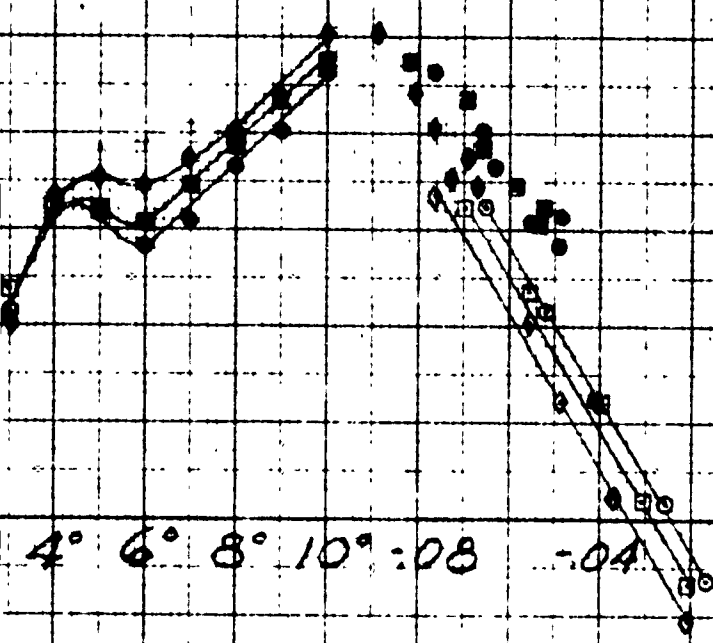


DRAG COEFF ~ C_D

ANGLE OF

WHIRLING TANK TEST NO 41
 BUSHIPS FOIL NO. 13
 M.A.C. = 152 FT AREA = 8.50 IN
 V = 40 KTS

ENED SYMBOLS
 TE VENTED FLOW

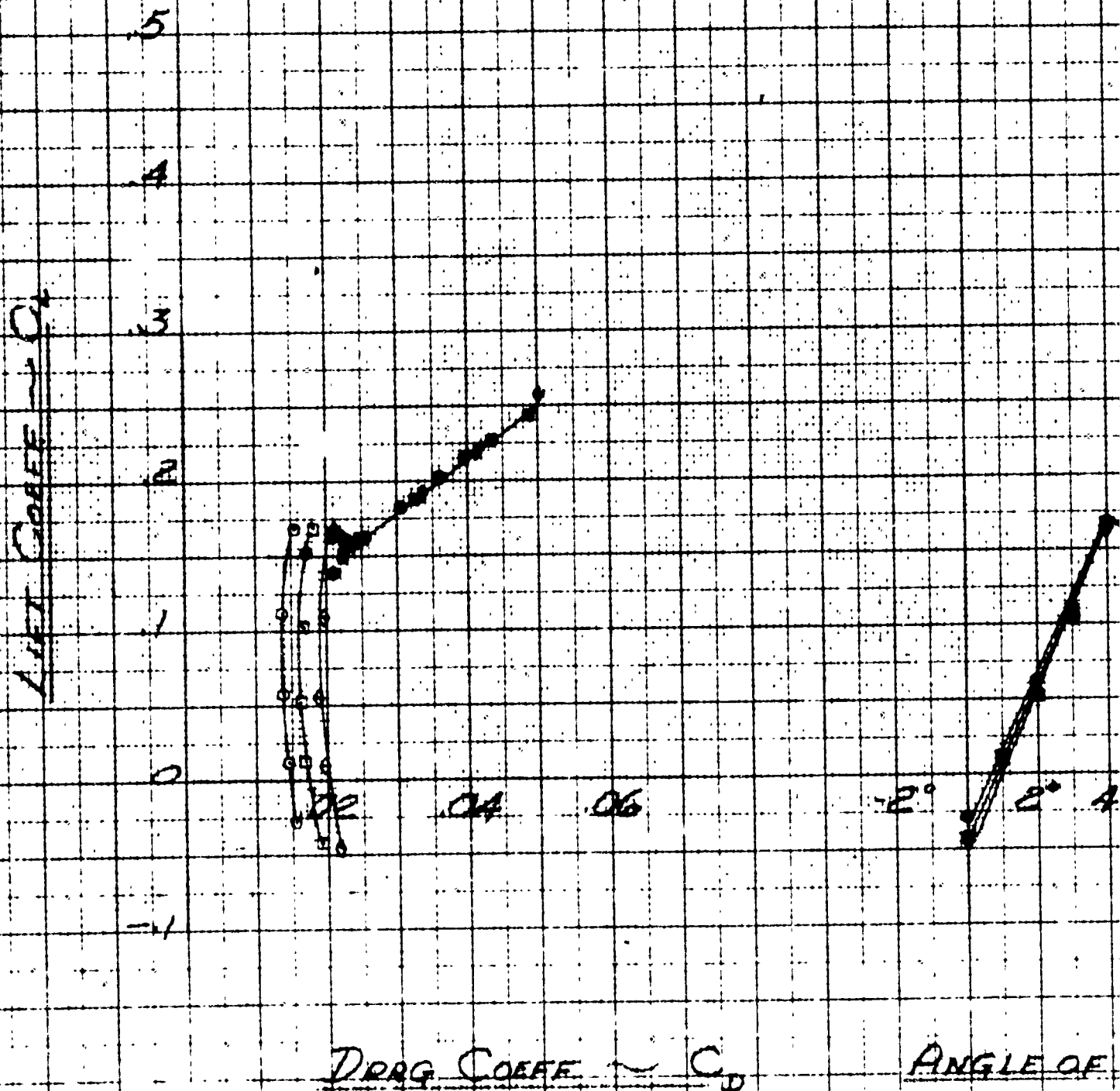


ATTACK ~ 4 ~ DEG. PITCHING MOMENT COEFF
 C_M ABOUT 9% OF MAC

MEASUREMENT

PROBE NO.	SYMBOL	$\frac{q}{\rho V_{\infty}^2}$	CAL NO.
6.1234	○	.75	4.06
5.3219	□	1.00	4.23
4.7705	●	1.25	4.37

NOTE:
1. DARKEN
INDICATE



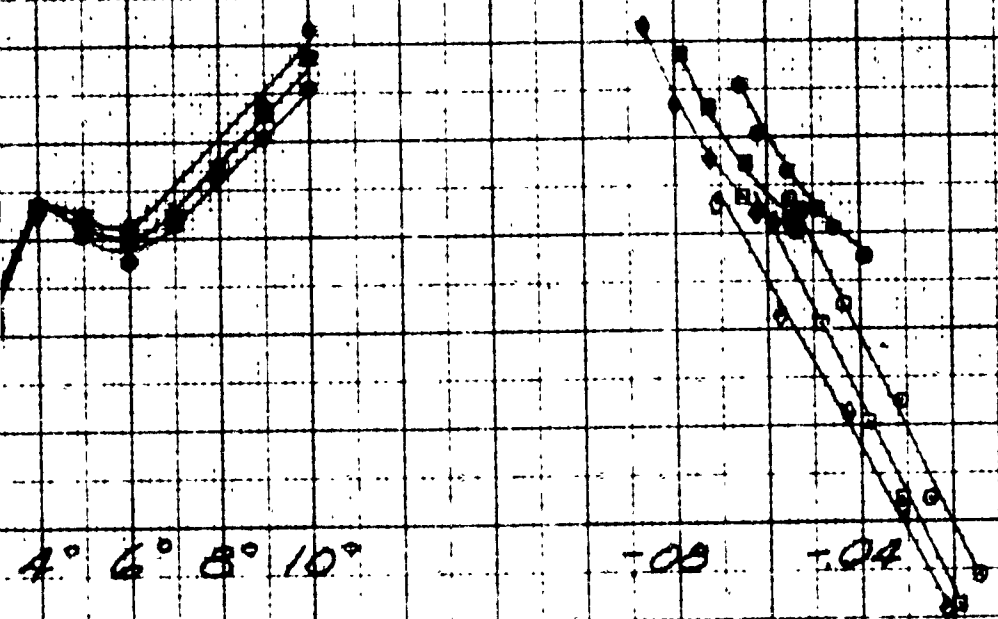
WHIRLING TANK TEST NO. 41

BUSHIPS FOIL NO. 13

MAC = 1.52 FT. AREA = 8.50 IN.

V = 45 KTS

UNED SYMBOLS
OF VENTED FLOW



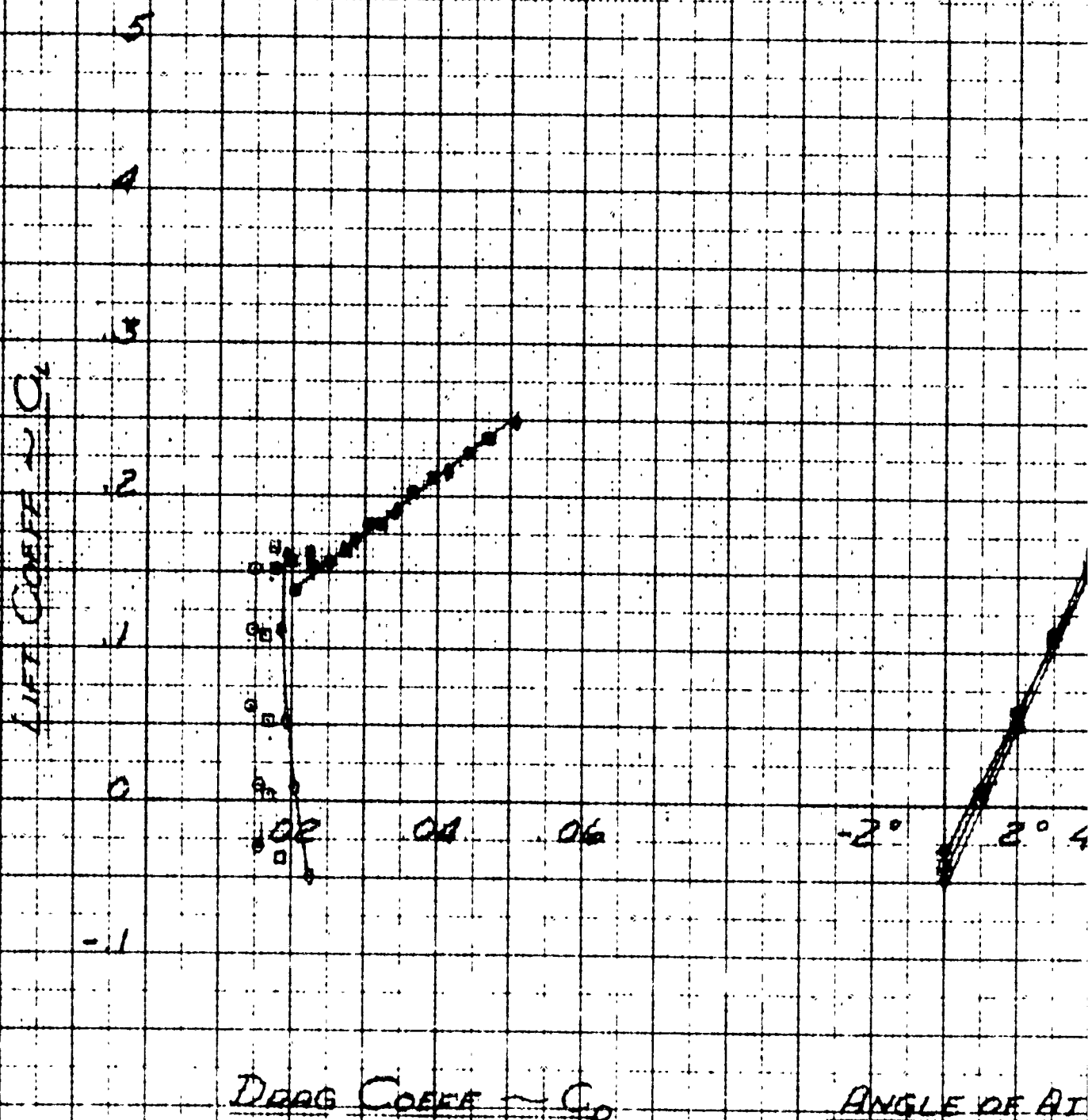
OF ATTACK ~ 4 ~ DEG.

PITCHING MOM. COEFF

C_M abt 1% of N₄C

SYMBOL	ρ/μ	FRONDE NO.	CAV NO.
o	.75	6.1234	3.41
o	1.00	5.3219	3.57
o	1.25	4.7785	3.74

NOTE:
1. DARKER
INDICAT



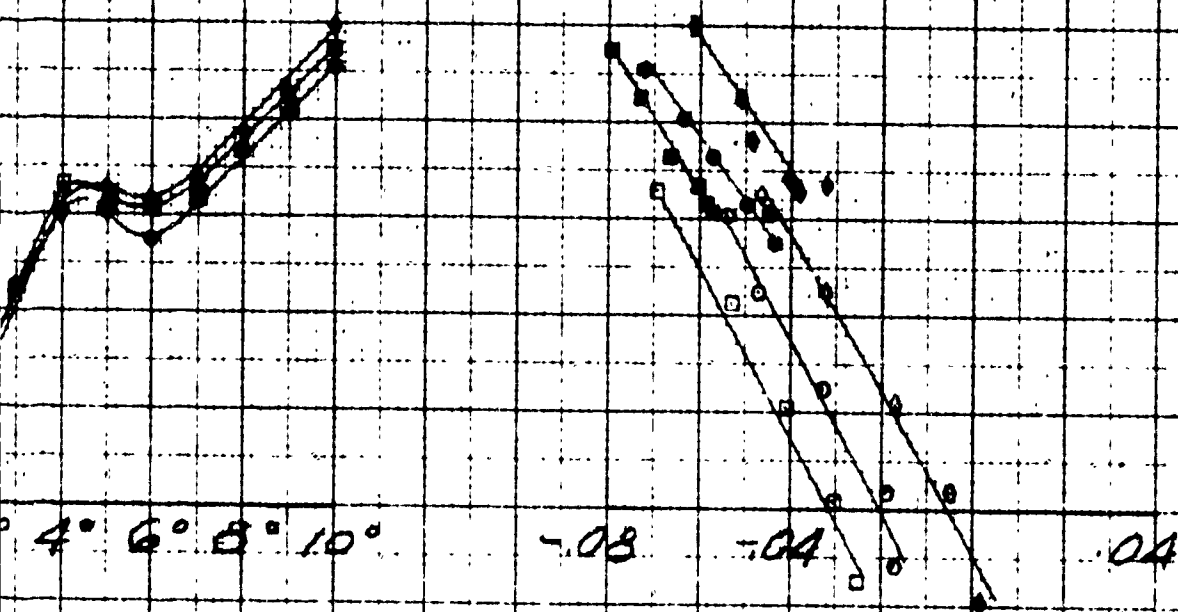
WHIRLING TANK TEST NO. 41

BUSHIPS FOIL NO. 13

M.A.C. = 152 FT AREA = 8'10"

$V = 50$ KTS

OPENED SYMBOLS
STATE VENTED FLOW



ATTACK α - DEG

PITCHING MOMENT COEFF - C_m

ABOUT 1/2 OF MGC

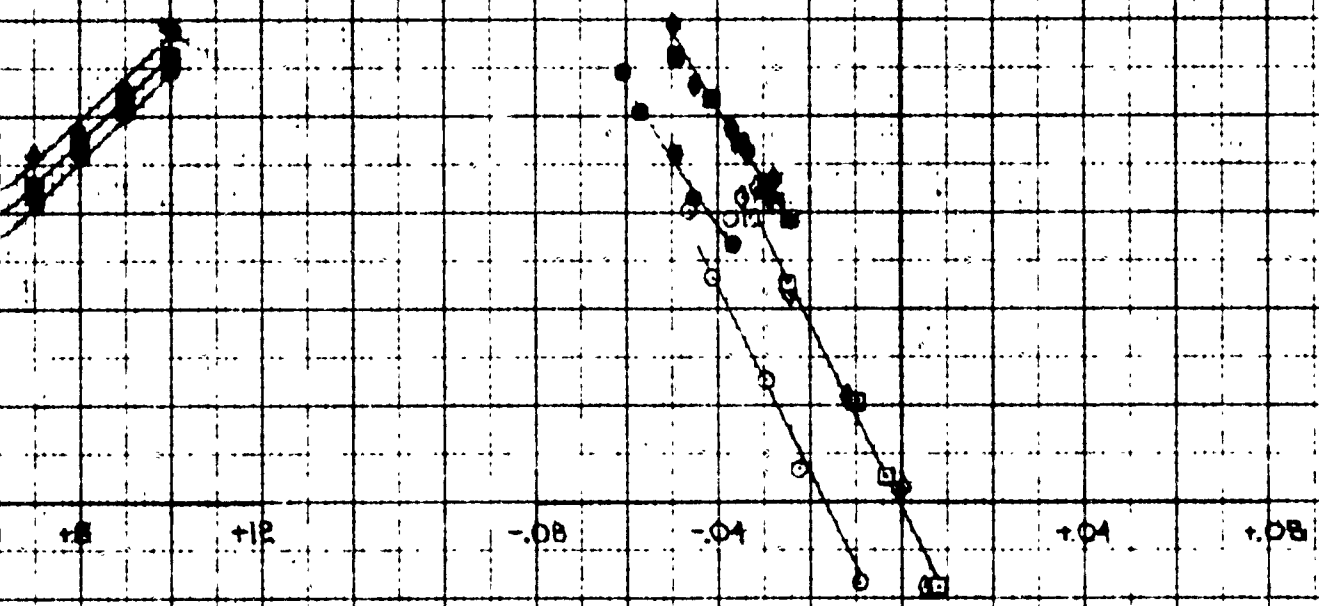
VELOCITY - 60 KTS

SYMBOL	LA	FRAUDE NO.	CAV NO.
●	.75	6.12	.253
■	1.00	5.32	.270
◆	1.05	4.78	.286



AT NO 41
13
S. 8.00 IN.
13

NOTE: DARKENED POINTS INDICATE
VENTED FLOW



DEGREES

PITCHING MOMENT COEFFICIENT - C_m

ABOUT 92 OR M.P.S.

WHIRLING TANK TEST. N

BU SHIPS FOIL # 13

MAX. DISC FT. 5.800

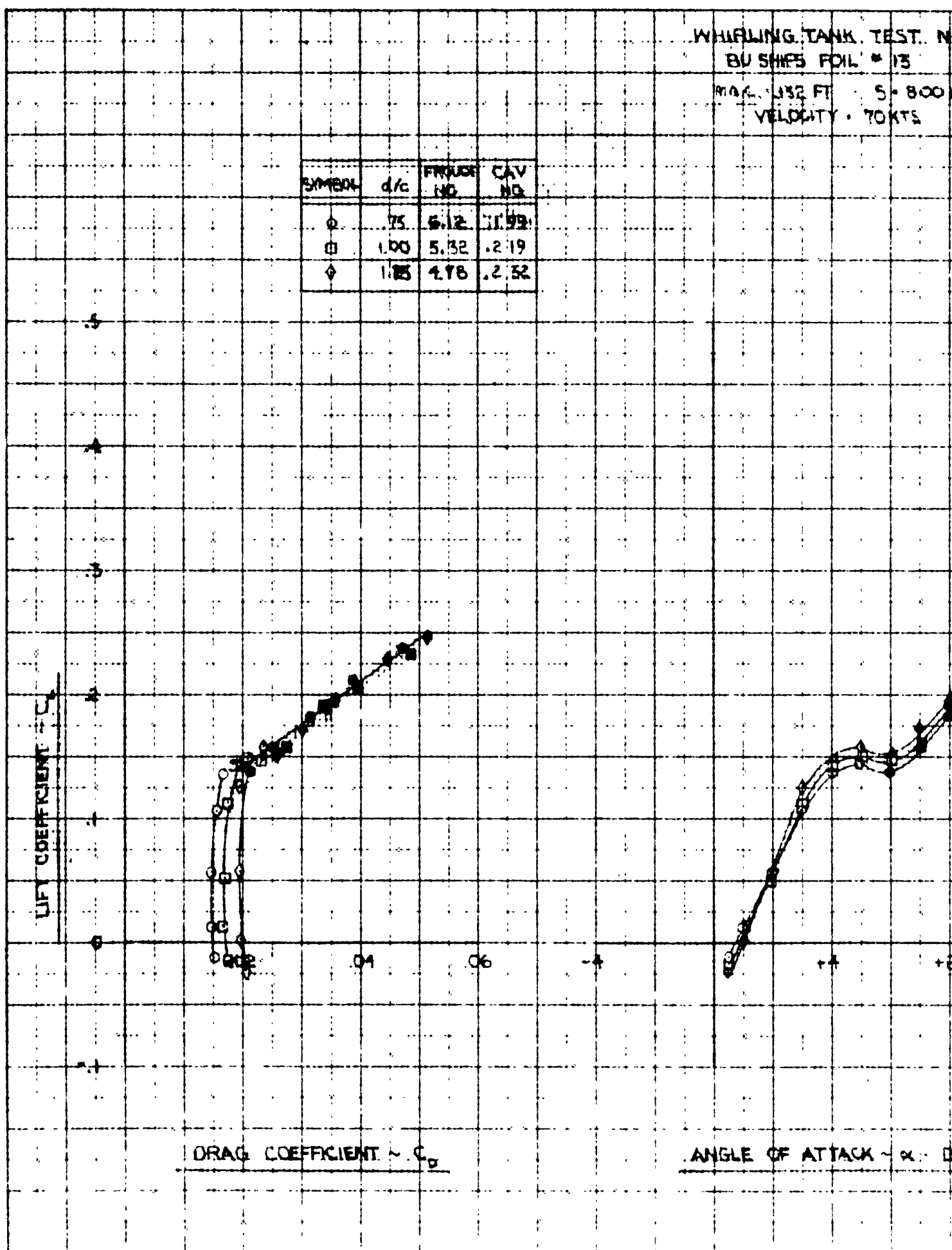
VELOCITY - TOKTS

SYMBOL	d/c	FROUDE NO.	CAV NO.
○	.75	5.12	11.99
□	1.00	5.32	12.19
◇	1.25	4.78	12.52

LIFT COEFFICIENT - C_L

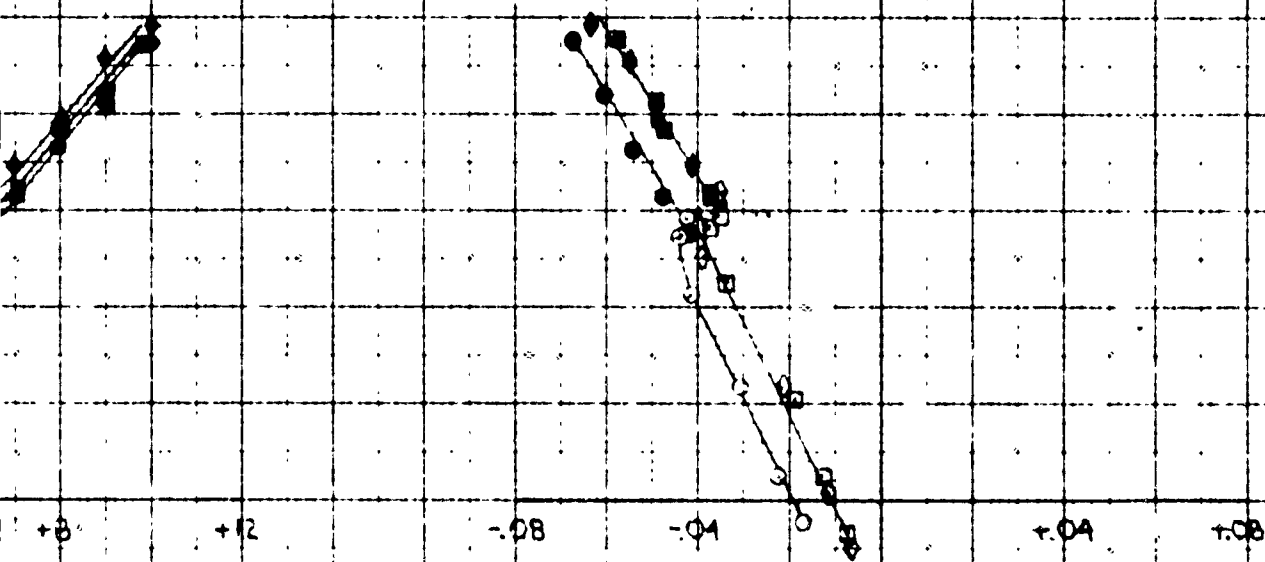
DRAG COEFFICIENT - C_D

ANGLE OF ATTACK - α - D



300 IN²
S

NOTE: CIRCLED POINTS INDICATE VENTED FLOW.



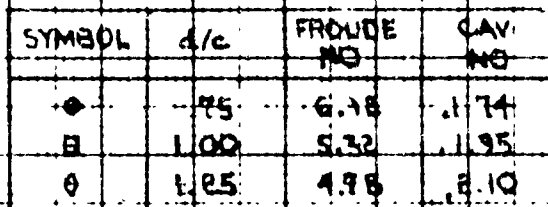
DEGREES

PITCHING MOMENT COEFFICIENT - C_m

ABOUT 2% OF AREA

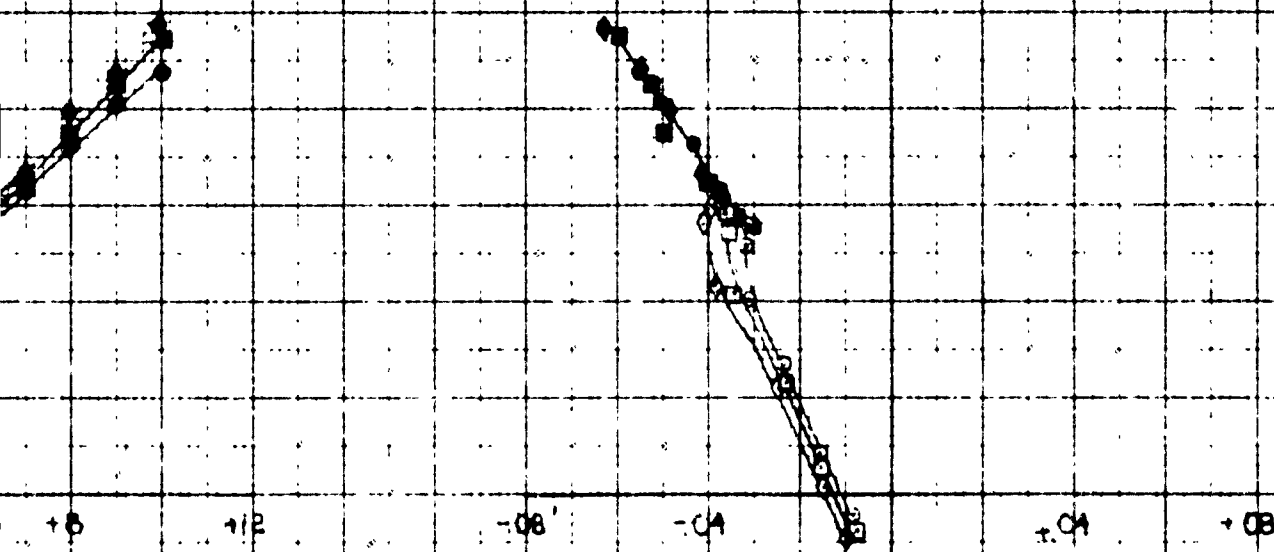
100. 30000 304

VELOCITY - 80 KTS



ONE
J

NOTE DARKENED POINTS INDICATE
VENTED FLOW



DEGREES

PITCHING MOMENT COEFFICIENT - C_m

SCALE 7% OF RANGE

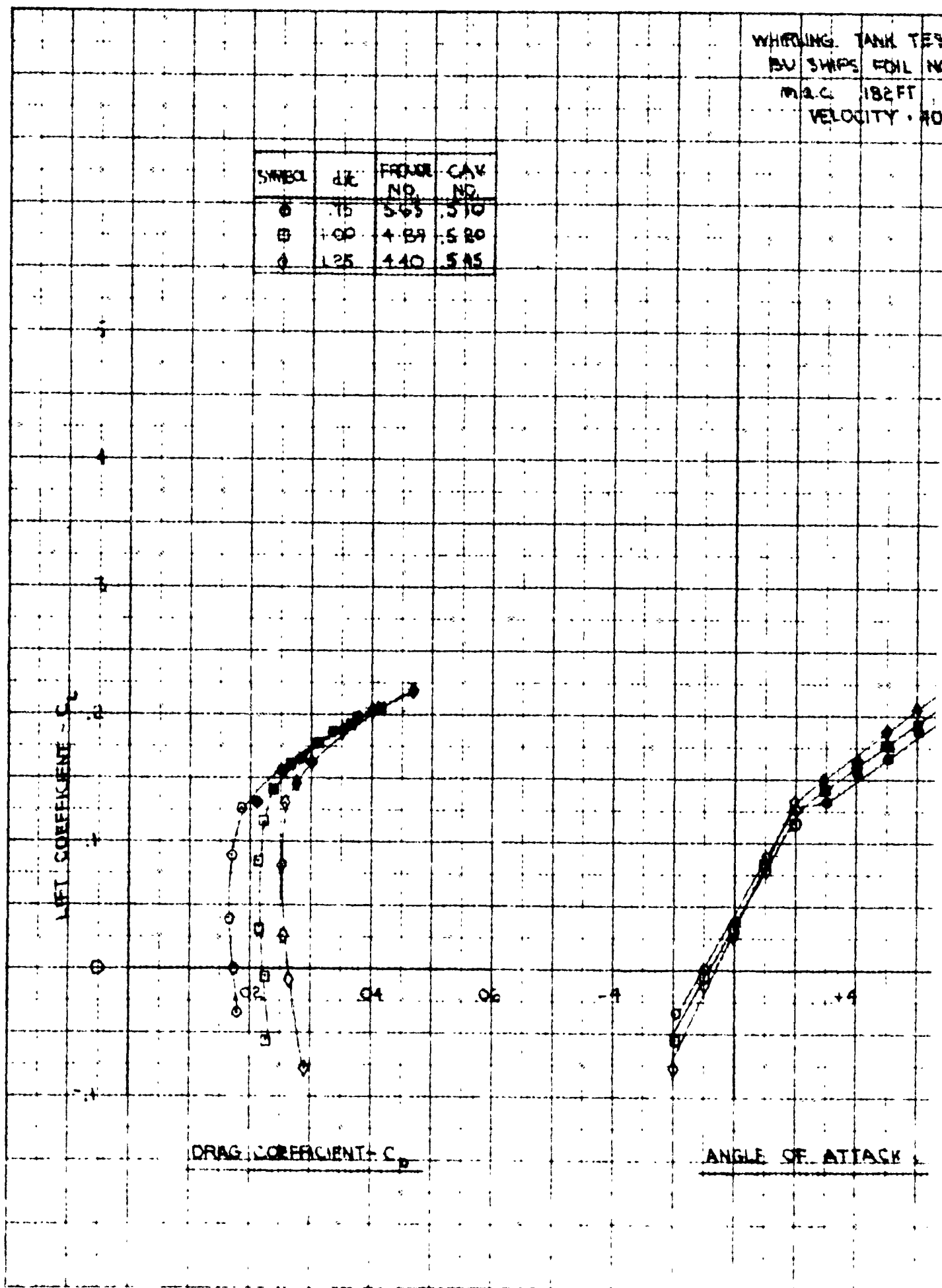
WHIRLING TANK TEST
BU SHIPS FOIL NO
M.A.C. 182 FT
VELOCITY .40

SYMBOL	d/c	FRONT NO.	CAV NO.
○	.75	5.65	5.10
□	.90	4.89	5.80
◇	1.25	4.40	5.45

LIFT COEFFICIENT - C_L

DRAG COEFFICIENT - C_D

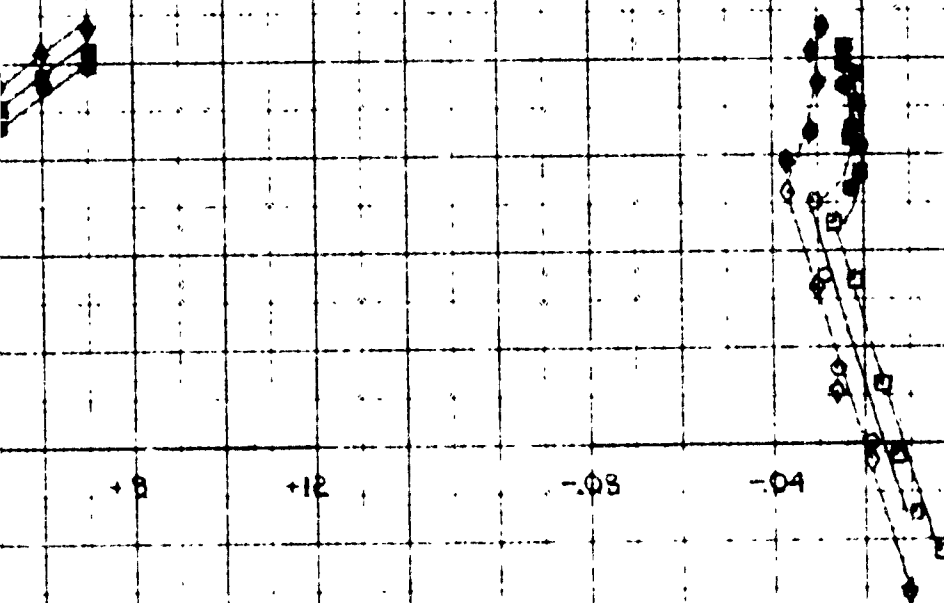
ANGLE OF ATTACK



TEST NO 43
 L NO 14
 FT 9.800 IN
 PORTS

PAGE IV-72

NOTE DARKENED POINTS INDICATE VENTED FLOW



ANGLE OF ATTACK - DEGREES

PITCHING MOMENT COEFFICIENT - $C_m = \frac{M}{\frac{1}{2} \rho V^2 S c}$

3-3
 3-3
 3-3

WHIRLING TANK TEST NO 4
 BU SHIPS FOIL NO 14
 WTC. 1.83 FT, 5.850 IN
 VELOCITY - 45 KTS

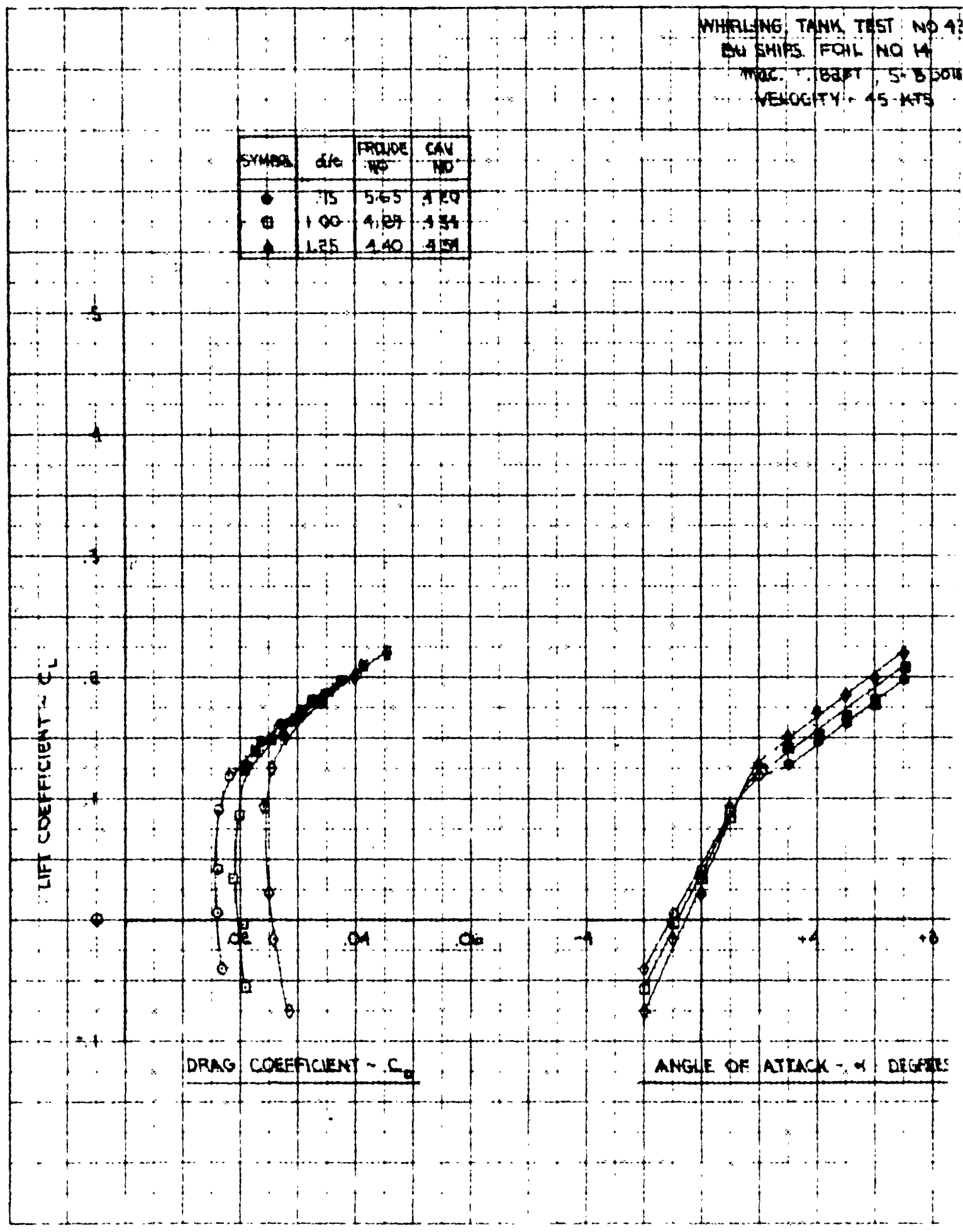
SYMBOL	d/c	FROUDE NO	CAV NO
●	.15	5.65	4.20
⊗	1.00	4.27	4.34
▲	1.25	4.40	4.51

LIFT COEFFICIENT - C_L

DRAG COEFFICIENT - C_D

ANGLE OF ATTACK - α DEGREES

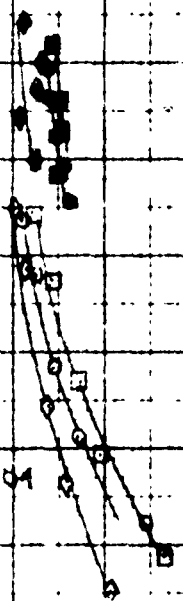
NO. 1



5018

5

NOTE: DARKENED POINTS INDICATE VENTED FLOW



PITCHING MOMENT COEFFICIENT - C_m

FELS

WHIRLING TANK TEST 43
 BU SHIPS FOIL # 14
 M.O.C. = .1825", S. BODIN
 VELOCITY = 50 KTS

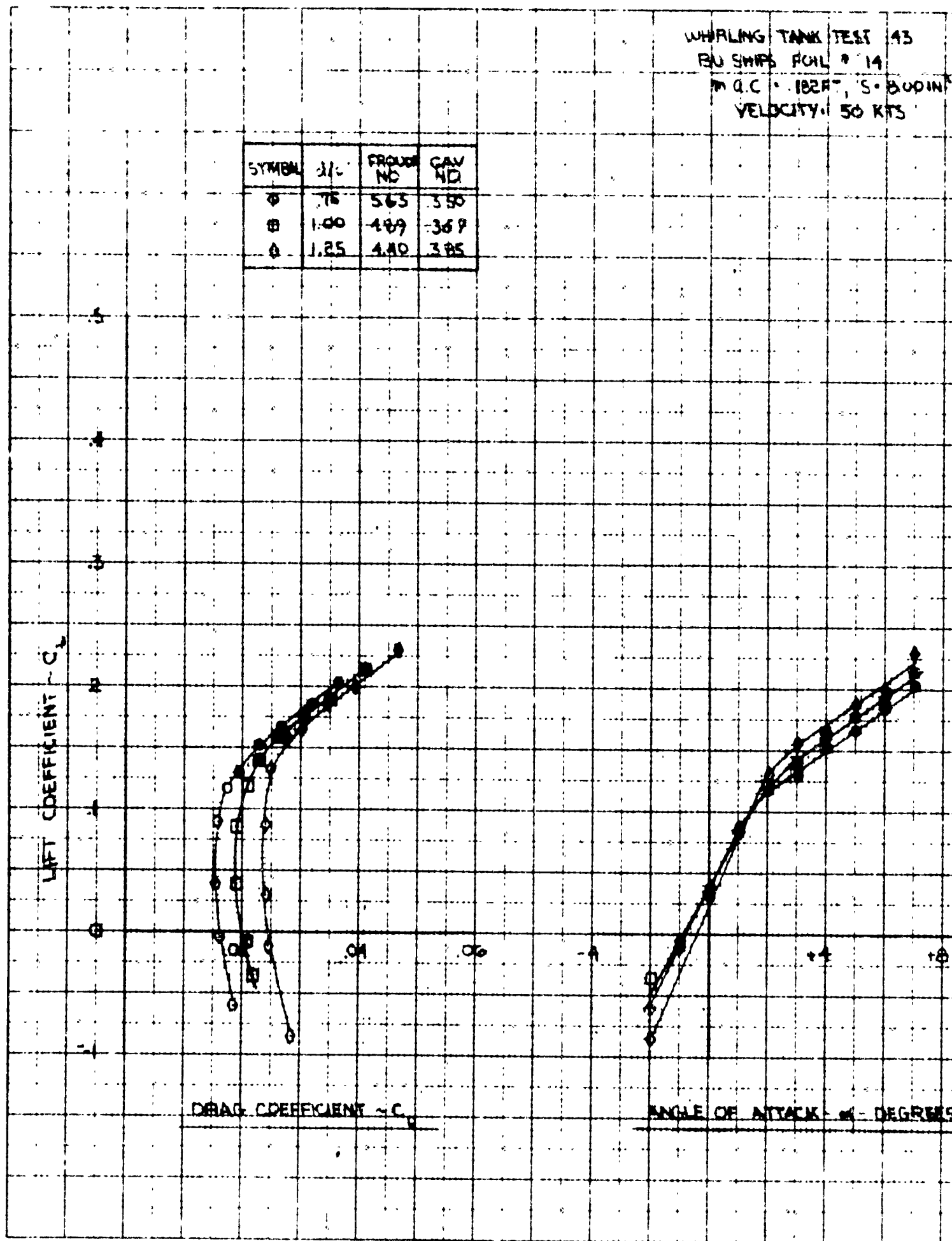
SYMBOL	β/c	FROUDE NO	CAV NO
○	.75	5.63	3.90
□	1.00	4.89	3.69
△	1.25	4.40	3.85

LIFT COEFFICIENT - C_L

DRAG COEFFICIENT - C_D

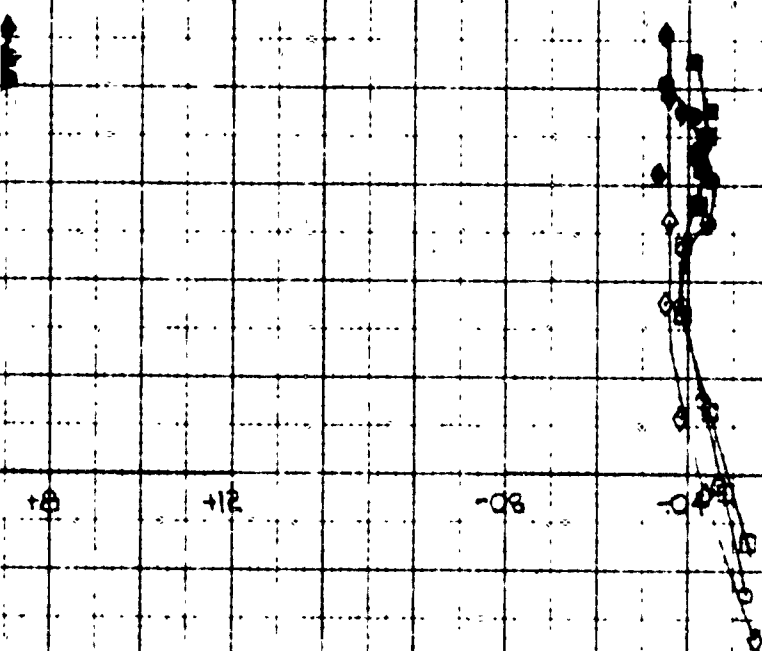
ANGLE OF ATTACK - α - DEGREES

K-2 AIRCRAFT OUT



POINT

NOTE DASHED POINTS INDICATE VENTED FLOW



REMS

PITCHING MOMENT COEFFICIENT $\sim C_{m-72}$

SYMBOL	γ_c	FROUDE NO.	CAV. NO.
•	.75	3.6280	.262
○	1.00	4.8924	.280
♦	1.25	4.4025	.300

LIFT COEFF - C_L

5
4
3
2
1
0
-1

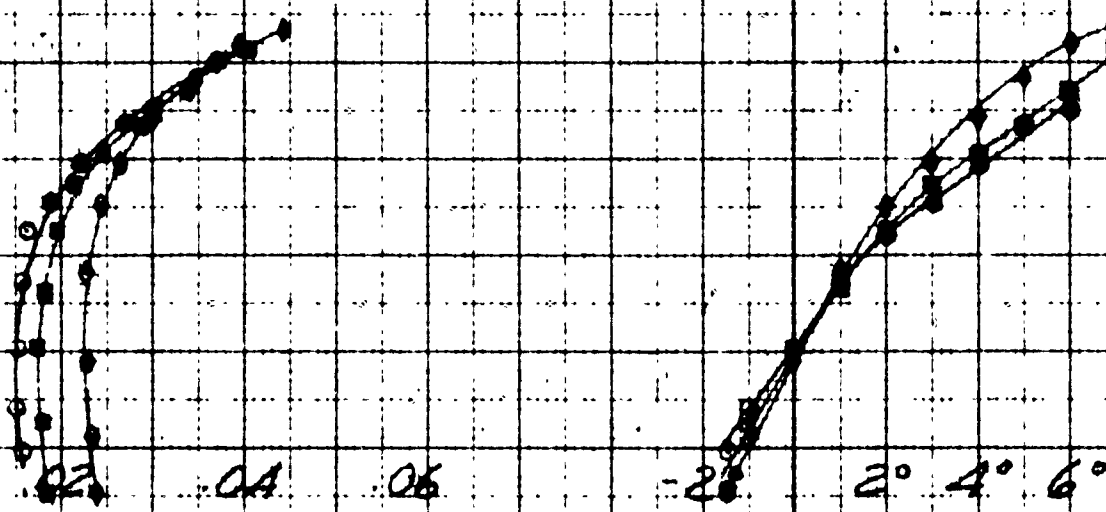
.02 .04 .06

-2° 2° 4° 6°

DRAW COEFF - C_D

ANGLE OF ATTACK

NO. 5 ALUMINUM WET

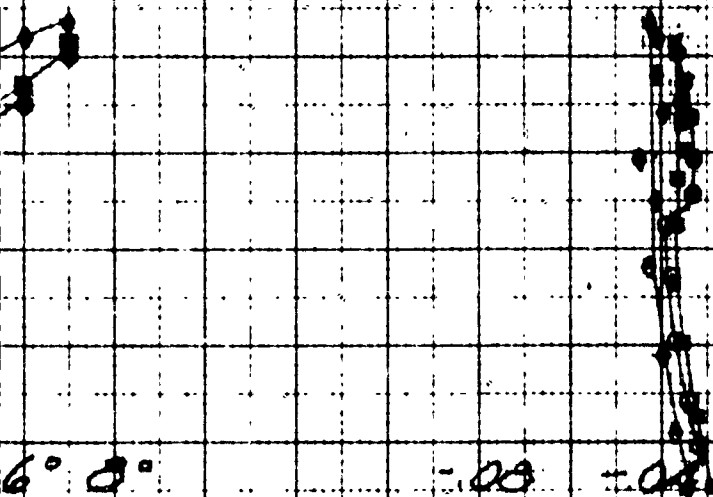


WIND TUNNEL TEST 43-BUSHIPS NO. 14

MACH = .182 FT AREA = 8.50 IN.

V = 60 KTS

NOTE: CARRIAGE POINTS INDICATE VENTED FLOW

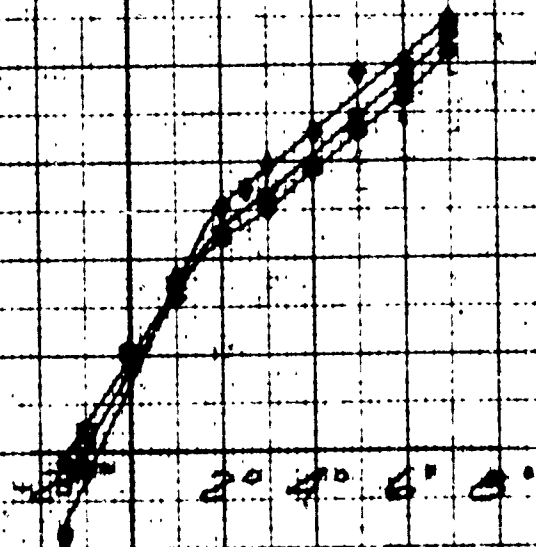
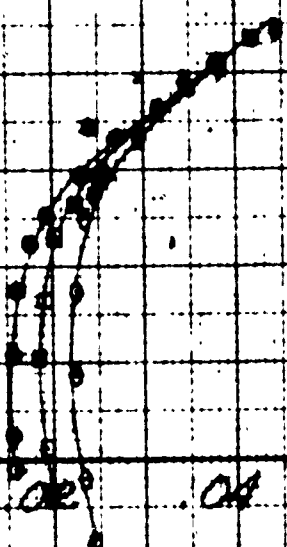


alpha - DEG

PITCHING MOMENT COEFF - C_m

SYMBOL	d/c	FROUDE NO.	CAN NO
○	.75	5.6280	.208
□	1.00	4.8924	.228
◇	1.25	4.4025	.247

LIFT COEFF - C_L



DRAG COEFF - C_D

ANGLE OF ATTACK - α

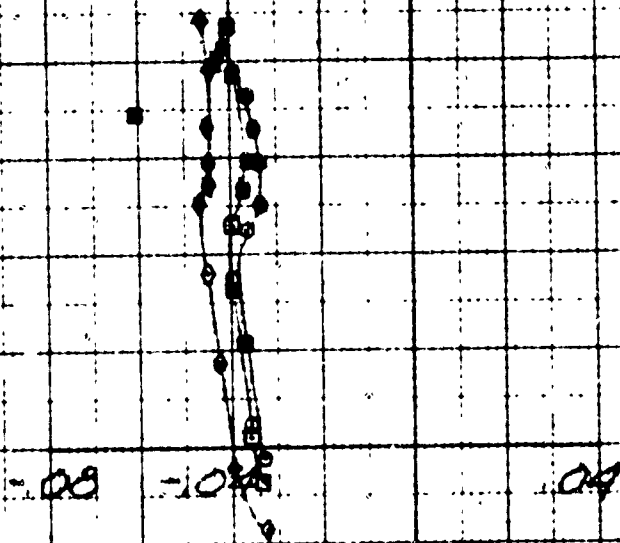
PAGE 14.76

WHIRLING TANK TEST #3 - BUSHIPS NO. 10

M.A.C. = .182 FT. ~ AREA = 8.50 IN.

V = 70 KTS

NOTE: DARKENED POINTS INDICATE VENTED FLOW

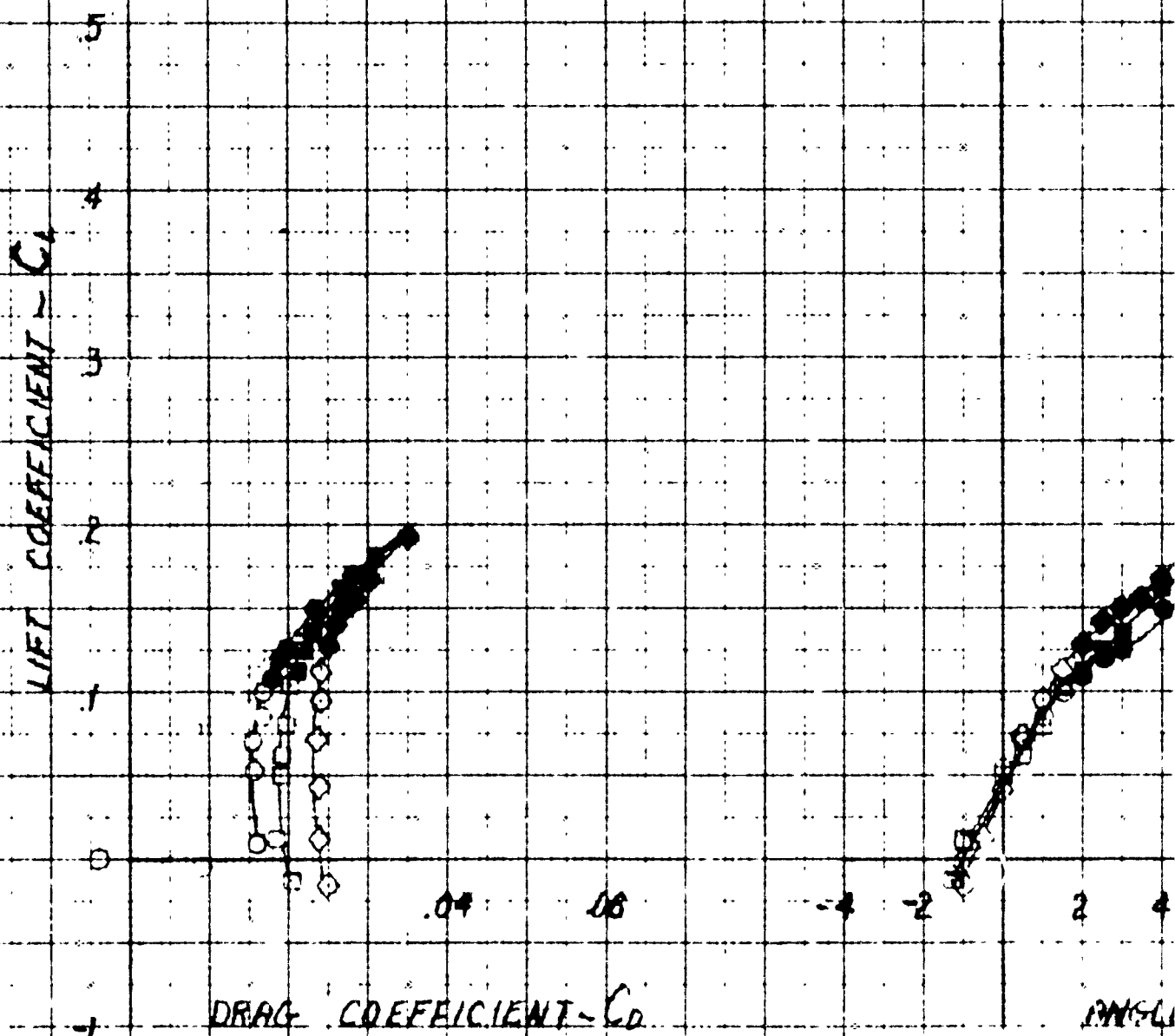


- DEG

PITCHING MOMENT COEFF ~ $C_{m\dot{\gamma}}$

SYMBOL	d/c	FROUDE NO.	CAV. NO.
⊕	.75	5.6280	184
⊞	1.00	4.8924	201
⊙	1.25	4.4025	219

NOTE



PAGE 1477

WHIRLING TANK TEST NO. 43

FG14 NO. 14 MAC = 0.182 FT

AREA = 8.00 SQ IN.

V = 80 KTS.

TE:

1 DARKENED SYMBOLS
INDICATE VENTED FLOW.



ANGLE OF ATTACK
 α ~ DEGREES

PITCHING MOMENT
COEFFICIENT ~ C_m

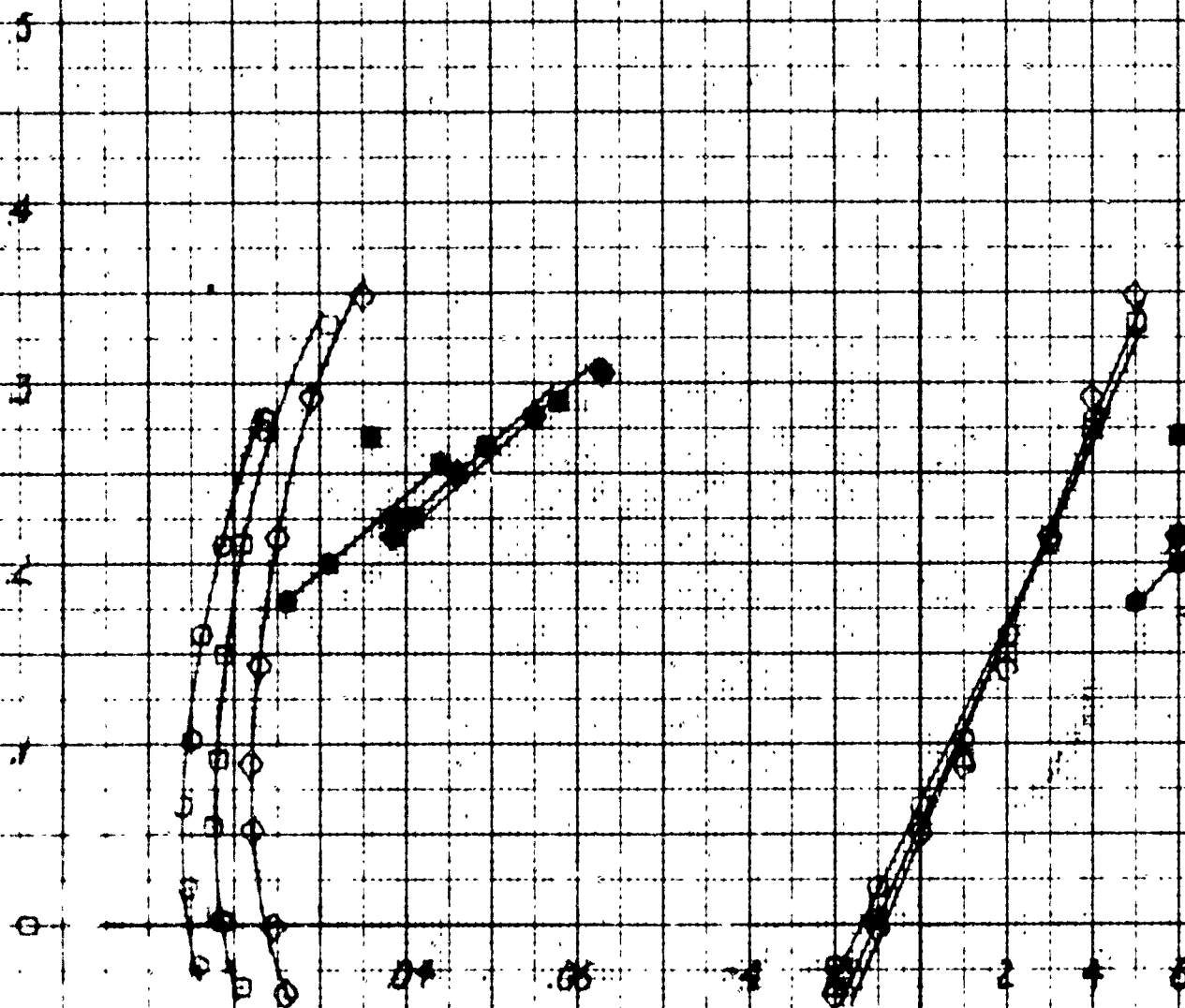
SYMBOLS	A/C	FROUDE NO.	CAN. NO.
○	75	6.1234	4.93
□	1.00	5.3219	5.13
◇	1.25	4.7785	5.42

NOTE: 1. DR
 VE.

COEFFICIENT OF LIFT - C_L

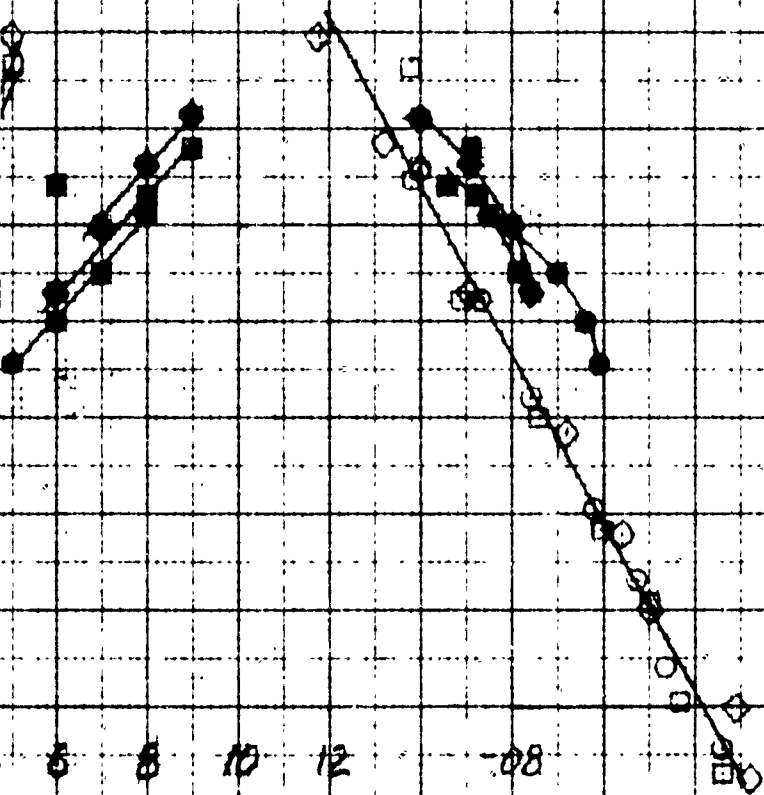
COEFFICIENT OF DRAG - C_D

ANGLE OF α



PAGE 147B
 WHIRLING TANK TEST NO. 44
 BUCHIPS FOIL NO. 15
 M.A.C. = 152 FT AREA = 80 IN.
 $V = 40$ KTS

DARKENED SYMBOLS INDICATE
 VENTED FLOW.



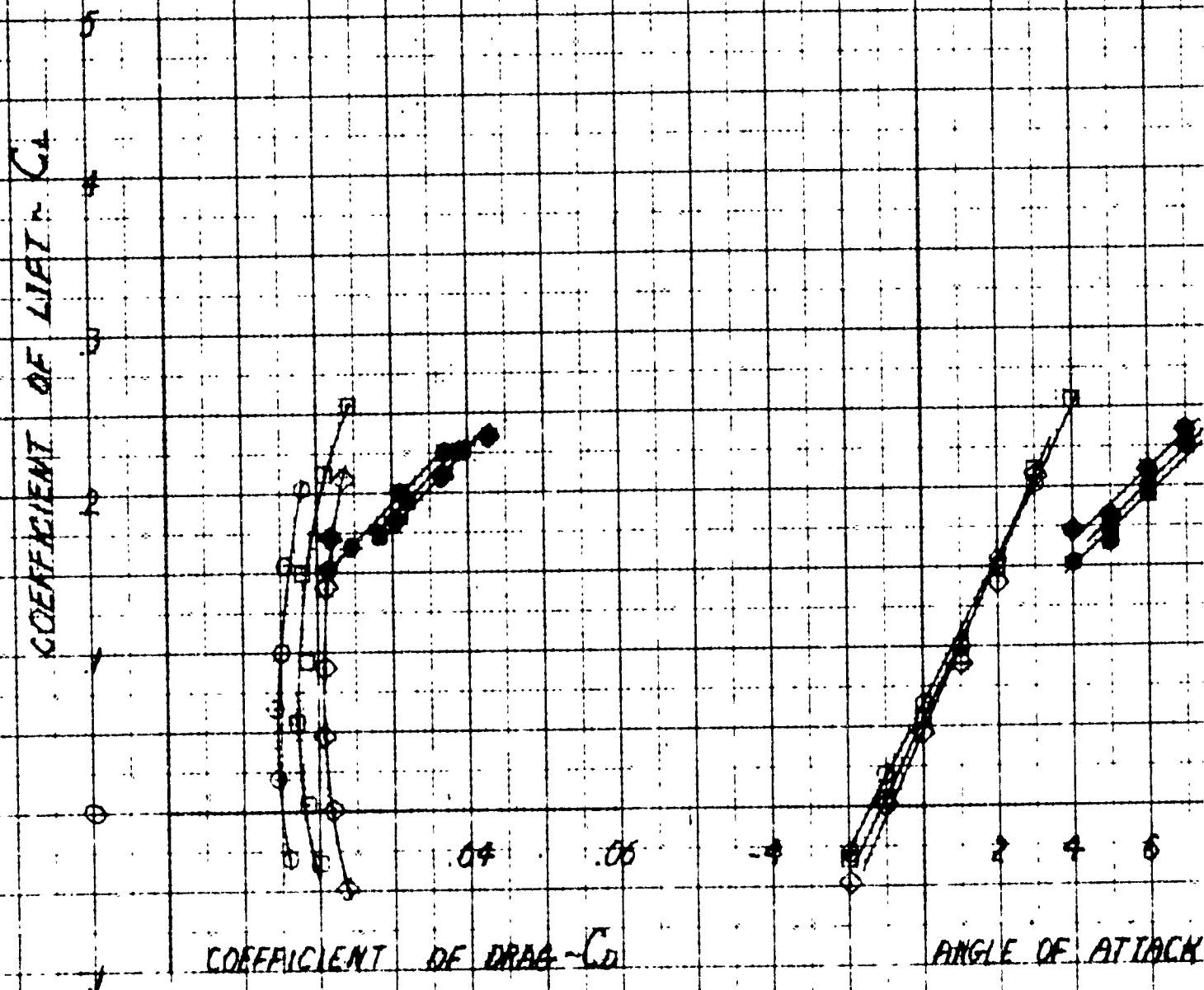
ATTACK α - DEGREES

COEFFICIENT OF
 PITCHING MOMENT C_m

NOTES: 1. 152 FT. AREA

NOTE:

1. DAA
VEA

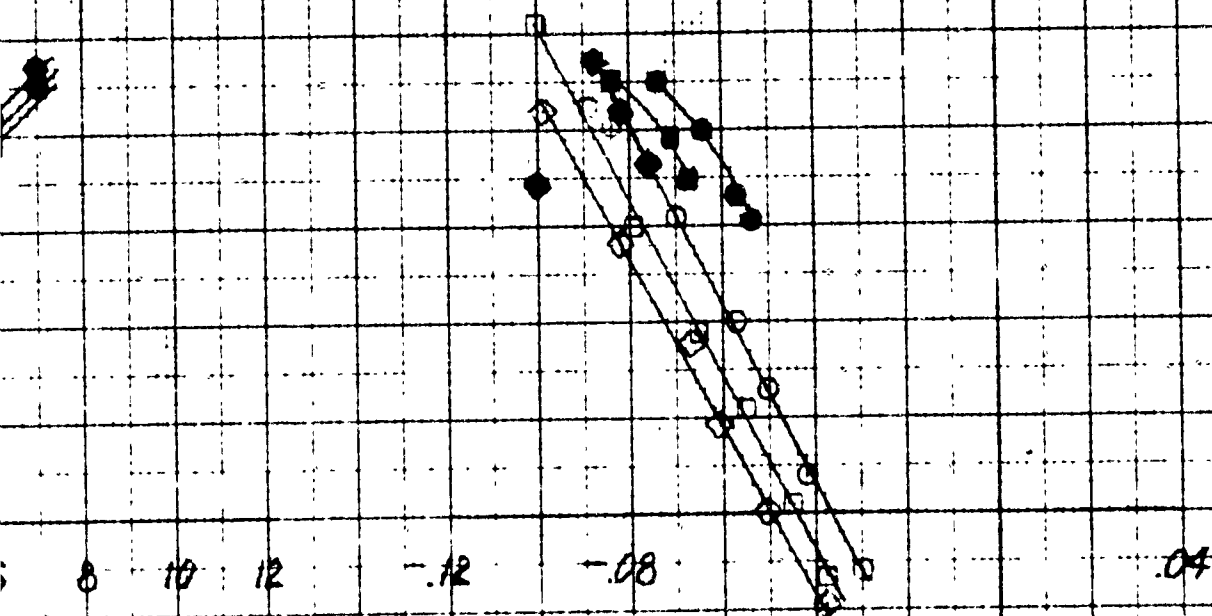


WHIRLING TANK TEST NO. 44

BU SHIPS FOIL NO. 15

MAC = .152 FT AREA = 8.00 SQ. IN.

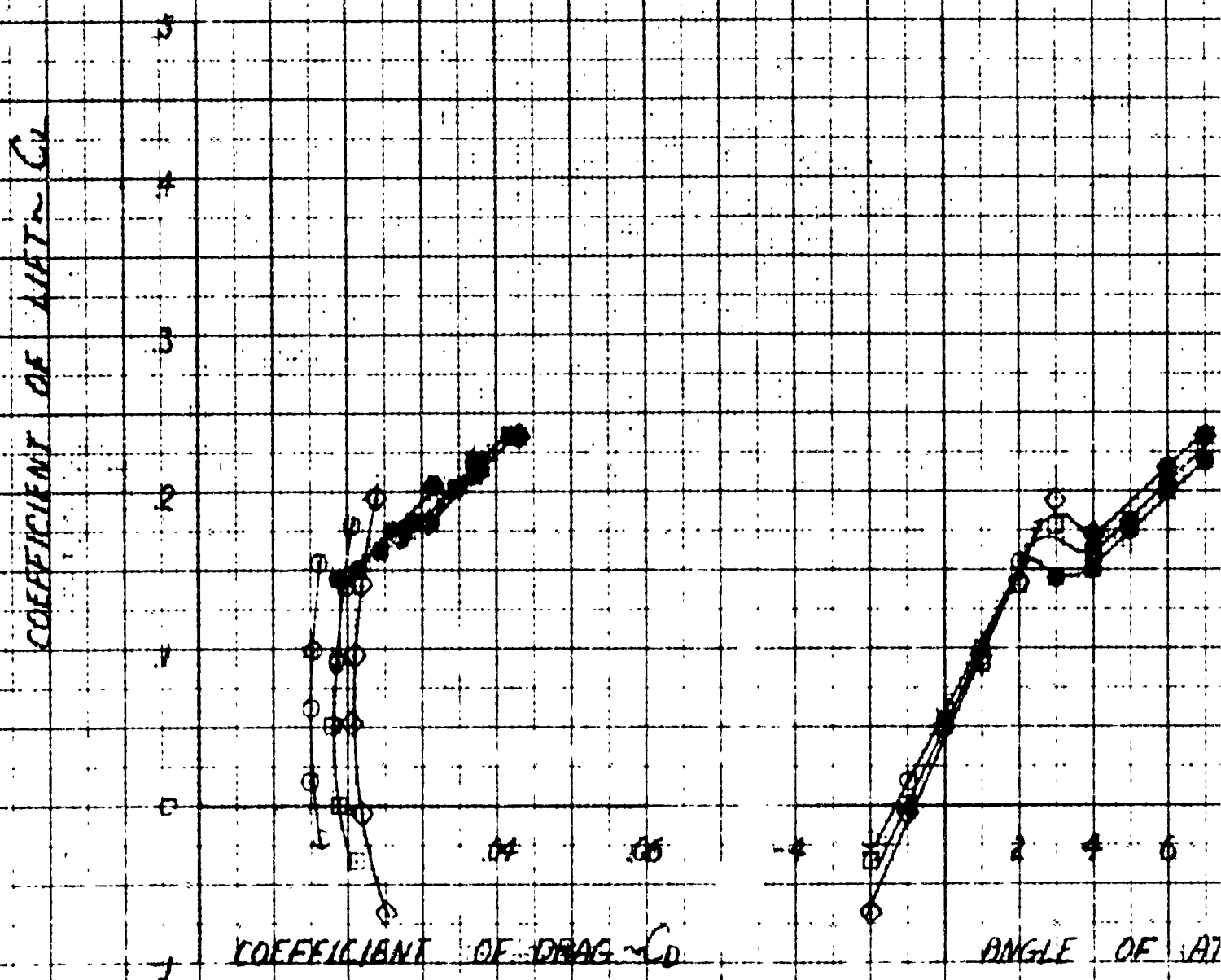
V = 45 M.T.S.

DARKENED SYMBOLS INDICATE
DISTURBED FLOW.ANGLE α - DEGREESCOEFFICIENT OF
PITCHING MOMENT $\sim C_m$ RESULT $\frac{1}{2} \rho V^2 S C_m$

SYMBOLS	d/c	FIGURE NO	CAL NO
⊙	.75	012	339
⊙	1.00	032	356
⊙	1.25	470	507

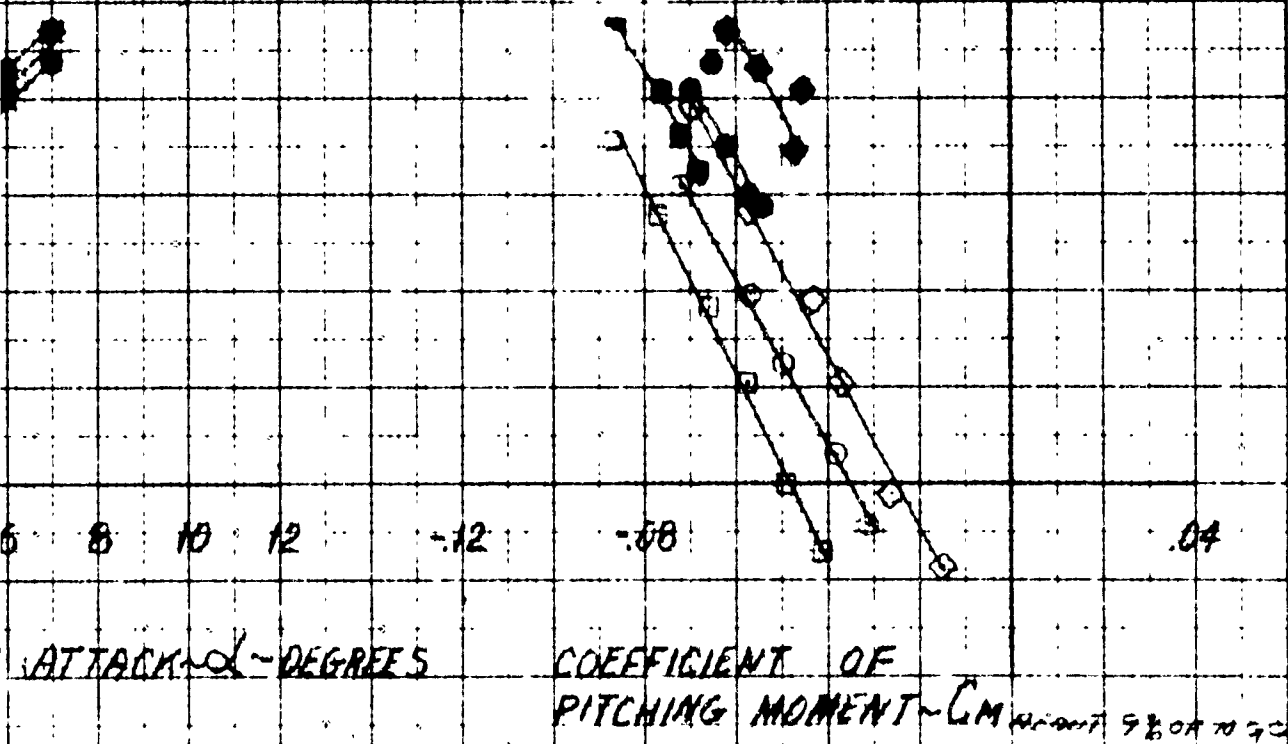
NOTES

1.0



PAGE IV.80
 WHIRLING TANK TEST NO. 44
 BU SHIPS FOIL NO. 15
 MAC = 152 AT AREA = 8.00 SQ. IN.
 V = 50 KTS.

1. DASHED SYMBOLS INDICATE
 VENTED FLOW



SYMBOL	d/c	FROUDE NO	CAN. NO
⊙	.75	6.12	2.53
⊕	1.00	5.32	2.68
⊗	1.25	4.78	2.85

NOTE:

1. DARKENED
VENTED F

COEFFICIENT OF LIFT - C_L

COEFFICIENT OF DRAG - C_D

ANGLE OF α

5

4

3

2

1

0

0.4

0.6

0.8

2

4

6

PAGE IV, B1

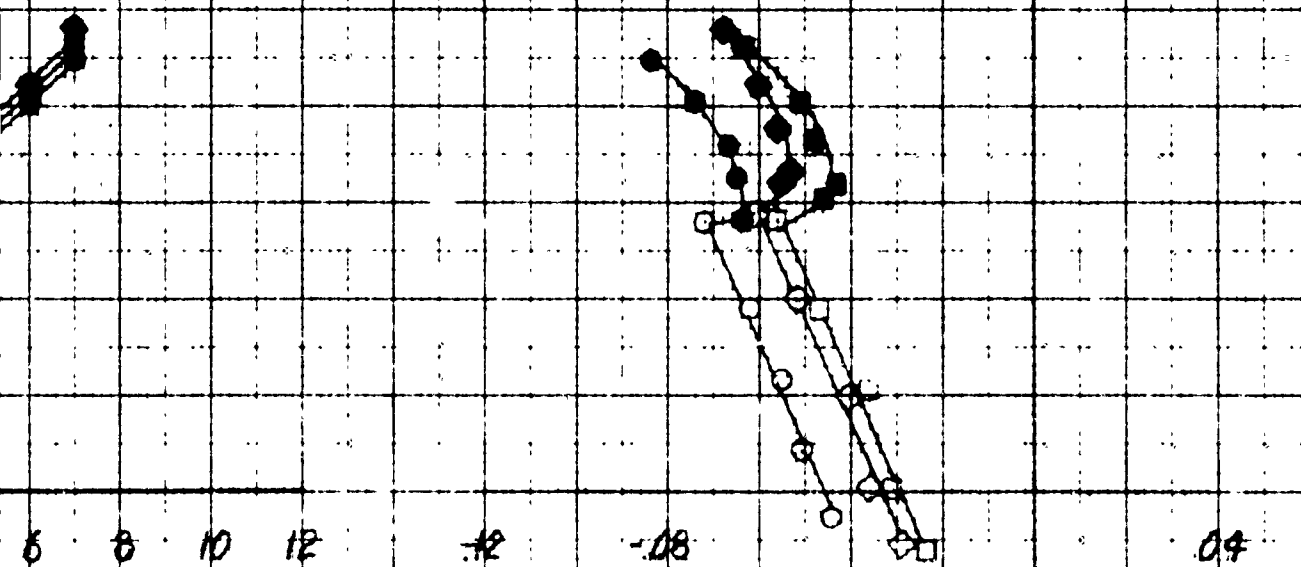
WHIRLING TANK TEST NO. 44

BU SHIPS FOIL NO. 15

MAC = .152 FT. AREA = 8.00 SQ IN.

V = 60 MTS.

ED SYMBOLS INDICATE
FLOW



ANGLE OF ATTACK - α - DEGREES

COEFFICIENT OF
PITCHING MOMENT - C_m (9% OF AREA)

10-2
Printed on 10-2

SYMBOL	$\frac{L}{c}$	FRAUDE NO	CAK NO
⊕	.75	6.12	199
⊞	1.00	5.32	210
⊙	1.25	4.78	232

NOTE:

1. DARKENED S
VENTED FLO

COEFFICIENT OF LIFT - C_L

5
4
3
2
1
0

COEFFICIENT OF DRAG - C_D

.04

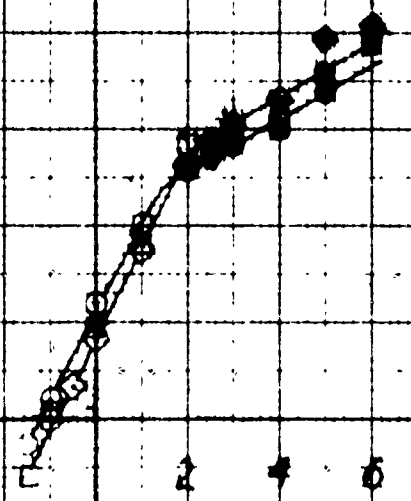
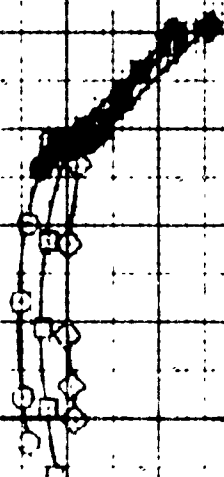
.06

ANGLE OF ATT

2

4

6



PAGE 1 KB2

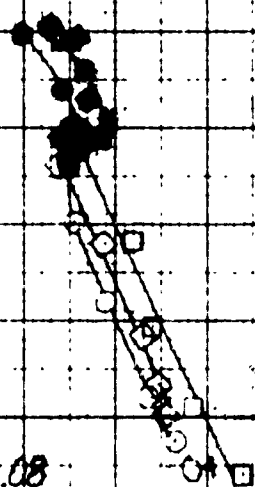
WHIRLING TANK TEST NO. 44

BU SHIPS FOIL NO. 15

M.A.C. = 152 FT. AREA = 8.00 SQ. IN.

V = 70 M.T.S.

SYMBOLS INDICATE
FLOW



ATTACK α - DEGREES

COEFFICIENT OF
PITCHING MOMENT C_m

SYMBOL	β/c	FROUDE NO.	CAL. NO.
o	.75	6.1234	E. 739
n	1.00	5.3219	L. 938
d	1.25	4.7785	2. 092

NOTE:

1. DARKENED \angle
INDICATE VE
FLOW

LIFT COEFF - C_L

.3

.2

.1

0

.02

.04

.06

-4

-2°

2°

4°

6°

DRAG COEFF - C_D

ANGLE OF ATTACK
DEGREES

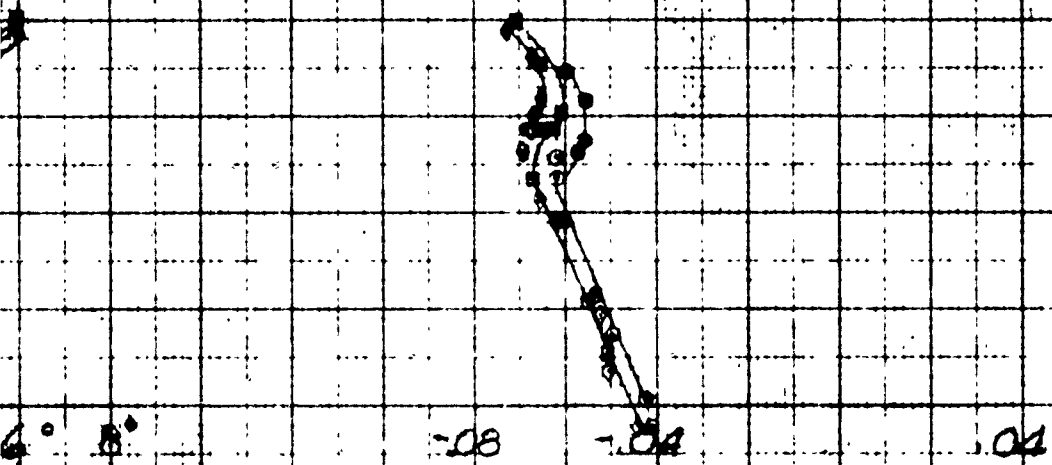
WHIRLING TANK TEST NO 44

BUSHIPS FOIL NO. 15

MAC = .152 FT AREA = 8 SQ. IN.

V = 80 KTS

SYMBOLS
VENTED



PITCHING MOMENT COEFF - C_M

ABOUT 92% OF THE MAC

ME 9-10000-1000 2307

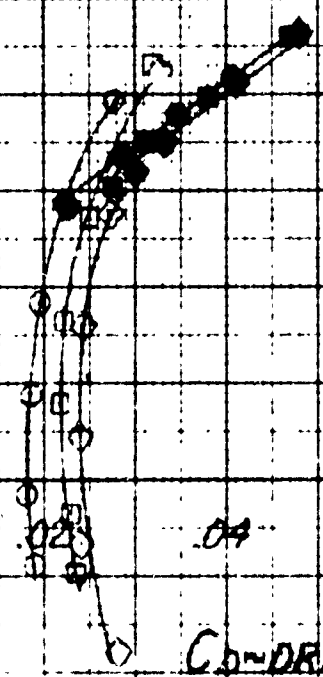
SYMBOL	α/c	FIGURE NO	CAV NO
○	.75	634	.488
□	1.00	551	.510
◇	1.25	495	.525

NOTE:

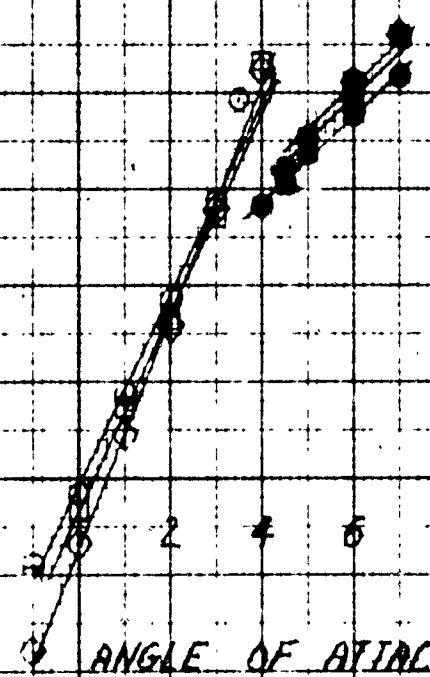
LOADING INDIC.

COEFFICIENT OF LIFT C_L

5
4
3
2
1
0
-1



C_D -DRAG COEFFICIENT



ANGLE OF ATTACK

PAGE IV.84

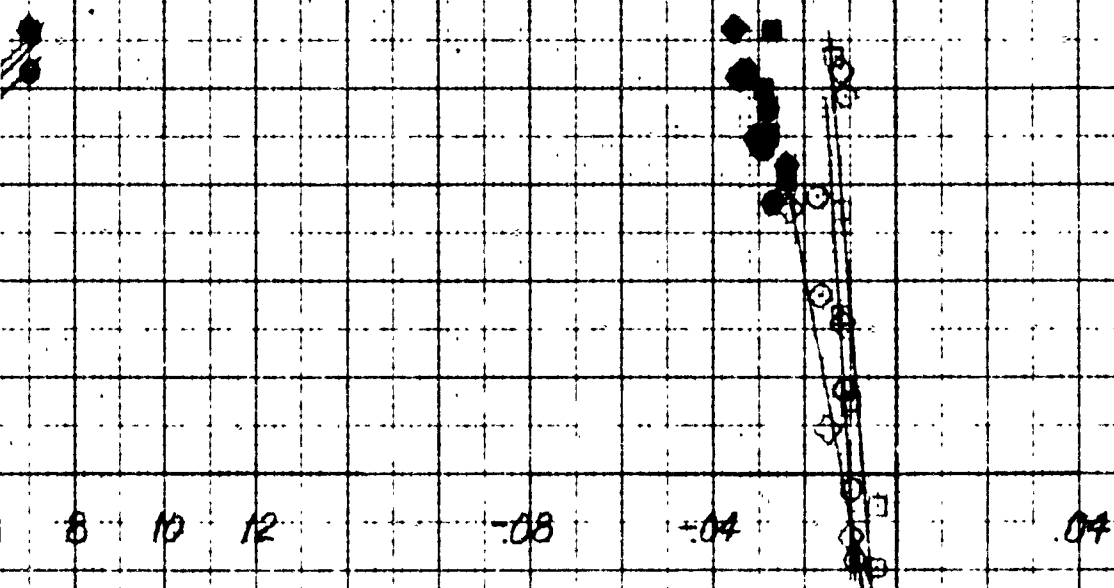
WHIRLING TANK TEST NO 45

BUL SHIPS FOIL NO. 16

M.A.C. = 142 FT AREA = 8.23 IN²

V = 40 MTS

ARMED SYMBOLS
INDICATE VENTED FLOW



ANGLE α DEGREES

COEFFICIENT OF
PITCHING MOMENT C_m

SYMBOL	α_H	FRANDE NO.	CAK NO.
○	75	534	402
□	100	551	417
◇	125	495	432

NOTE:

1. DAME
 INDICA

COEFFICIENT OF LIFT C_L

5

4

3

2

1

0

-1

DRAG COEFFICIENT C_D

.04

.06

-2

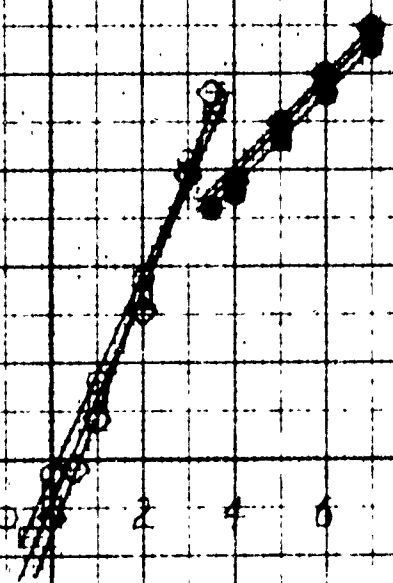
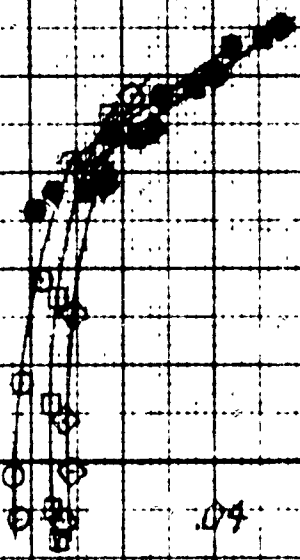
2

4

6

8

ANGLE OF ATTACK



PAGE IV. 85

WHIRLING TANK TEST NO. 46

BU SHIPS FOIL NO. 16

MAC = .142 FT. AREA = 8.23 IN²

V = 45 KTS.

NEWED SYMBOLS
LATE VENTED FLOW.

0 10 12

18

24

30

α - DEGREES

COEFFICIENT OF
PITCHING MOMENT - $C_{M\frac{1}{2}}$

SYMBOL	d/c	FROUDE NO.	CAH NO.
○	.75	8.34	.385
□	1.00	5.51	.380
◇	1.25	4.95	.360

NOTE:

1. DARKENED
INDICATE VEM

COEFFICIENT OF LIFT ~ C_L

5

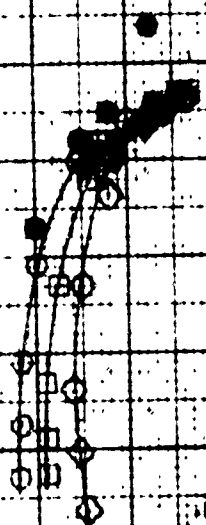
4

3

2

1

0



.04

.06

.2

2

4

6

8

DRAG COEFFICIENT ~ C_D

ANGLE OF ATTACK ~

WHIRLING TANK TEST NO. 46

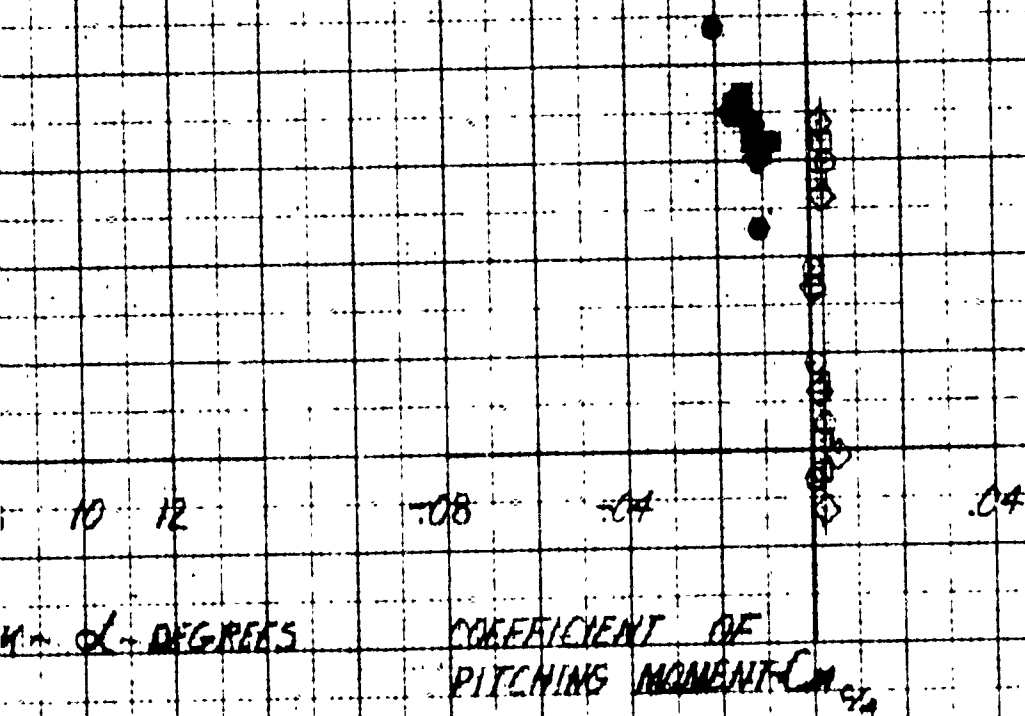
BLI SHIPS FOIL NO. 16

MAC = 1.42 FT AREA = 8.23 in²

V = 50 KTS.

SYMBOLS

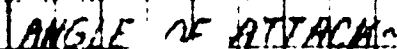
VENTED FLOW



SYMBOL	d/c	FROUDE NO.	CAV. NO.
⊙	75	6 34	.2 48
⊠	100	5 51	.2 63
⊙	125	4 95	.2 78

NGTE:

1. DARMENED
INDICATE



PAGE 1487

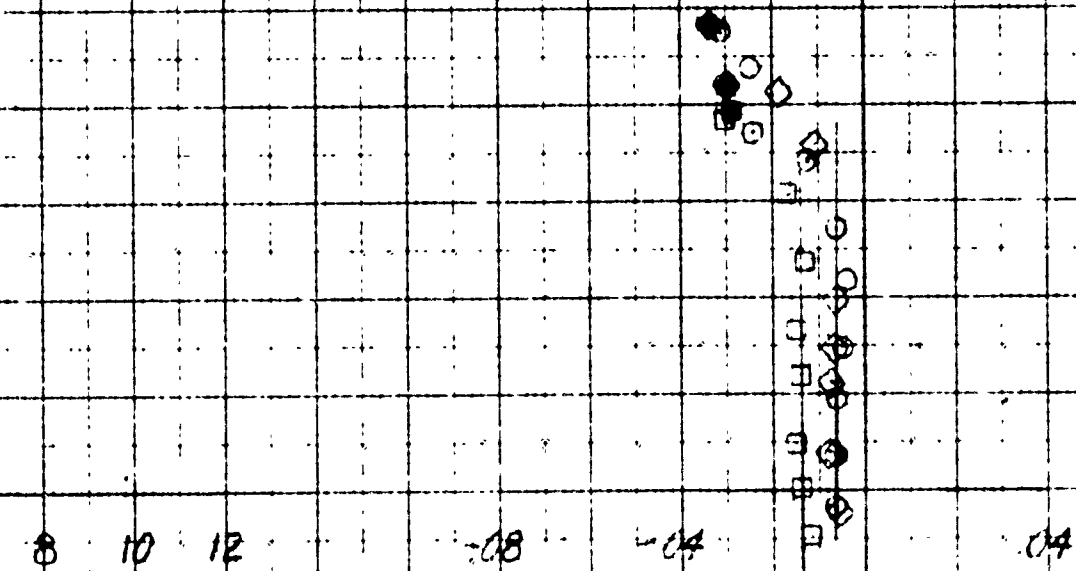
WHIRLING TANK TEST NO. 46

BU SHIPS FOIL NO. 16

M.A.C. = 142 FT. AREA = 8.23 IN²

V = 60 KTS.

ED SYMBOLS
F VENTED FLOW



α - DEGREES

COEFFICIENT OF
PITCHING MOMENT - Cm

SYMBOL	d/c	FROUDE NO.	CAV. NO.
○	.75	6.34	.195
□	1.00	5.51	.211
◇	1.25	4.95	.226

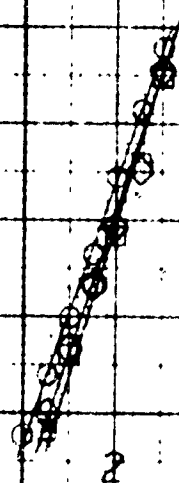
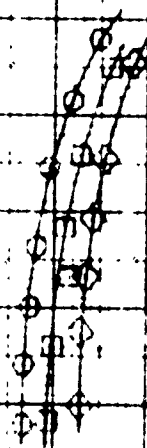
NOTE:

1. DASHED
INDICATE W

COEFFICIENT OF LIFT - C_L

DRAG COEFFICIENT - C_D

ANGLE OF ATTACK - α



WHIRLING TANK TEST NO. 46

BU SHIPS FOIL NO. 16

MAC = 142 FT AREA = 8.23 m²

V = 70 KTS.

2 SYMBOLS
VENTED FLOW



10 12

-08

-04

04

α - DEGREES

COEFFICIENT OF
PITCHING MOMENT - $C_{m\frac{\gamma}{A}}$

NO. 1000

SYMBOL	d/c	FROUDE NO.	CAV. NO.
→	75	—	—
□	100	5.51	.177
◇	125	4.95	.192

NOTE:
1. DARKENED
INDICATE

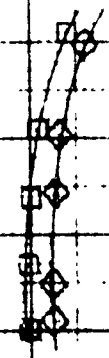
COEFFICIENT OF LIFT - C_L

DRAG COEFFICIENT - C_D

ANGLE OF ATTACK

5
4
3
2
1
0

.04 .05 .2 2 4 6 8



WHIRLING TANK TEST NO. 46

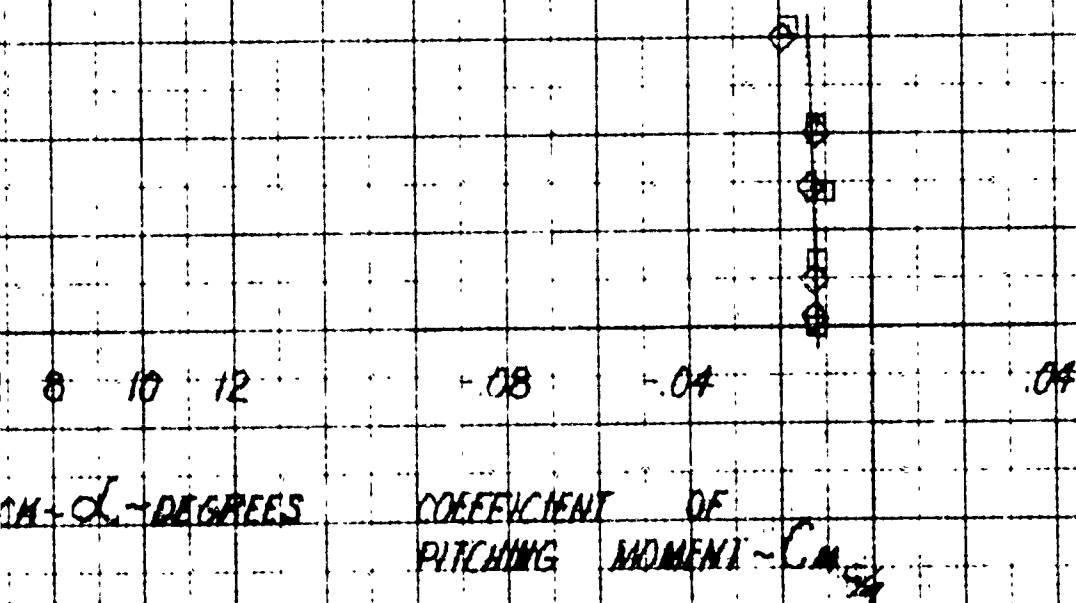
BU SHIPS FOIL NO. 16

MAC = .142 FT. AREA = 8.23 in²

V = 80 KTS.

SYMBOLS

VENTED FLOW



SYMBOL	d/c	FRAUDE NO	CAV. NO
○	75	6.34	4.96
□	100	5.51	5.09
△	125	4.85	5.22

NOTE:

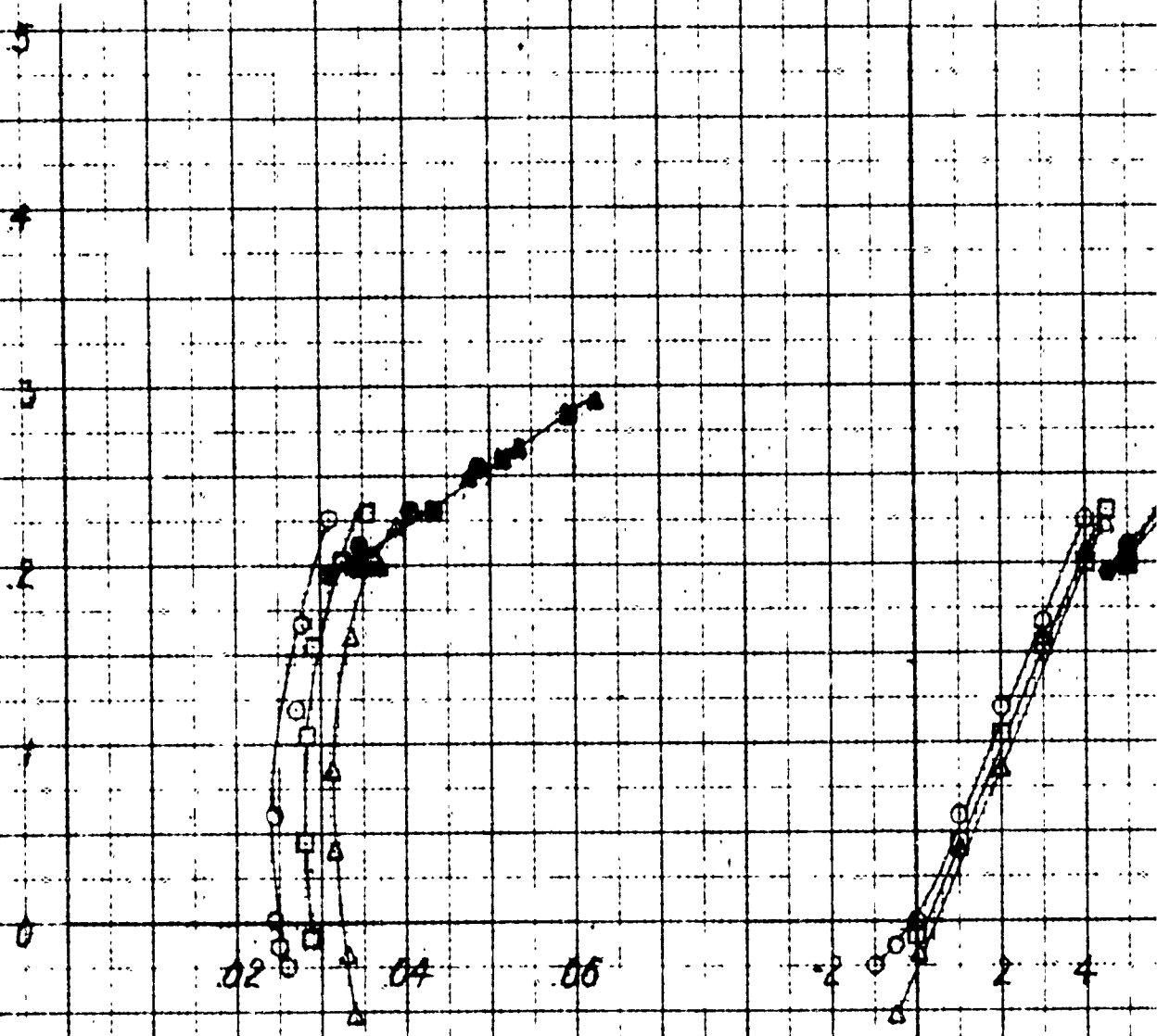
1. DARM
VENT

LIFT COEFFICIENT $\sim C_L$

DRAG COEFFICIENT $\sim C_D$

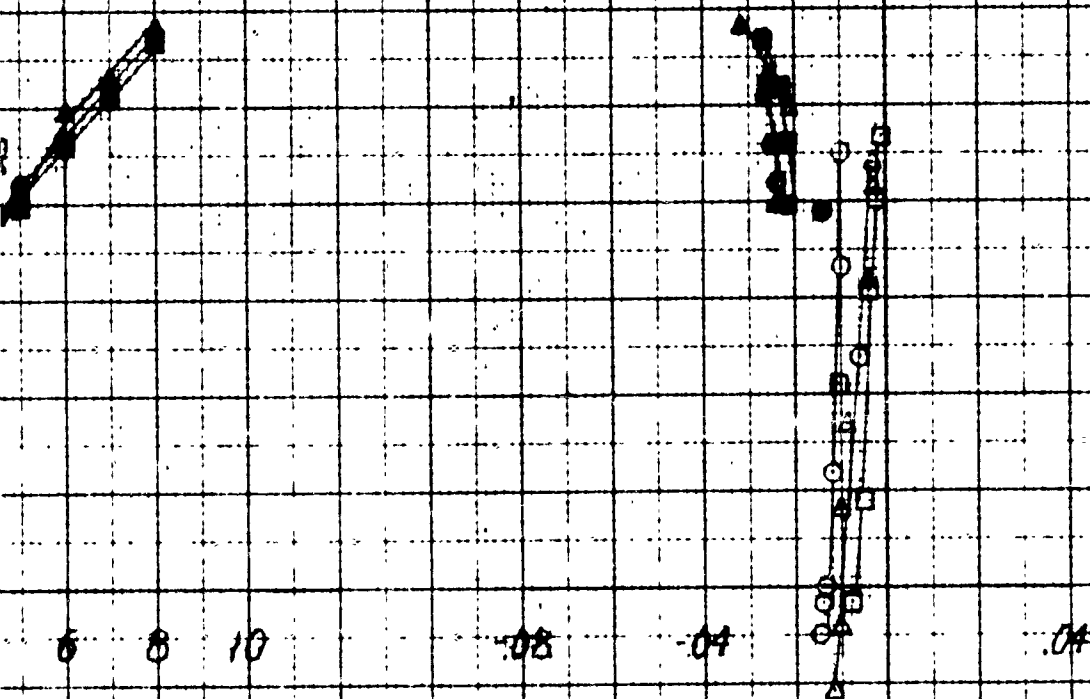
ANGLE OF A

K-2 aligned 1007



WHIRLING TANK TEST NO. 47
 BU SHIPS FOIL NO. 17
 $S = 8.23 \text{ IN}^2$; $MAC = 142 \text{ FI}$
 $V = 40 \text{ MTS}$

ARMED POINTS INDICATE
 STALLED FLOW



ATTACK - α - DEGREES PITCHING MOMENT
 COEFFICIENT - C_m - $\frac{1}{4}$

WHIRLING TANK TEST
 BU SHIPS FOIL NO 17
 M.A.L. 142 FT 5.32
 VELOCITY = 45 K

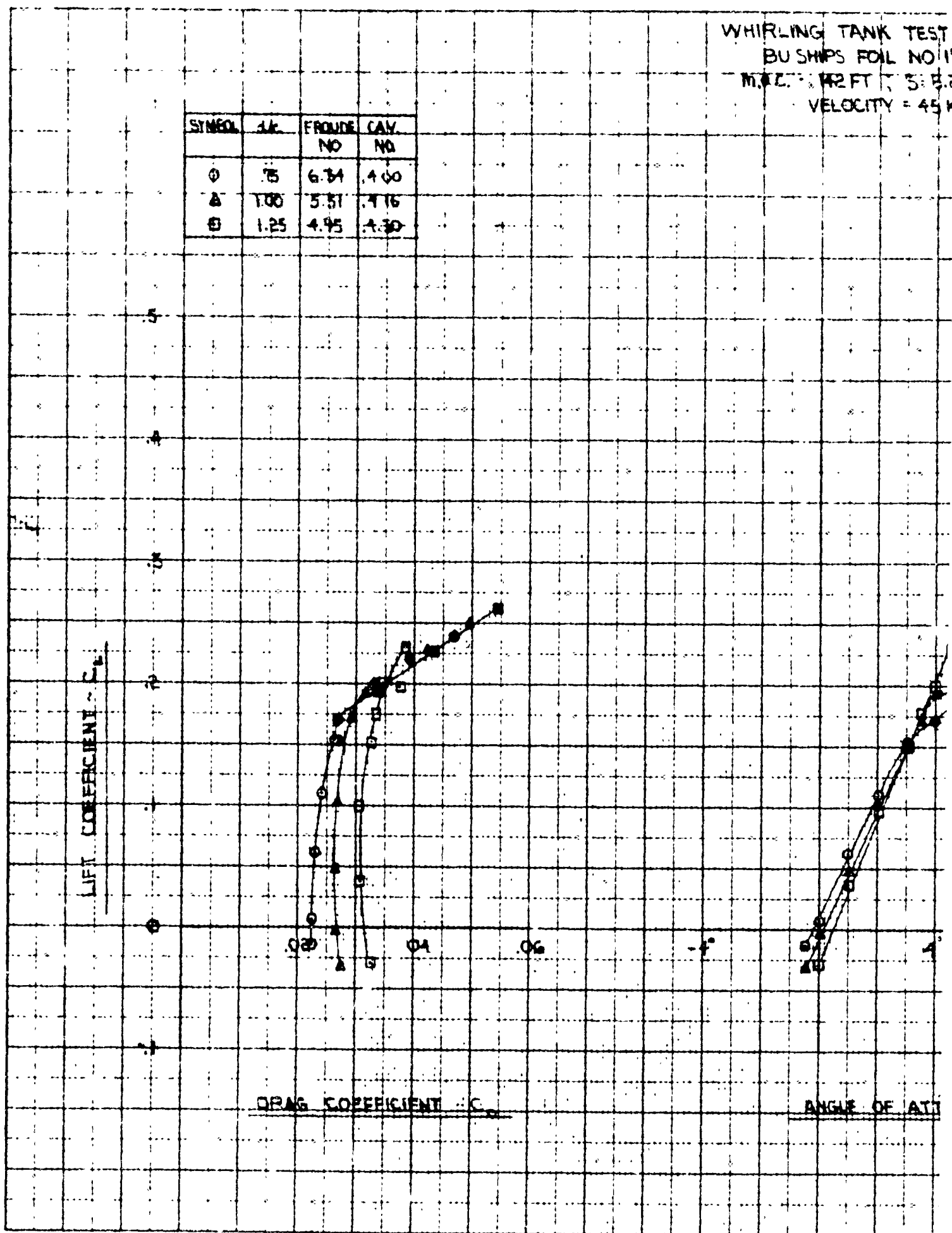
SYMBOL	SL	FROUDE NO	CAV. NO
○	.75	6.34	.400
△	1.00	5.31	.416
□	1.25	4.95	.430

LIFT COEFFICIENT - C_L

DRAG COEFFICIENT - C_D

ANGLE OF ATT

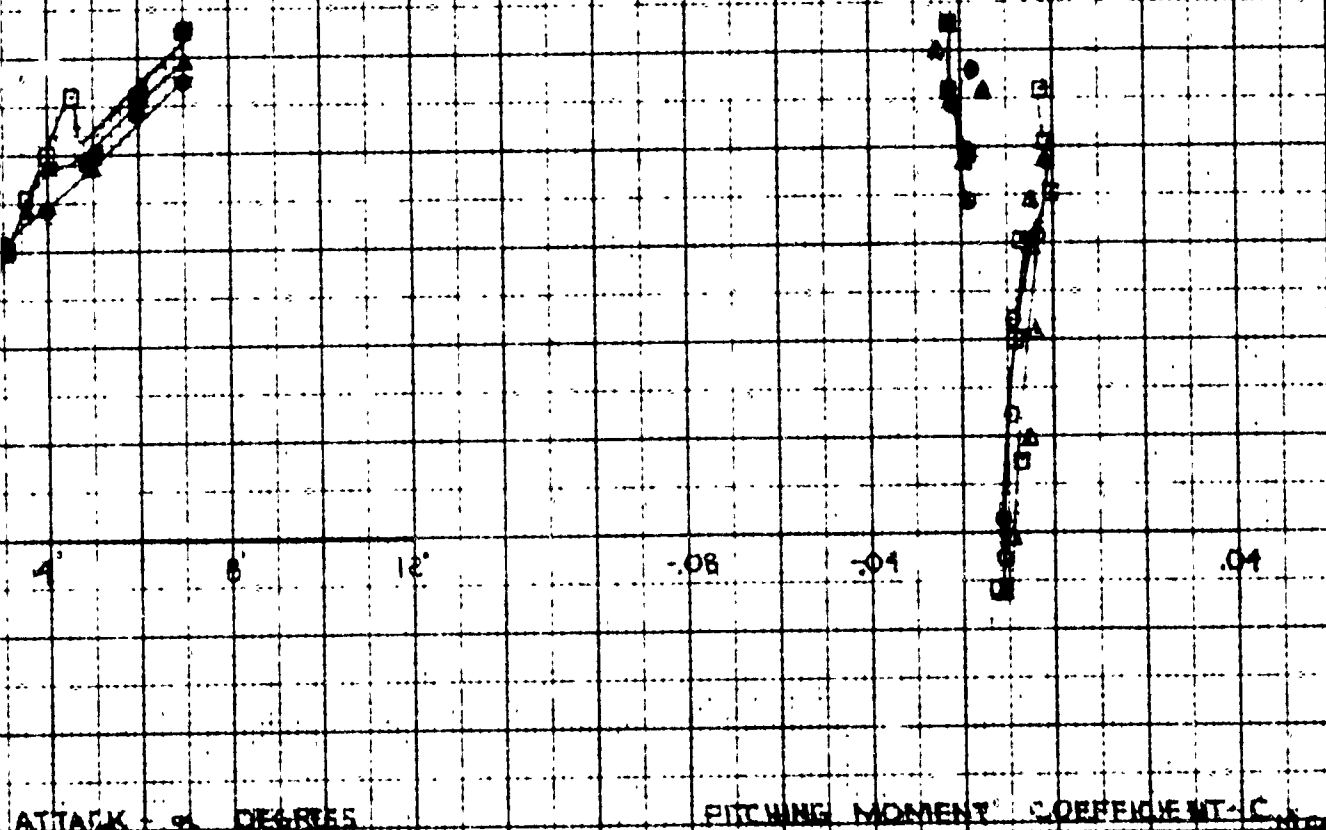
NOE ALBANESE 1947 5215



EST NO 47
NO 17
E 2350 IN
45 KTS

PAGE 14.91

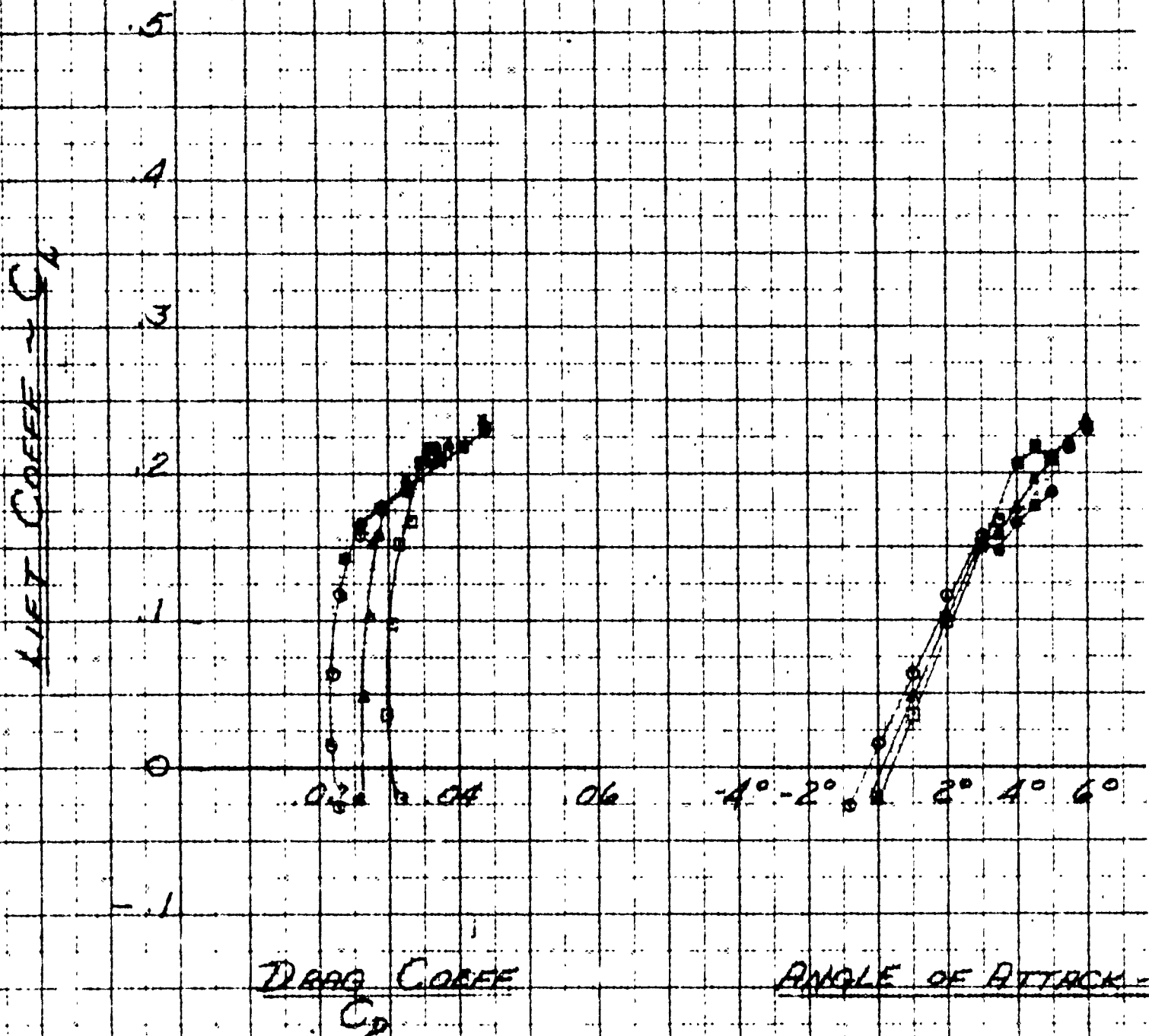
NOTE: DARKENED POINTS INDICATE VENTED FLOW



SYMBOL		d/c	FROUDE No.	OAV. No.
◎		75	6.3416	3.36
▲		1.00	5.5147	3.52
□		1.25	4.9527	3.65

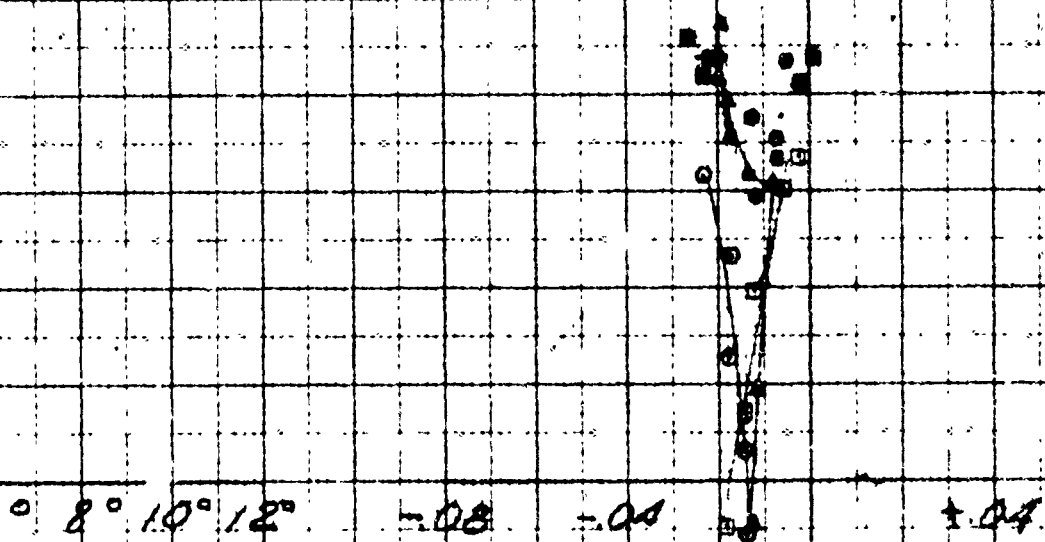
NOTES:

1. DARKE,
INDICA:



W.I.T. No 47 BUSSHIPS FOIL No. 17
M.A.C. .142 FT. AREA = 8.23 sq in
VEL = 50 KTS

KEYED SYMBOLS
CATE VENTED FLOW



α - DEGREES

PITCHING MOMENT COEFF - C_m

SYMBOL	d/c	FROUDE NO	CAN. NO
○	.75	6.34	.248
□	1.00	5.51	.263
△	1.25	5.22	.278

NOTE:

1. DARMENEL
VENTED

LIFT COEFFICIENT $-C_L$

DRAG COEFFICIENT $-C_D$

ANGLE OF ATTACK

W-E alignment test

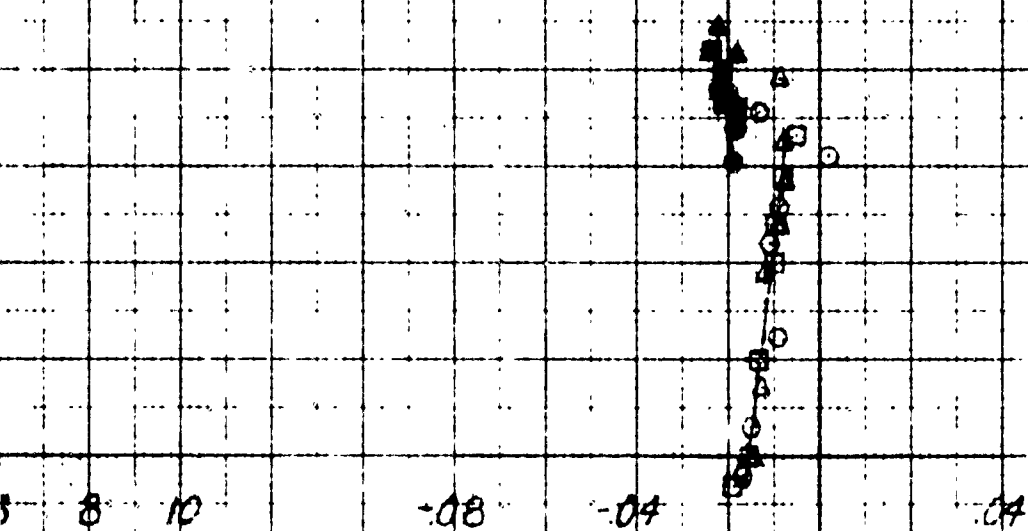
WHIRLING TANK TEST NO. 47

BU SHIPS FCIL NO. 17

MAC = .142 FT ; S = 8.23 IN²

V = 60 KTS. PAGE 11, 23

ENED POINTS INDICATE
ED FLOW.

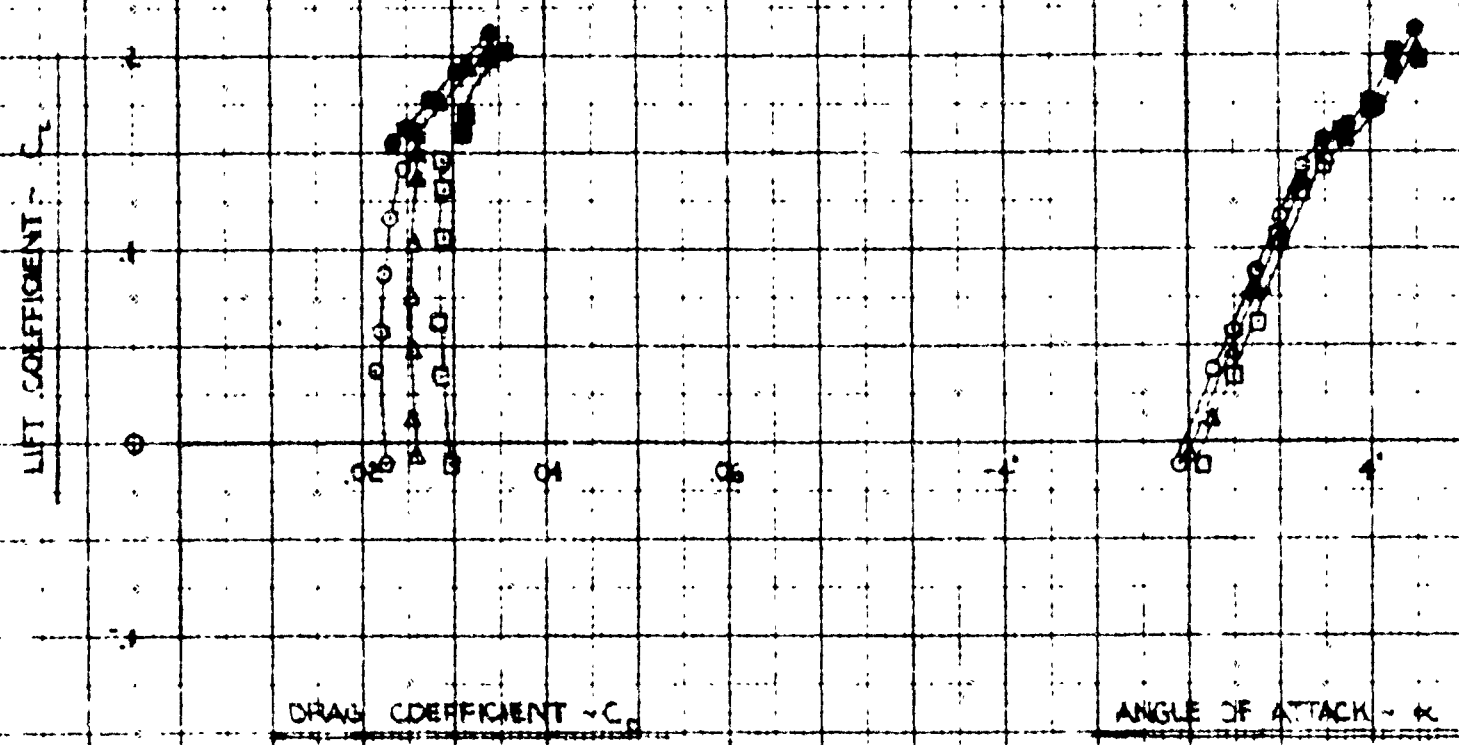


ATTACK ~ α ~ DEGREES

PITCHING MOMENT
COEFFICIENT ~ C_m

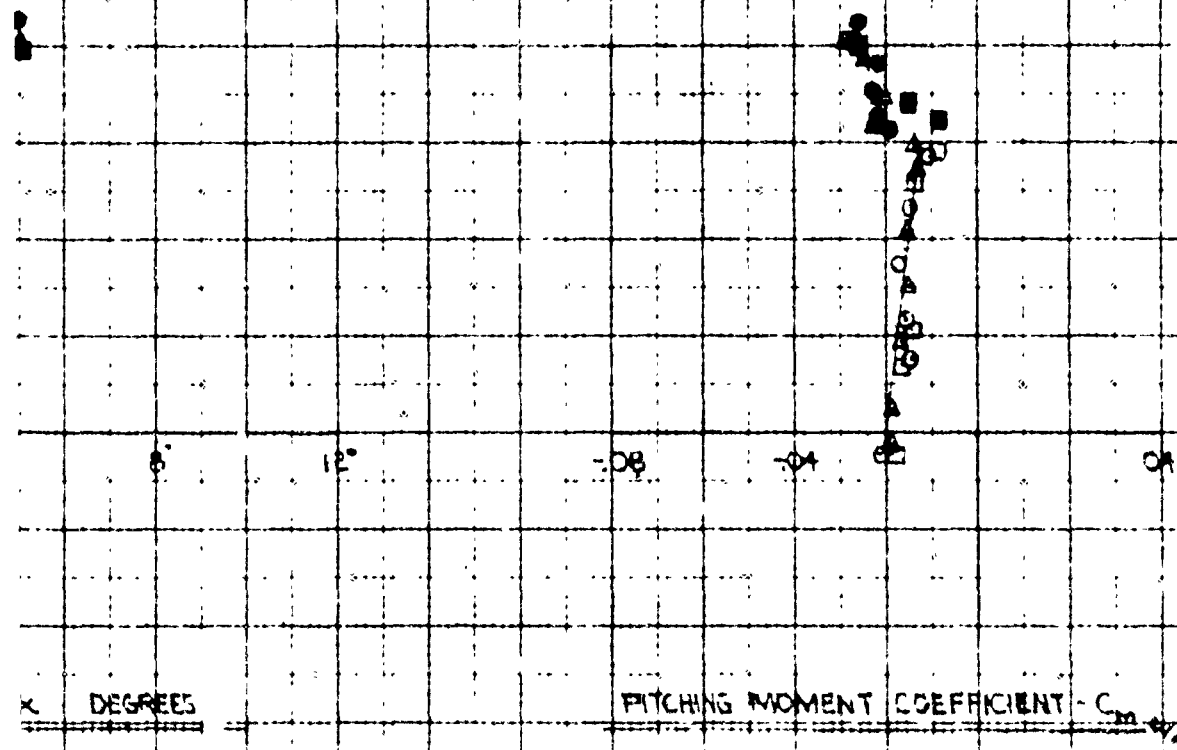
WHIRLING TANK TEST NO. 47
 BU SHIPS FOIL NO. 17
 M.O.C. .142 FT S.B. 23 SQ IN
 VELOCITY . 70 KTS

SYMBOL	W/C	FRONT NO	CAV. NO
○	.75	6.34	.196
△	1.00	5.51	.24
□	1.25	4.95	.276



W/C = 0.75

NOTE: DARKENED POINTS INDICATE VENTED FLOW



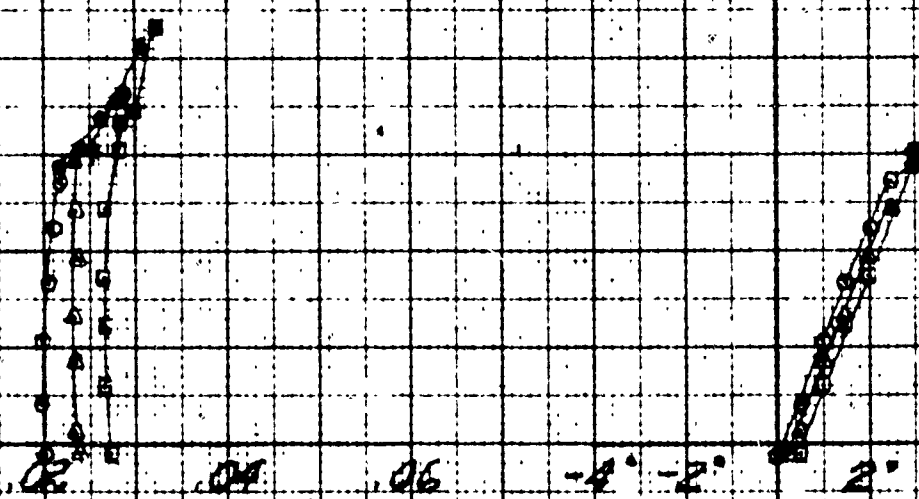
MOE ALBANY 1947 12705

SYMBOL	R/c	FROUDE NO.	CAV. NO.
○	.75	6.3416	1.585
△	1.00	5.5147	1.750
□	1.25	4.9527	1.915

NOT
1

LIFT COEFF ~ C_L

5
4
3
2
1
0
-1



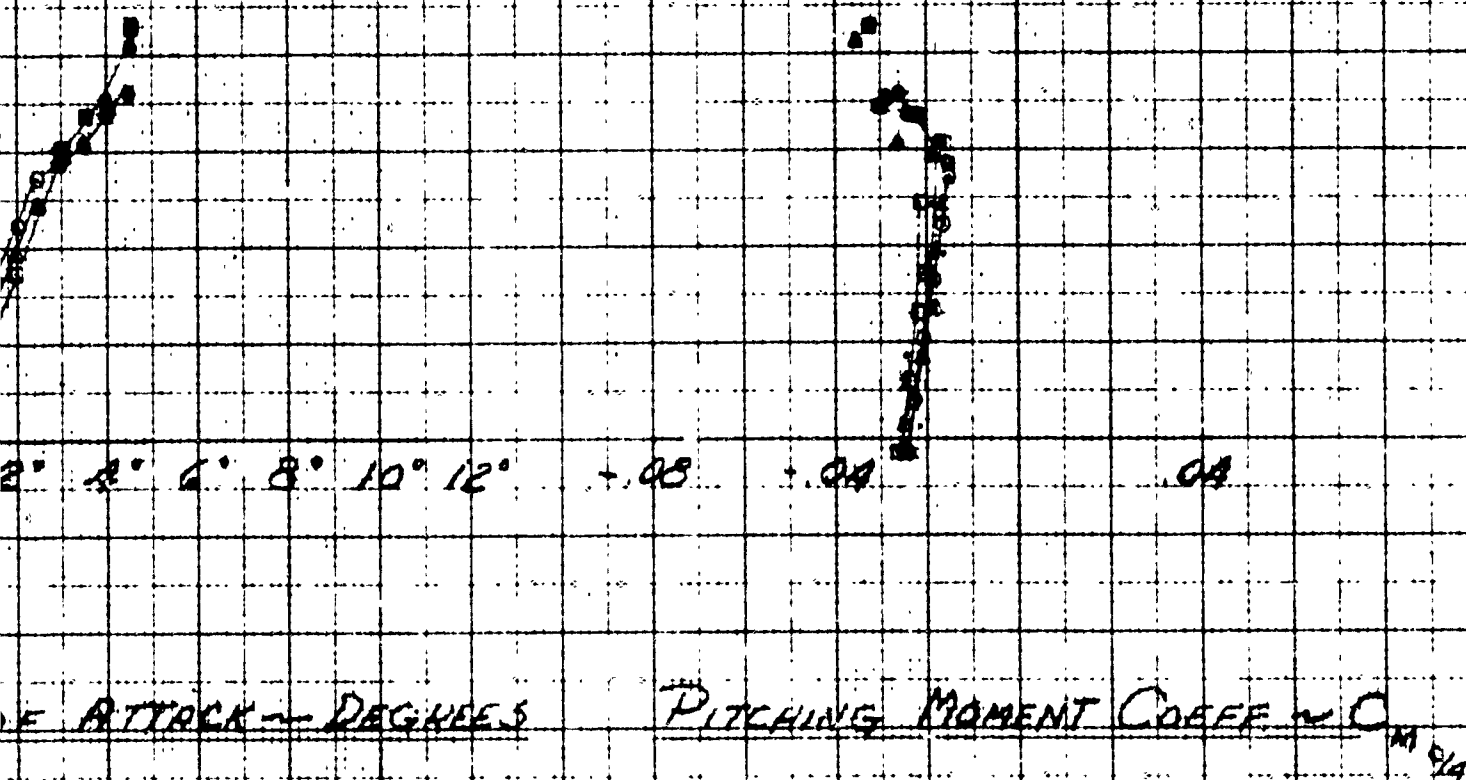
DRAG COEFF ~ C_D

ANGLE OF

W.T.T. No. 47 BUSHIPS FOIL No. 17
 M.A.C. 142 FT. AREA = 8.23 π "
 VEL = 80 KTS

NOTES:

1. DARKENED SYMBOLS
 INDICATE VENTED FLOW.



SYMBOL	α/c	FROUDE NO.	CAV. NO.
O	.75	NOT TESTED	—
Δ	1.00	5.517	.152
E	1.25	4.9527	.149

No.

LIFT COEFF - C_L

5

4

3

2

1

0

-1

.02

.04

.06

-4° - 2°

2

DRAG COEFF - C_D

ANGLE OF A

W.T.T. NO. 47 BUSHIPS FOIL NO. 17

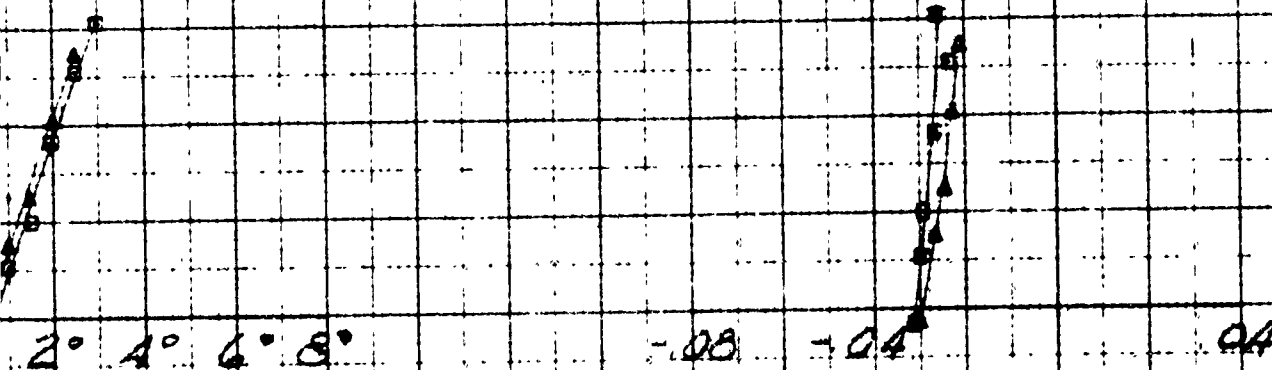
MAC = 182 FT. AREA = 8.230"

VEL = 90 KTS

NOTES:

1. DARKENED SYMBOLS

INDICATE VENTED FLOW



ATTACK - DEGREES

PITCHING MOMENT COEFF - $C_{M_{CP}}$

WHIRLING TANK TEST
 BU SWIPS FOIL NO.
 MAX. 1.142 FT 5.
 VELOCITY 96 K

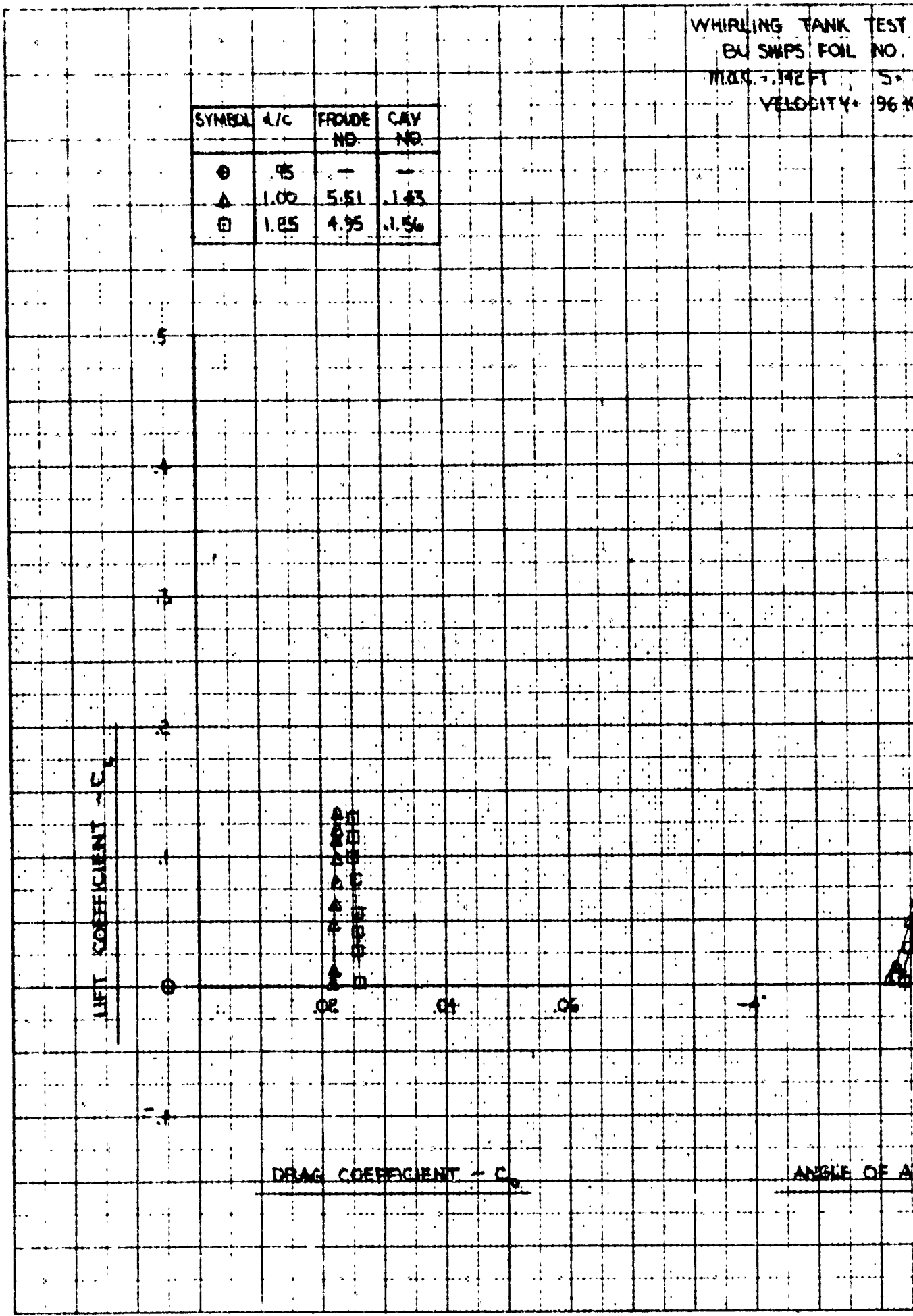
SYMBOL	A/C	FROUDE NO.	CAY NO.
⊙	.75	—	—
Δ	1.00	5.51	1.43
⊞	1.25	4.95	1.56

K-2 ALBANY 1007 2325
 3

LIFT COEFFICIENT - C_L

DRAG COEFFICIENT - C_D

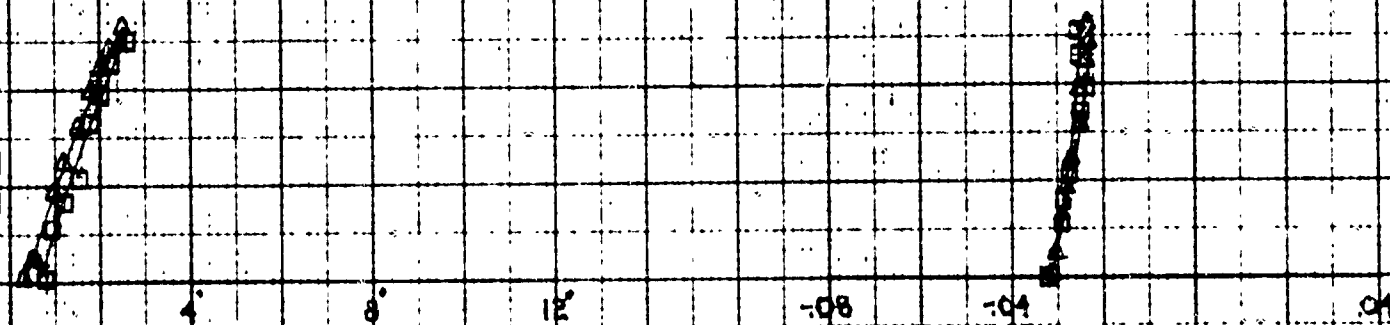
ANGLE OF A



TEST NO 41
NO. 17
5.8.55 SQIN.
96 KTS

NOV 14 1957

NOTE $\alpha/c = .75$ WAS NOT TESTED DUE TO EXTREME WATER
ROUGHNESS.



ANGLE OF ATTACK - DEGREES

PITCHING MOMENT COEFFICIENT - C_m

WHIRLING TANK TEST
 BU SHIPS FOIL NO 1
 M.W.C. .142 FT 3.8
 VELOCITY . 40 M

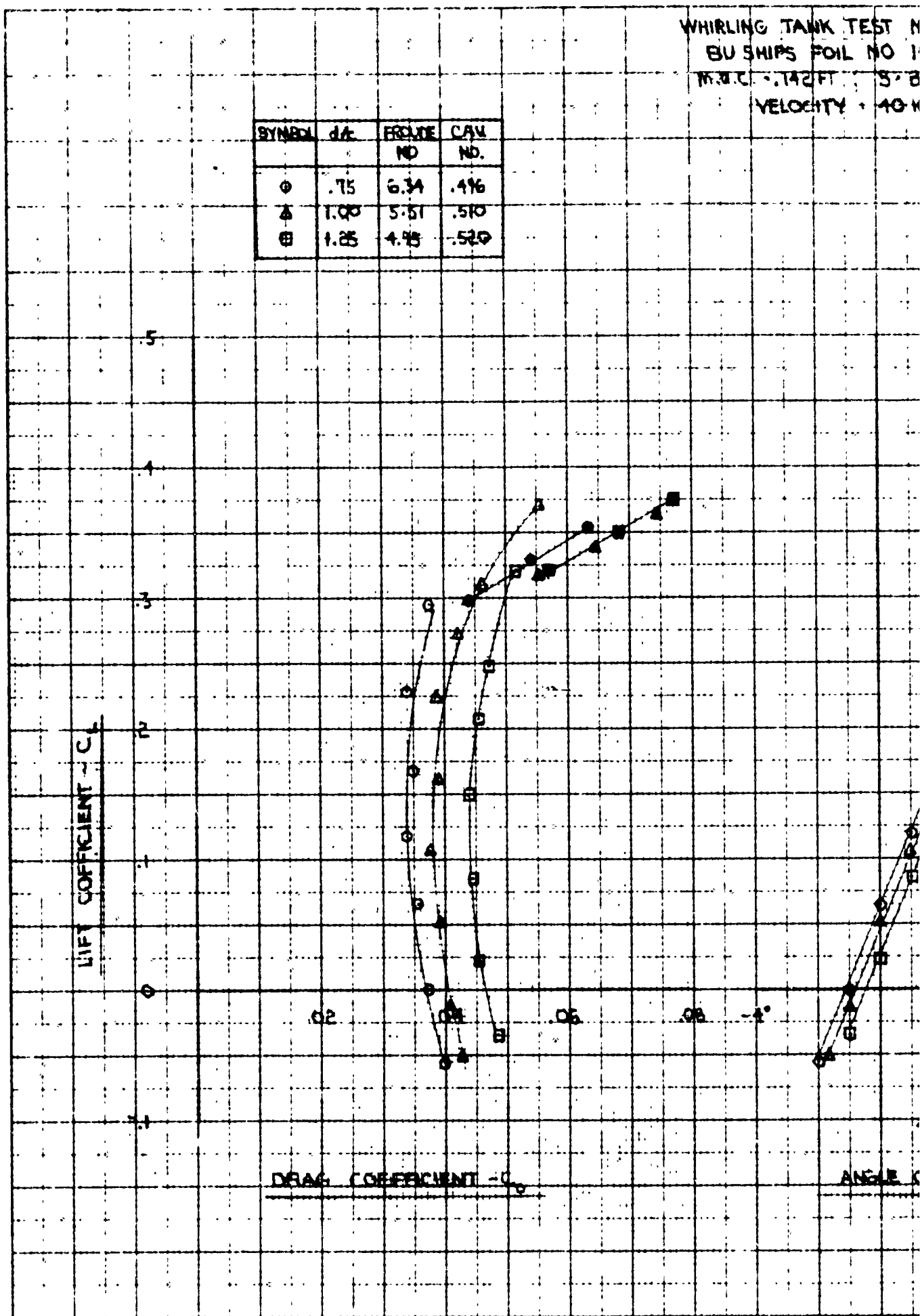
SYMBOL	dA	FROUDE NO	CAV NO.
○	.75	6.34	.496
△	1.00	5.51	.510
⊞	1.25	4.95	.520

LIFT COEFFICIENT - C_L

DRAG COEFFICIENT - C_D

ANGLE α

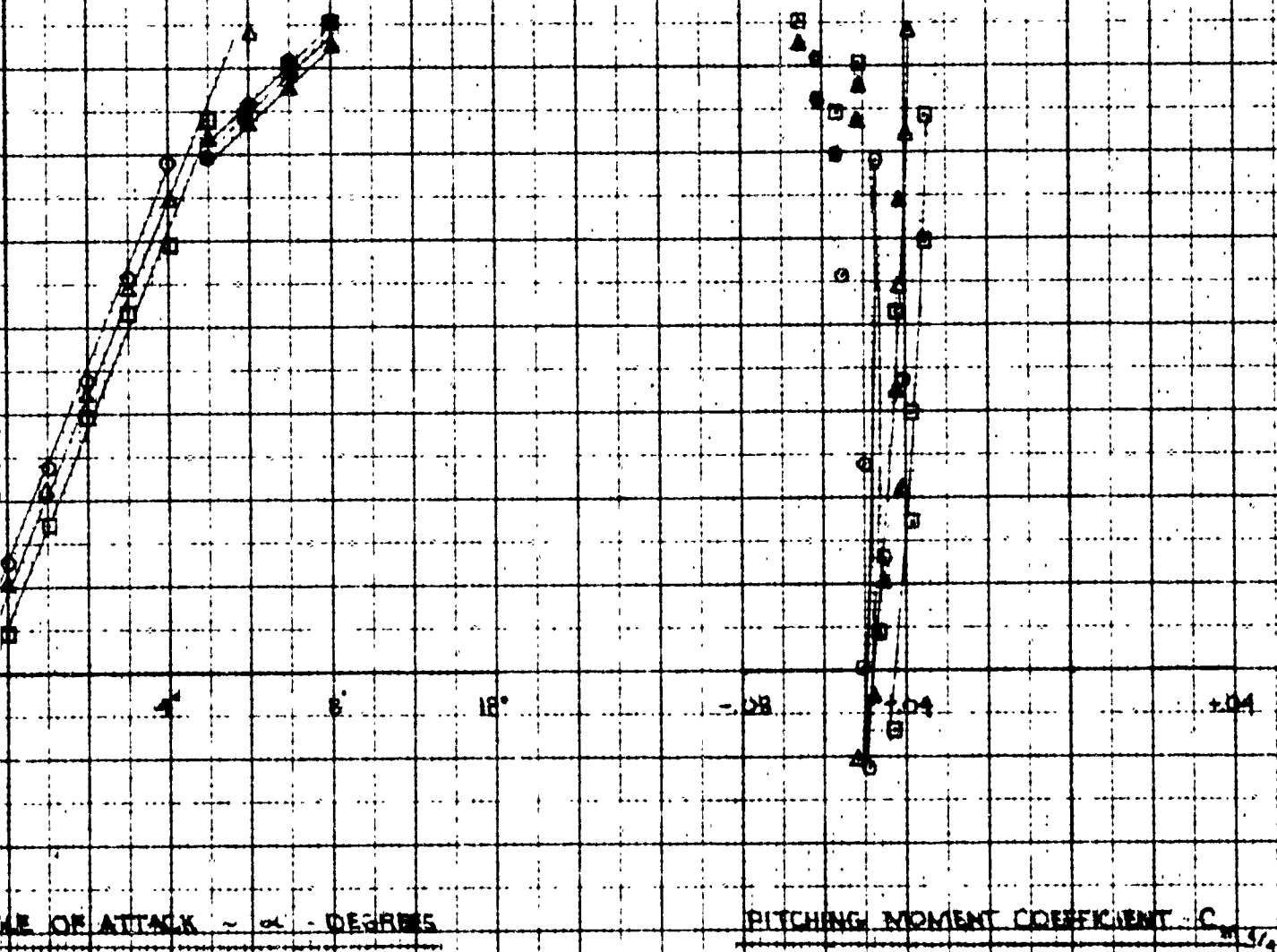
NO. 1
 5200



ST NO 48
10 18
5.83 SQ IN.
10 KTS

PAGE IV. 98

NOTE: DARKENED POINTS INDICATE VENTED FLOW



SYMBOL	q/c	FROUDE NO.	CAV. NO.
○	.75	6.3416	.335
△	1.00	5.5147	.353
□	1.25	4.9527	.364

LIFT COEFF - C_L

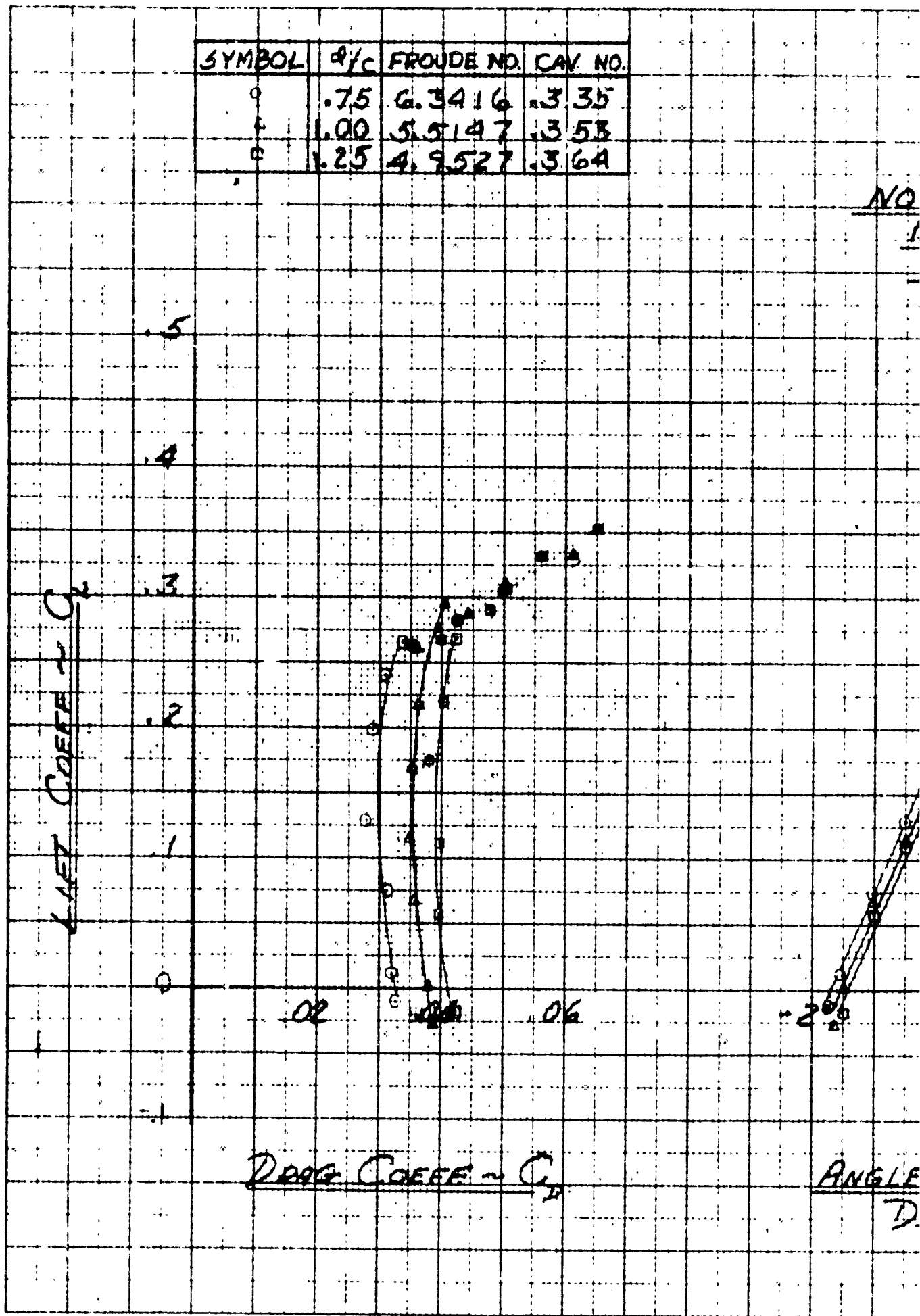
DRAG COEFF - C_D

ANGLE
D

NO

1

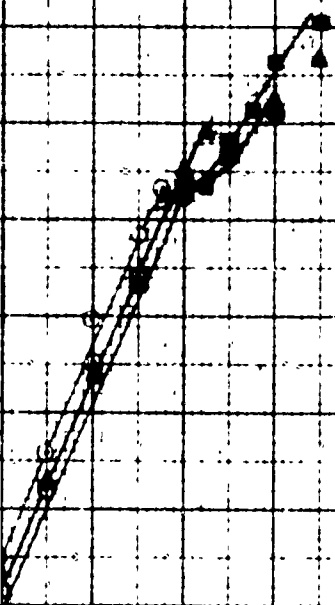
W-E ALBANY 1957



W.T.T. NO. 48 BUSHING R. & L. 100
 N.A.C. = .142 FT. AREA = 8230
VEL = 50 KTS

NOTE:

1. DARKENED SYMBOLS
INDICATE VENTED FLOW



2° 4° 6° 8°

0.08

-0.04

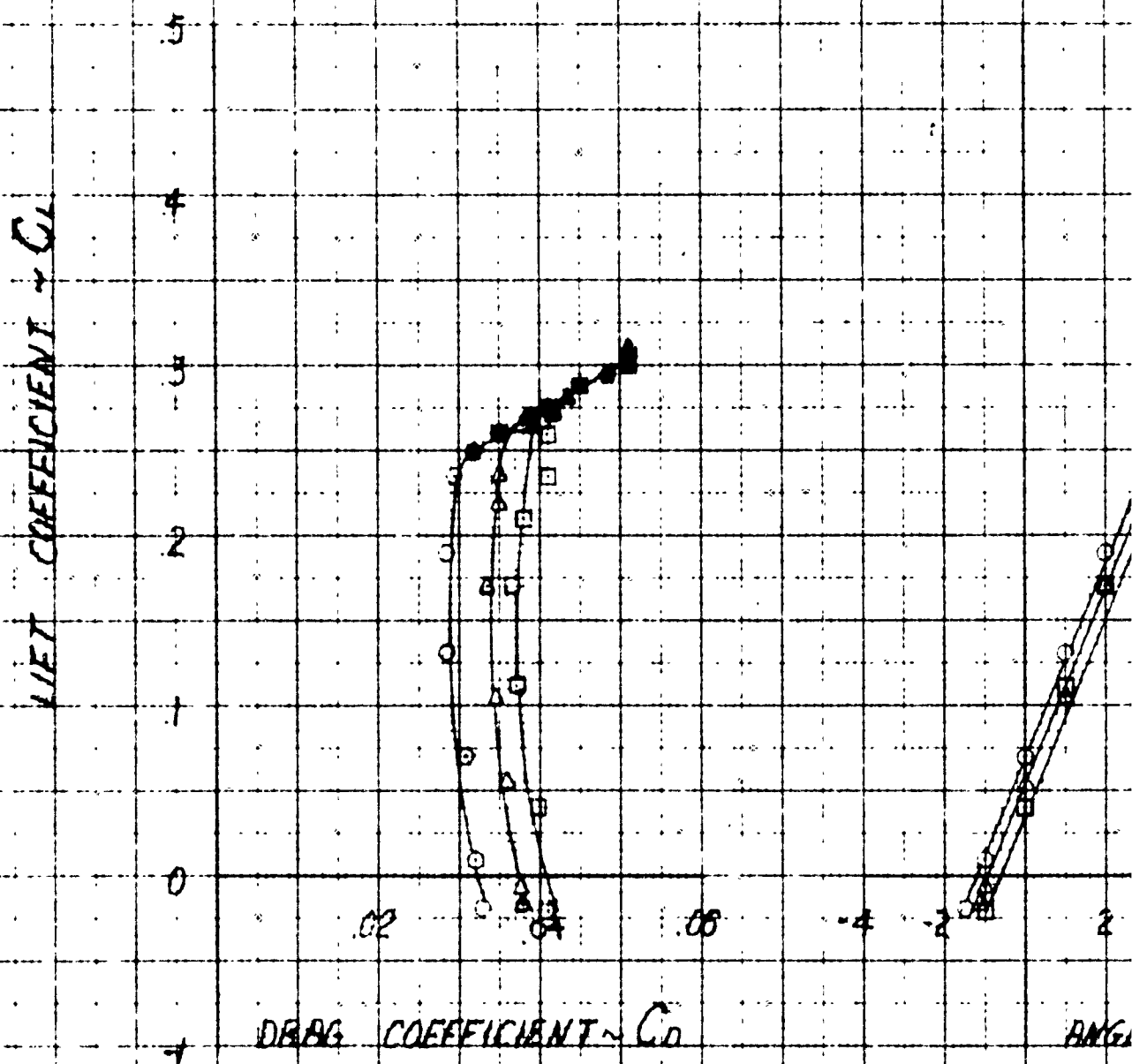
0.04

ANGLE OF ATTACK
 DEGREES

PITCHING MOMENT COEFF - C_m

SYMBOL	d/c	FRONDE NO.	CAV. NO.
⊕	75	6 34	.2 47
⊖	100	5 51	.2 62
⊗	125	4 95	.2 79

NOTE

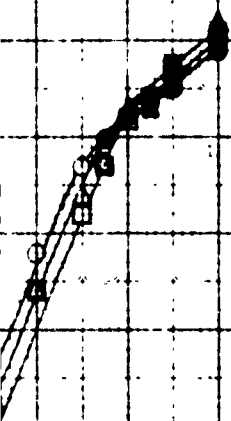


WHIRLING TANK TEST NO. 48

BU SHIPS FOIL NO. 18

 $S = 8.23 \text{ in}^2$ $MAC = .142 \text{ ft.}$ $V = 50 \text{ MTS.}$

NOTE:

DARKENED POINTS INDICATE
VENTED FLOW.ANGLE OF ATTACK - α - DEGREESPITCHING MOMENT
COEFFICIENT - $C_{m, \alpha}$

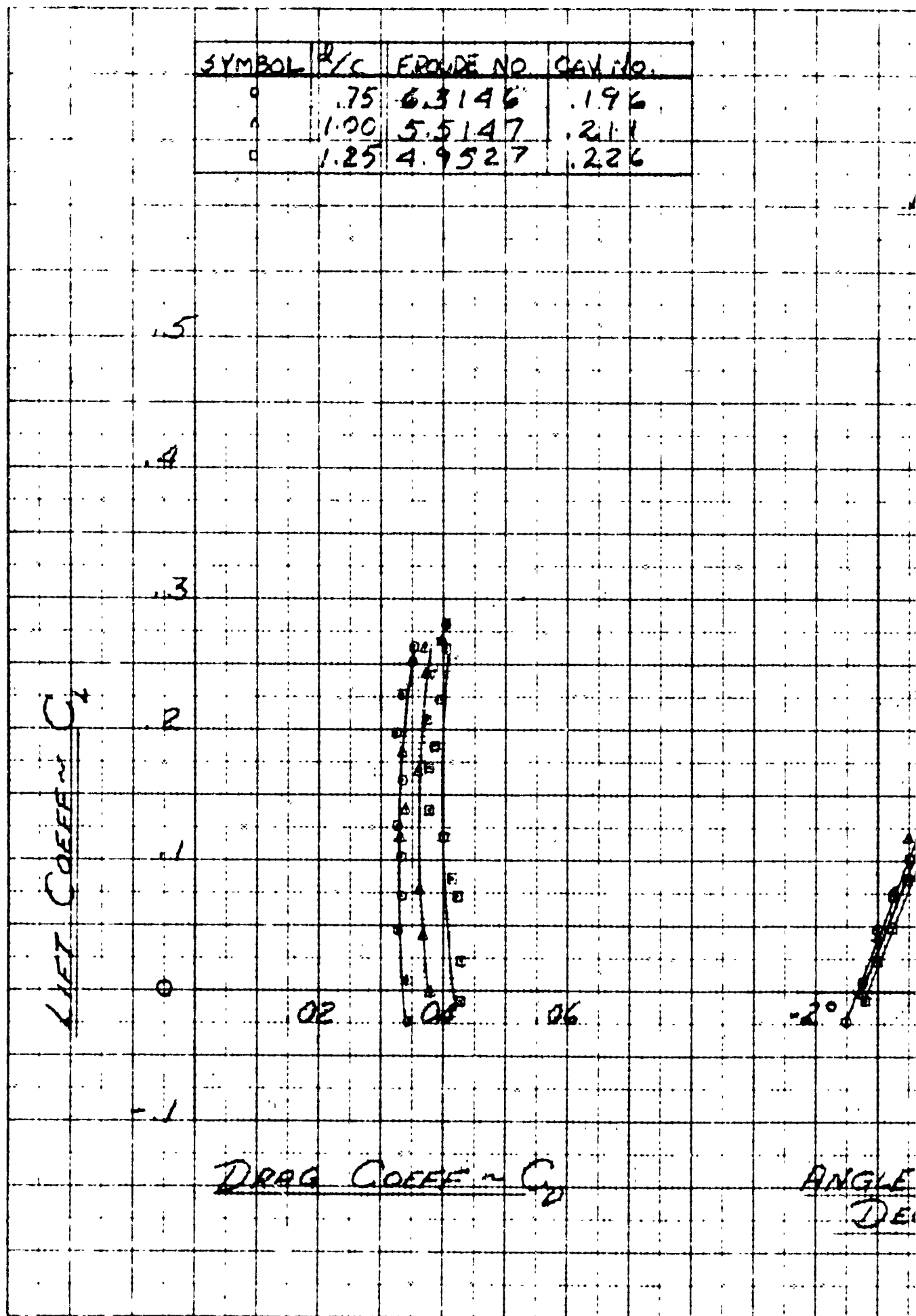
WAE 11/20/67 521.4

SYMBOL	R/C	FRONDE NO	CAV. NO.
o	.75	6.3146	.196
^	1.00	5.5147	.211
□	1.25	4.9527	.226

LIFT COEFF ~ C_L

DRAG COEFF ~ C_D

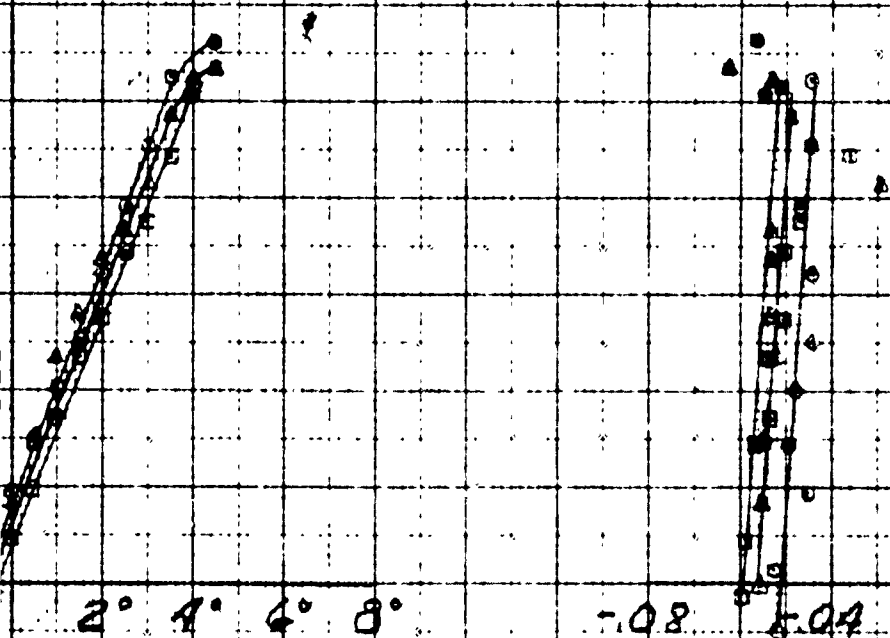
ANGLE
DEG



W.T.I. No 48 BUSHIPS Foil No. 18

M.A.C. = .142 FT AREA = 8.230"

VEL = 70 KTS

NOTE:1. DARKENED SYMBOLSINDICATE VENTED FLOWANGLE OF ATTACK
DEGREESPITCHING MOMENT
COEFF. ~ C_m

SYMBOL	$\frac{h}{c}$	FROUDE NO	CAV. NO.
o	.75	6.3416	.158
A	1.00	5.5147	.175
P	1.25	4.9537	.192

NOTE:

1. DI

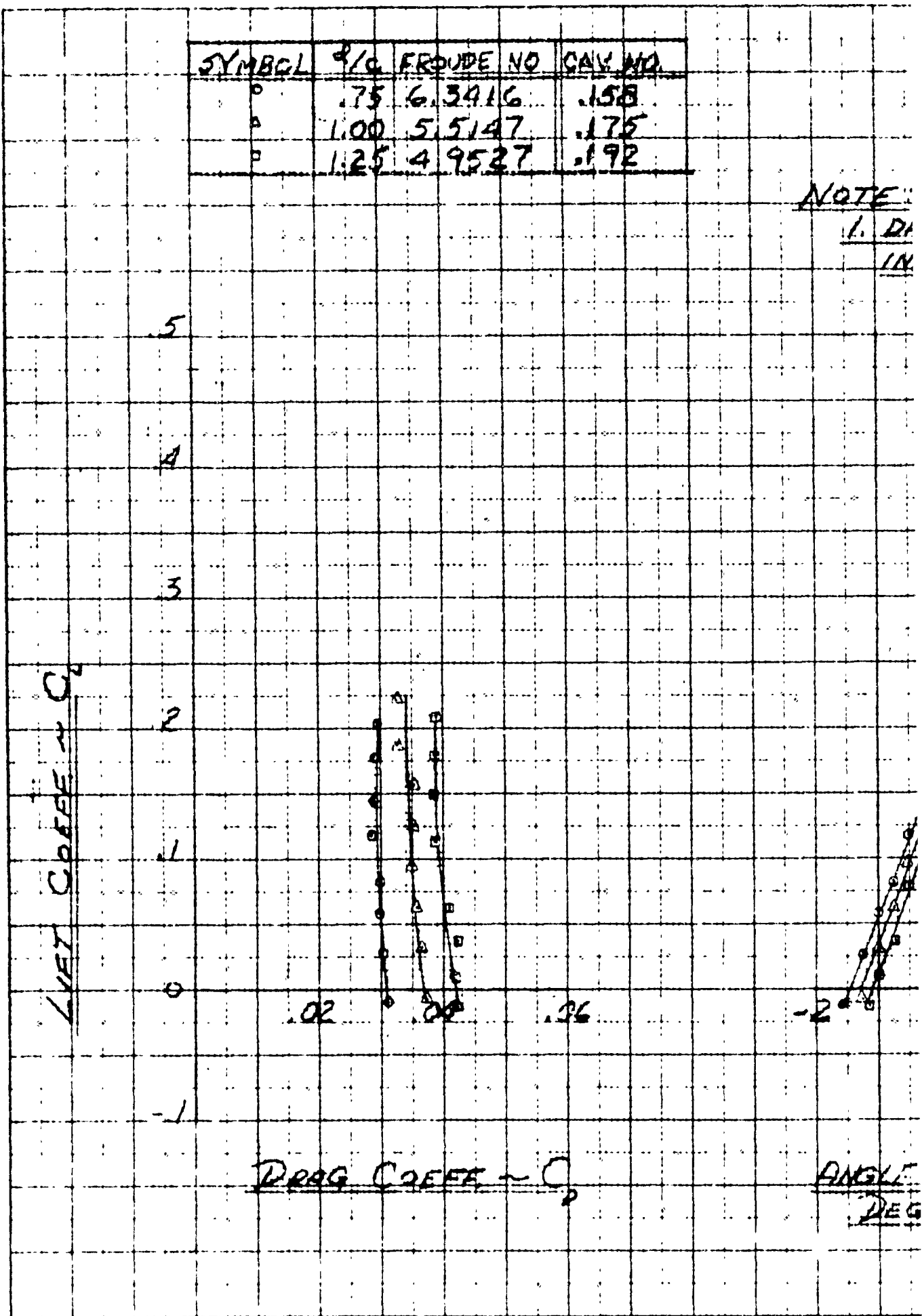
IN

W-E ALBANY 1907

LIFT COEFF - C_L

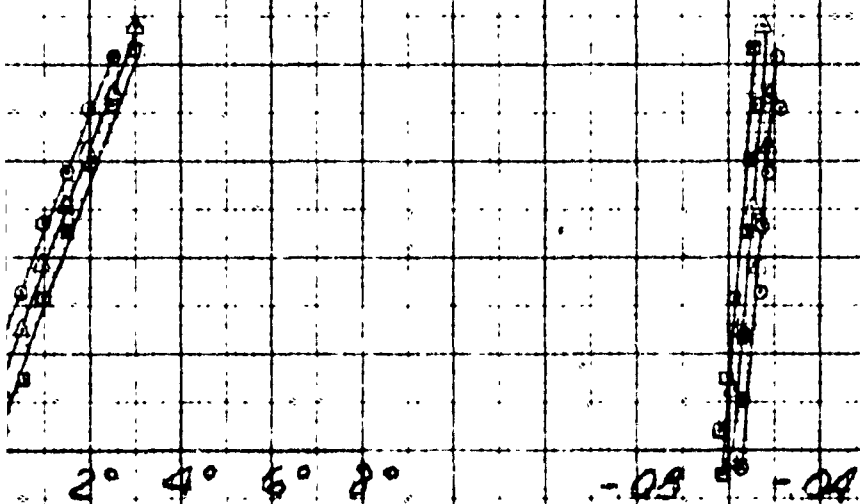
DRAG COEFF - C_D

ANGLE
DEG



W. T. T. 110. 48 BUSHIPS FOIL NO. 18
M.A.C. = 1.42 FT AREA = 8.23 sq ft
VEL. = 80 KTS

DARKENED SYMBOLS
INDICATE VENTED FLOW



ANGLE OF ATTACK
DEGREES

PITCHING MOMENT
COEFFICIENT C_m

WHIRLING TANK TEST
 BU SHIPS FOIL NO
 T.M.C. 112 FT 5°
 VELOCITY: 30 KT

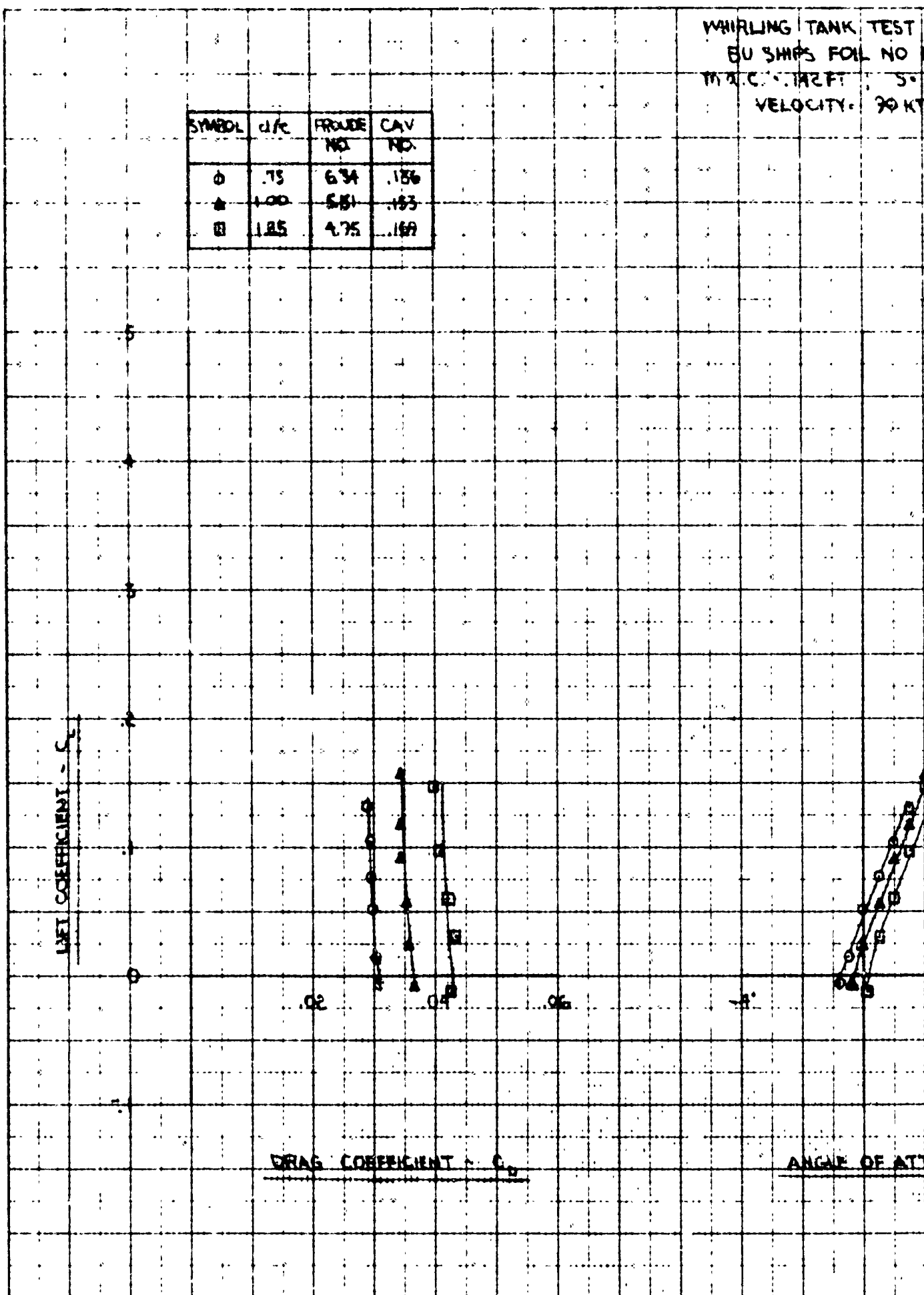
SYMBOL	d/c	FROUDE NO.	CAV NO.
○	.75	6.34	.186
▲	1.00	5.81	.183
□	1.25	4.75	.169

14-E 1/10/48 1000

LIFT COEFFICIENT - C_L

DRAG COEFFICIENT - C_D

ANGLE OF ATTACK



EST NO 48
NO 18
5.8.23 30 IN
0 KTS

PAGE IV. 103



ATTACK - DEGREES

PITCHING MOMENT COEFFICIENT - C_m

SYMBOL	d/c	FROUDE NO	CAV. NO
○	75	6.34	.130
△	100	5.51	.144
□	125	4.95	.159

LIFT COEFFICIENT C_L

DRAG COEFFICIENT C_D

ANGLE

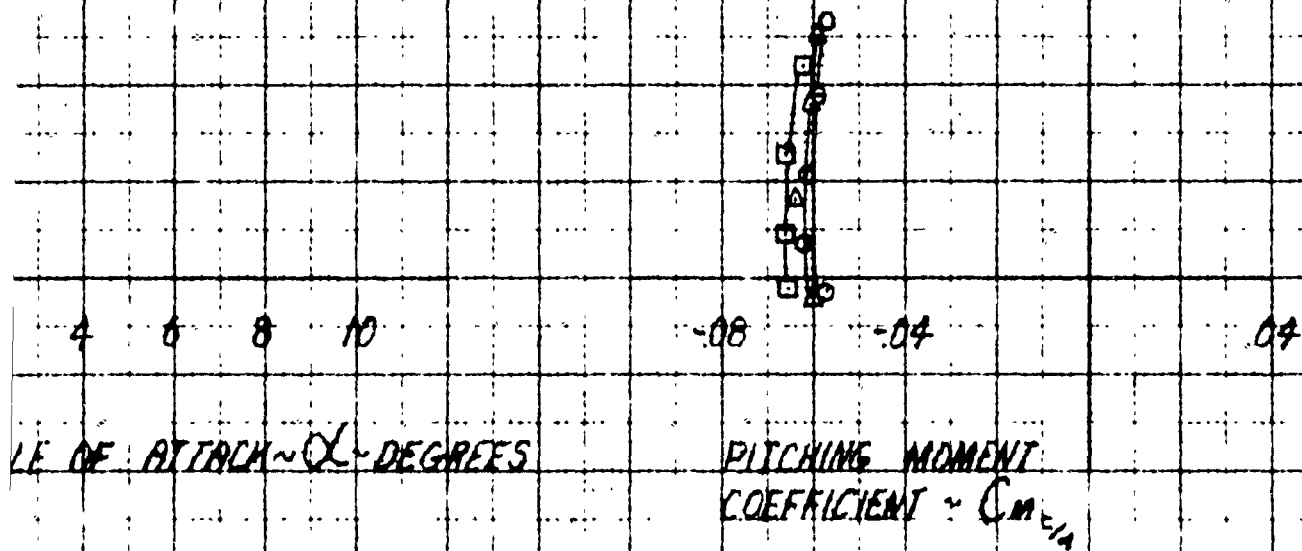
10-2-50

WHIRLING TANK TEST NO. 48

BU SHIPS FOIL NO. 18

$S = 8.23 \text{ m}^2$; $MAG = 142 \text{ AT}$

$V = 9.5 \text{ KTS.}$



SYMBOL	α/c	FROUDE NO.	CAV NO.
○	.75	7.1767	4.82
△	1.00	6.2353	4.93
□	1.25	5.5949	5.10

W&E ALBANY 1007 2-73

LIFT COEFF - C_L

.5
 .4
 .3
 .2
 .1
 0
 -.1

.02

.04

.06

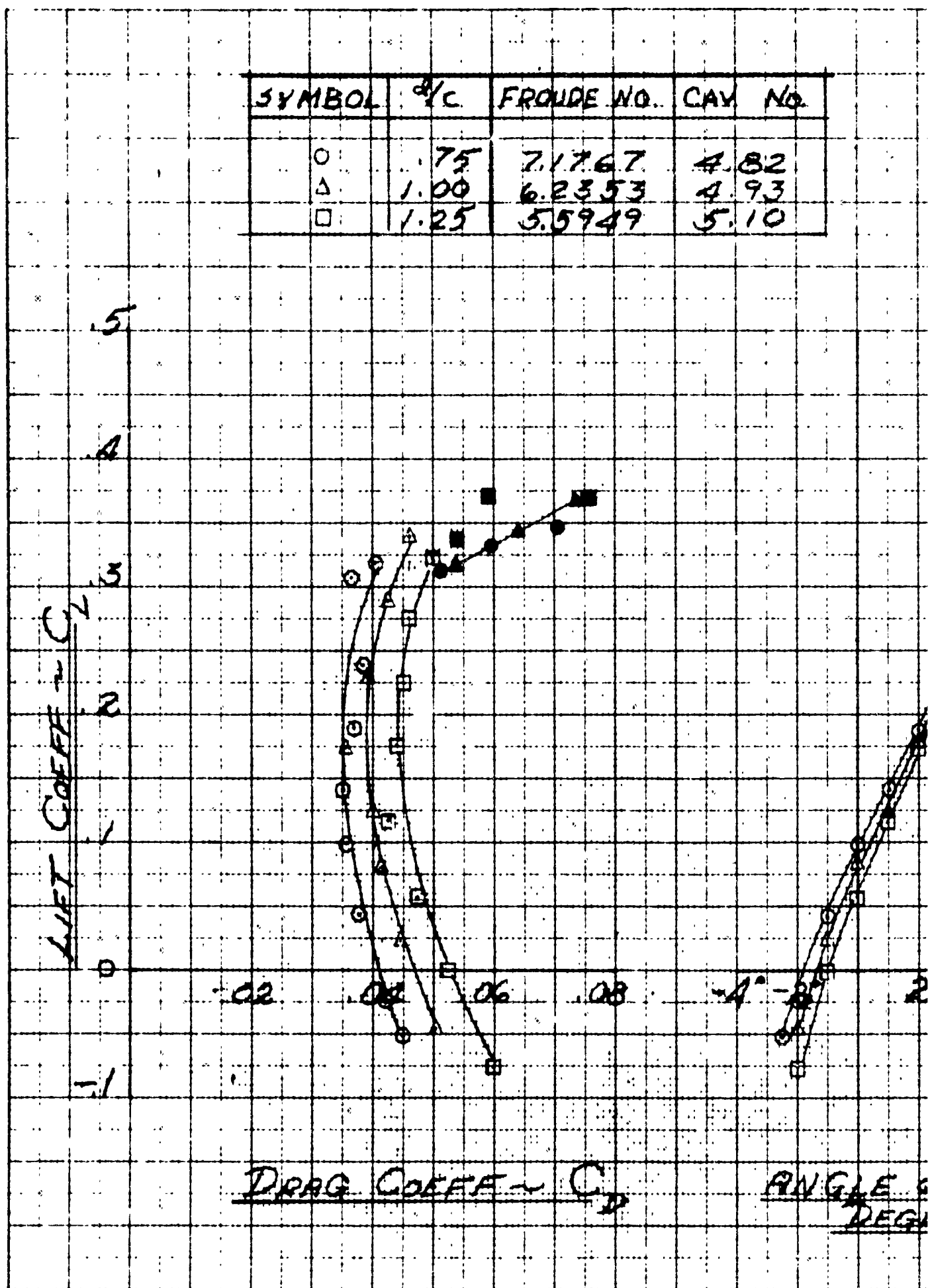
.08

.1

.12

DRAG COEFF - C_D

ANGLE
 DEGR



WIT No. 49 BUSHIPS FOIL No. 19

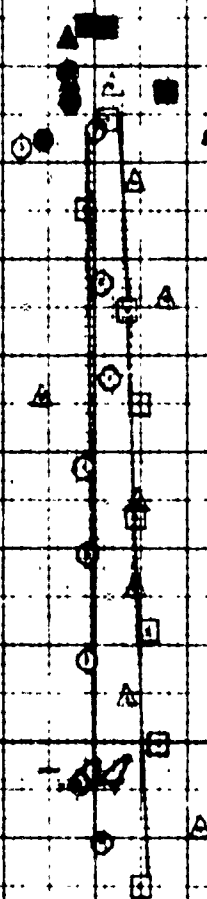
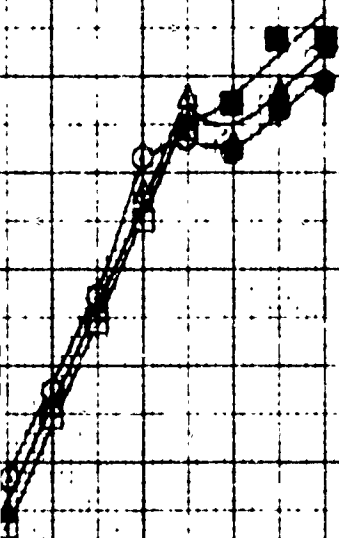
M.A.C. = .11 FT. AREA = 8.94 sq in

VEL = 40 KTS

NOTE:

1 DARKENED SYMBOLS

INDICATE VENTED FLOW.



ANGLE OF ATTACK
DEGREES

PITCHING MOMENT COEFF
 $C_{M'}$

SYMBOL	a/c	FRAUDE NO	CAR NO
○	.75	7.18	327
△	1.00	8.23	337
□	1.25	5.59	349

NOTE:

COEFFICIENT OF LIFT ~ C_L

3

4

3

2

1

0

02

04

06

-4

-2

2

DRAG COEFFICIENT ~ C_D

ANGLE

-1

0

1

2

3

4

5

6

7

8

9

10

11

12

13

14

15

16

17

18

19

20

21

22

23

24

25

26

27

28

29

30

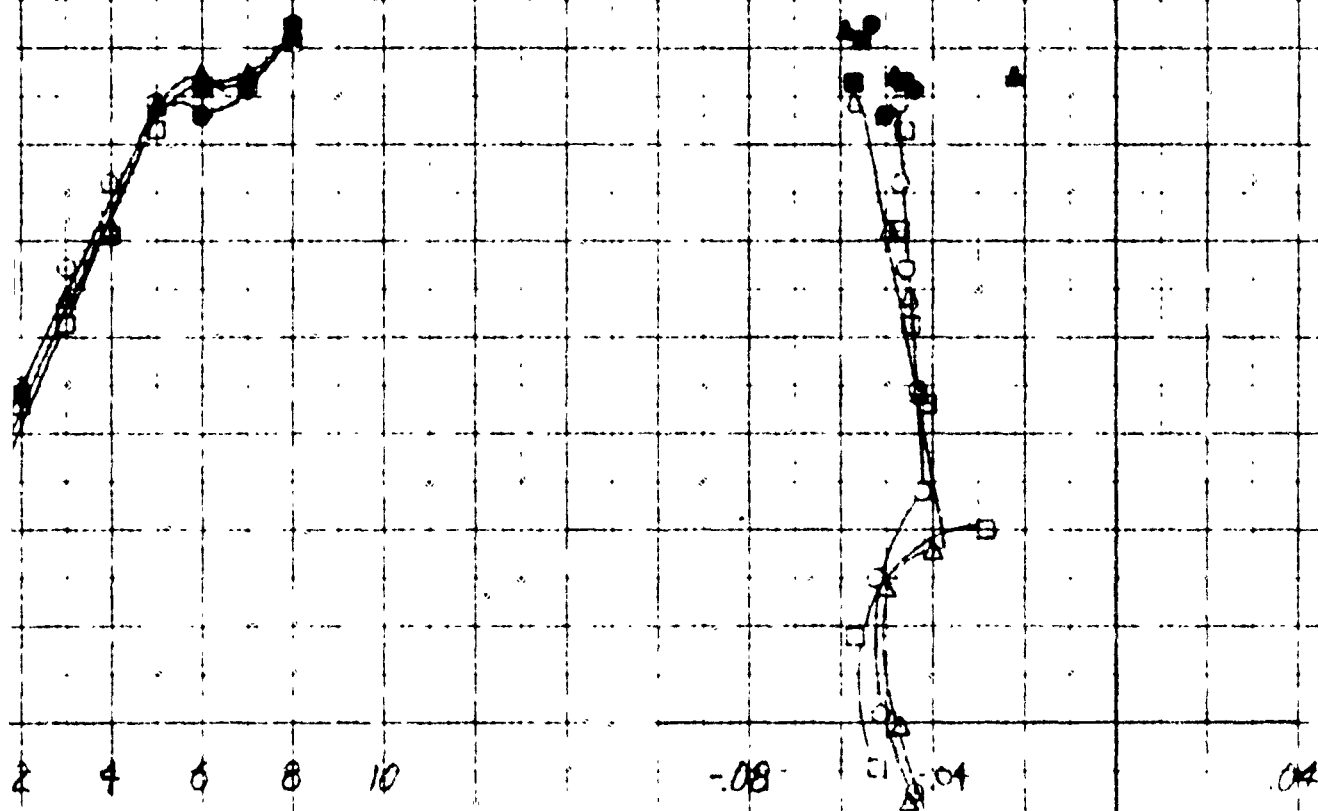
WHIRLING TANK TEST NO. 4.9

BU. SHIPS FOIL NO. 19

$S = 4.94 \text{ in}^2$, $MAC = .110 \text{ in}$

$V = 56 \text{ KTS}$

1. DARKENED SYMBOLS
INDICATE VENTED FLOW.



ANGLE OF ATTACK - α - DEGREES.

PITCHING MOMENT
COEFFICIENT - C_m %

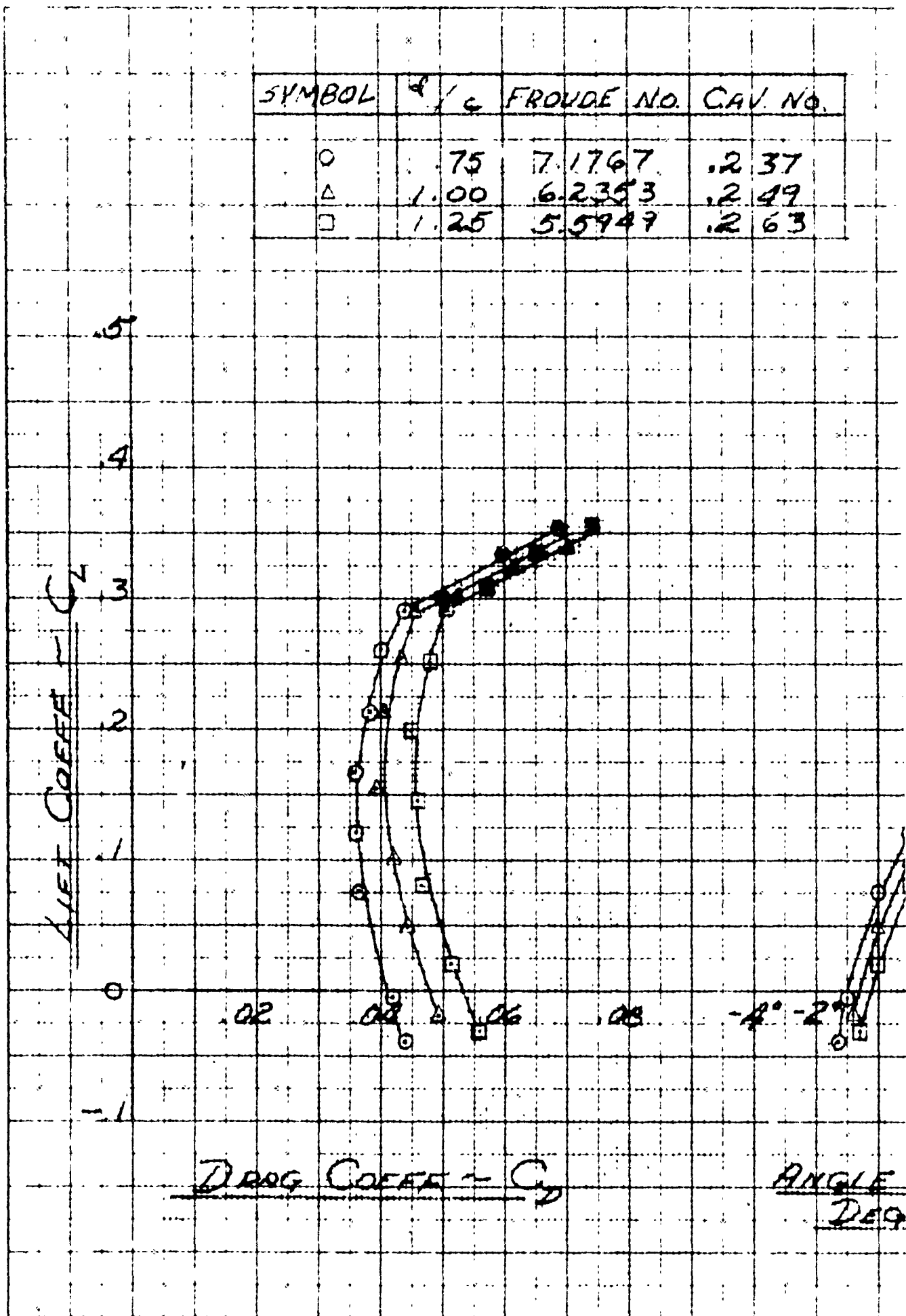
SYMBOL	$\frac{h}{c}$	FROUDE NO.	CAV. NO.
○	.75	7.1767	.237
△	1.00	6.2353	.249
□	1.25	5.5949	.263

NO. 3

LIFT COEFF - C_L

DRAG COEFF - C_D

ANGLE
DEG.



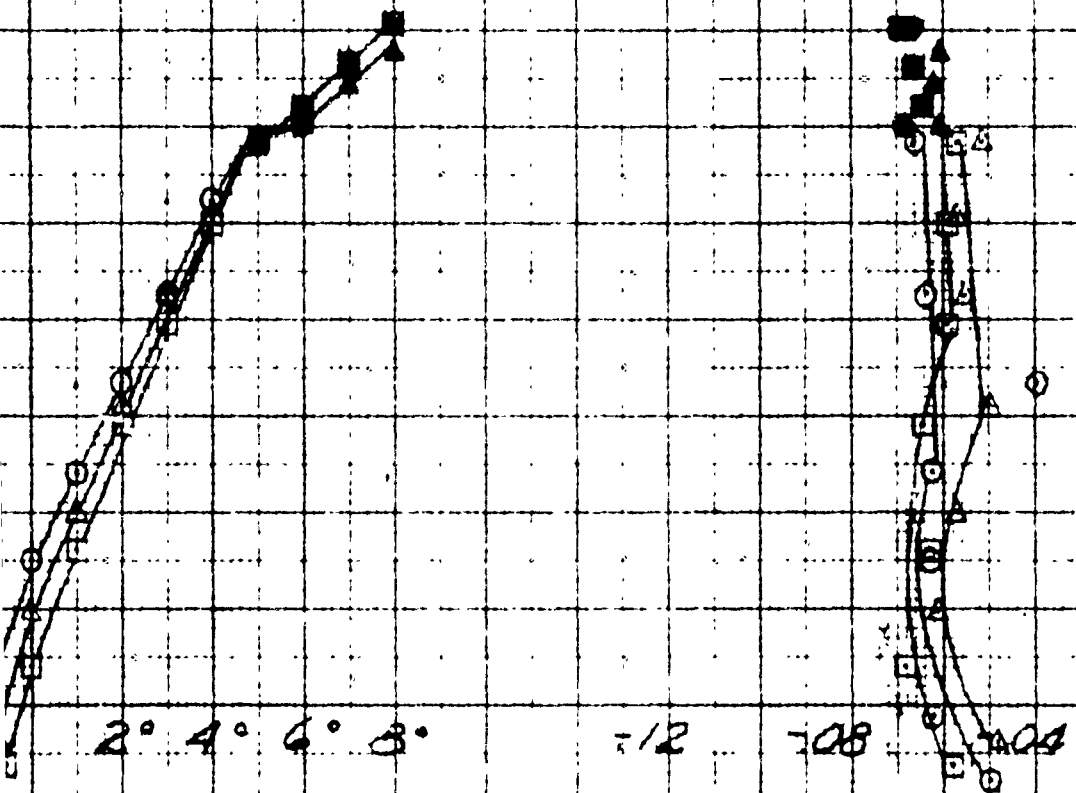
W.T.I. NO. 49 BUSHIPS FEEL NO. 19

MAC = .11 FT. AREA = 4.94 FT²

VEL = 60 KTS

NOTE:

1. DARKENED SYMBOLS
INDICATE VENTED FLOW



ANGLE OF ATTACK
DEGREES

PITCHING MOMENT COEFF
 $C_{M \frac{c}{4}}$

SYMBOL	d/c	FROUDE NO.	CAV. NO.
○	.75	7.18	.185
△	1.00	6.23	.197
□	1.25	5.59	.209

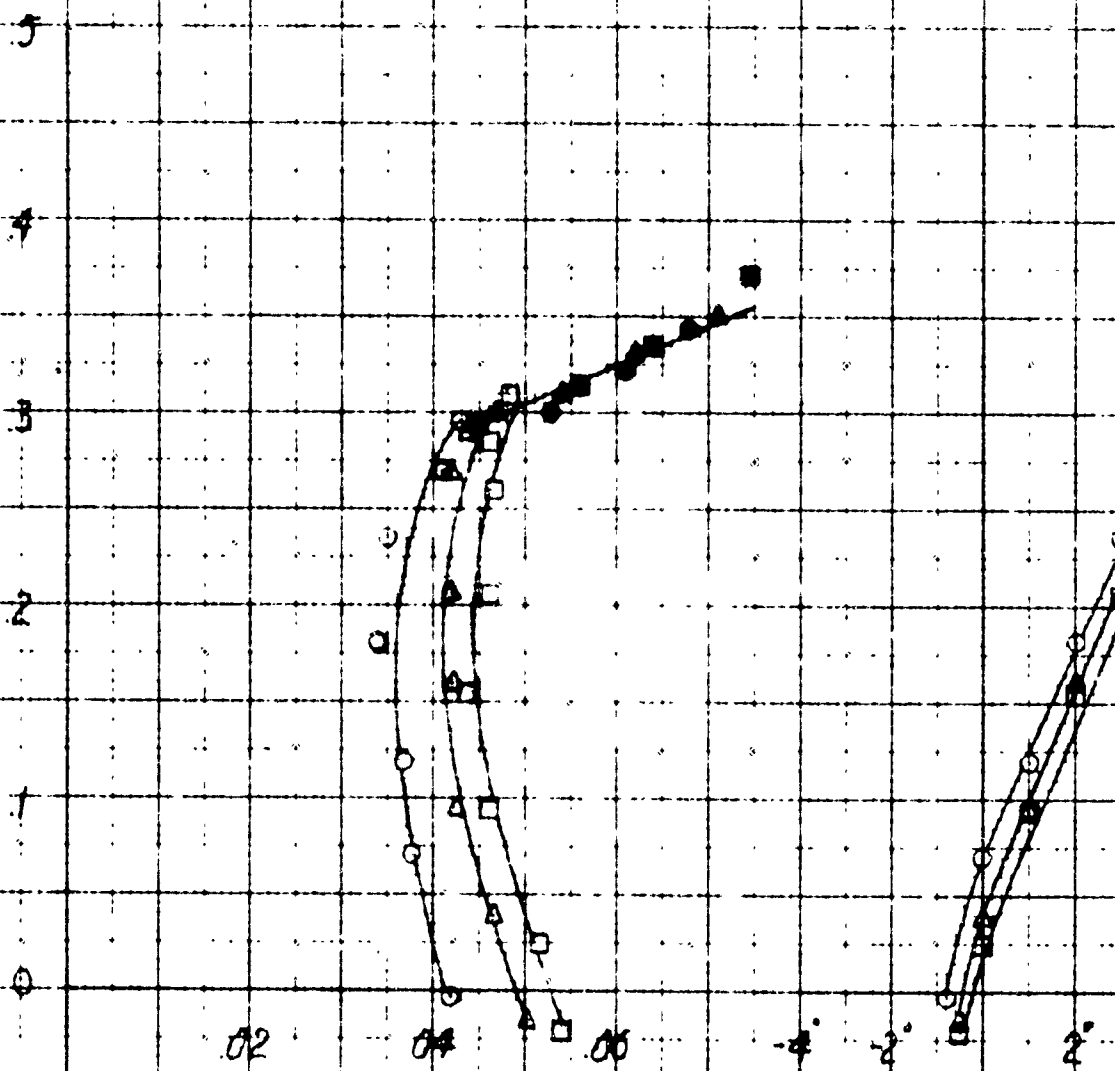
NOTE:

COEFFICIENT OF LIFT $\sim C_L$

DRAG COEFFICIENT $\sim C_D$

ANGLE

NO. 34



WHIRLING TANK TEST NO. 49

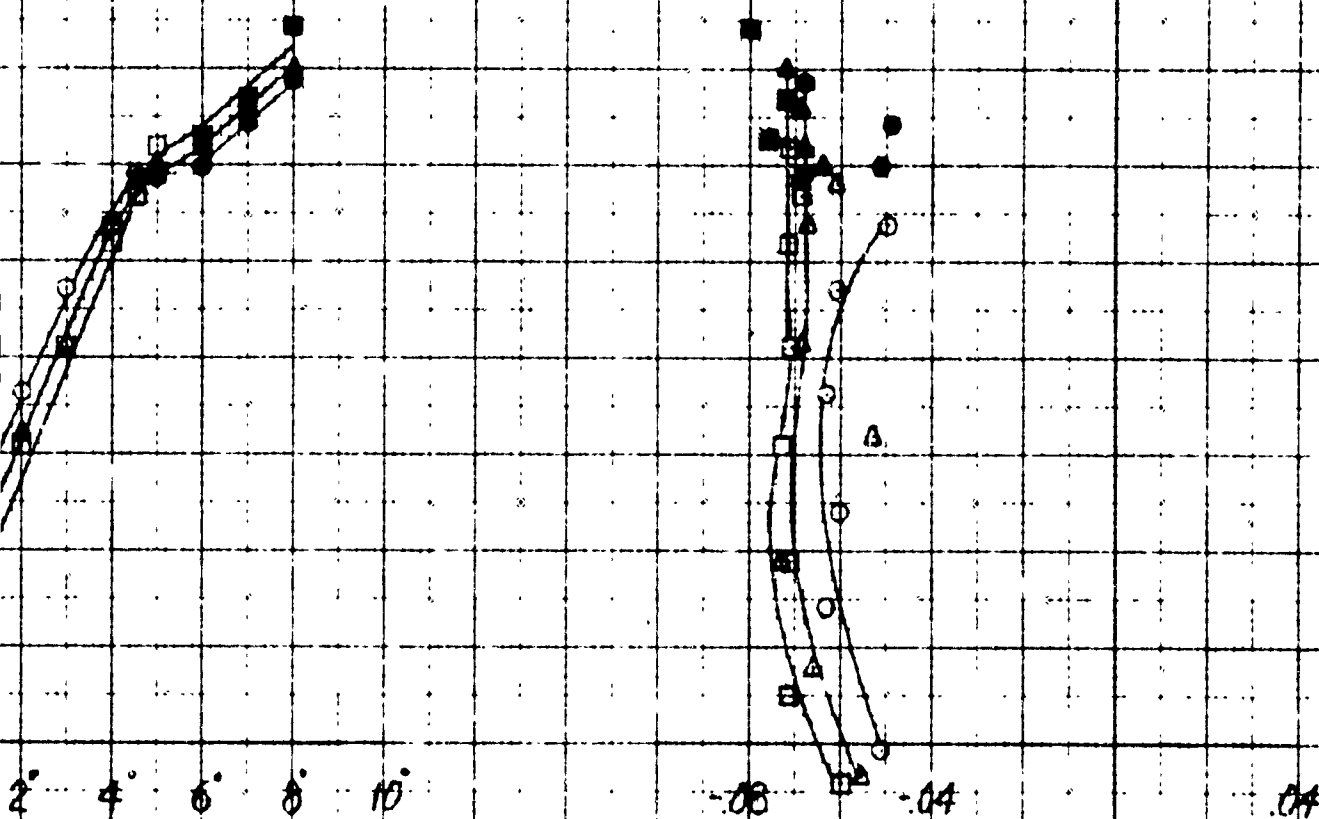
BU SHIPS FOIL NO 19

$S = 4.94 \text{ IN}^2$; $MAC. = .110 \text{ FT}$

$V = 70 \text{ KTS}$

E:

1. DARKENED SYMBOLS
INDICATE VENTED FLOW.

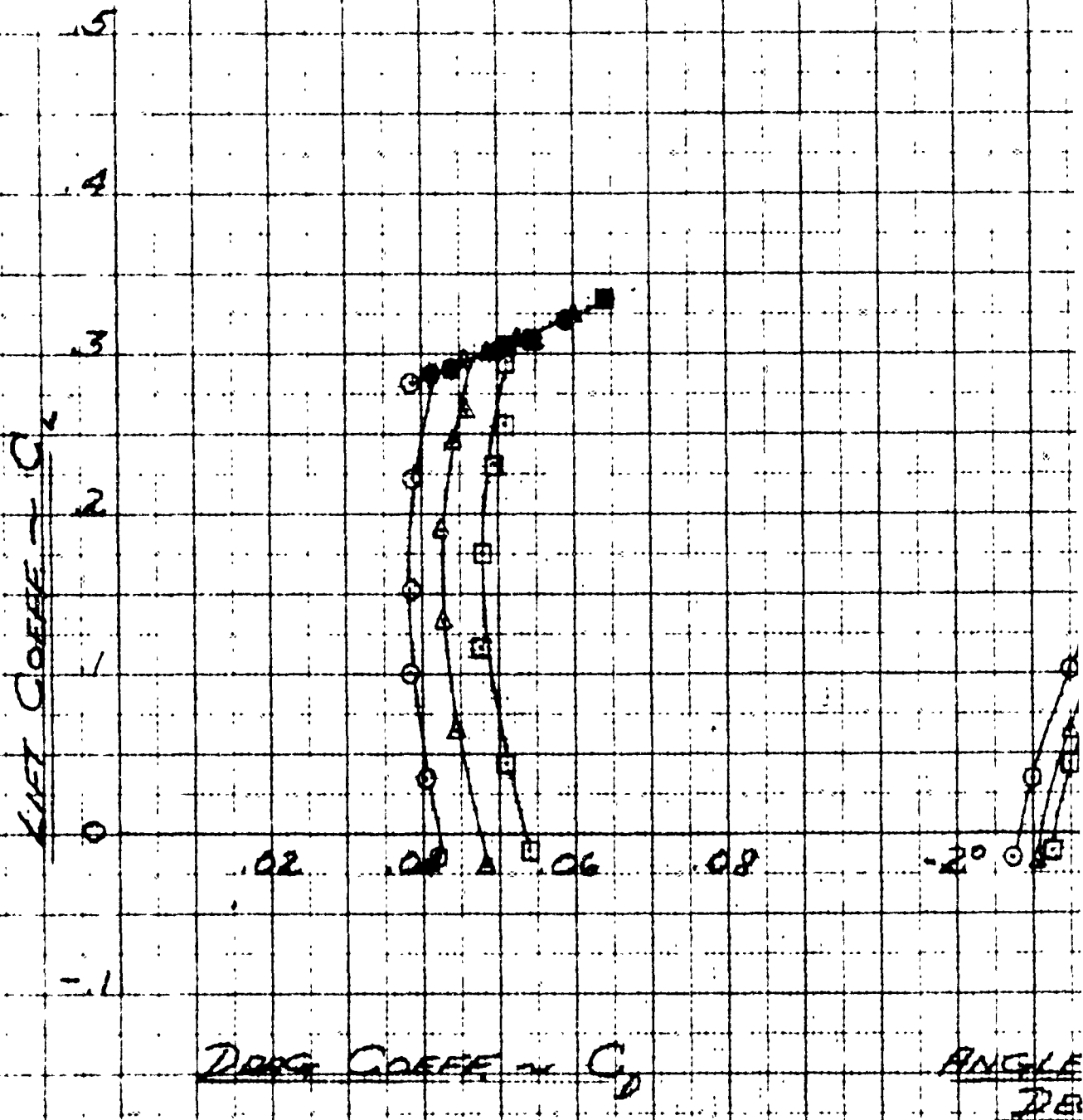


ANGLE OF ATTACK - α - DEGREES

PITCHING MOMENT
COEFFICIENT - C_m

SYMBOL	d/c	FROUDE No.	CAN. No.
O	.75	7.1767	.150
Δ	1.00	6.2353	.162
□	1.25	5.5949	.175

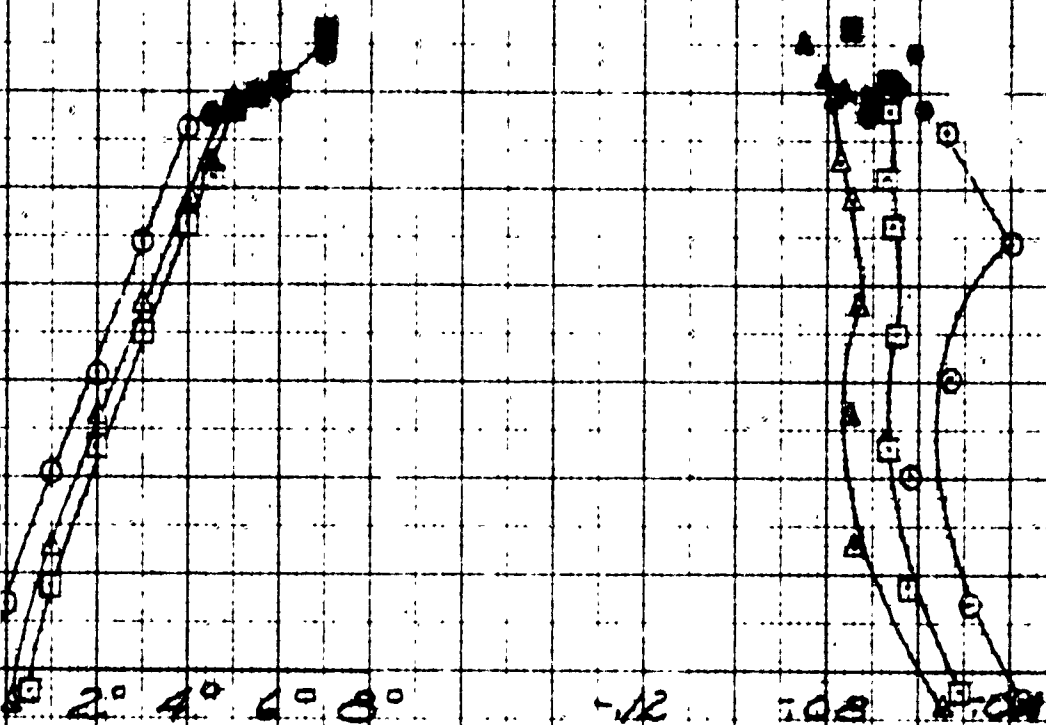
MS 1/16" x 1/16" x 1/16"



W.T.T. NO. 49 BUSHIPS NO. 19
 M.A.C. = 11 FT. AREA = 4.940 "
 VEL = 80 KTS

NOTE:

1. DARKENED SYMBOLS
 INDICATE VENTED FLOW



ANGLE OF ATTACK
 DEGREES

PITCHING MOMENT COEFF.
 $C_{m \frac{1}{2}}$

SYMBOL	d/c	FROUDE NO.	CAV. NO.
○	.75	7.18	.127
△	1.00	6.23	.140
□	1.25	5.59	.152

COEFFICIENT OF LIFT C_L

DRAG COEFFICIENT C_D

ANGLE

NO. 3, 1947, 1948

.5

.4

.3

.2

.1

0

-.1

.02

.04

.06

-.4°

-.2°

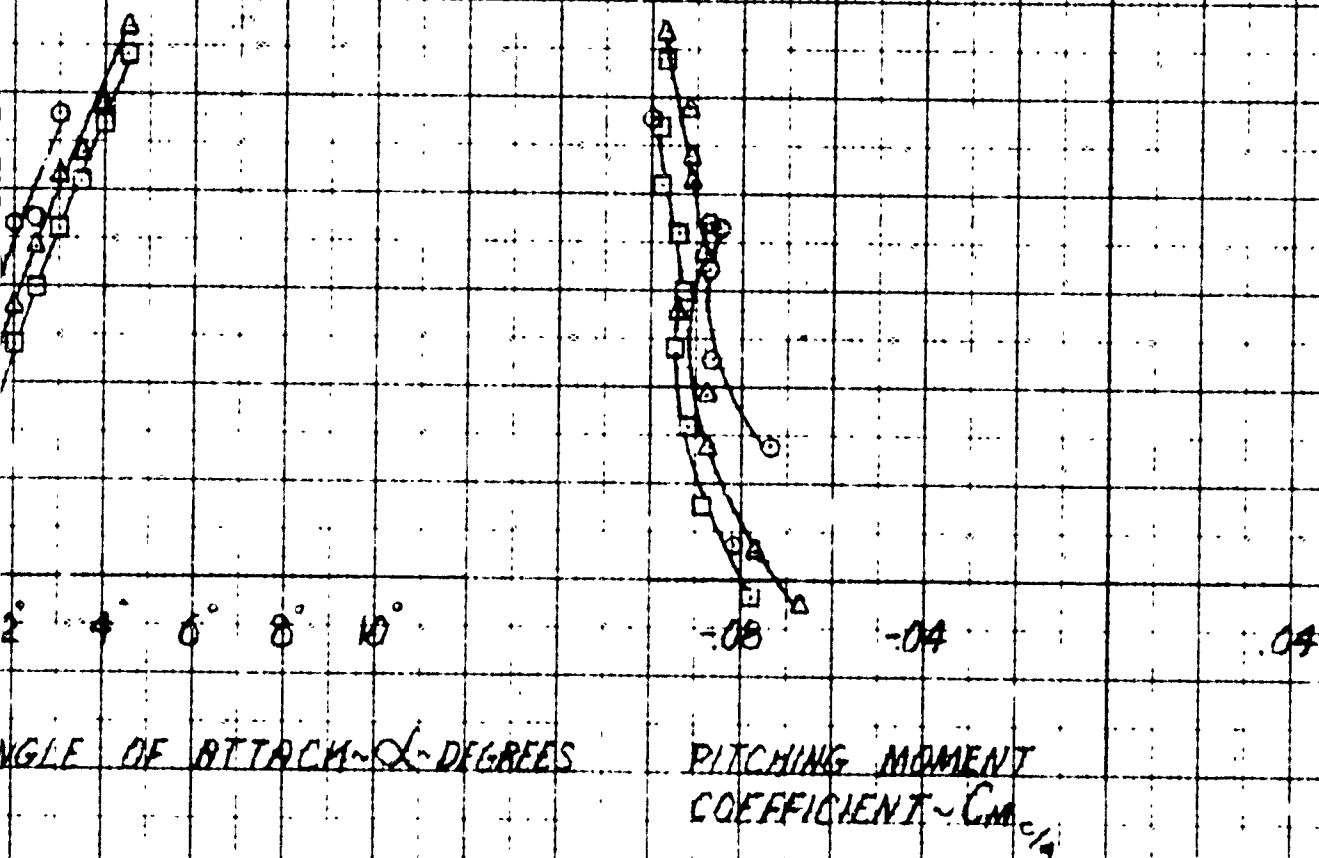
2°

WHIRLING TANK TEST NO. 49

BU SHIPS FOIL NO. 19

$S = 4.94 \text{ in}^2$; $MAC = .110 \text{ ft}$

$V = 90 \text{ MTS.}$



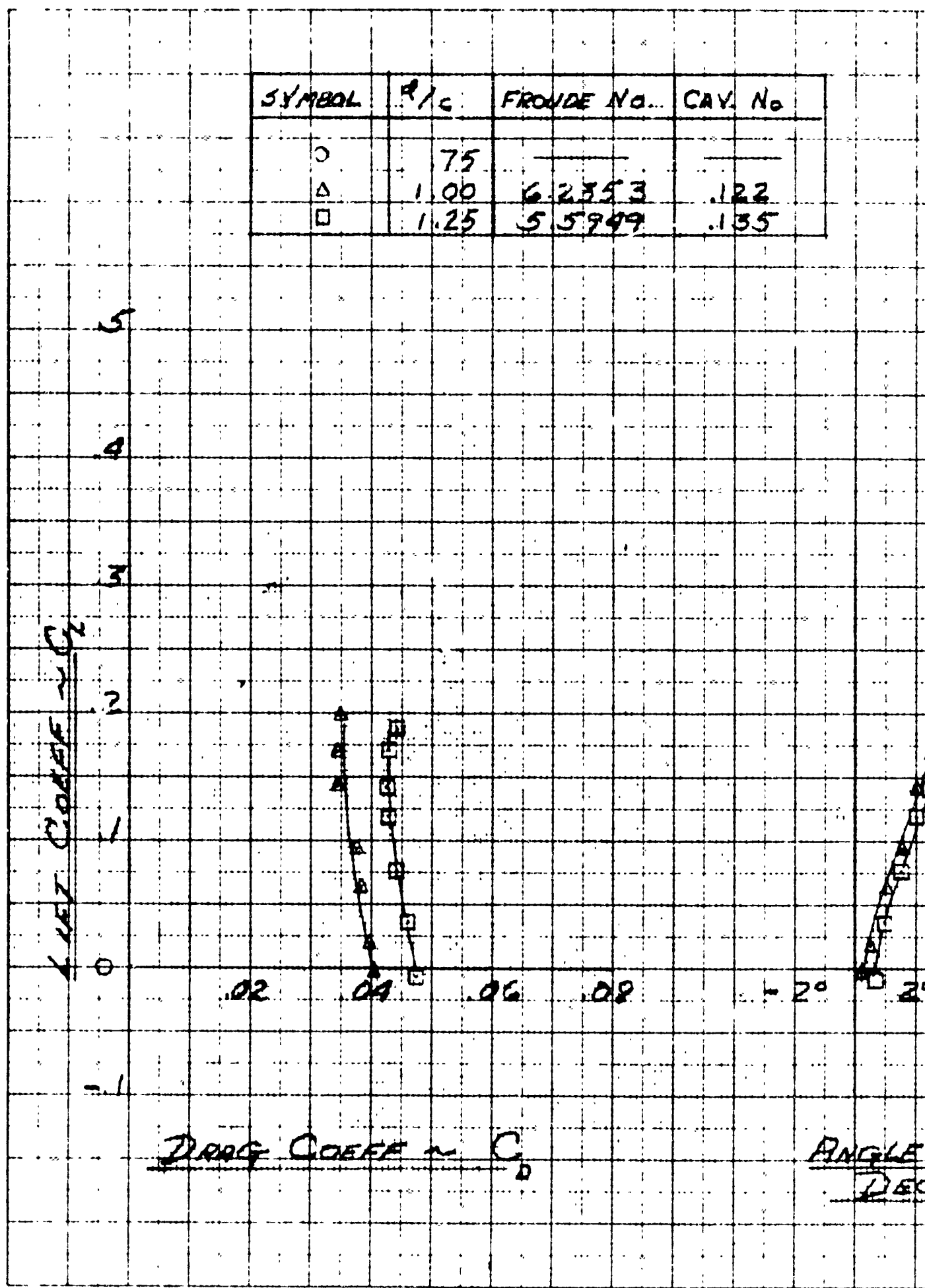
SYMBOL	R/C	FROUDE No.	CAY. No.
○	.75	—	—
△	1.00	6.2353	.122
□	1.25	5.5949	.135

W-E, bearing from 21.5
to 21.5

LIFT COEFF - C_L

DRAW COEFF - C_D

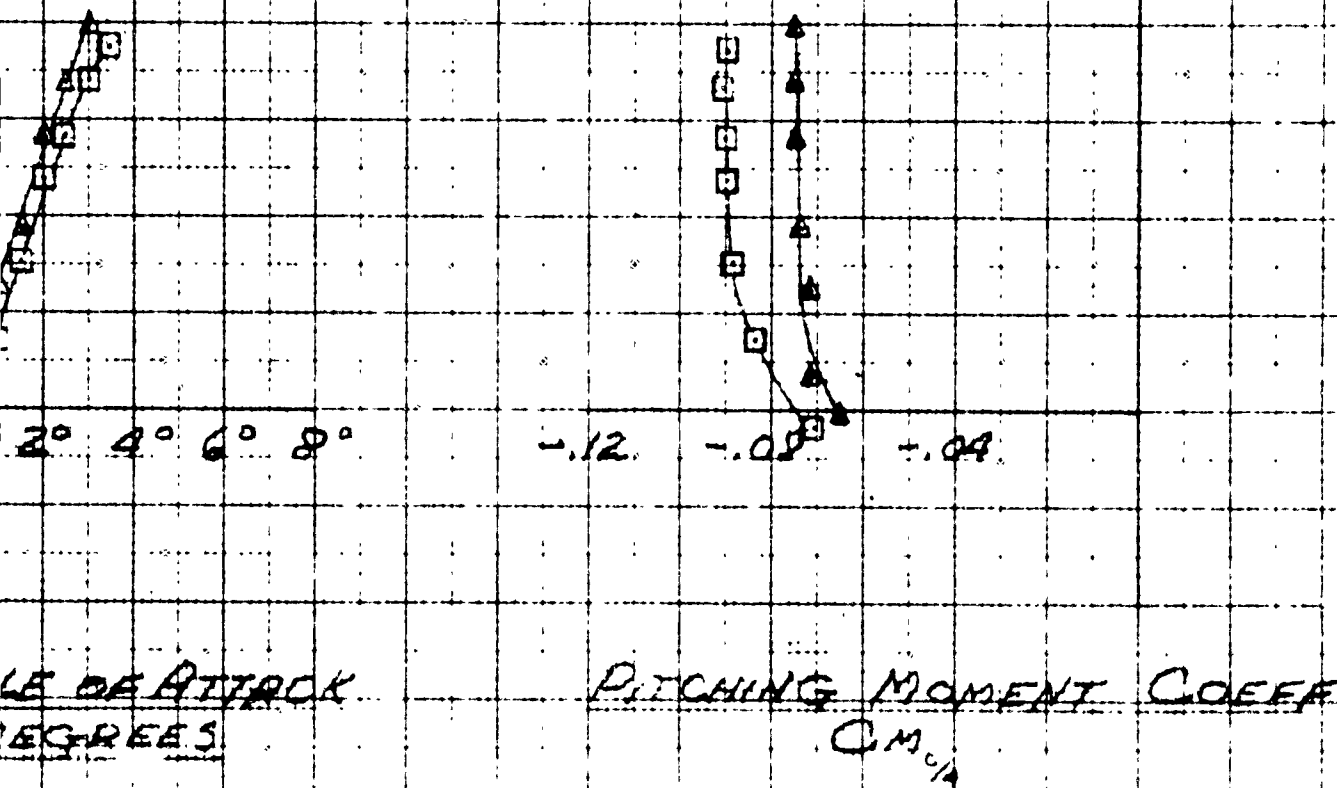
ANGLE
DEG



W.I.T. NO. 49 BUSHIPS FOIL NO. 19

M.A.C. = 11 FT AREA = 4.940

VEL = 96 KTS

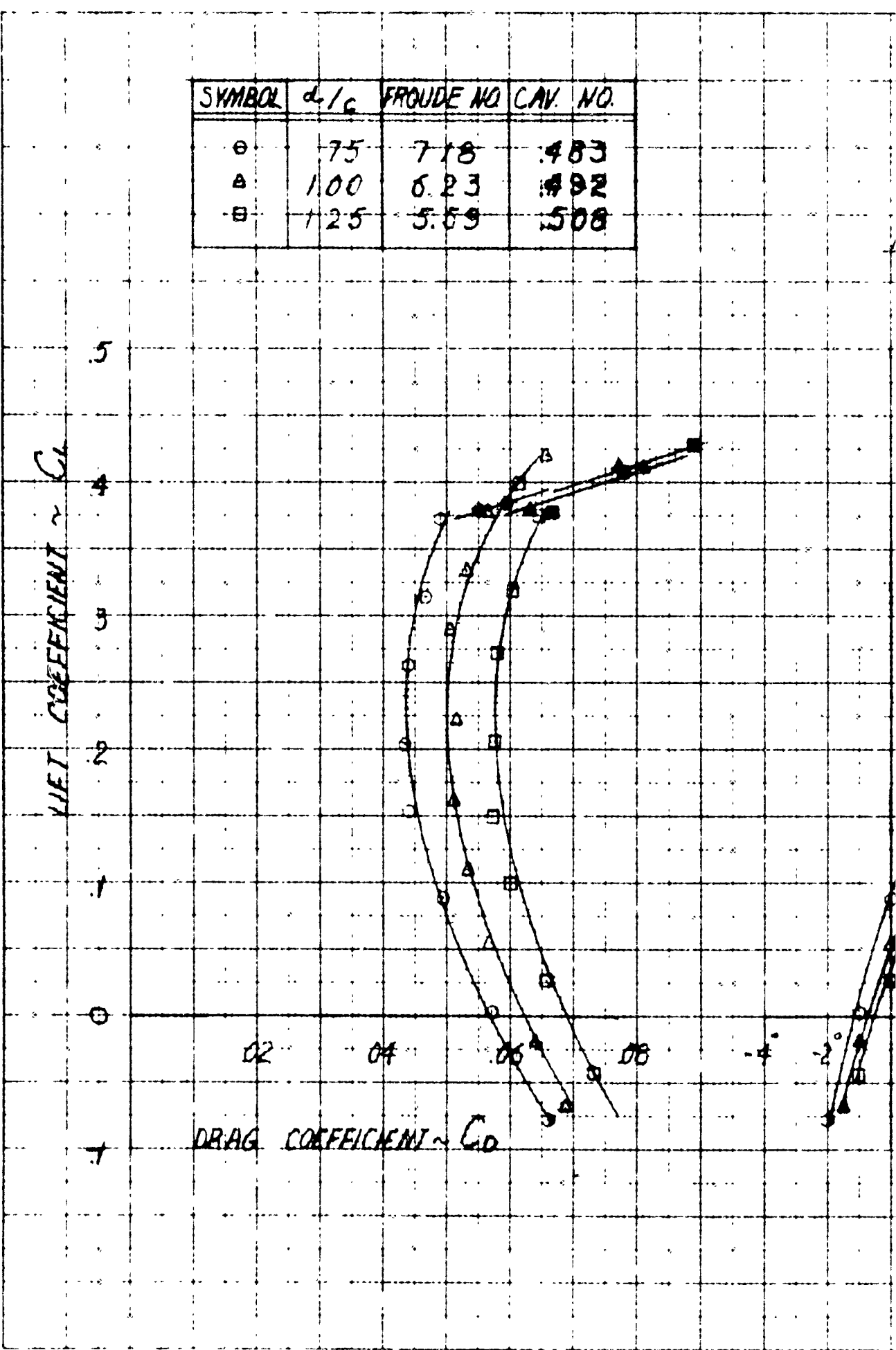


SYMBOL	d/c	FROUDE NO	CAV. NO.
○	.75	7.18	.483
△	1.00	6.23	.492
□	1.25	5.59	.508

LIFT COEFFICIENT $\sim C_L$

DRAG COEFFICIENT $\sim C_D$

NOE - BUREAU OF NAVY



WHIRLING TANK TEST NO. 50

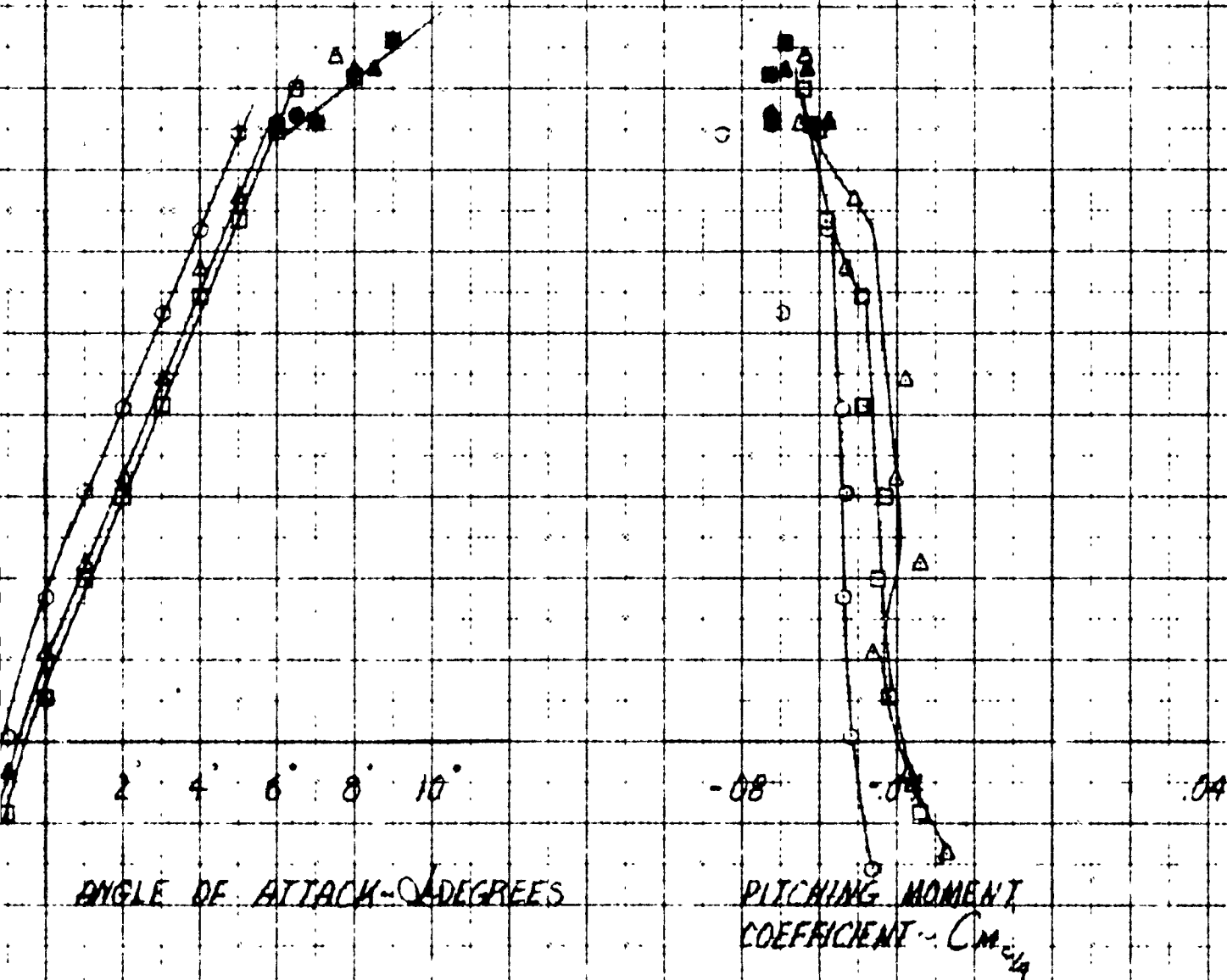
BU SHIPS FOIL NO. 20

M.A.C. = .110 FT. ; $S = 4.94 \text{ IN}^2$

$V = 40 \text{ MTS.}$

NOTE:

1 DARKENED POINTS INDICATE
VENTED FLOW



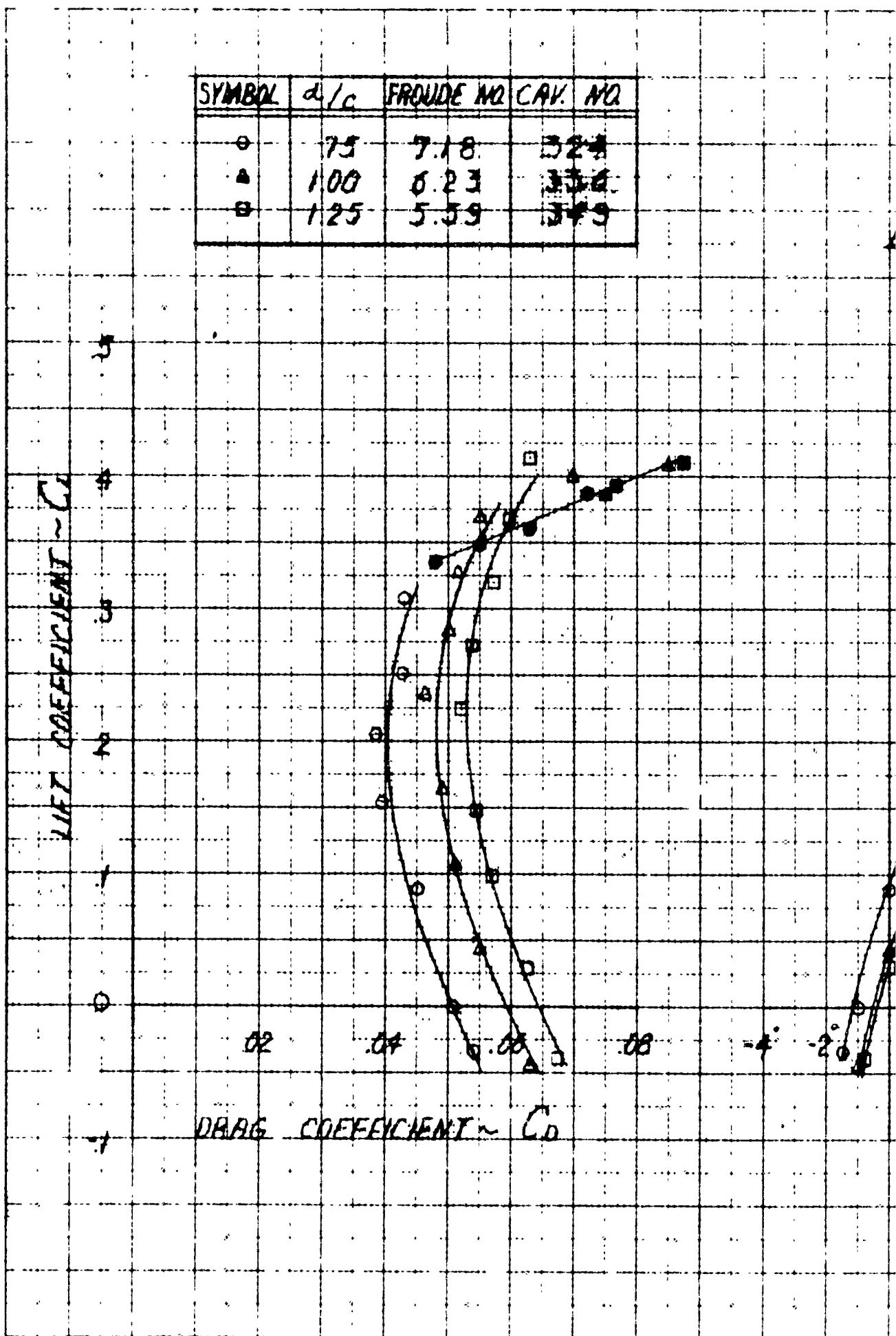
SYMBOL	d/c	FROUDE NO	CAV. NO
○	75	7.18	324
△	100	6.23	336
□	125	5.59	349

LIFT COEFFICIENT $\sim C_L$

DRAFF COEFFICIENT $\sim C_D$

K-E MEASUREMENT 1967

8 11 14 1967



WHIRLING TANK TEST NO. 50

BU SHIPS FOIL NO. 20

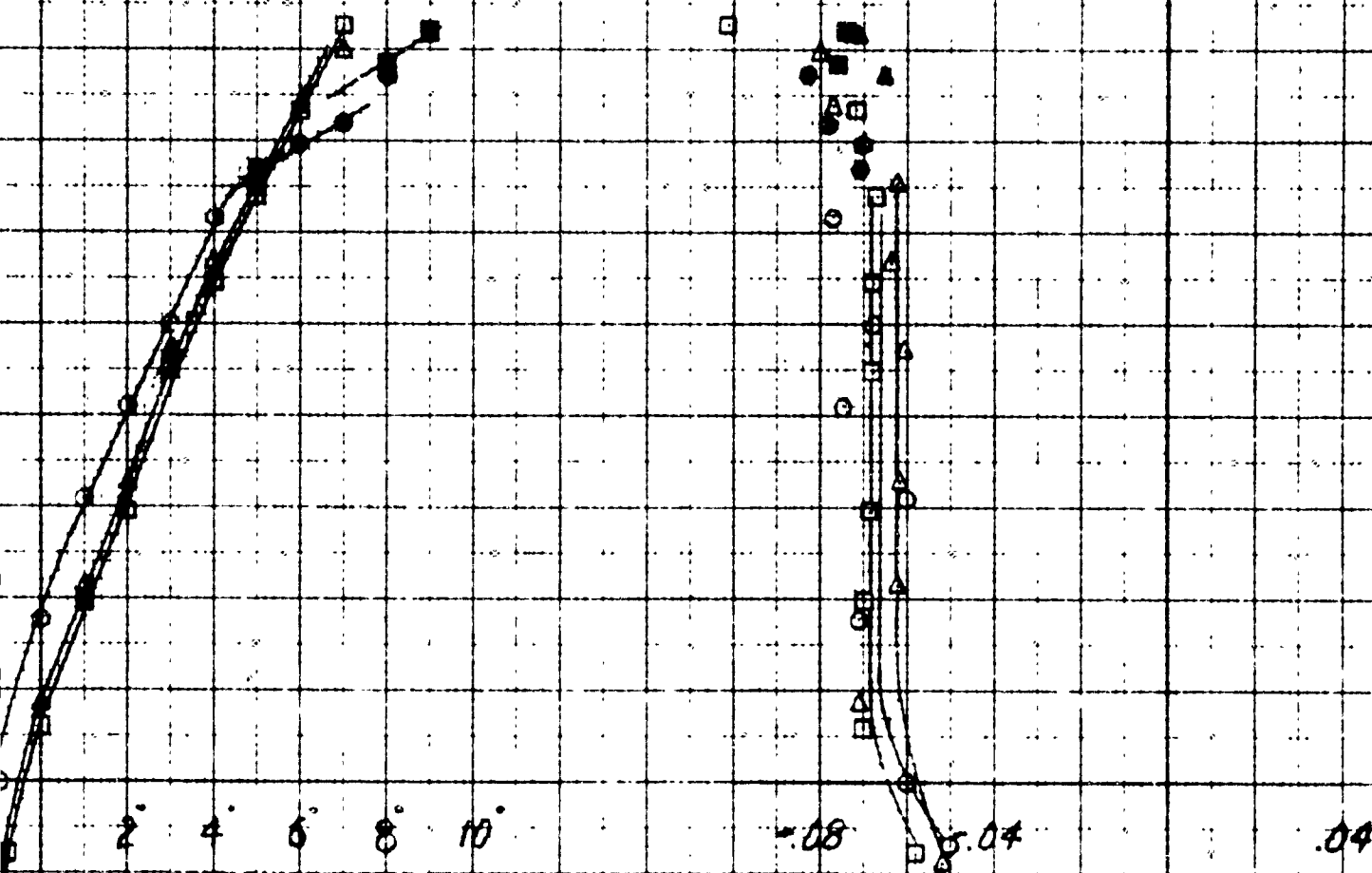
MAC = .110 FT ; S = 4.94 IN²

V = 50 KTS.

NOTE:

1. DARNED POINTS

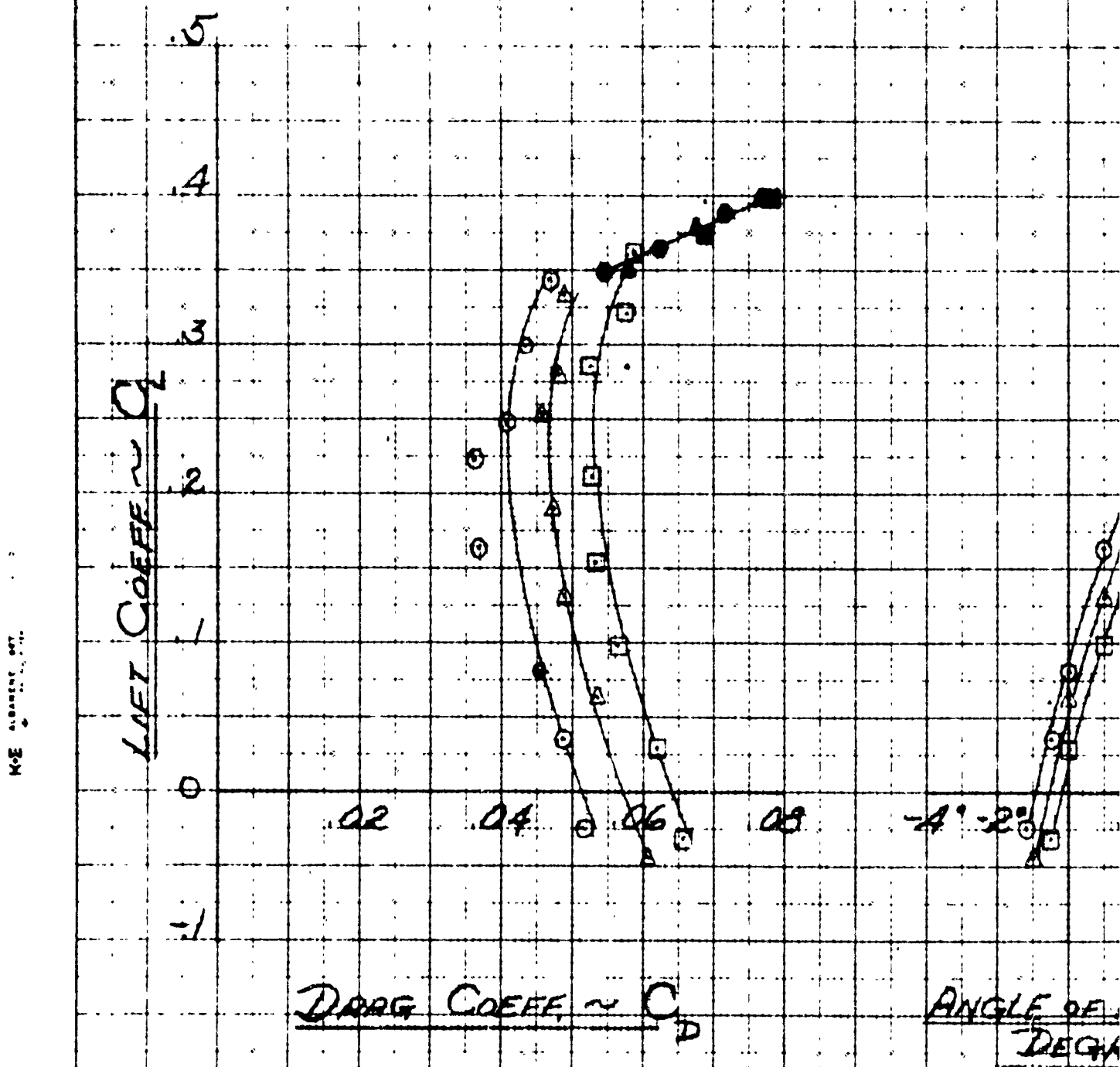
INDICATE REVERSED FLOW.



ANGLE OF ATTACK ~ X - DEGREES

PITCHING MOMENT
COEFFICIENT ~ CM

SYMBOL	d/c	FROUDE NO.	CAV. NO.
O	.75	7.1767	.237
Δ	1.00	6.2353	.249
□	1.25	5.5949	.261



W. I. T. No. 50, BUSHIPS FOIL No. 20

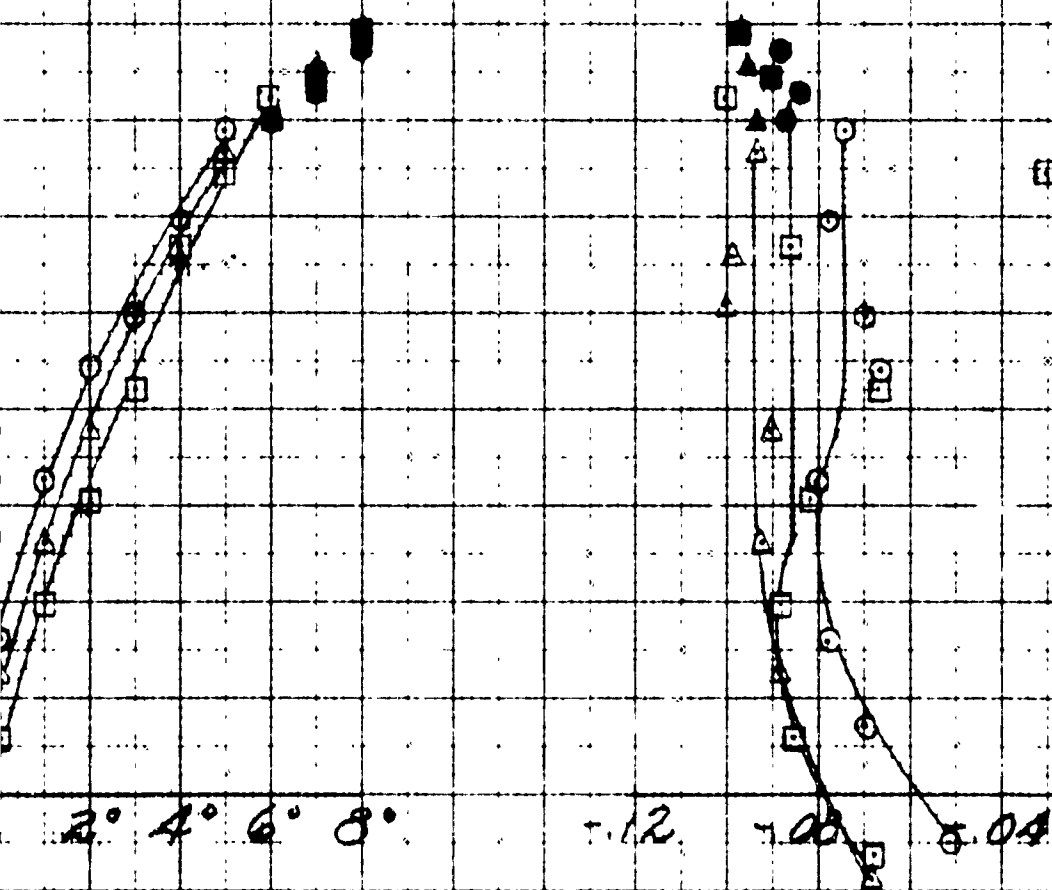
M.A.C. = 110 FT. AREA = 4.94 sq ft

VEL. = 60 KTS

NOTE:

1. DARKENED SYMBOLS

INDICATE VENTED FLOW



Angle of Attack
Degrees

Pitching Moment Coeff

Cm/14

SYMBOL	d/c	FROUDE NO.	CAV. NO.
○	.75	7.1767	.184
△	1.00	6.2353	.196
□	1.25	5.5949	.809

No.

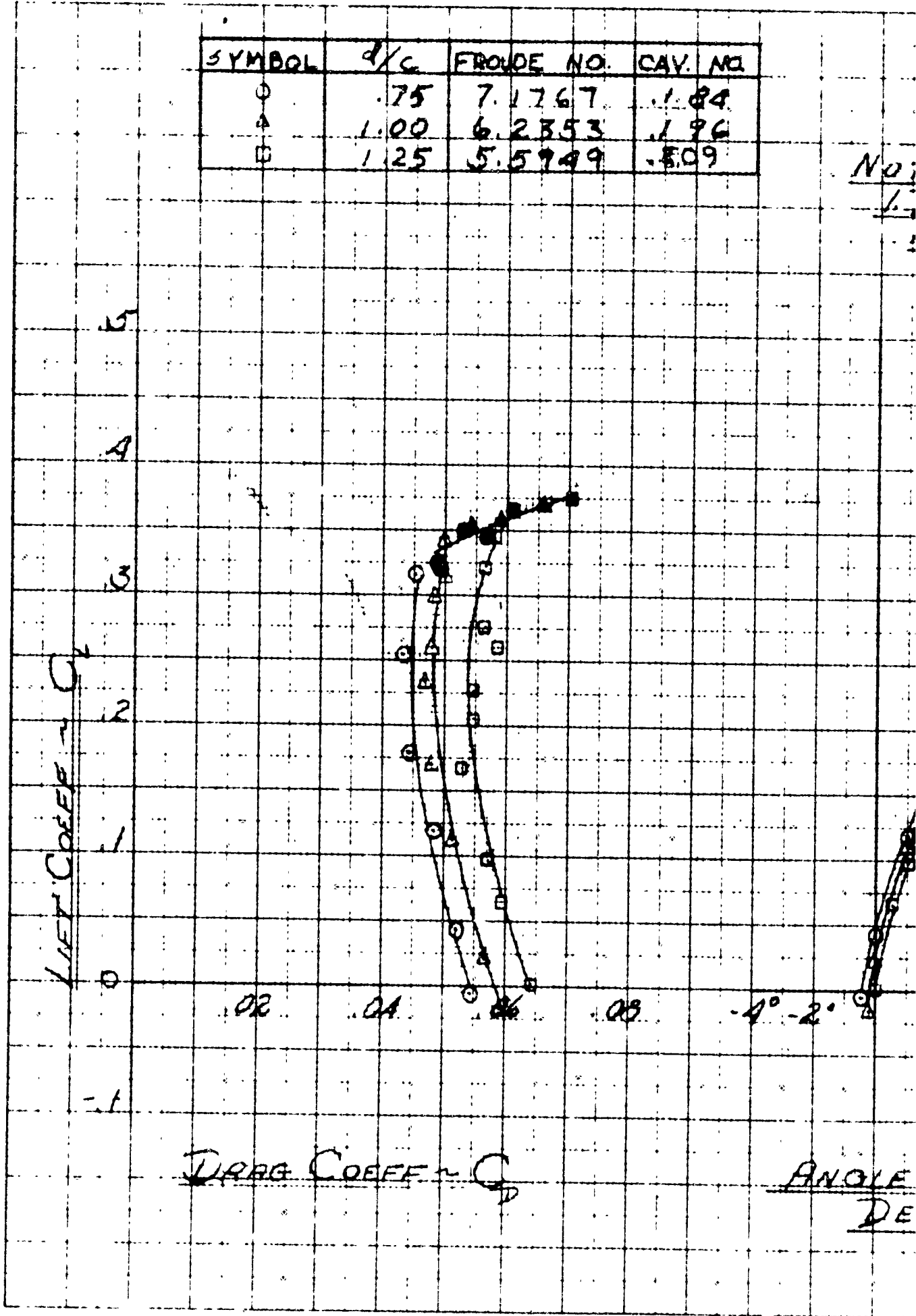
1.

MoE cleaned 1007 52.00

LIFT COEFF ~ C_L

DRAW COEFF ~ C_D

ANGLE
DE



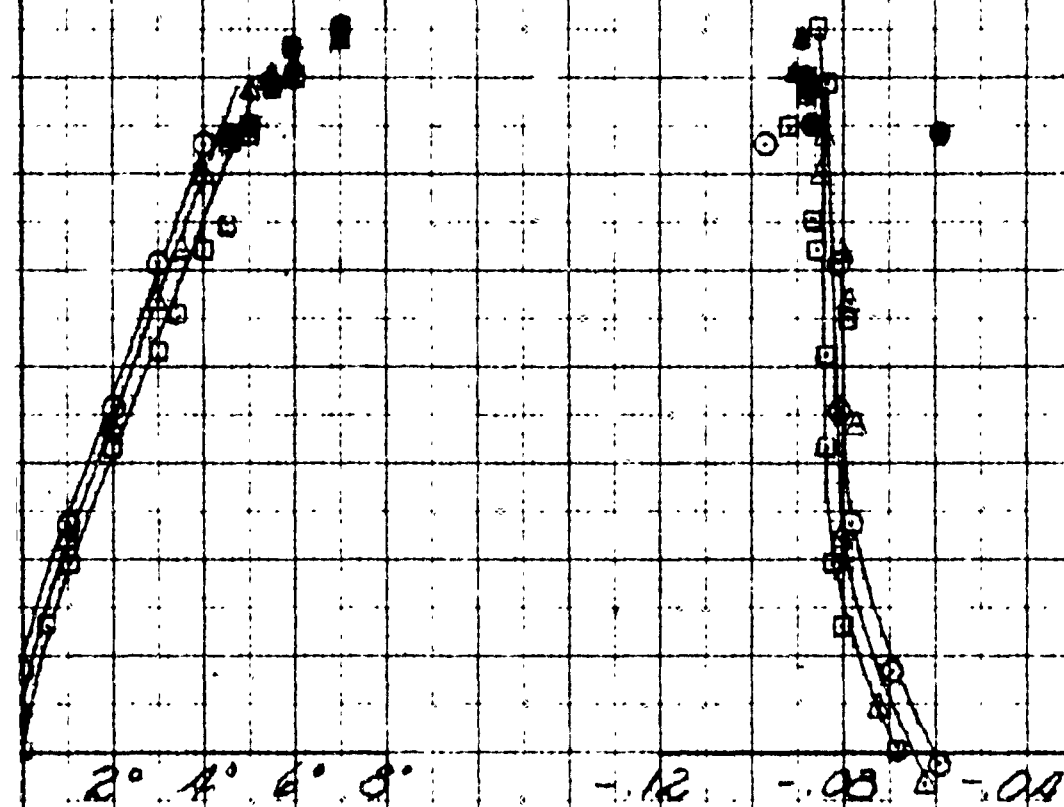
W. T. T. No. 50 BU SHIPS FOIL No. 20

M. A. C. = .110 FT AREA = 4.94 sq in

VEL = 20 KTS

NOTE:

1. DARKENED SYMBOLS
INDICATE VENTED FLOW.

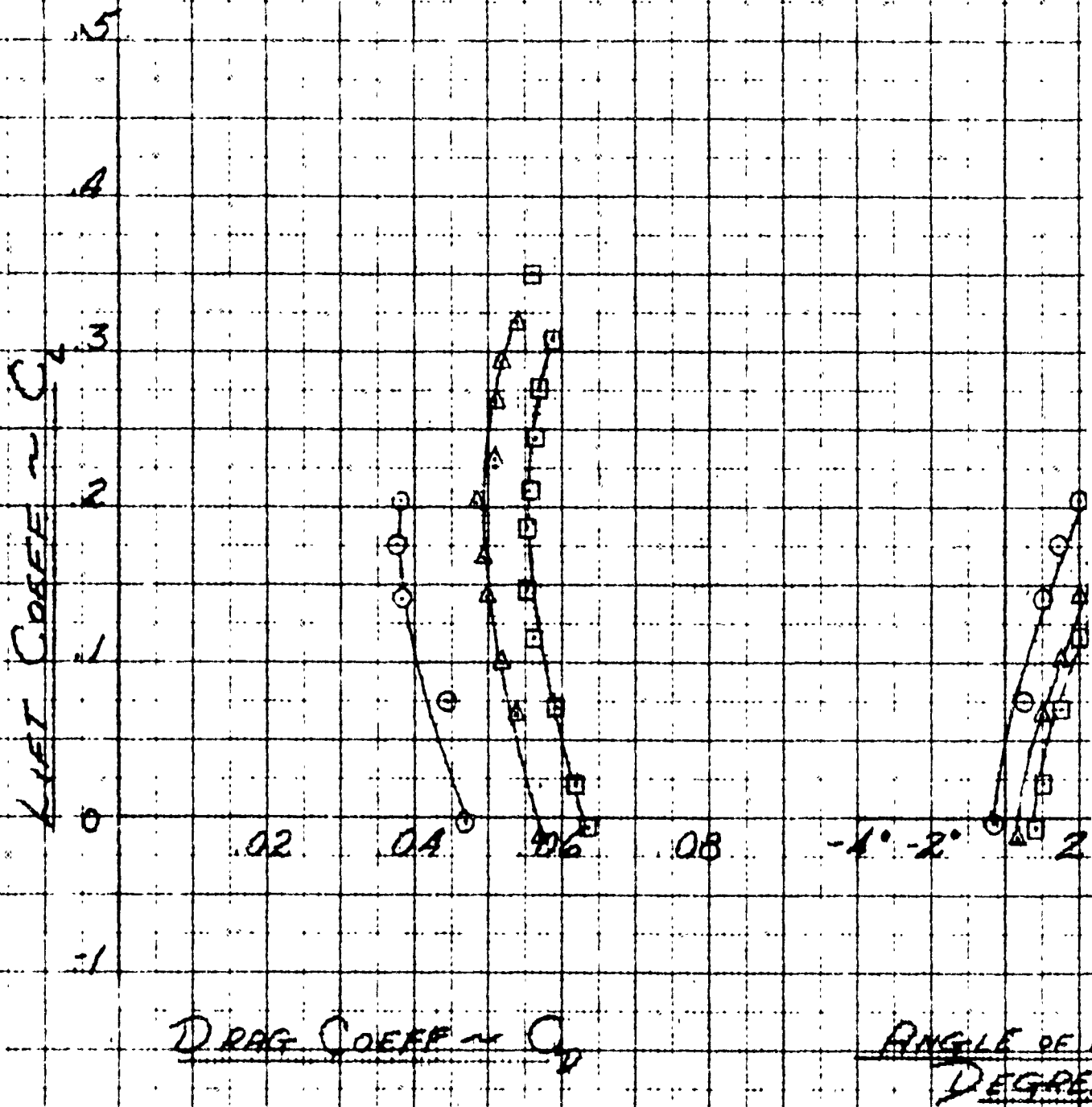


ANGLE OF ATTACK
DEGREES

PITCHING MOMENT COEFF

C_m

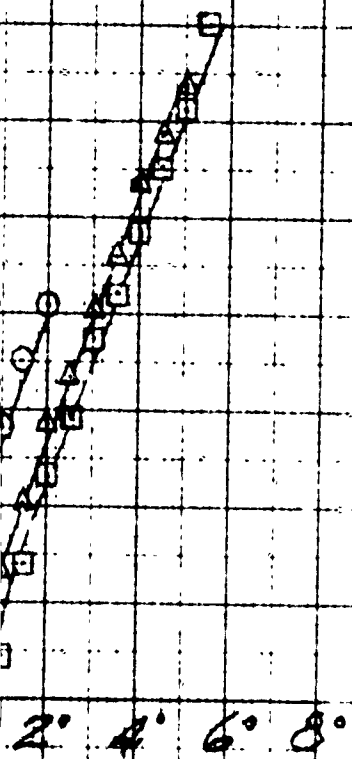
SYMBOL	d/c	FROUDE NO.	CAV. NO.
○	.75	7.1767	.150
△	1.00	6.2353	.163
□	1.25	5.5949	.175



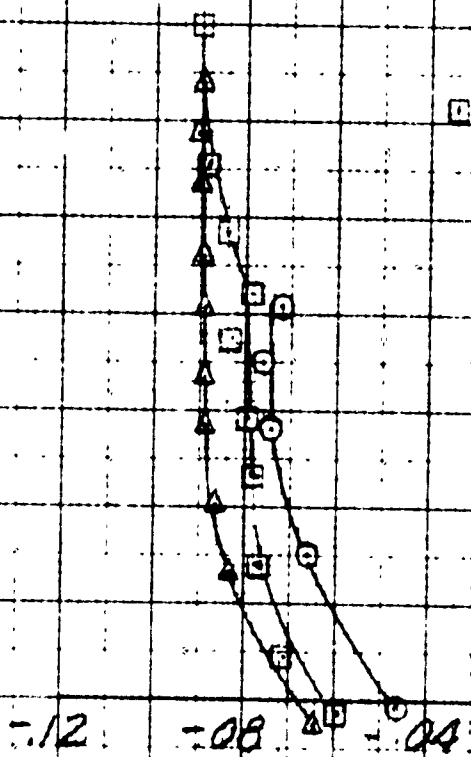
W. T. T. No. 50 BUSKIPS FOIL No. 20

M.A.C. = .110 FT AREA = 4.94 sq ft

VEL = 80 KTS

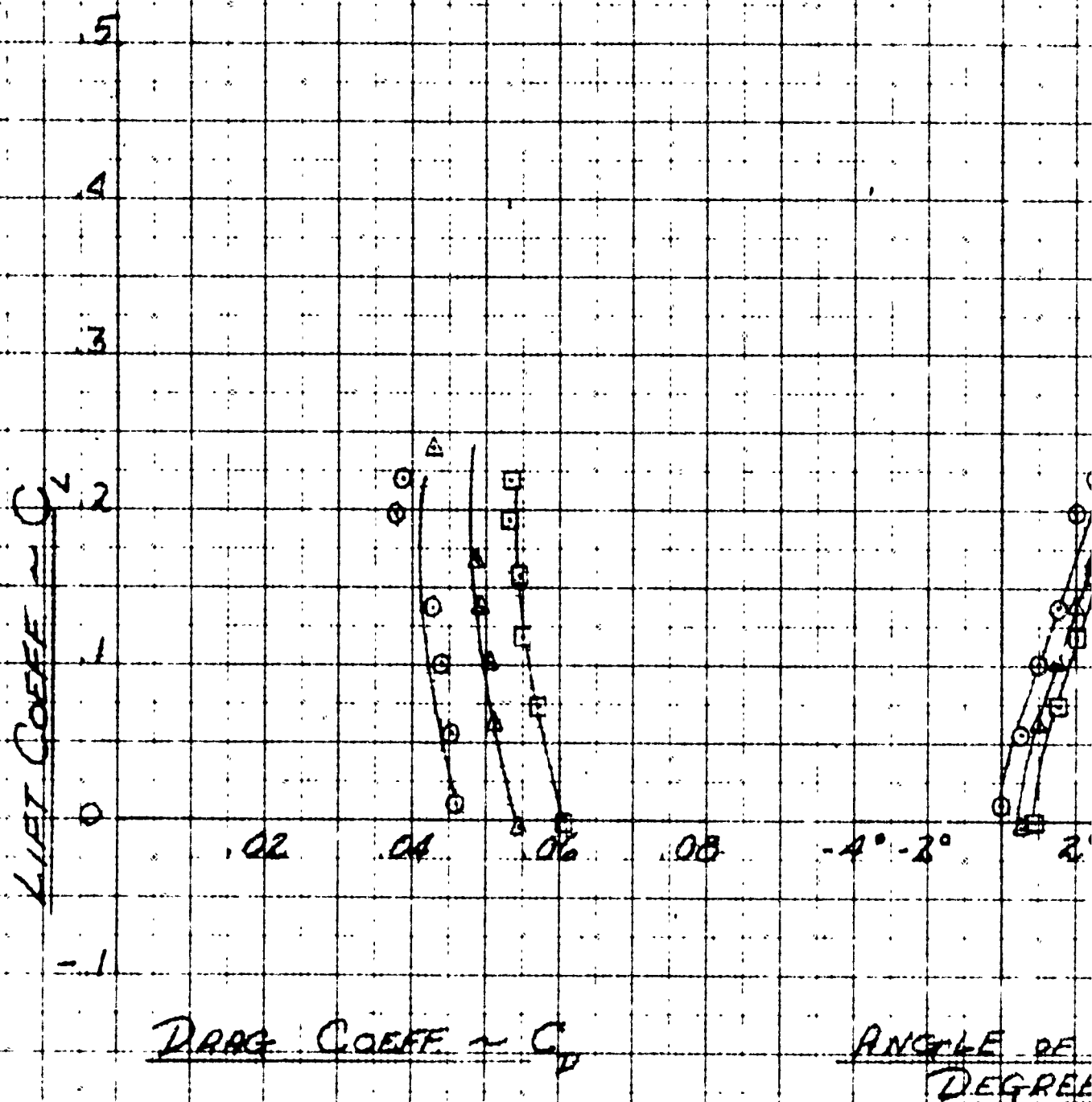


ANGLE OF ATTACK
DEGREES



PITCHING MOMENT COEFF
 $C_{m, \frac{1}{4}}$

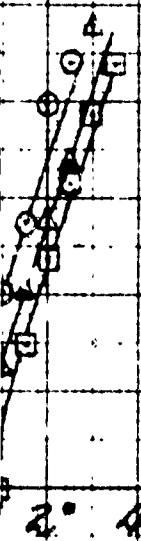
SYMBOL	R/C	FROUDE NO.	CAV. NO.
○	.75	7.1767	.127
△	1.00	6.2353	.139
□	1.25	5.5949	.152



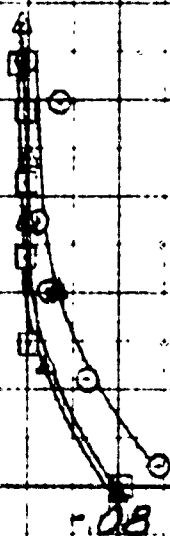
W.T.T. No. 50 BUSHIPS Foil No. 20

M.A.C. = 110 FT. AREA = 4.94 sq ft

VEL. = 90 KTS



ANGLE OF ATTACK
DEGREES



PITCHING MOMENT COEFF
 C_{m/c_a}

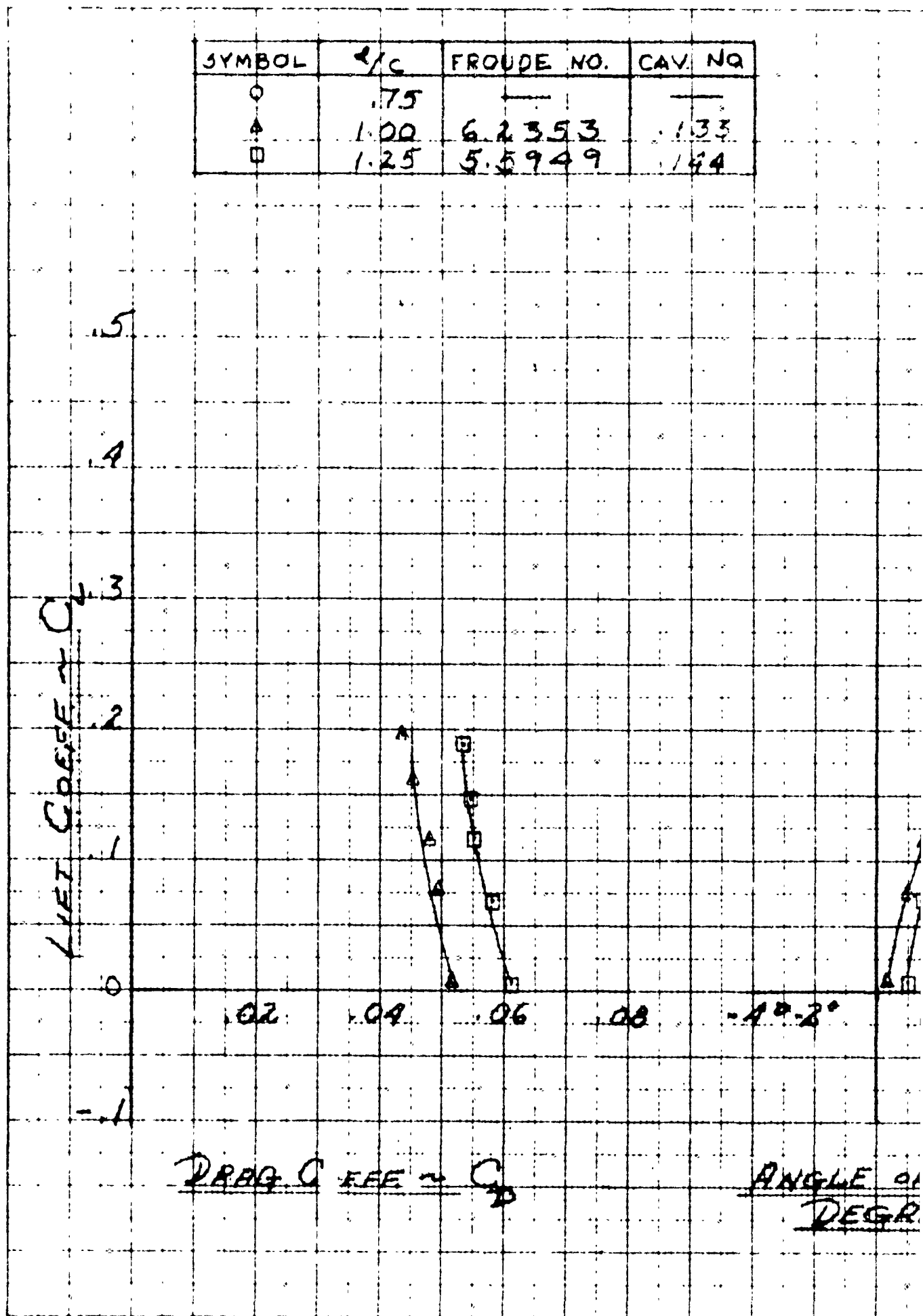
SYMBOL	$\frac{L}{C}$	FROUDE NO.	CAV. NO.
○	.75	—	—
△	1.00	6.2353	.133
□	1.25	5.5949	.144

NO. 1, 2, 3, 4, 5, 6, 7, 8, 9, 10, 11, 12, 13, 14, 15, 16, 17, 18, 19, 20, 21, 22, 23, 24, 25, 26, 27, 28, 29, 30, 31, 32, 33, 34, 35, 36, 37, 38, 39, 40, 41, 42, 43, 44, 45, 46, 47, 48, 49, 50, 51, 52, 53, 54, 55, 56, 57, 58, 59, 60, 61, 62, 63, 64, 65, 66, 67, 68, 69, 70, 71, 72, 73, 74, 75, 76, 77, 78, 79, 80, 81, 82, 83, 84, 85, 86, 87, 88, 89, 90, 91, 92, 93, 94, 95, 96, 97, 98, 99, 100

LIFT COEFF - C_L

DRAW COEFF - C_D

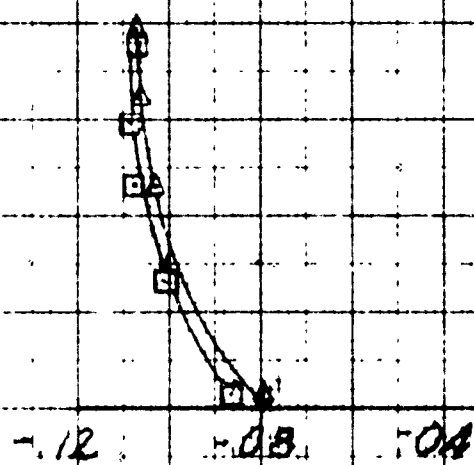
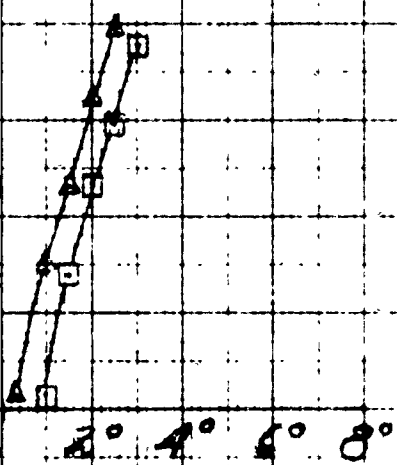
ANGLE OF
DEGR



W.I.T. No. 50 BU SHIPS FOIL No. 20

M.A.C. = 110 FT AREA = 4.940 "

VEL = 95 KTS

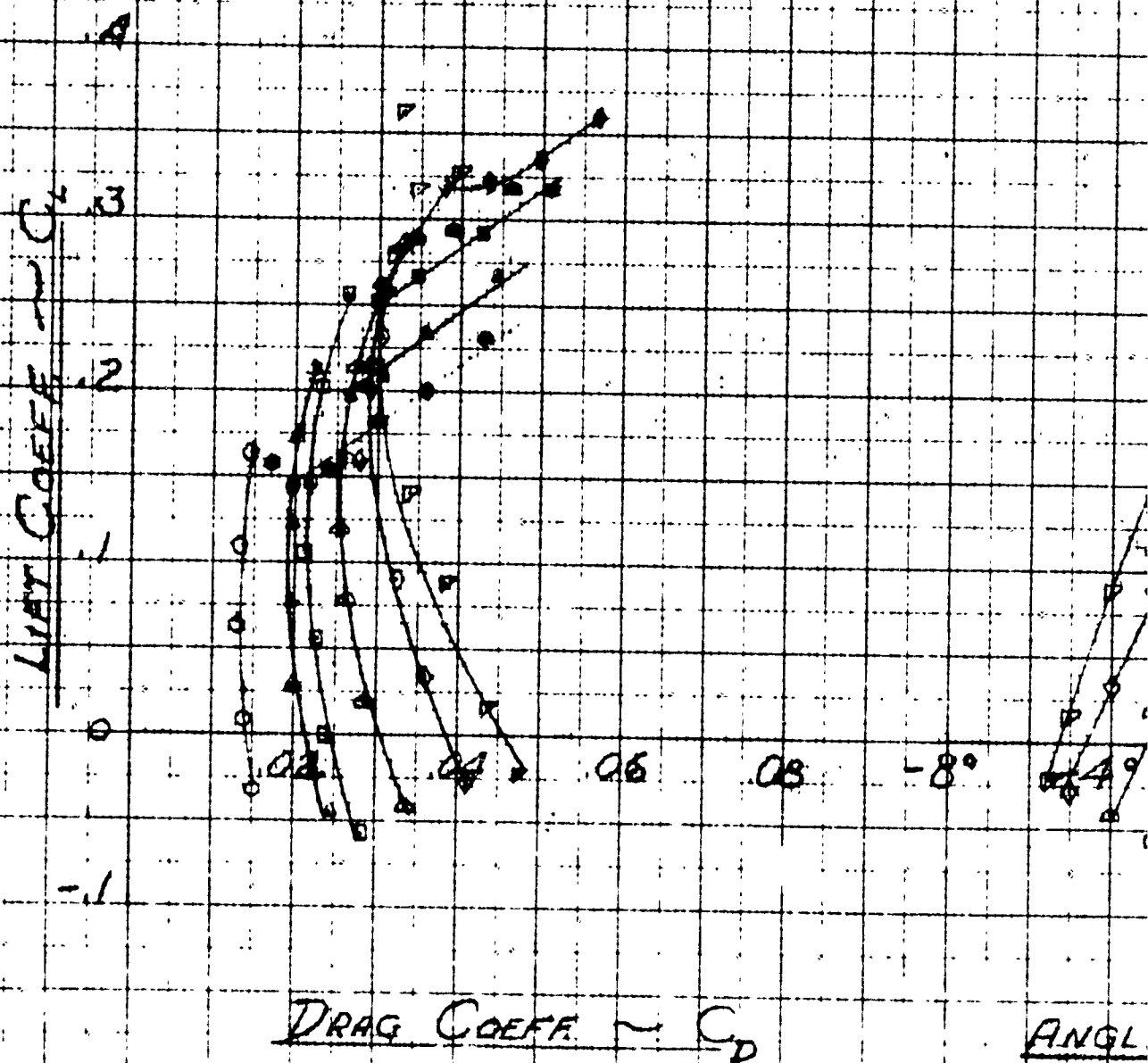


ANGLE OF ATTACK
DEGREES

PITCHING MOMENT COEFF.
CM CM

SYMBOL	FLAP SET	PROVIDE NO.	CAN NO.
○	0°		
△	4°	6	4
□	8°	1	7
◇	12°	2	8
◇	16°	3	5
▽	20°	4	

NO



ANGL

WHIRLING TANK TEST NO. 42

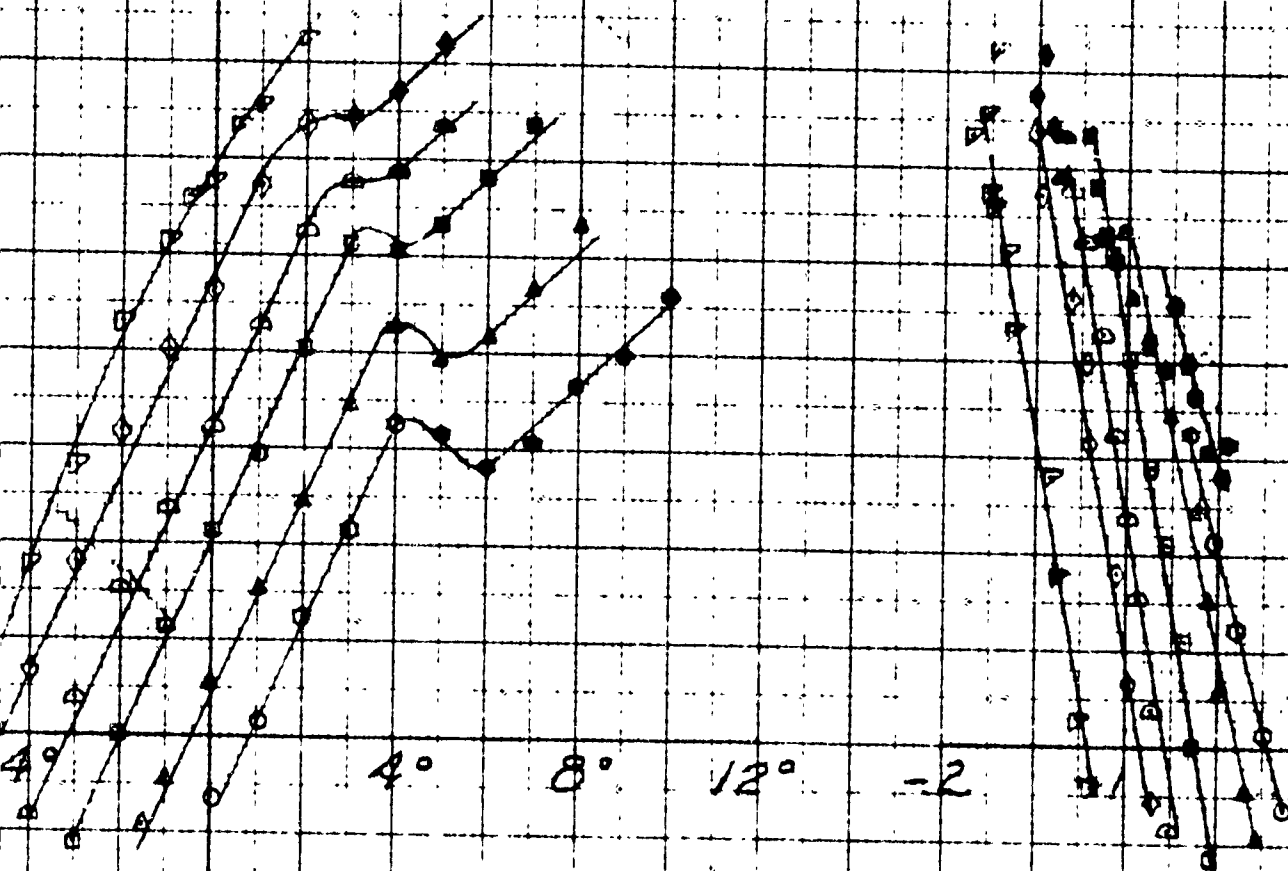
BUSHIPS FOIL NO. 13

MAR. = 152 FT AREA = 80" $d/c = .75$

VEL = 40 KTS

NOTE:

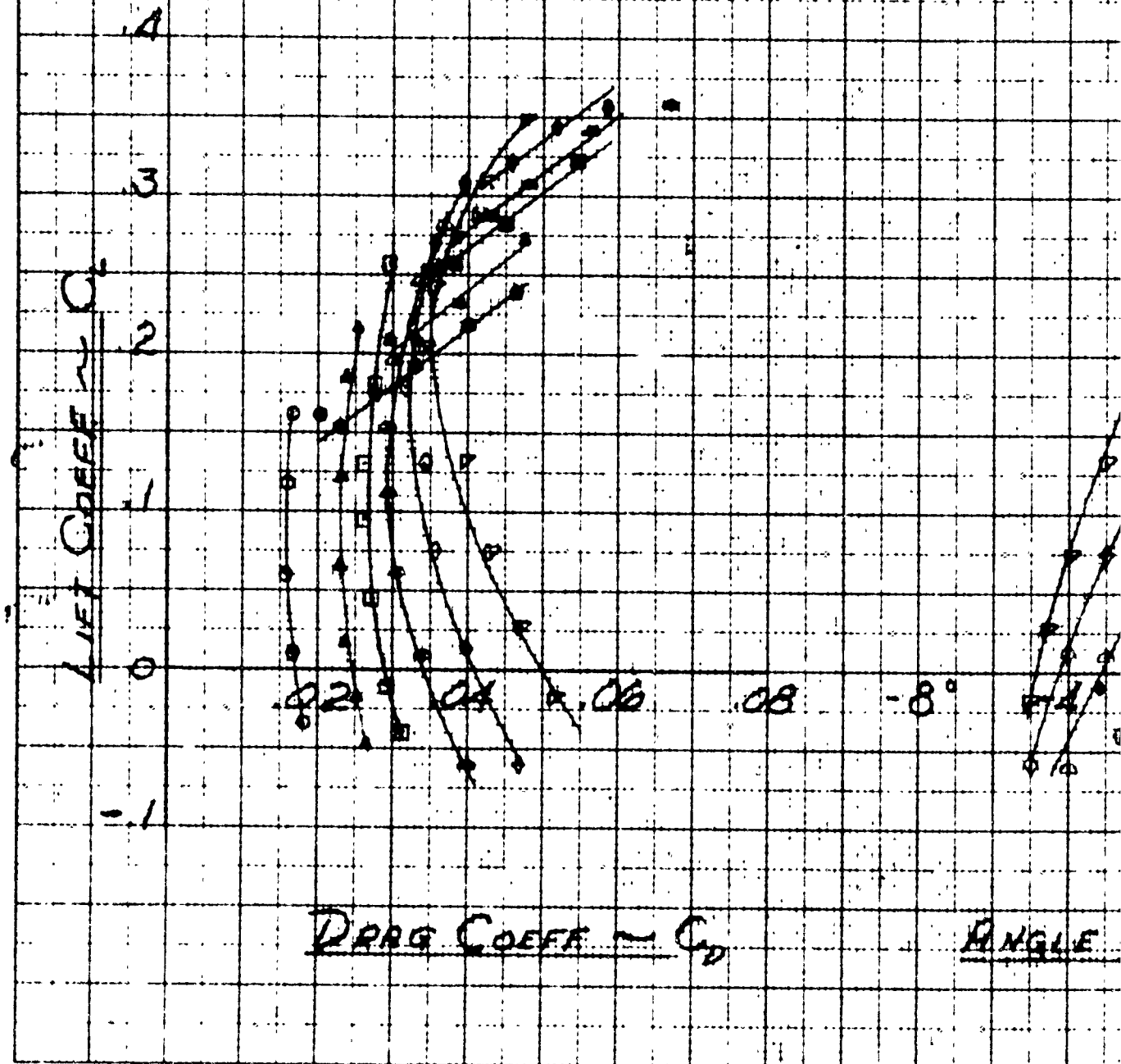
1. DARKENED SYMBOLS
INDICATE VENTED FLOW



ANGLE OF ATTACK - DEGREES

PITCHING MOMENT
COEFF - CM 94

SYMBOL	FLAP SET	FROUDE NO.	CAV. NO.
○	0°		
△	4°	5.3	5.1
□	8°	2	1.5
◇	12°	1.9	7
◇	16°		
◇	20°		



WHIRLING TANK TEST No. 42

BUSHIPS FOIL No. 13

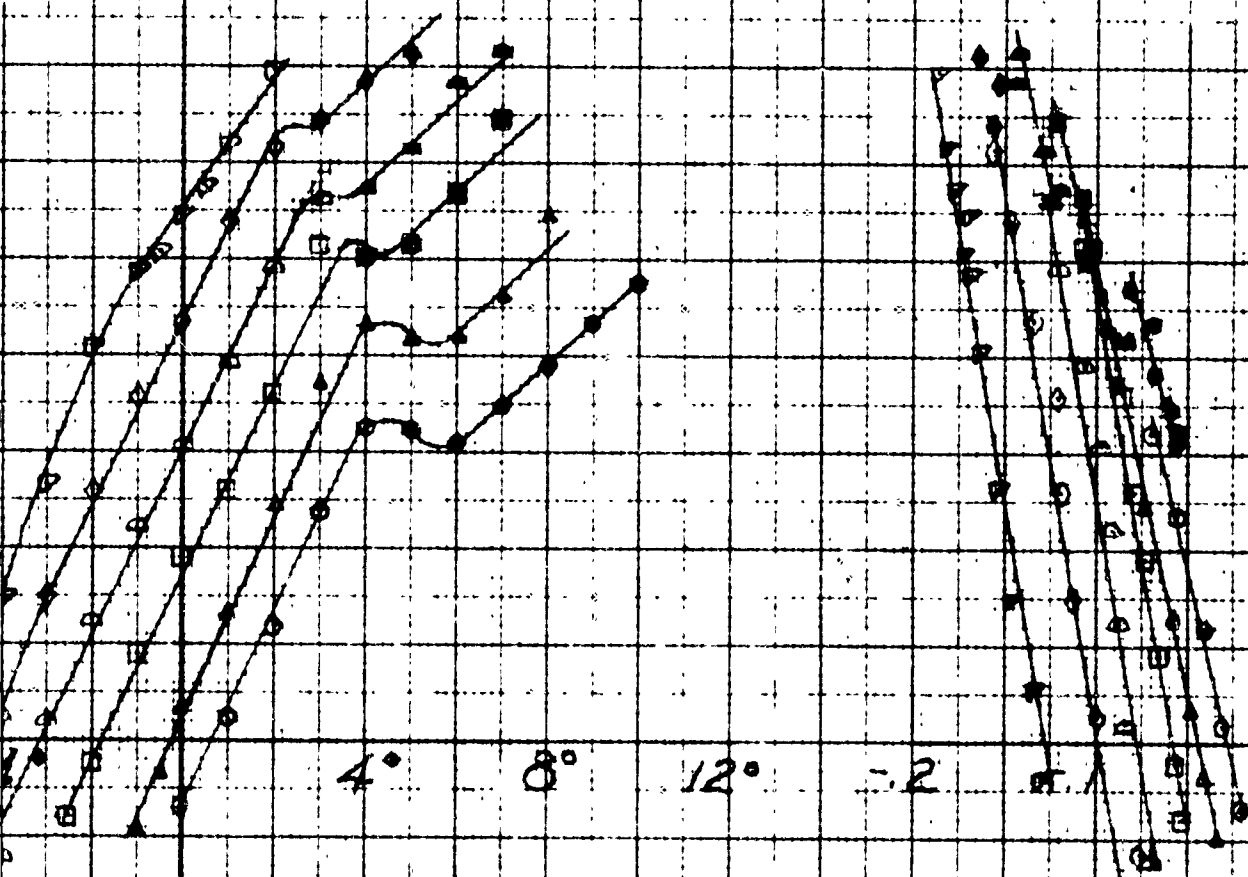
M.P.C. = 152 FT AREA = 80" $d/c = 1.00$

VEL = 40 KTS

NOTE:

1. DARKENED SYMBOLS

INDICATE VENTED FLOW.

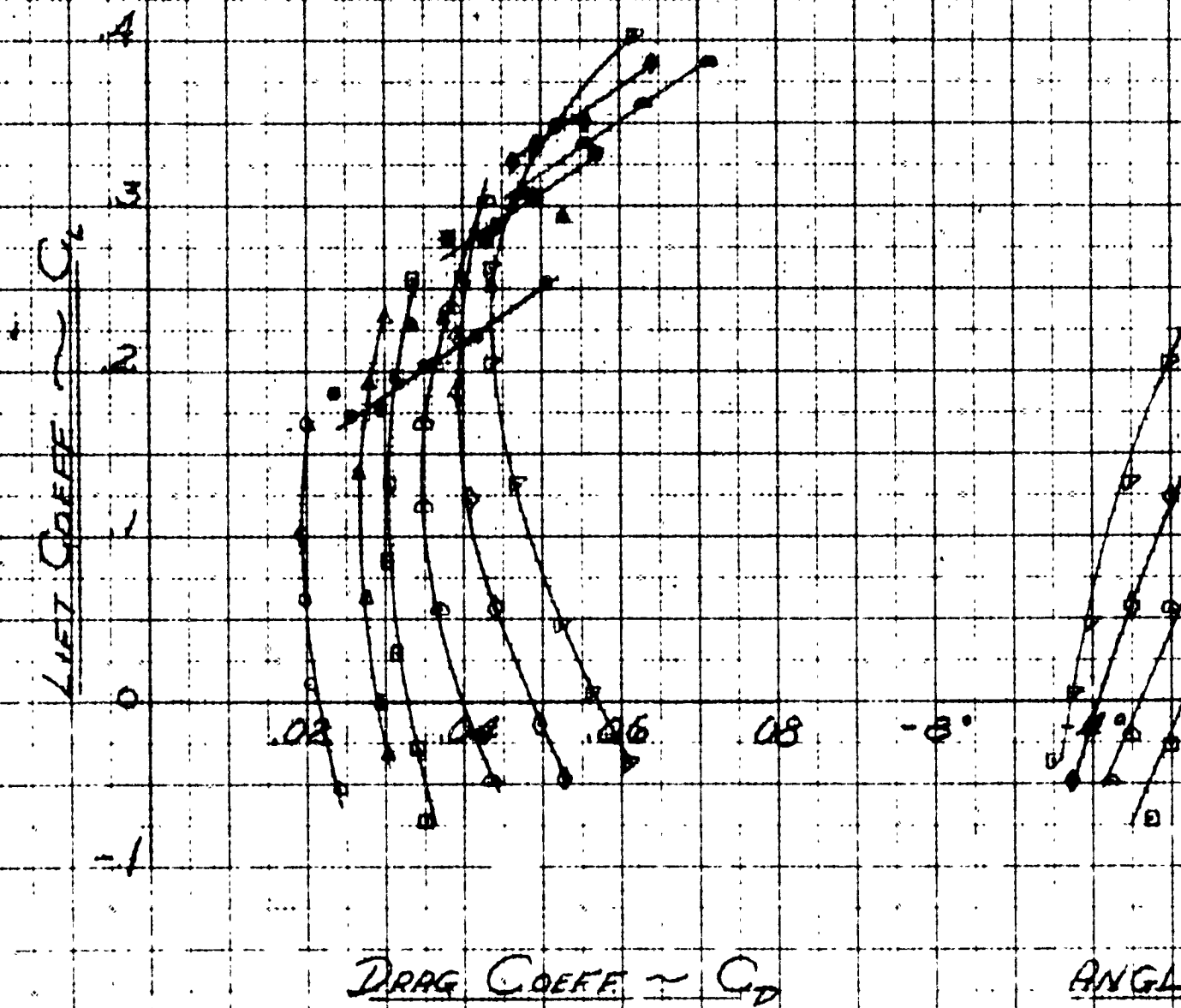


ANGLE OF ATTACK - DEGREES

PITCHING MOMENT
COEFF. ~ $C_{M/C}$

SYMBOL	FLAP SET	FRONDE NO.	DAY NO.
○	0°		
×	4°	1.7	12
□	8°	7	8
+	12°	8.5	6
●	16°		
△	20°		

NOTE:
1. DRAG
INDI



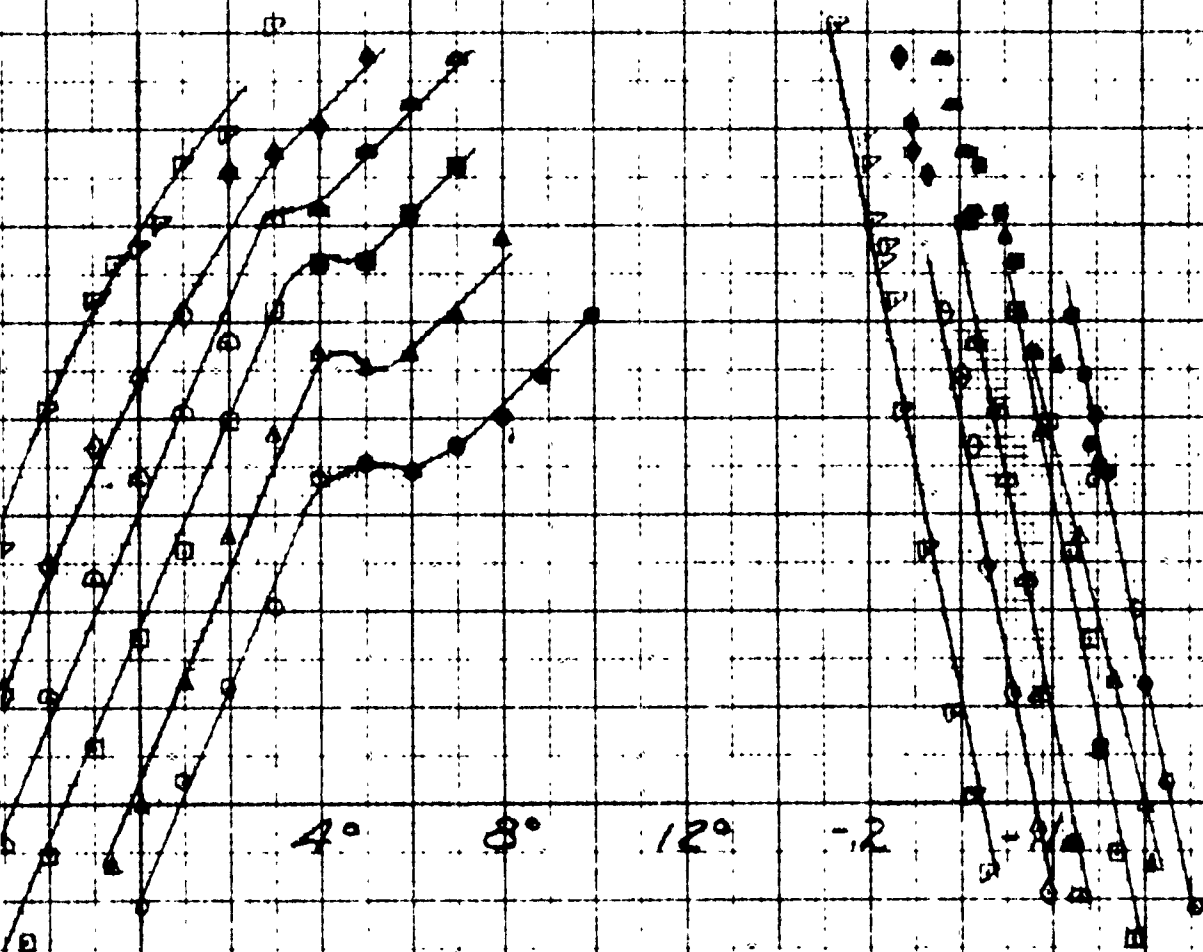
WHIRLING TANK TEST NO. 42.

BUSHIPS FOIL NO. 13

M.A.C. = 152 FT. AREA = 89' $\frac{1}{2}$ = 125

VEL. = 40 KTS

DARKENED SYMBOLS
INDICATE VENTED FLOW.



ANGLE OF ATTACK - DEGREES

PITCHING MOMENT
COEFF. - $C_{M/4}$

NO. 13

WHIRLING TA
BU SHIPS F
MBC = .152 FT
VELOC

SYMBOL	δ	FROUDE NO.	CAV NO.
\circ	0°	6.123	10%
Δ	4°		
\square	8°		
\diamond	12°		
∇	16°		
\times	20°	6.123	10%

LIFT COEFFICIENT $\rightarrow C_L$

DRAG COEFFICIENT $\rightarrow C_D$

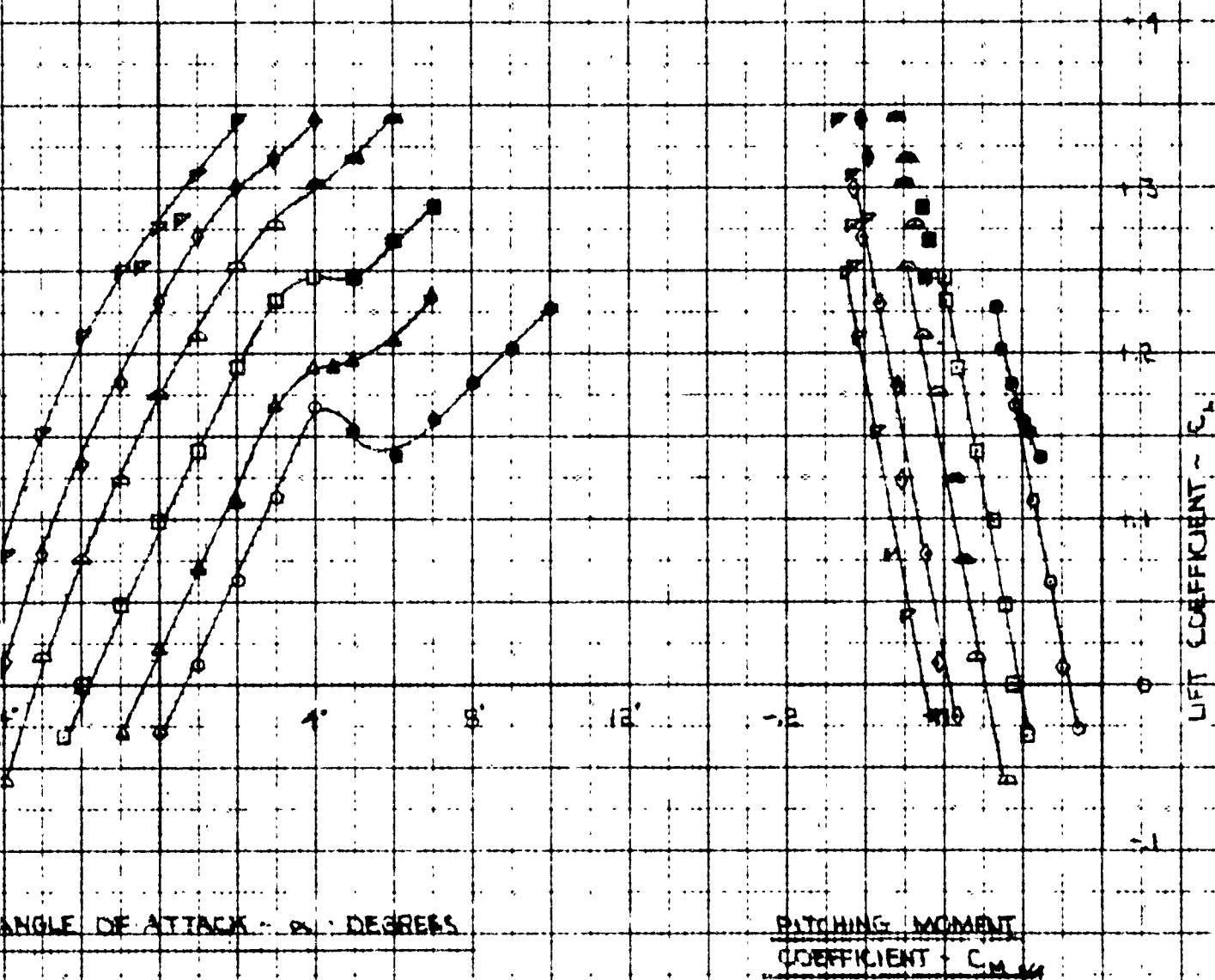
ANGL

M-2 ELEMENT TEST

TANK TEST NO 42
S FOIL NO 13
FT 5-80050711
VELOCITY 45 KTS

PAGE IV. 122

NOTE DARKENED POINTS INDICATE VENTED FLOW



WHIRLING TANK TEST
 BU SHIPS FOIL
 T.D.C. = .152 FT, S.E.
 VELOCITY =

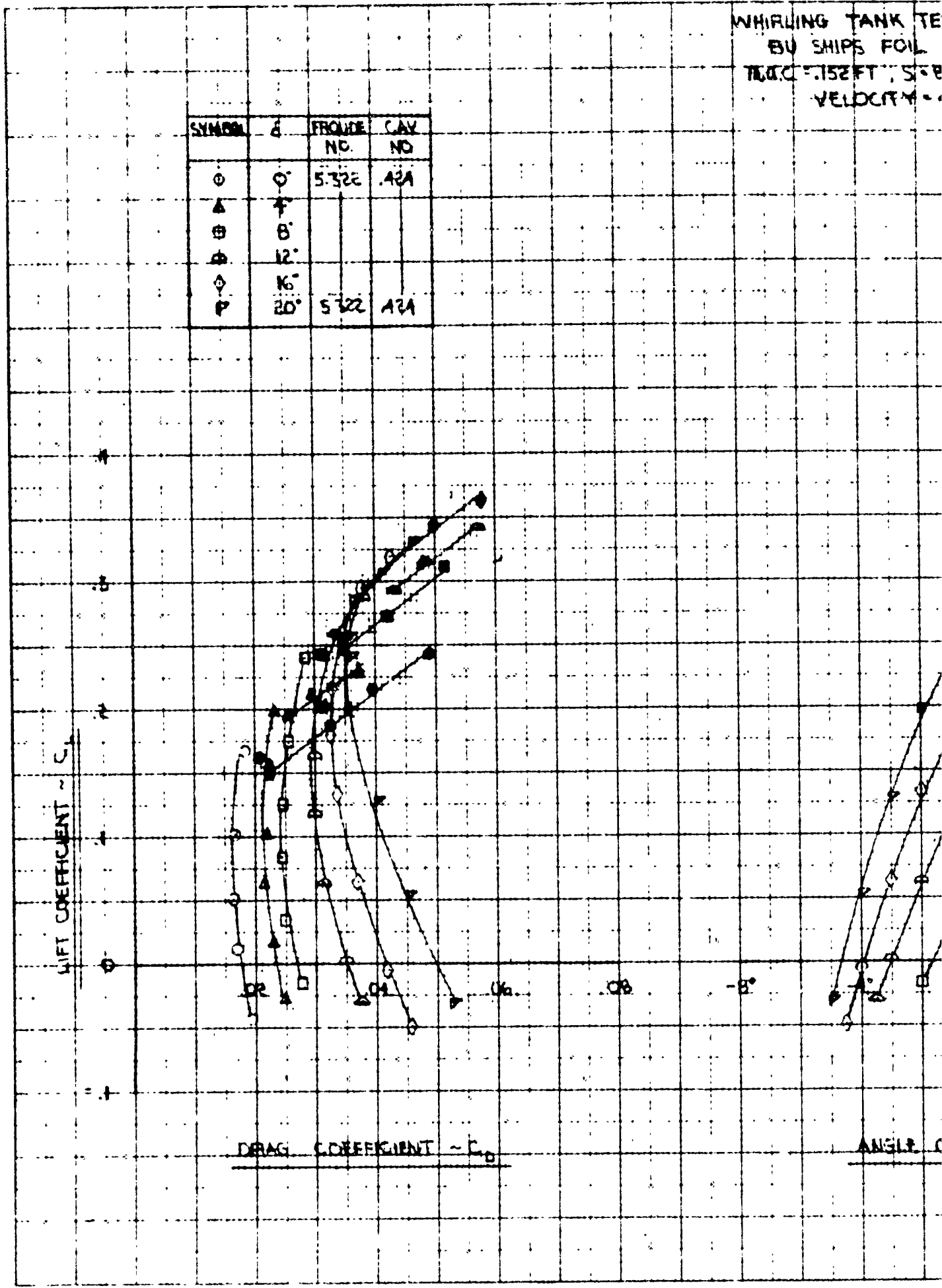
SYMBOL	δ	FROUDE NO.	CAV NO.
○	0°	5.322	.424
▲	4°		
⊕	8°		
⊙	12°		
◇	16°		
□	20°	5.322	.424

M-E ALUMINUM 1001
 1/2" x 1/2" x 1/2"

LIFT COEFFICIENT - C_L

DRAG COEFFICIENT - C_D

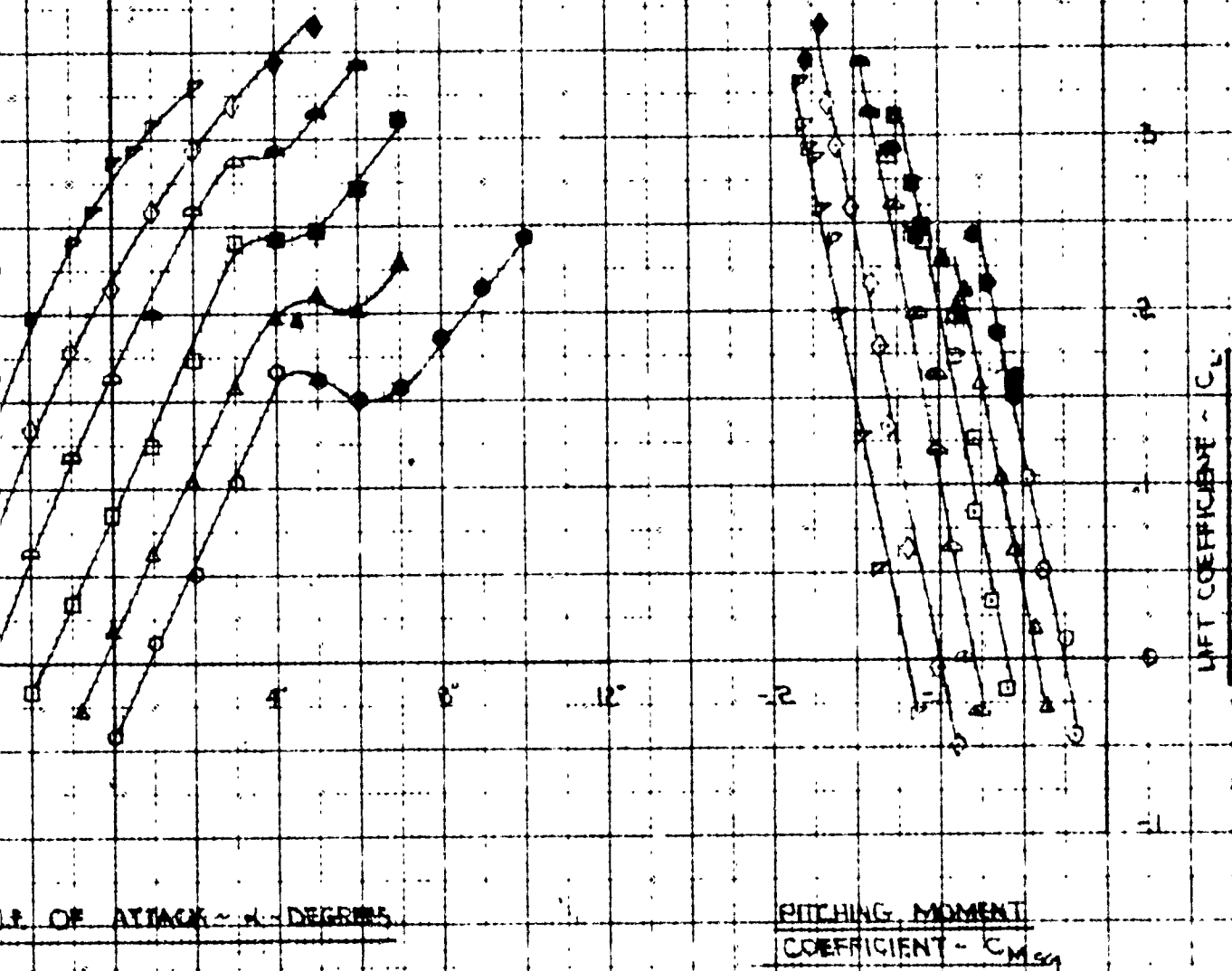
ANGLE OF ATTACK



TEST NO 42
OIL NO 13
S = 8.00 SQ IN, $AR = 1.00$
M = 45 KTS

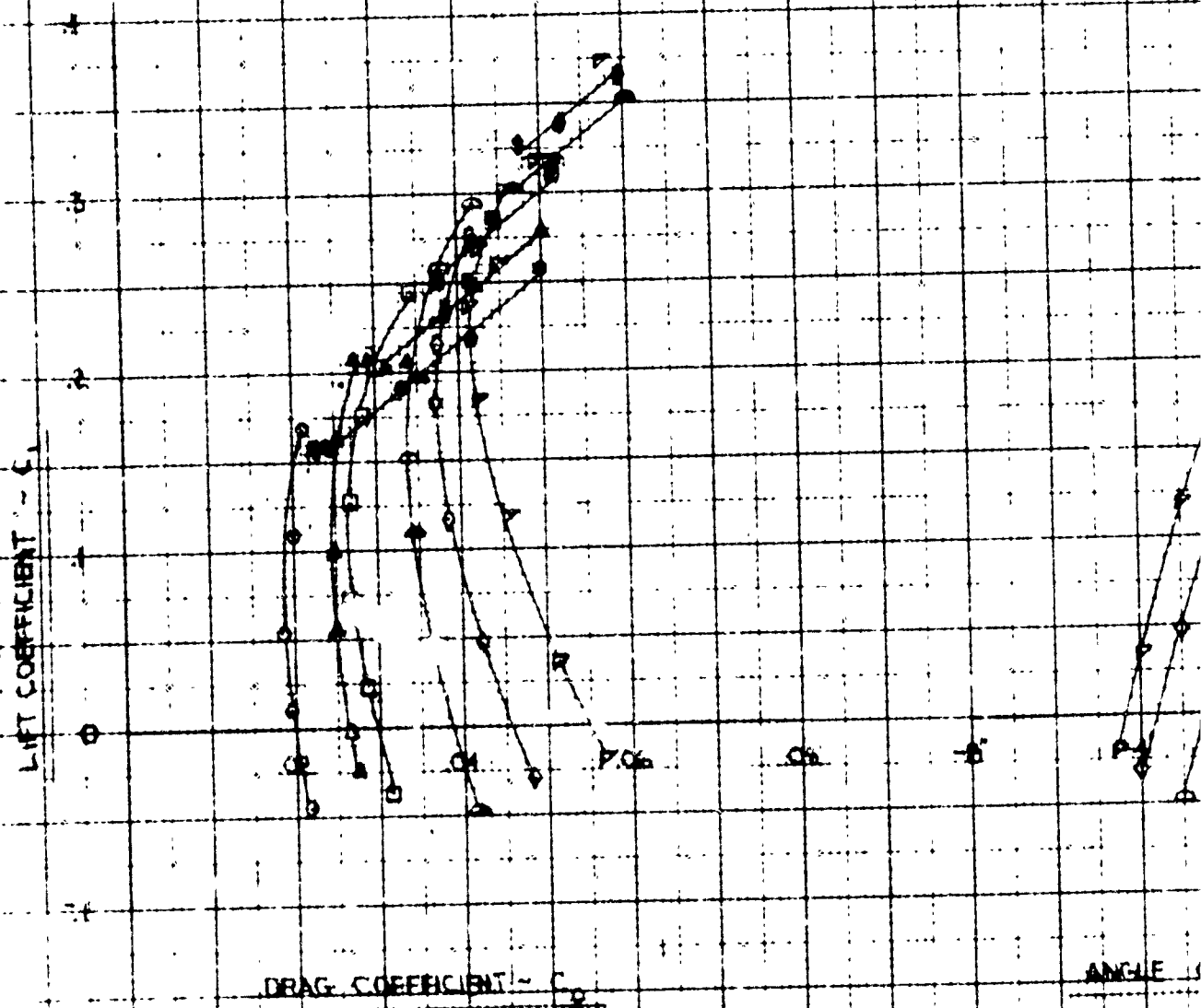
PAGE IV.123

NOTE: DARKENED POINTS INDICATE VENTED FLOW



WHIRLING TANK
BU SHIPS FO
M.O.C. .152 FT
VELOCITY

SYMBOL	δ	FRODO NO.	CAV NO.
○	0°	4.718	438
△	4°		
□	8°		
◇	12°		
●	16°		
▽	20°	4.718	438



NO. 100-200000-1

TANK TEST NO 42

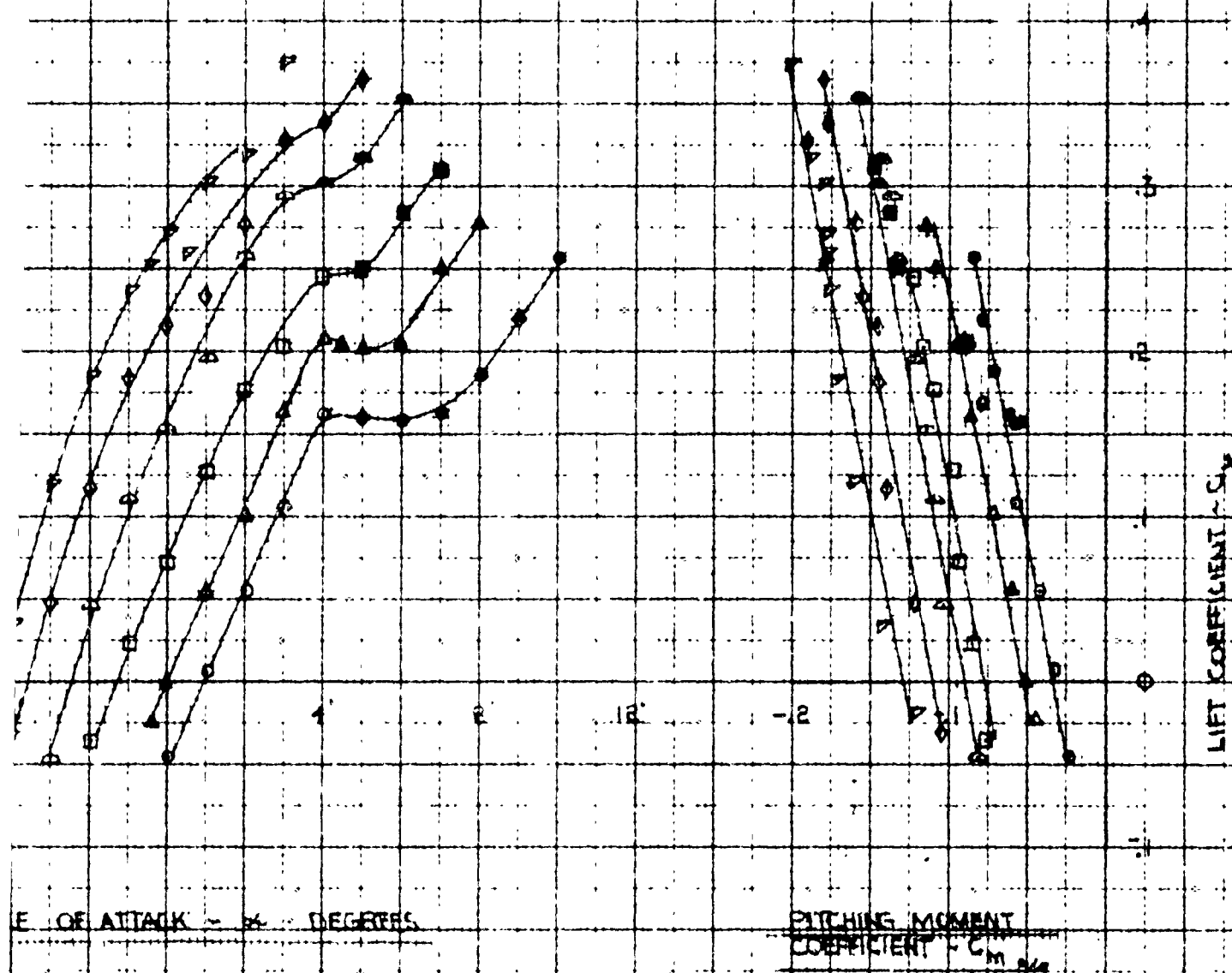
FOIL NO 13

T, 5.8005014, $\alpha_c = 1.25$

ITY: 45 KTS

PAGE 18124

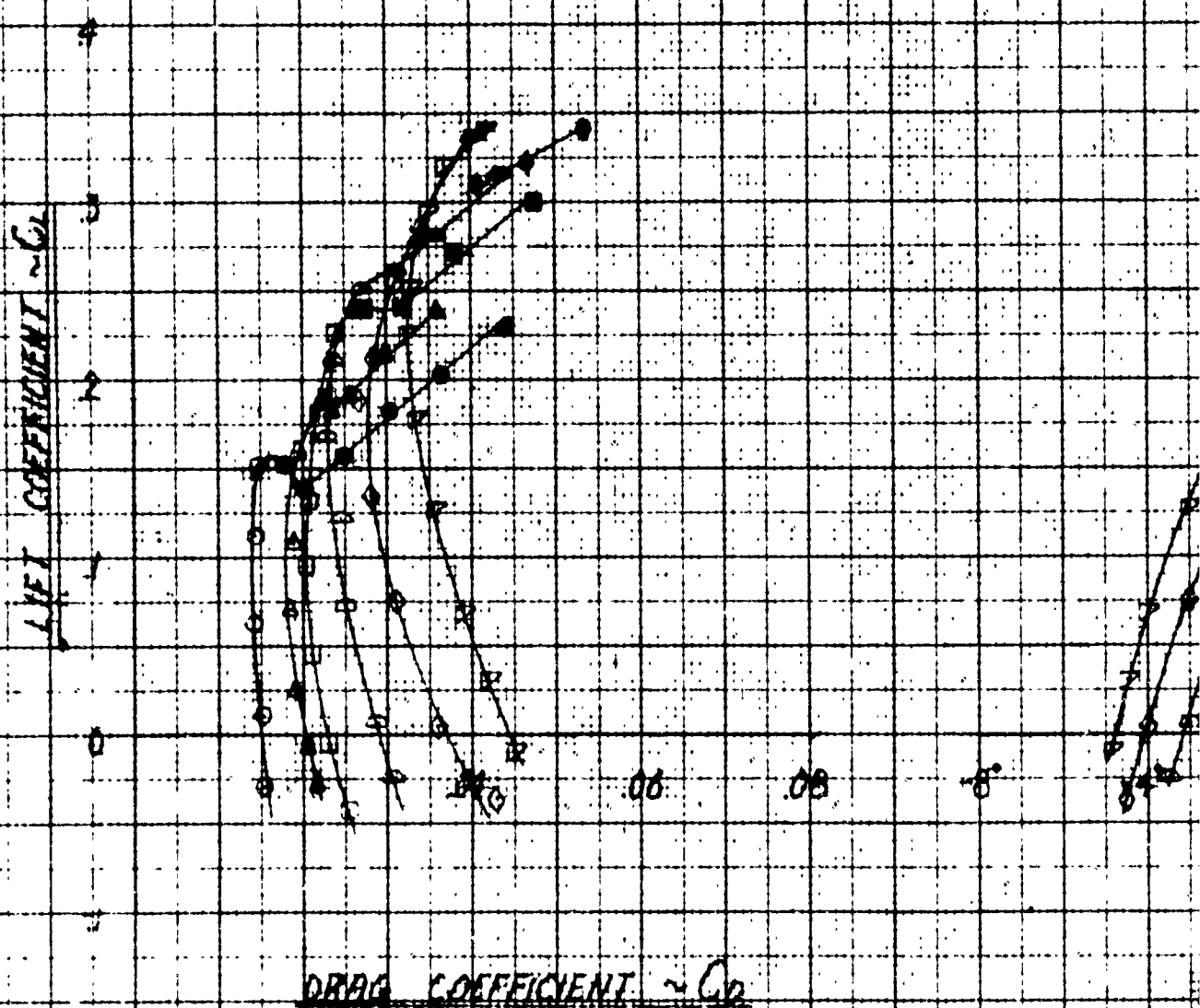
NOTE: DARKENED POINTS INDICATE VENTED FLOW



W.E. ALLEN 1987 52303

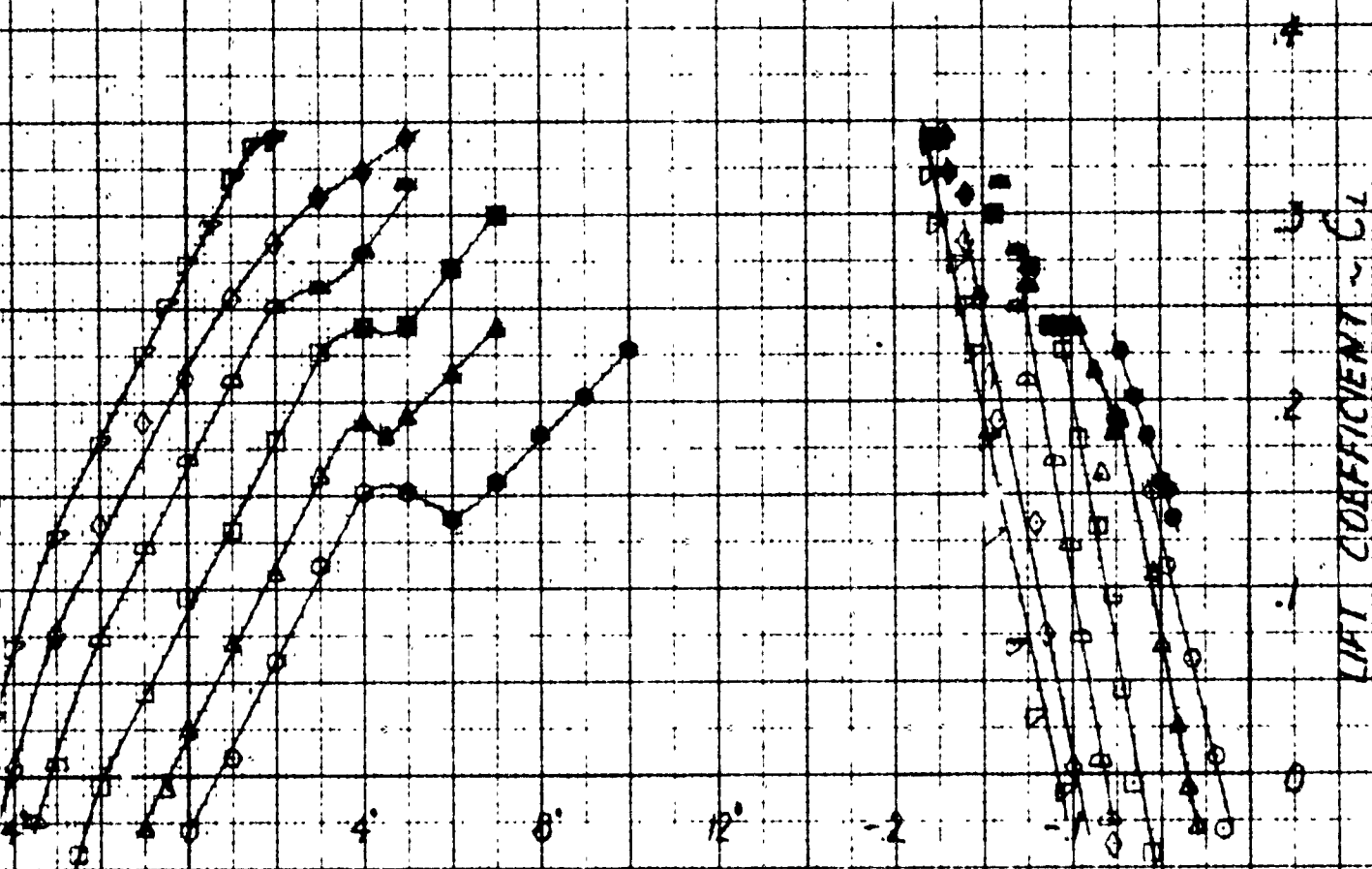
WHIRLING TA
BU SHAPS
MAC = 152 FT
VELOC

SYMBOL	δ	FROUDE NR	CAN NR
○	0°		
△	4°	6	3
□	8°	1	4
◆	12°	2	1
◇	16°		
⊙	20°		



TANK TEST NO. 42
 APS ROLL NO. 13
 REF. $S = 8.00 \text{ IN}^2$, $\alpha_c = 75$
 VELOCITY = 50 MTS.

NOTE: DARKENED SYMBOLS INDICATE VENTED FLOW.



ANGLE OF ATTACK $\sim \alpha \sim$ DEGREES

PITCHING MOMENT
 COEFFICIENT $\sim C_{M_{ac}}$

LIFT COEFFICIENT $\sim C_L$

WHIRLING
BY SHIPS
MAC = 152 FT.
VELOC

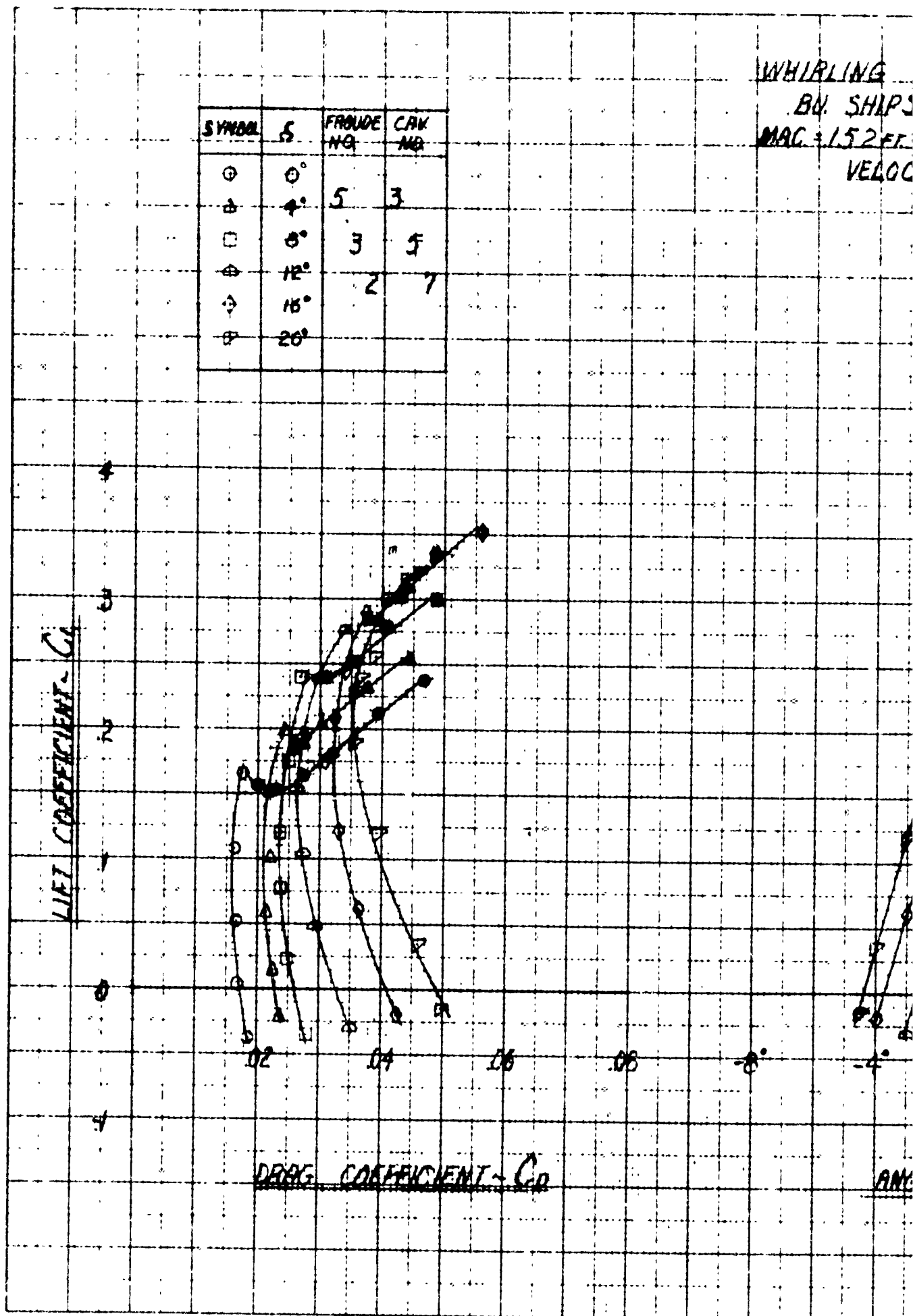
SYMBOL	δ	FROUDE NO	CAY NO
○	0°		
△	4°	5	3
□	8°	3	5
⊕	12°	2	7
◇	16°		
⊗	20°		

NOE 1.000000000 0.000000000

LIFT COEFFICIENT - C_L

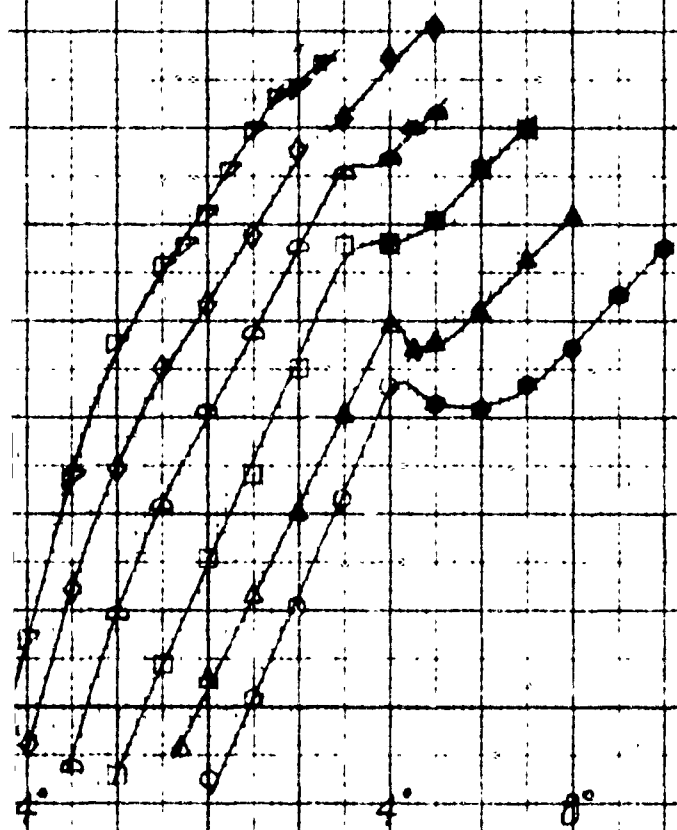
DRAG COEFFICIENT - C_D

ANY



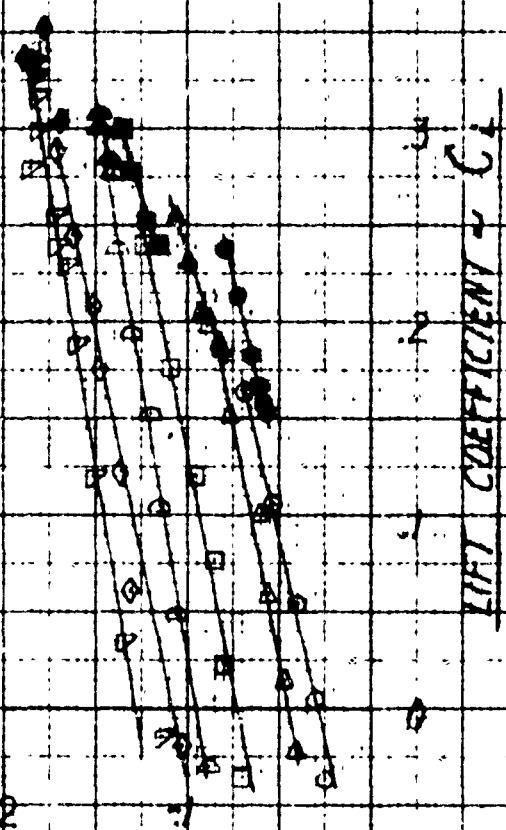
G TANK TEST NO. 42
 WIPS FOIL NO. 13
 REF. S = 800 IN²; $\frac{A}{c} = 1.00$
 VELOCITY = 50 MTS.

NOTE: DARKENED SYMBOLS INDICATE VENTED FLOW.



ANGLE OF ATTACK - α - DEGREES

PITCHING MOMENT
 COEFFICIENT - $C_{m, \frac{1}{4}}$



LIFT COEFFICIENT - C_L

McE 11/20/50 1007 2.5

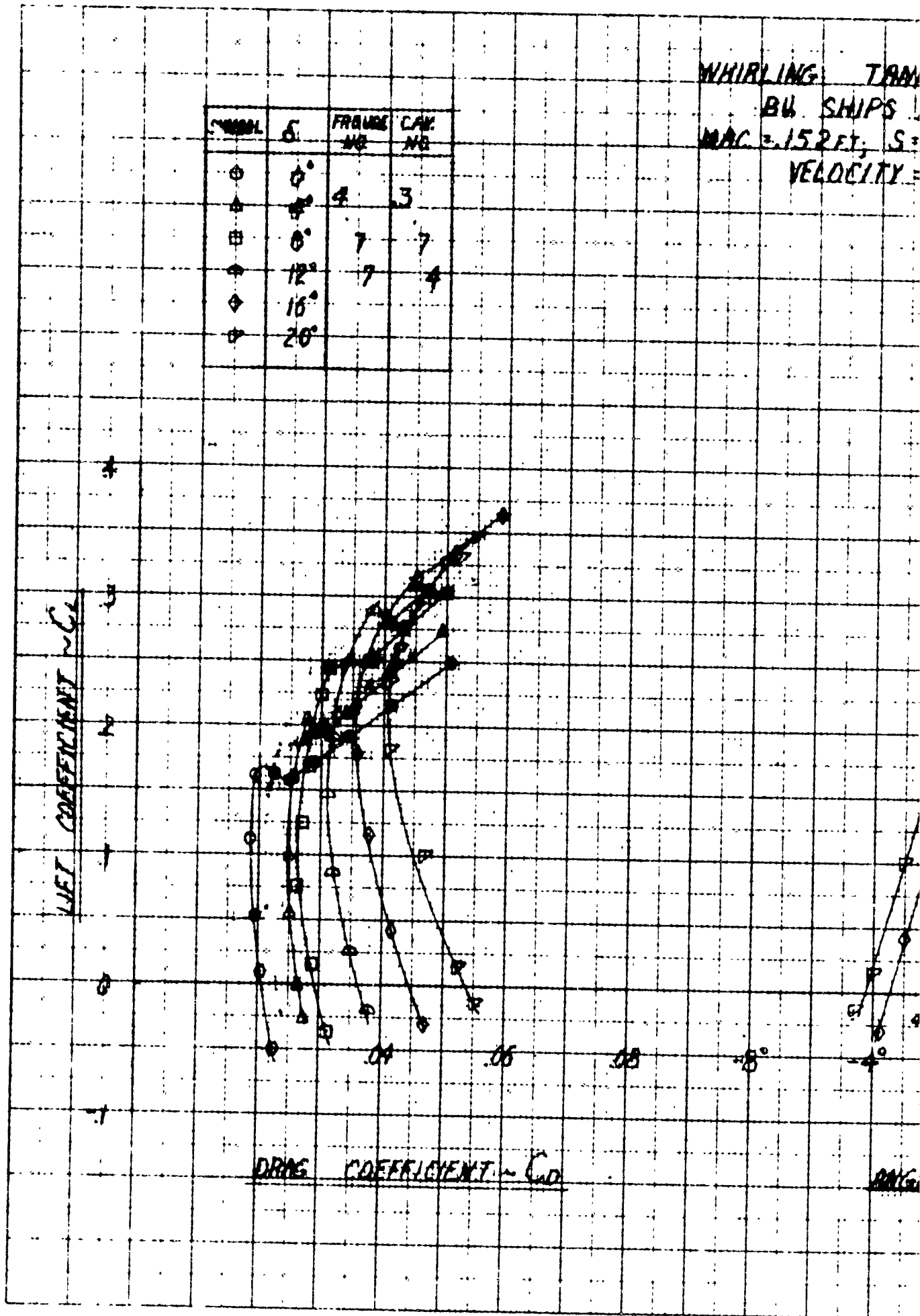
WHIRLING TRAM
BU SHIPS
MAC = 1.52 FT; S =
VELOCITY =

SYMBOL	δ	FRAME NO	CAN NO
○	0°		
△	4°	4	3
□	8°	7	7
◇	12°	7	4
◇	16°		
◇	20°		

LIFT COEFFICIENT $\sim C_L$

DRAG COEFFICIENT $\sim C_D$

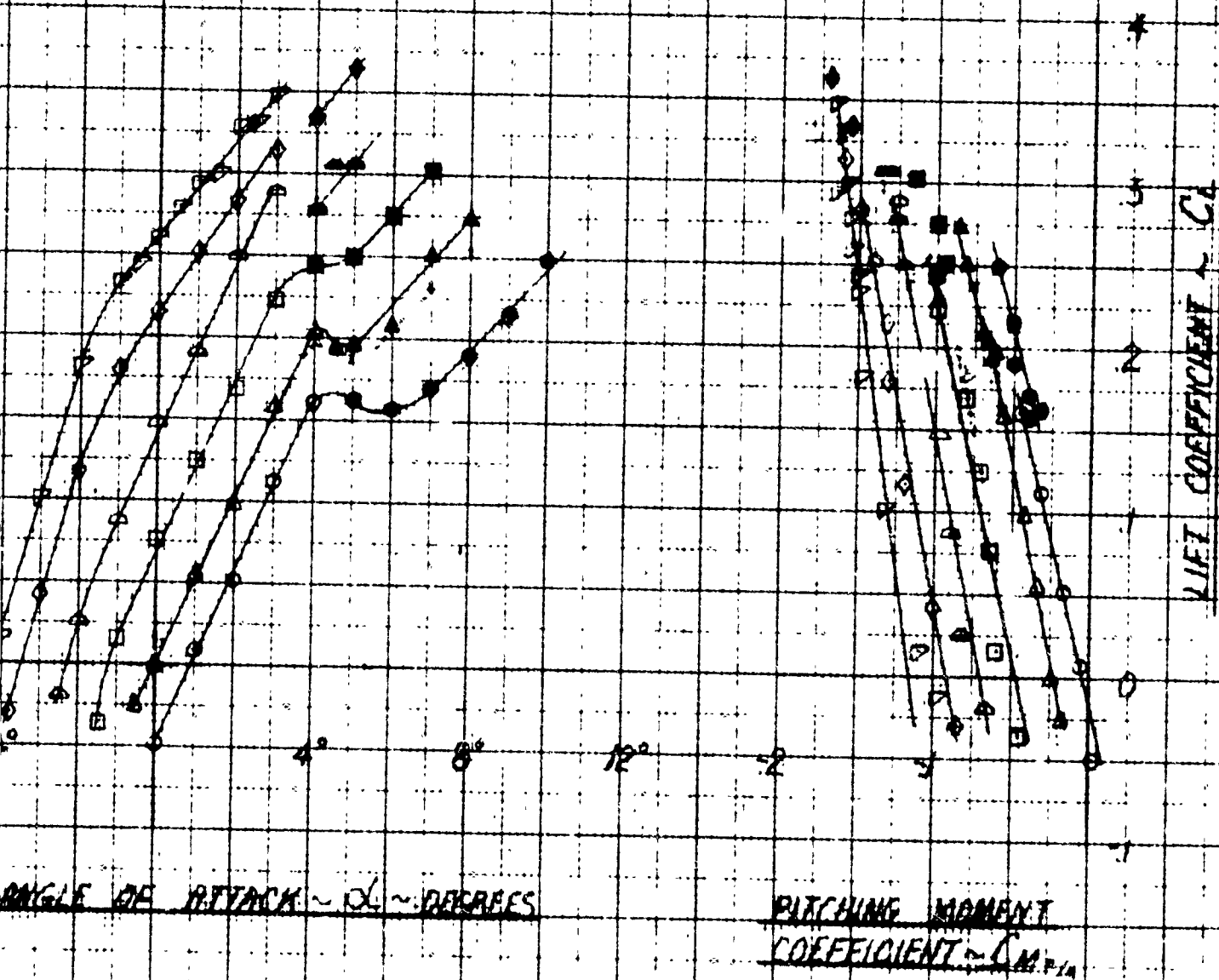
ANGLE



TRAW TEST NO 42
PS FOIL NO 13
S = 8.000M²; $d/c = 1.25$
TY = 50 KTS

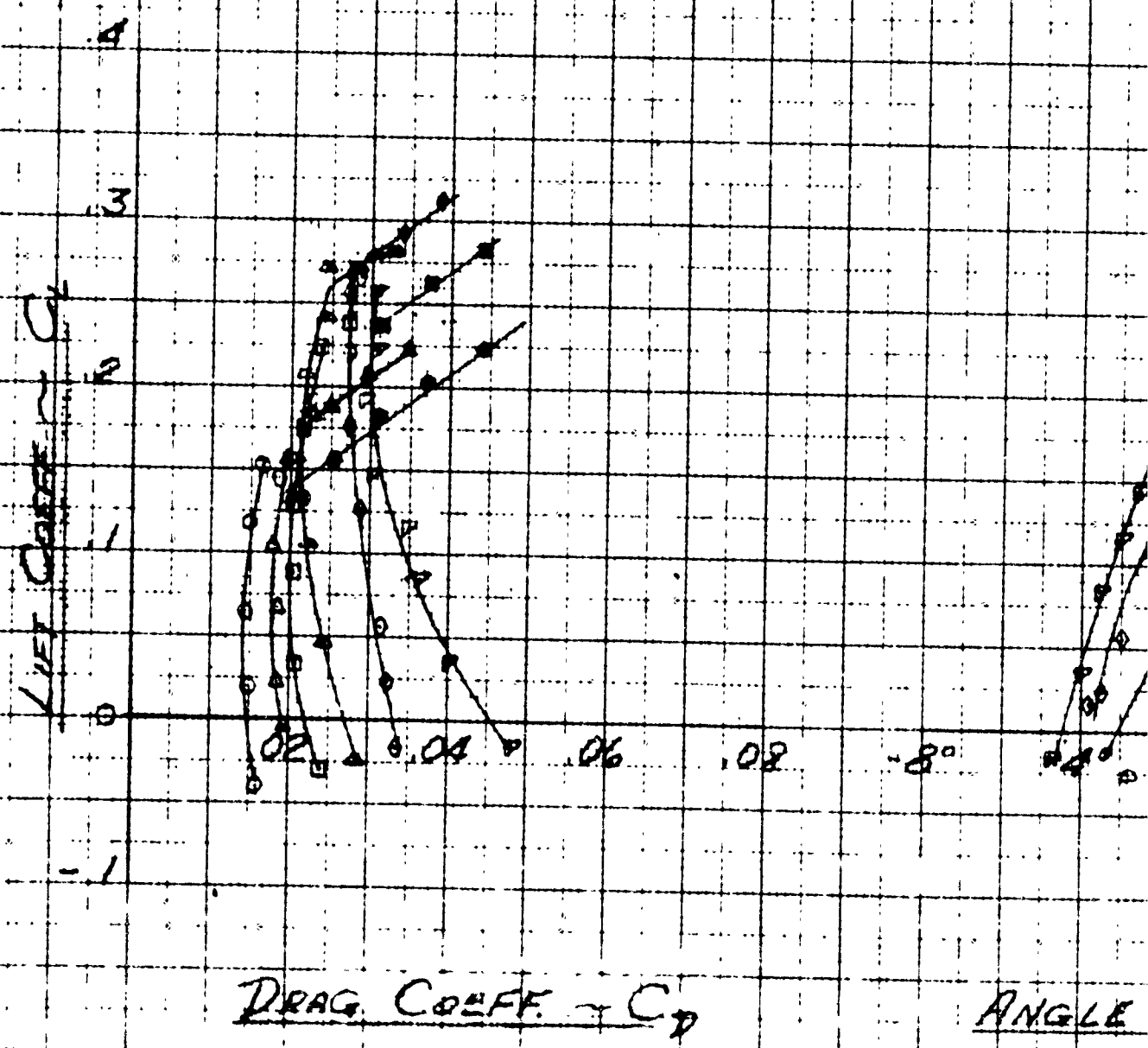
PAGE IV. 127

NOTE: DARKENED SYMBOLS INDICATE VENTED FLOW.



SYMBOL	FLAP SET	FROUDE NO.	CAY. NO.
○	0°		
△	4°	6	2
□	8°	1	5
◇	12°	2	8
▽	16°	3	
▼	20°	4	

NOTE
1. D
1A



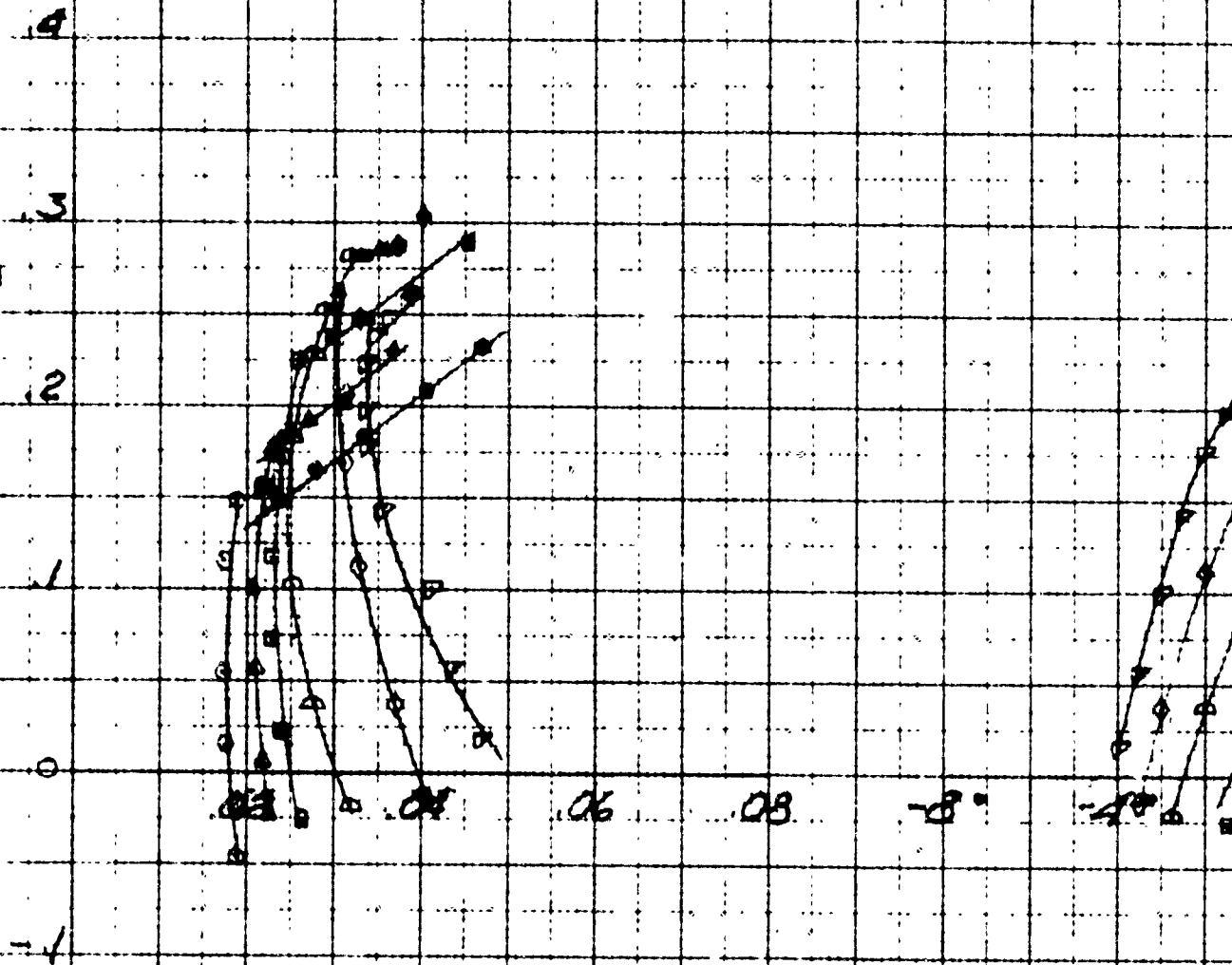
SYMBOL	FLAP SET	FROUDE No.	CAV. No.
○	0°		
△	4°	5.3	27
□	8°	2.9	03
▲	12°		
◇	16°		
▽	20°		

NOTE:
1. DA
INL

LIFT COEFF. $\sim C_L$

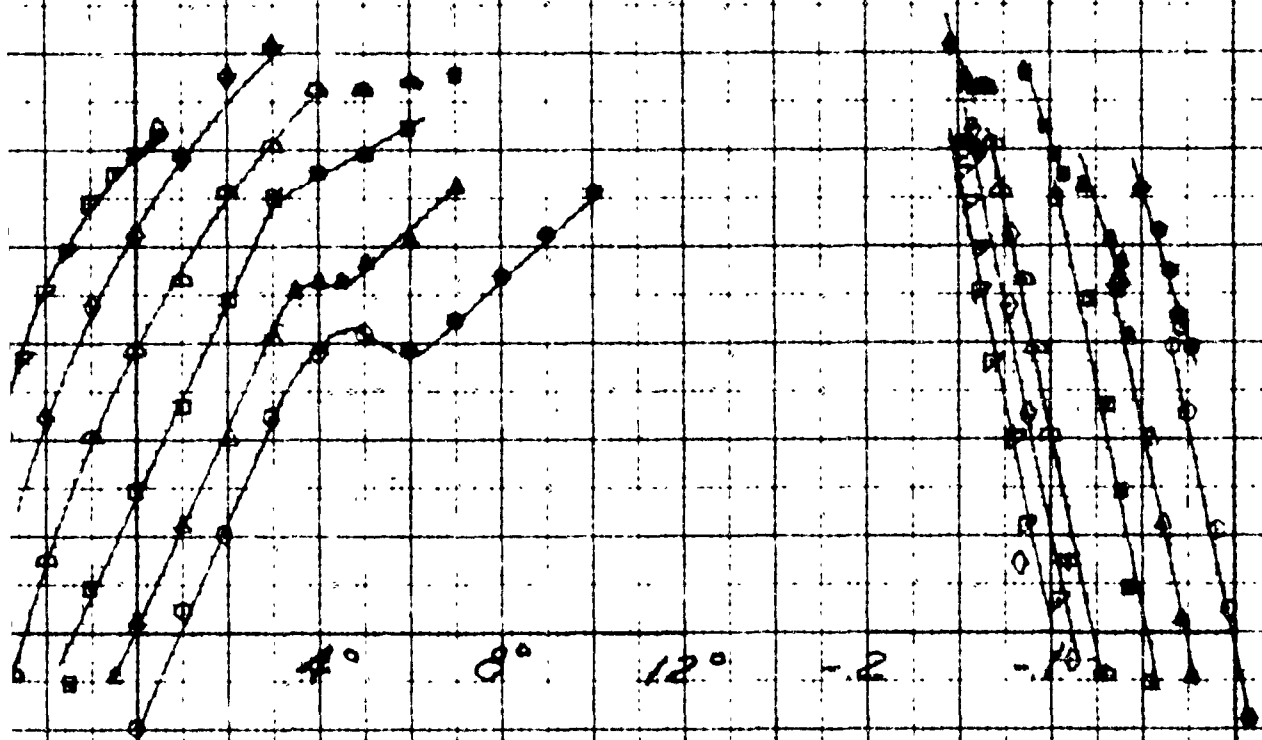
DRAG COEFF. $\sim C_D$

ANGLE α



WHIRLING TANK TEST NO. 42
 BUSHIPS FOIL NO. 13
 M.A.C. = 152 FT AREA = 88 $\frac{1}{2}$ 1.00
 VEL = 60 KTS

DARKENED SYMBOLS
INDICATE VENTED FLOW



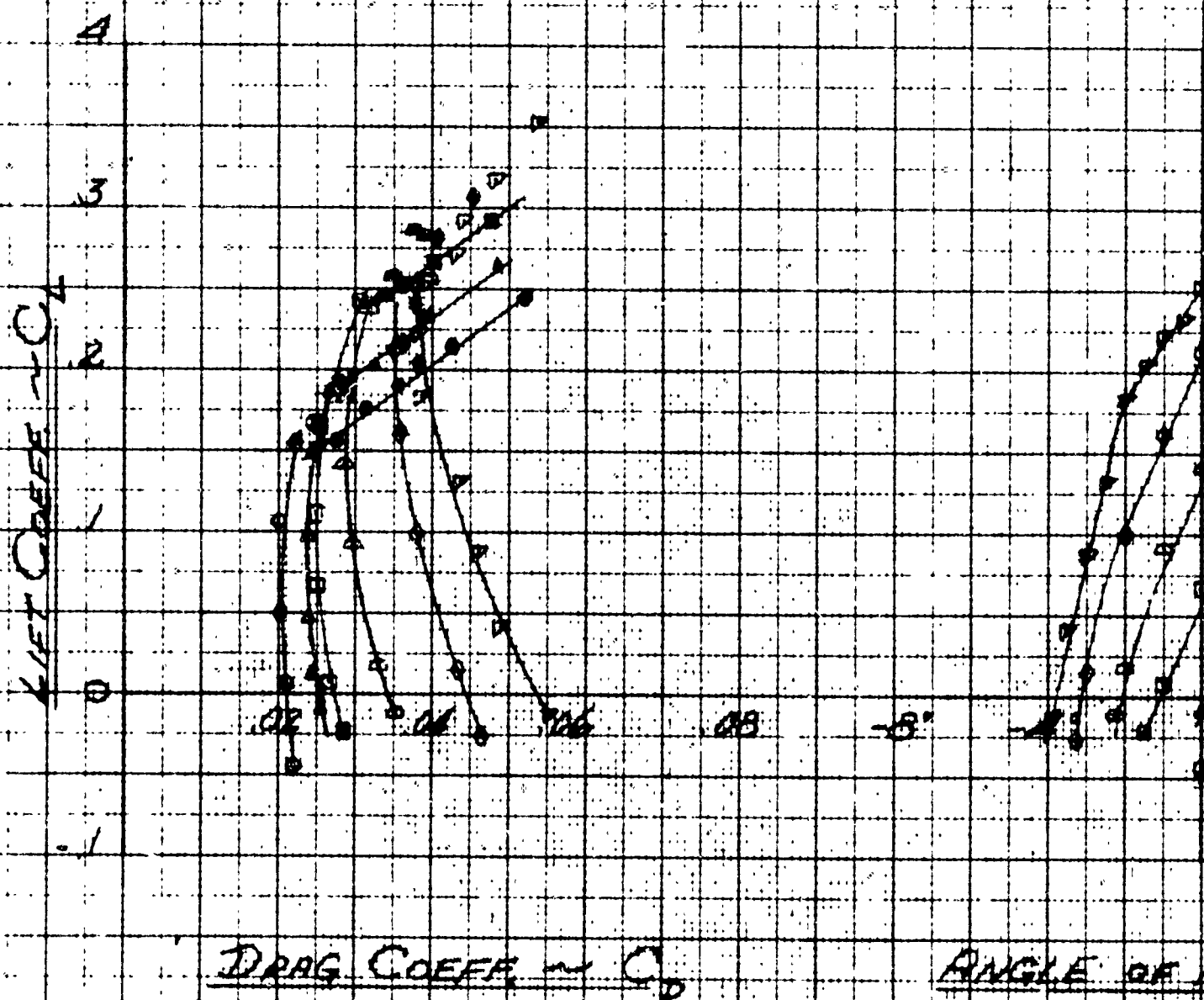
ANGLE OF ATTACK - DEGREES

PITCHING MOMENT
COEFF - C_m

SYMBOL	FLAP SET	FROUDE NO.	CAN. NO.
○	0°		
▲	4°	4.7	2851
□	8°		
△	12°		
◇	16°		
●	20°		

NOTE

1. D
1A
FD



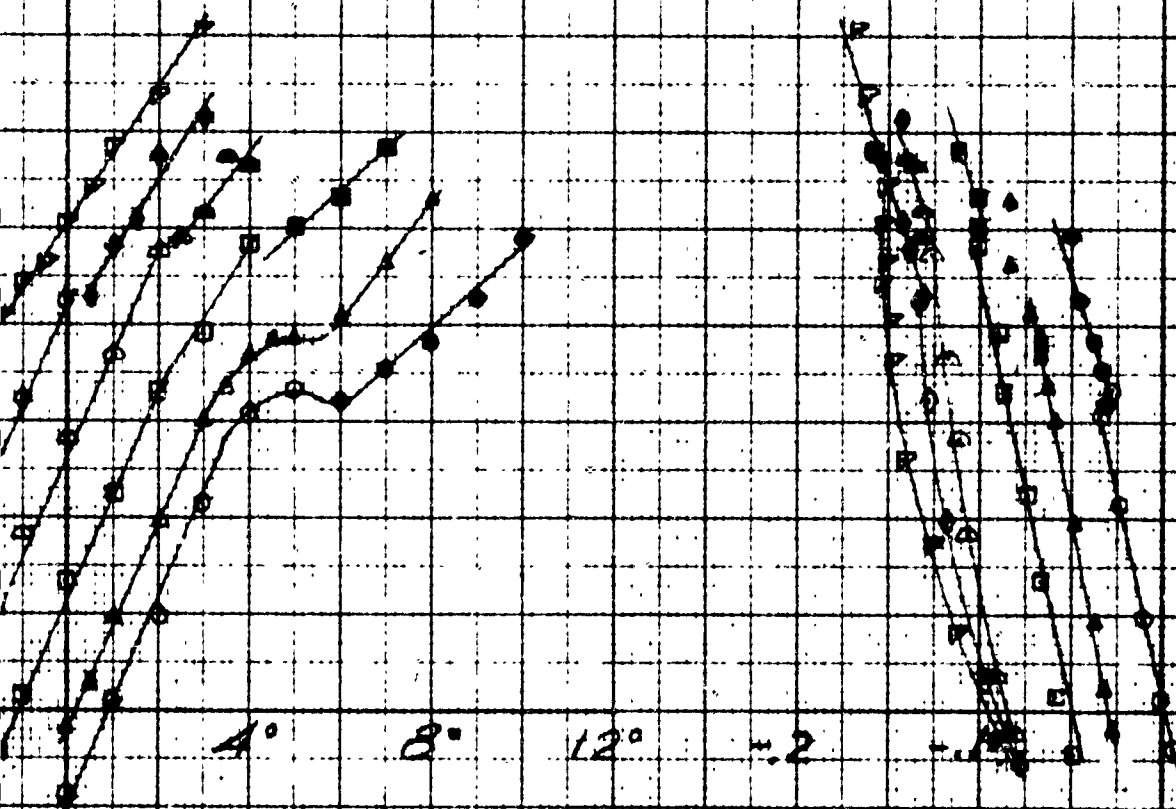
ANGLE OF

WHIRLING TANK TEST NO. 42

BUSHIPS FOIL NO. 13

M.A.C. = .152 FT AREA = 87" $d/c = 1.25$ VEL. = 60 KTS

TE
 1. DARKENED SYMBOLS
 INDICATE VENTED
 FLOW.

ANGLE OF ATTACK - DEGREESPITCHING MOMENTCOEFF - C_m

WHIRLING TANK
BU SHIPS FOL
W.C. 1.15E FT, S
VELOCITY =

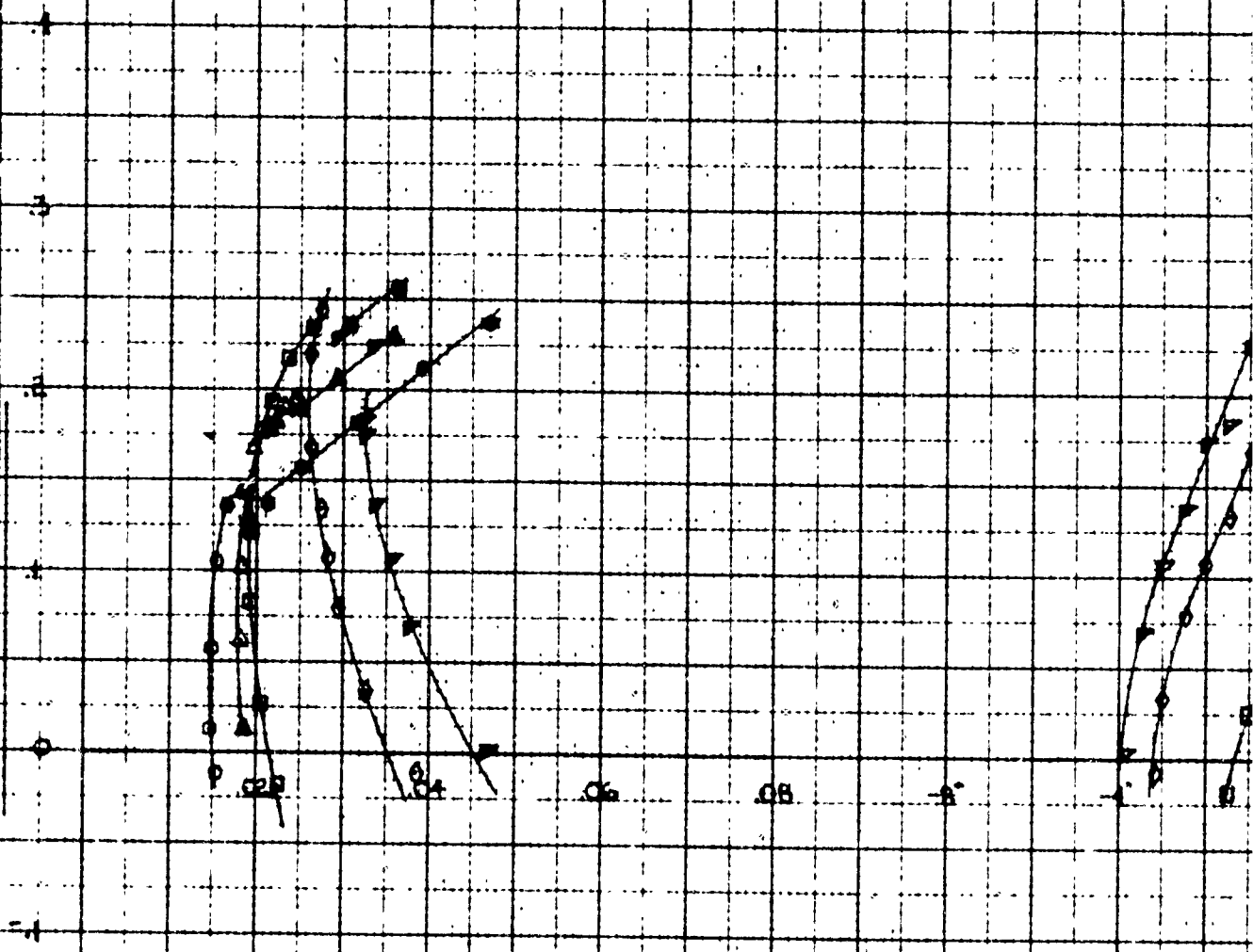
SYMBOL	δ	FROUDE NO	CAV NO
○	0°	6.123	199
▲	4°		
◻	8°		
△	12°		
◇	16°		
▽	20°	6.123	199

NOE ALUMINUM TUB 52155
8 1/2 IN DIA

LIFT COEFFICIENT - C_L

DRAG COEFFICIENT - C_D

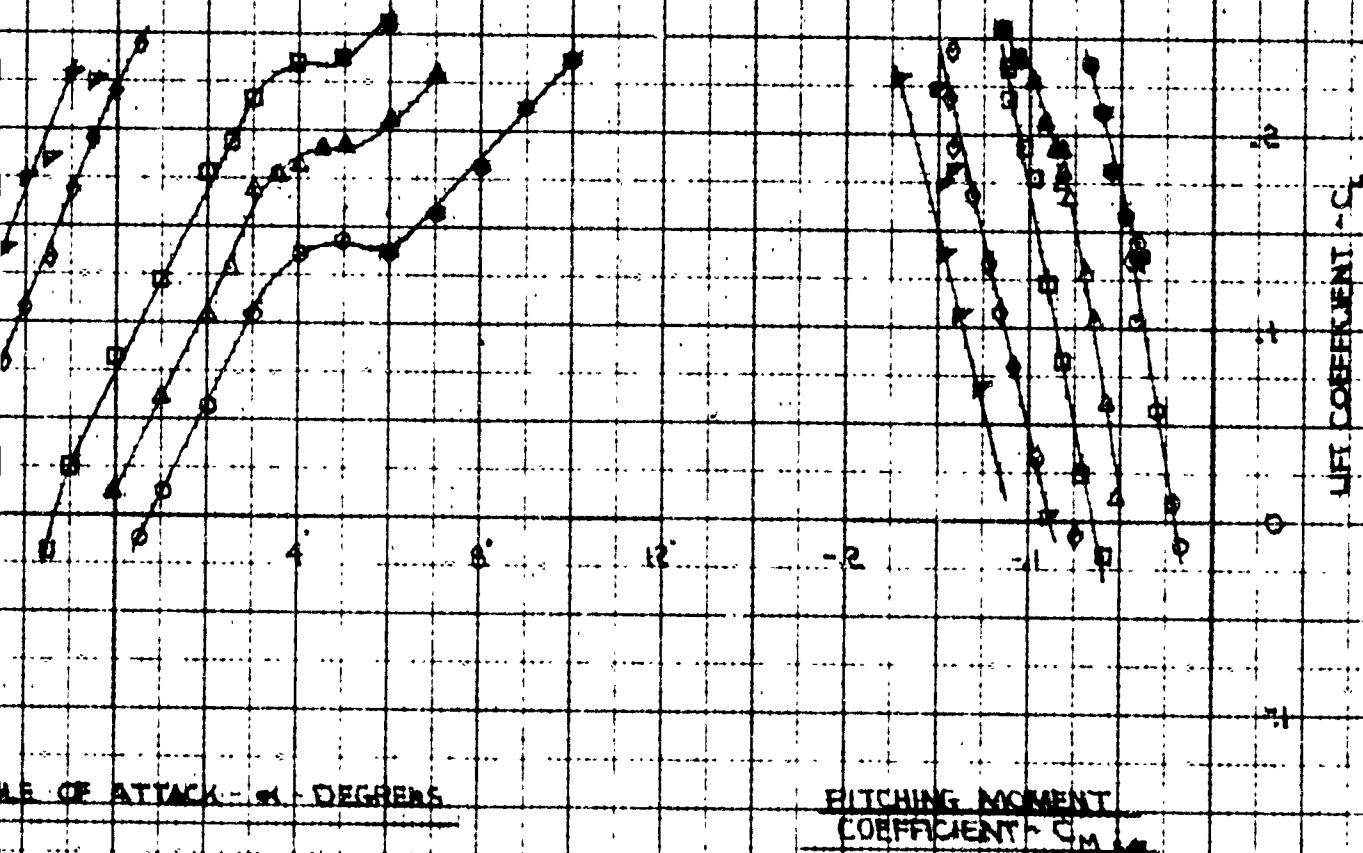
ANGLE α



TANK TEST NO 42
POL NO 13
S = 800 SQ IN, $\rho/\sigma = .75$
TY: 40 KTS

PAGE IV.131

NOTE: DARKENED POINTS INDICATE VENTED FLOW



WHIRLING TANK TEST
BU SHIPS FOIL
H. 12' 52 FT 5.80
VELOCITY 704

SYMBOL	δ	FROUDE NO	CAN NO
◇	0°	5.322	216
▲	4°		
□	8°		
△	12°		
▽	16°		
●	20°	5.322	216

LIFT COEFFICIENT - C_L

DRAG COEFFICIENT - C_D

ANGLE OF A

ME ALABAMA 1907 1233

ME ALABAMA 1907 1233

TEST NO 42

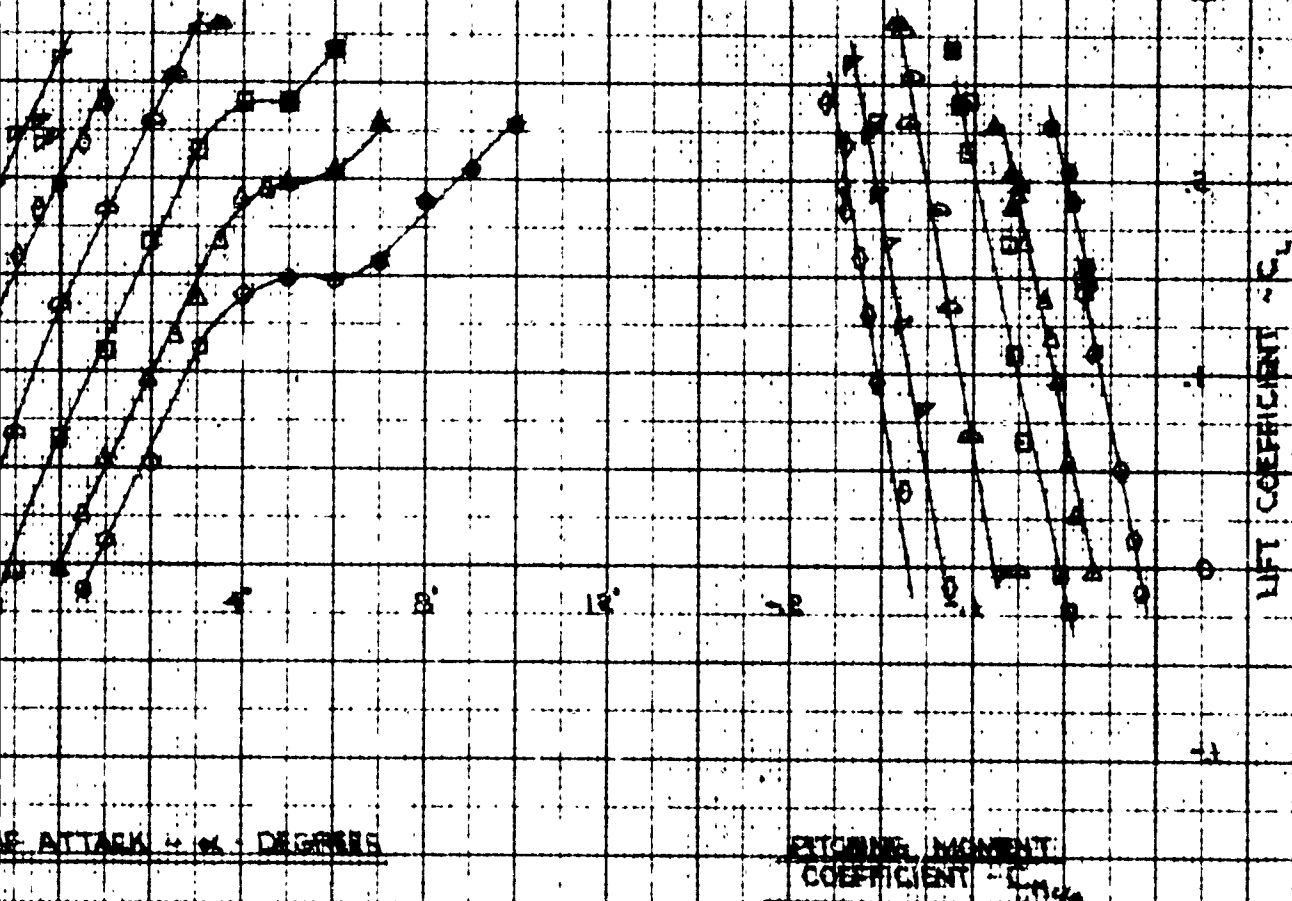
OIL NO 13

78.05 SQ IN; $\Delta E = 1.00$

70 KTS

PAGE IV. 132

NOTE DARKENED POINTS INDICATE VENTED FLOW



WHIRLING TANK TEST
 BU SHIPS FOIL NO
 M.O.C. 152 FT, S.B. 00 50
 VELOCITY 70 KTS

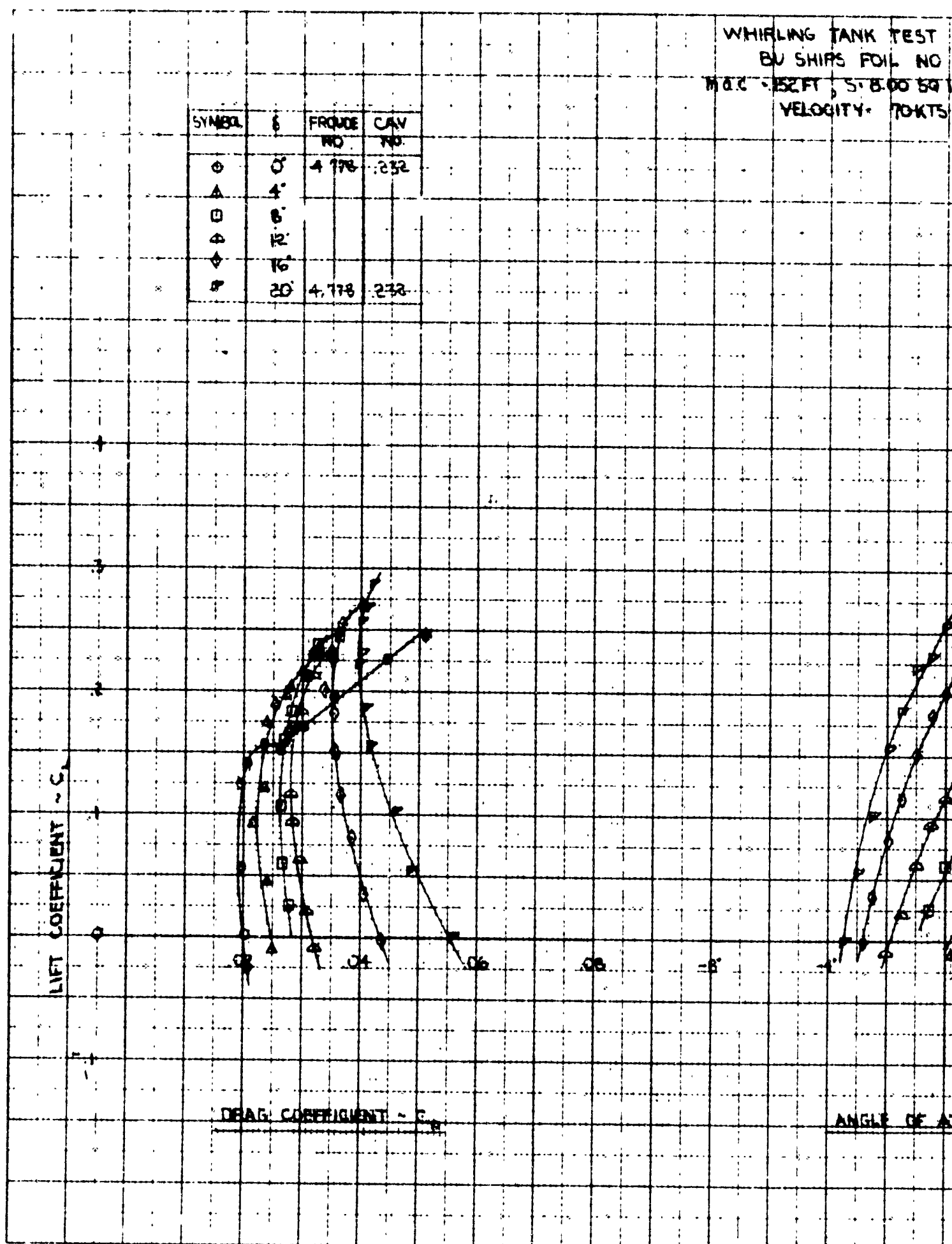
SYMBOL	δ	FROUDE NO	CAV NO
○	0°	4.778	232
△	4°		
□	8°		
◇	12°		
◆	16°		
●	20°	4.778	232

LIFT COEFFICIENT - C_L

DRAG COEFFICIENT - C_D

ANGLE OF A

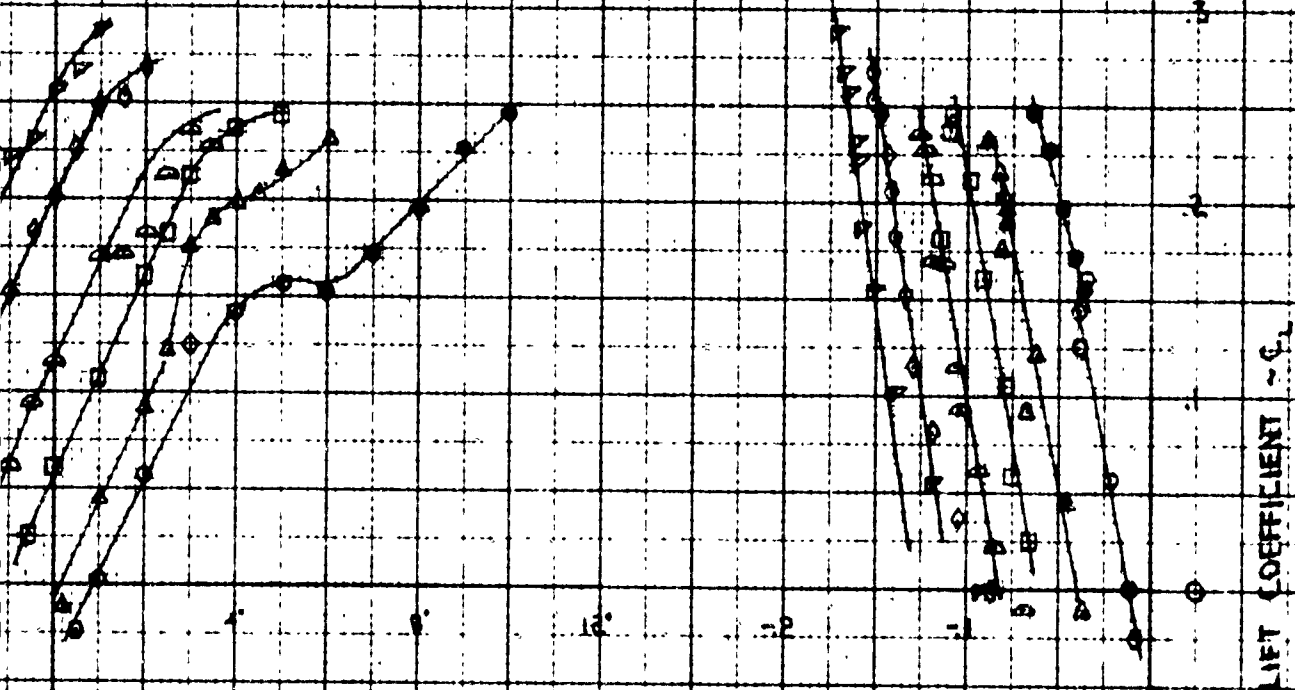
M-O-E ALBARENE 1007 2215



ST NO 42
NO 13
50 IN, 2/6-11.25
KTS

PAGE IV.133

NOTE: DARKENED POINTS INDICATE VENTED FLOW



ANGLE OF ATTACK - α - DEGREES

PITCHING MOMENT
COEFFICIENT - C_m

LIFT COEFFICIENT - C_L

WHIRLING TANK TEST NO 42
 BU SHIPS FOR NO 13
 W.A.C. 1152 FT. S. 800 SQ. IN. 4/6/75
 VELOCITY: 30 KTS

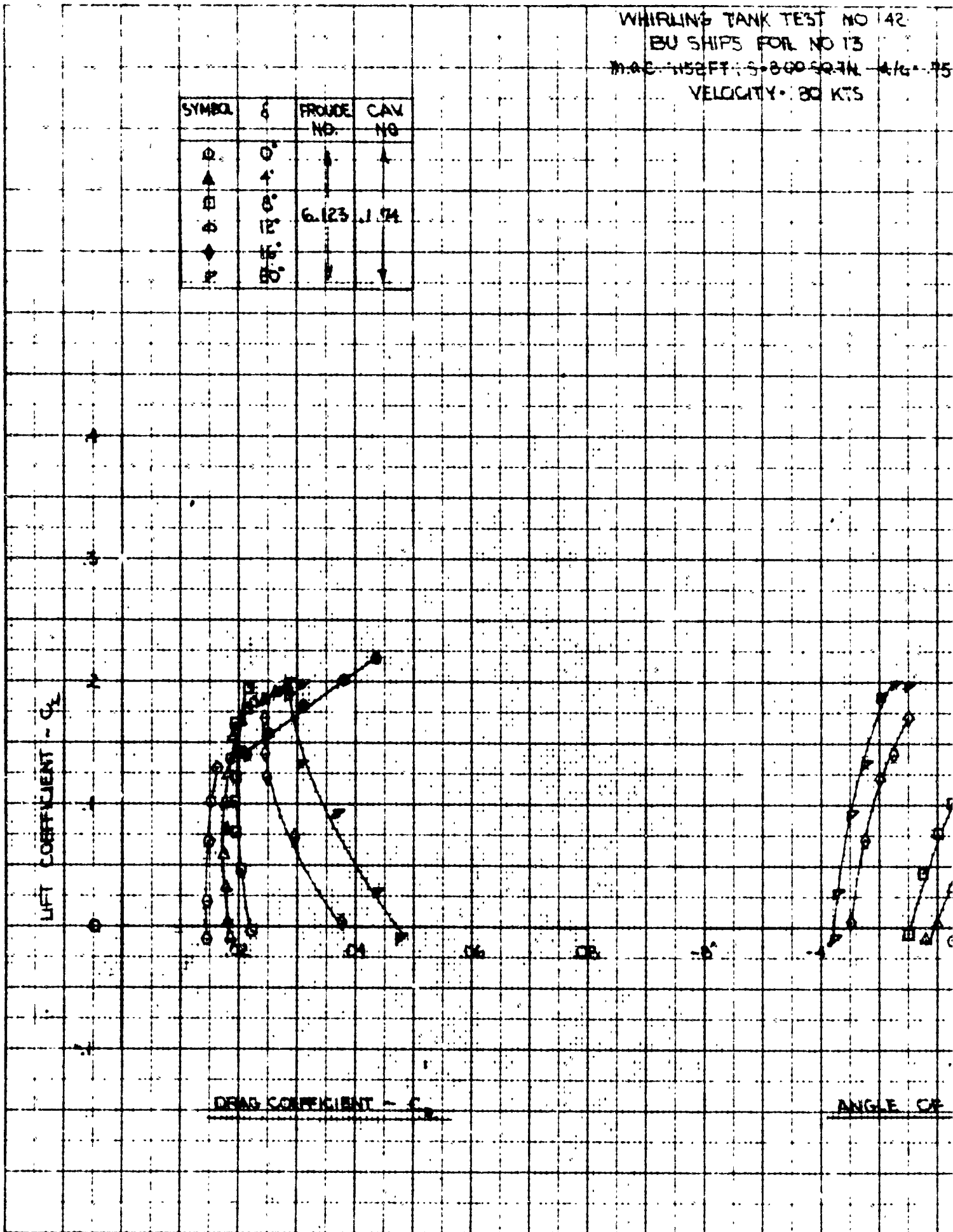
SYMBOL	δ	FROUDE NO.	CAV NO.
○	0°	↓	↑
▲	4°	↓	↑
□	8°	6.123	1.174
◇	12°	↓	↑
◆	16°	↓	↑
●	20°	↓	↑

W.A.C. 1152 FT. S. 800 SQ. IN. 4/6/75

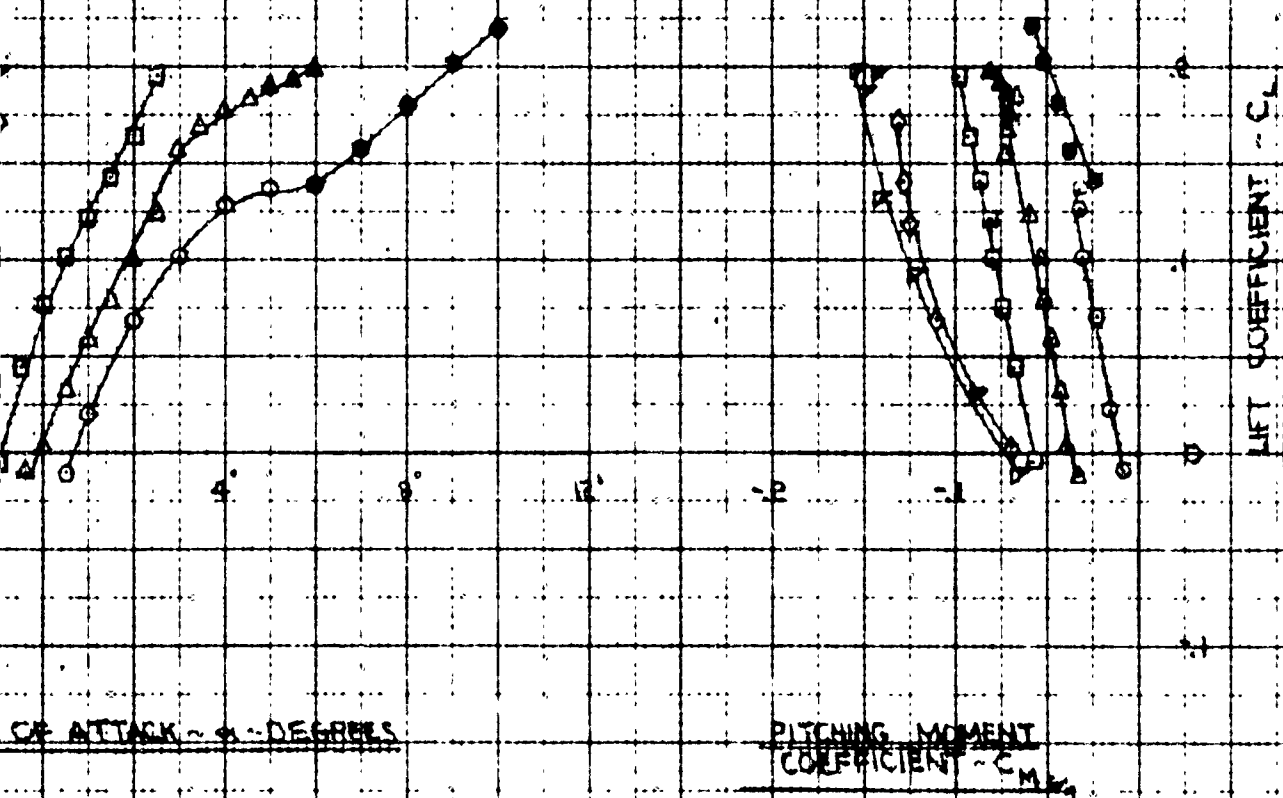
LIFT COEFFICIENT - C_L

DRAG COEFFICIENT - C_D

ANGLE OF



NOTE DARKENED POINTS INDICATE VENTED FLOW



WHIRLING TANK TEST N
 BU SHIPS FOIL NO 1
 MAC .152, S. 8.00 SQ IN
 VELOCITY 80 KTS

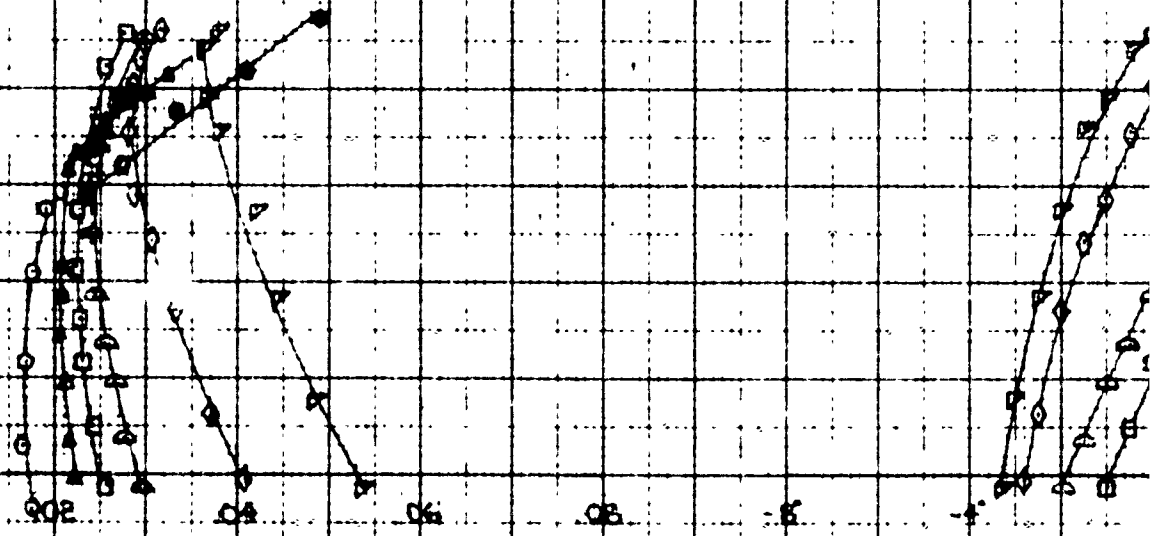
SYMBOL	G	FROUDE NO	CAV NO
○	0°		
▲	4°		
⊗	8°	5.32	195
⊕	12°		
◇	16°		
▽	20°		

MEASUREMENT 1007

NET COEFFICIENT - C_L

DRAG COEFFICIENT - C_D

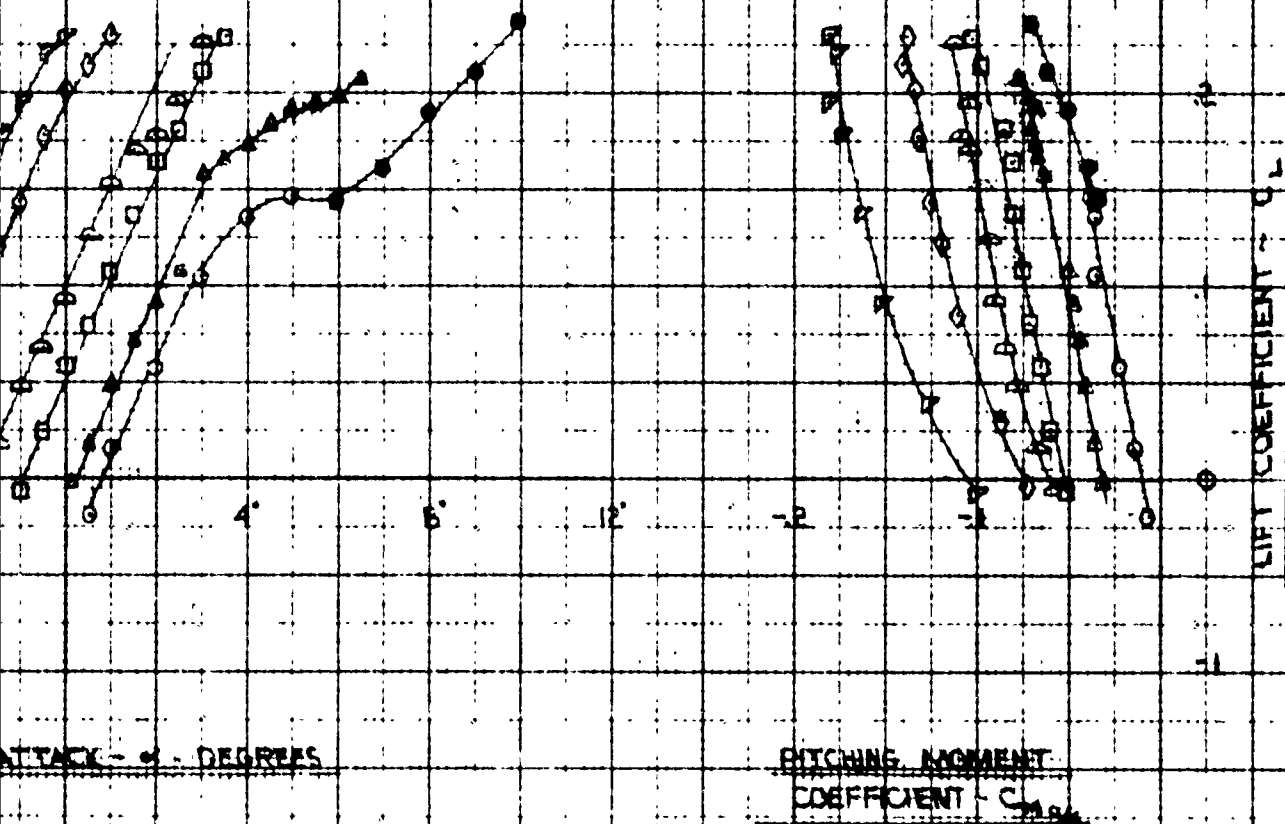
ANGLE OF ATTACK



T NO 42
NO 13
SQ IN 270.00
KTS

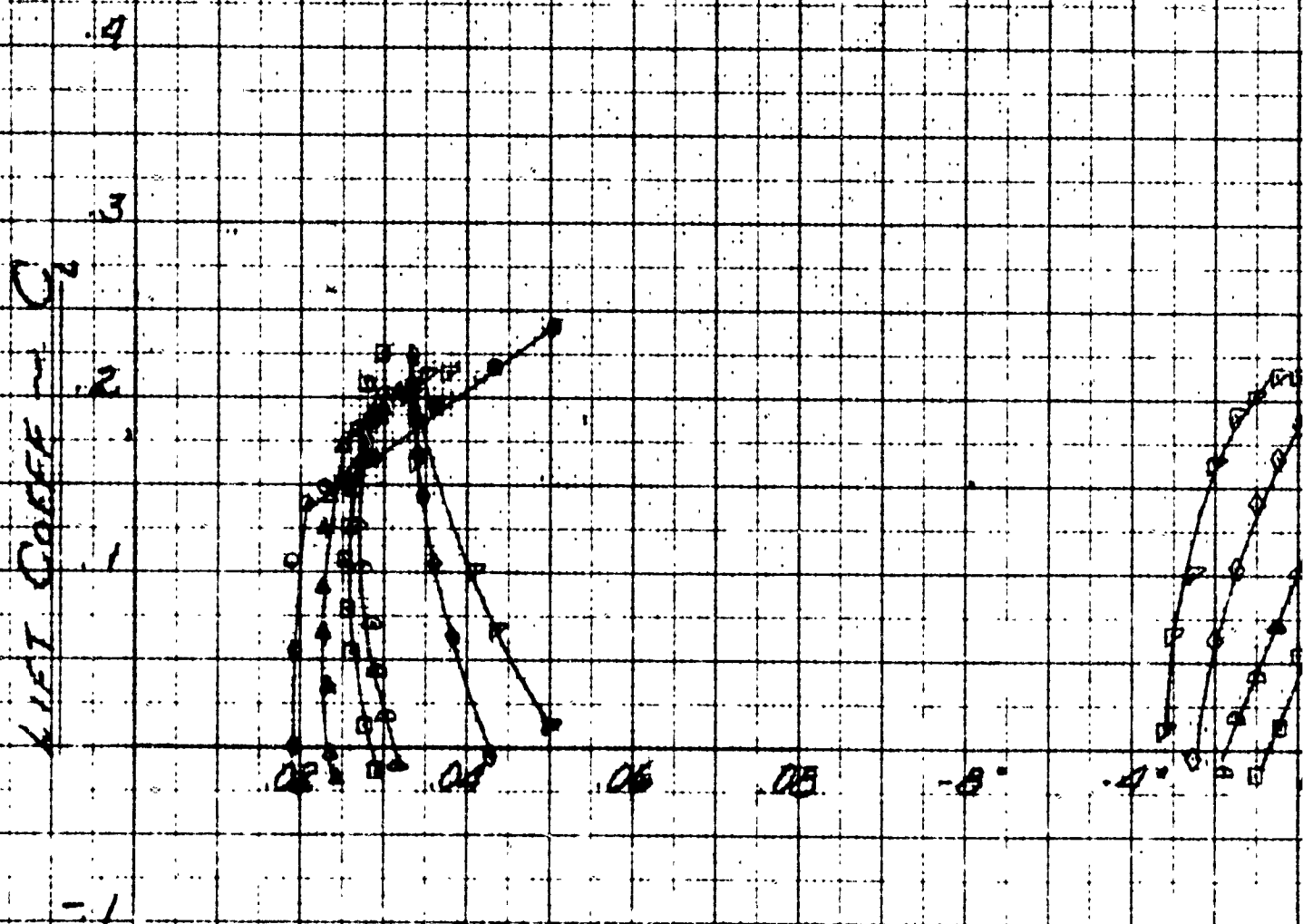
PAGE IV.135

NOTE DARKENED POINTS INDICATE VENTED FLOW



SYMBOL	FLAP SET	FROUDE NO.	CAV. NO.
○	0°		
△	4°	4.7	
□	8°	7.8	8
⊠	12°	8.5	9
◇	16°		
▽	20°		

NOTE
1



DRAG COEFF ~ C_D

ANGLE OF A

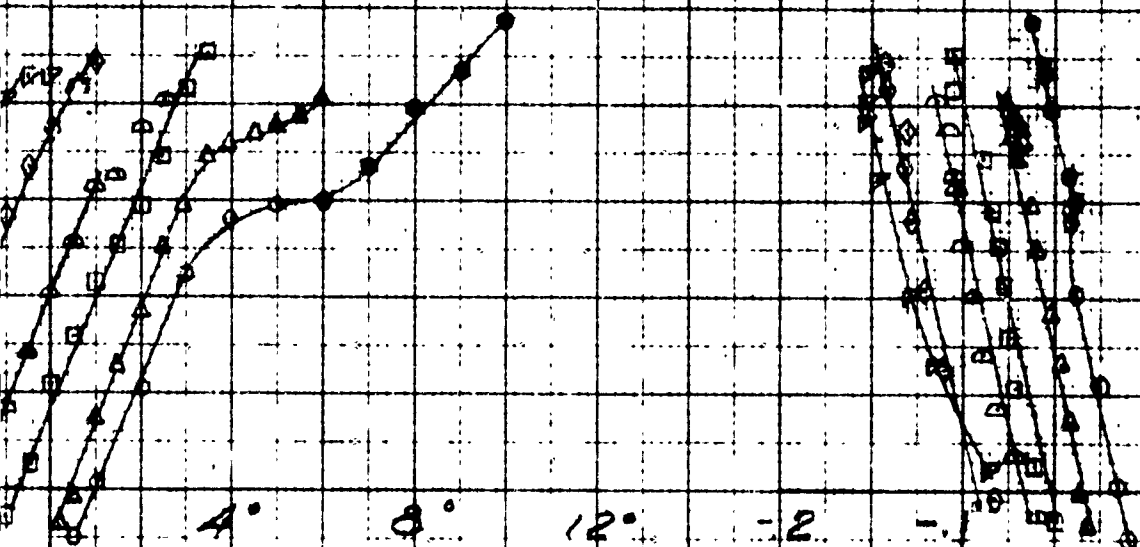
WHIRLING TANK TEST NO 42

BUSHIPS FOIL NO. 13

M.A.C. = 152 FT AREA = 850" $q/c = 1.25$

VEL = 80 KTS

TE:
1. DARKENED SYMBOLS
INDICATE VENTED
FLOW.



ATTACK ~ DEGREES

PITCHING MOMENT
COEFF. ~ $C_{M \frac{1}{4}}$

LIFT COEFF - C_L

DEEG COEFF - C

ANGLE DE
Dessin

SYMBOL	FLAP SET	FROUDE NO.	CAY. NO.
○	0°		
△	4°	4	5
□	8°	8	2
◇	12°	9	1
◊	16°	2	
◈	20°	4	

No 7

4

WHIRLING TANK TEST NO. 43

BUSHIPS FOIL NO. 14

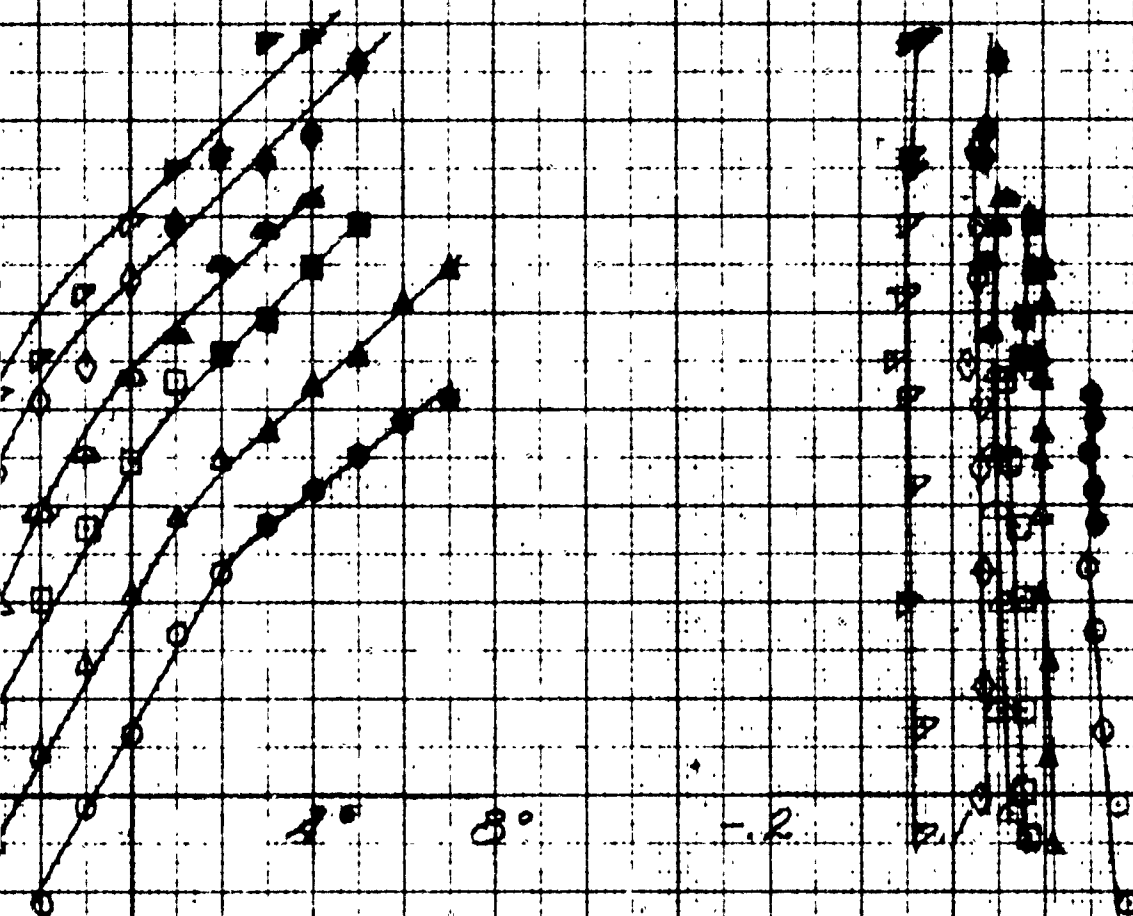
M.A.C. = 1.182 FT AREA = 8.04

$\mu/c = 1.00$

$V = 40$ KTS

NOTE

1. DARKENED SYMBOLS
INDICATE VENTED FLOW.

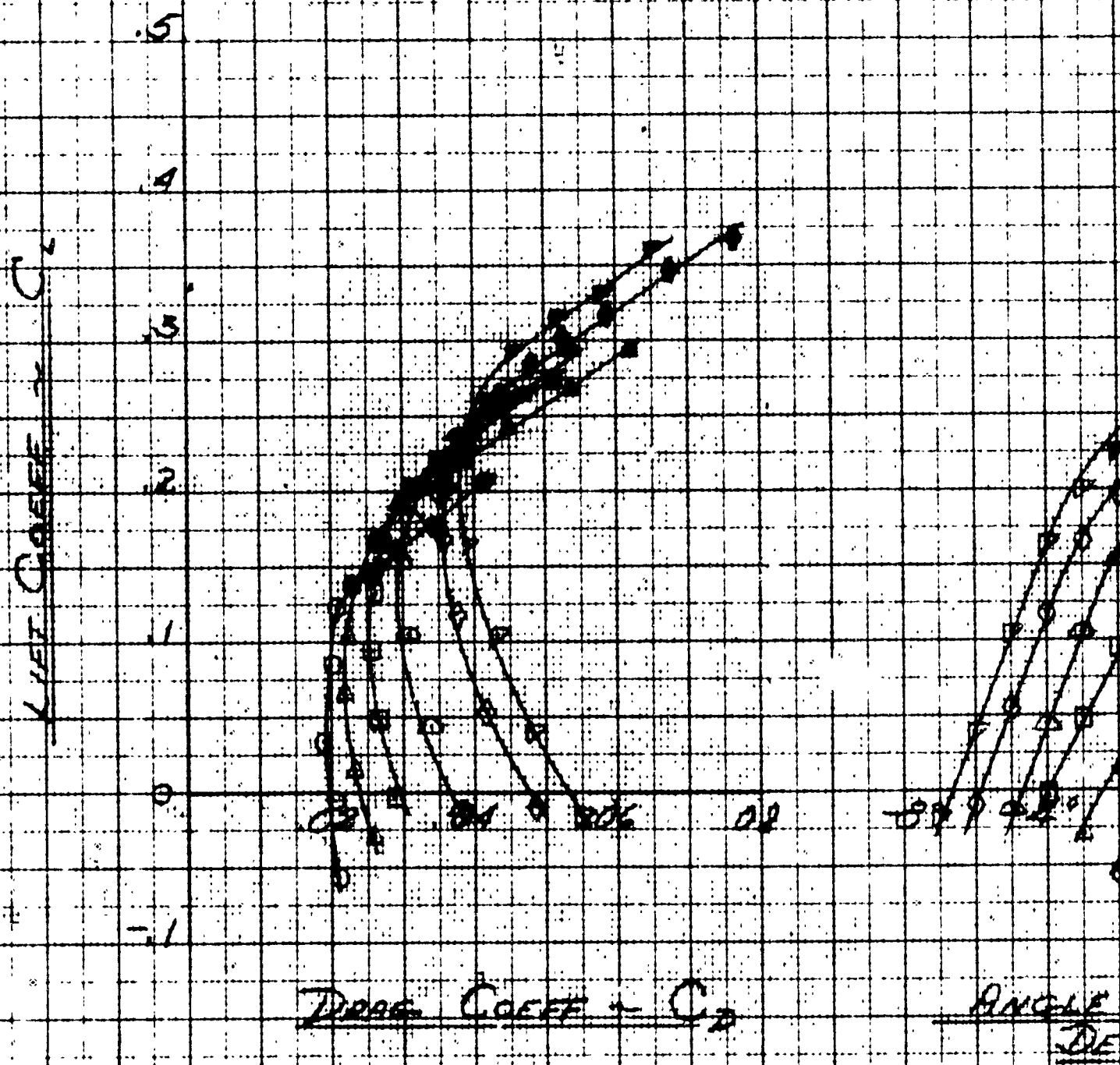


RE ATTACK
VALUES

PITCHING MOMENT
COEFF ~ C_M

SYMBOL	FLAP SET	FROUDE NO.	CAV. NO.
○	0°	4	
△	4°	8	
□	8°	9	
◇	12°	2	3
◇	16°	4	
▽	20°		

NOTE
1. DR
IN



WHIRLING TANK TEST NO 43

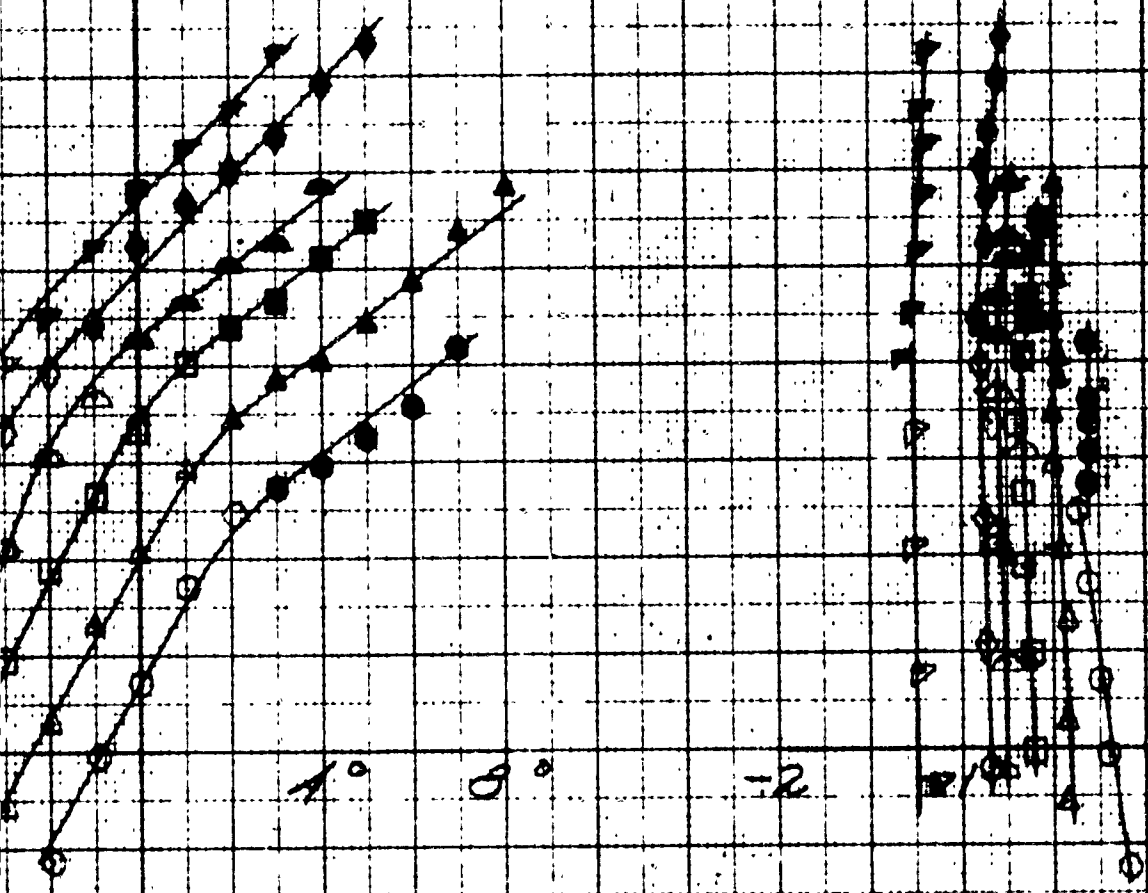
BUSHIPS FOIL NO 1A

MAC = 1.22 FT AREA = 8.7"

$\rho/c = 1.00$

VEL = 45 KTS

DARKENED SYMBOLS
INDICATE VENTED FLOW

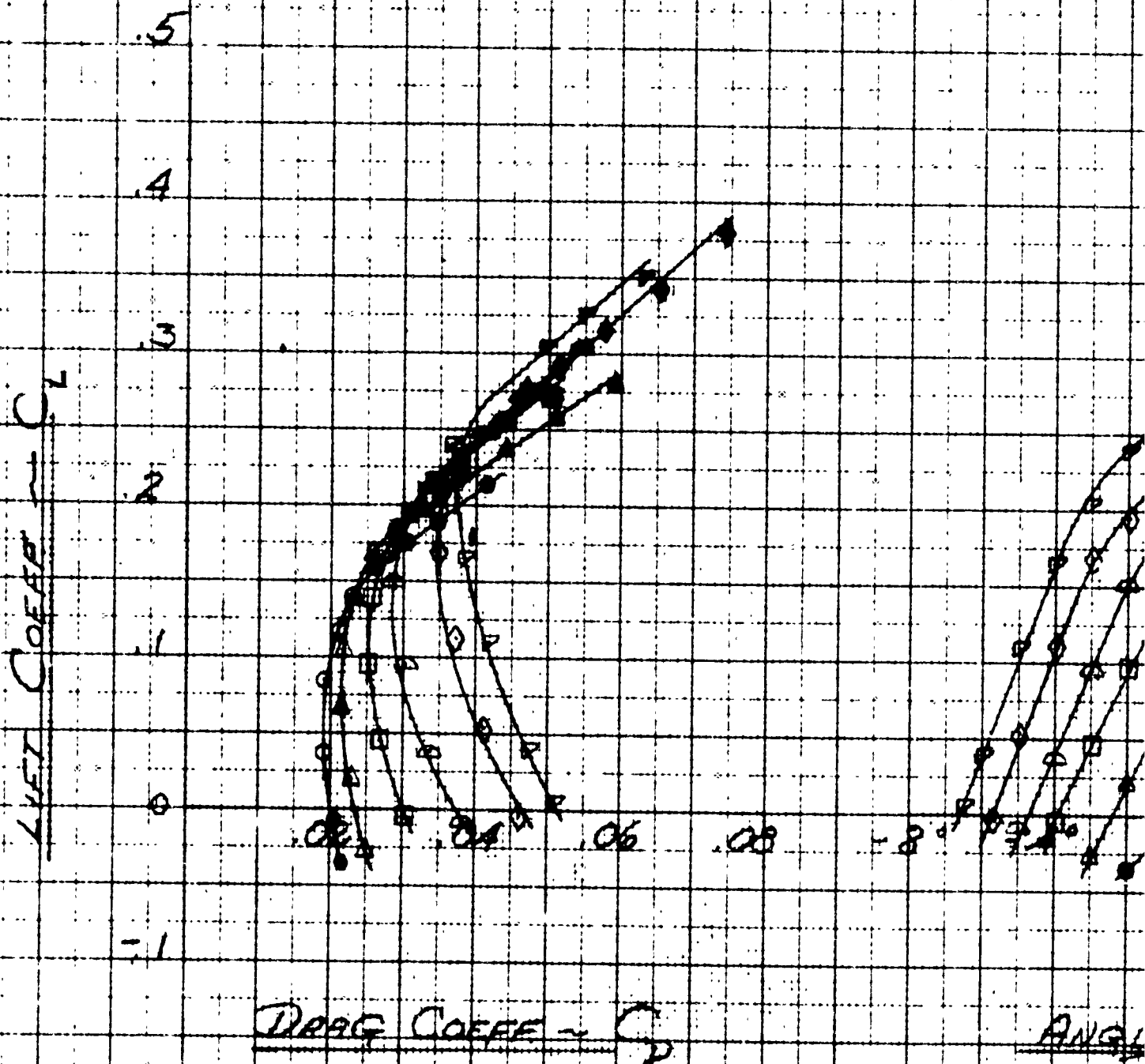


ANGLE OF ATTACK
DEGREES

PITCHING MOMENT
COEFF. ~ C_M

SYMBOL	FLAP SET	FROUDE NO.	CAV. NO.
○	0°		
△	2°	1	3
□	8°	8	6
◐	12°	9	
◑	16°	2	
◇	20°	4	

NOTE:
1. DAA
IND



WHIRLING TANK TEST NO. 43

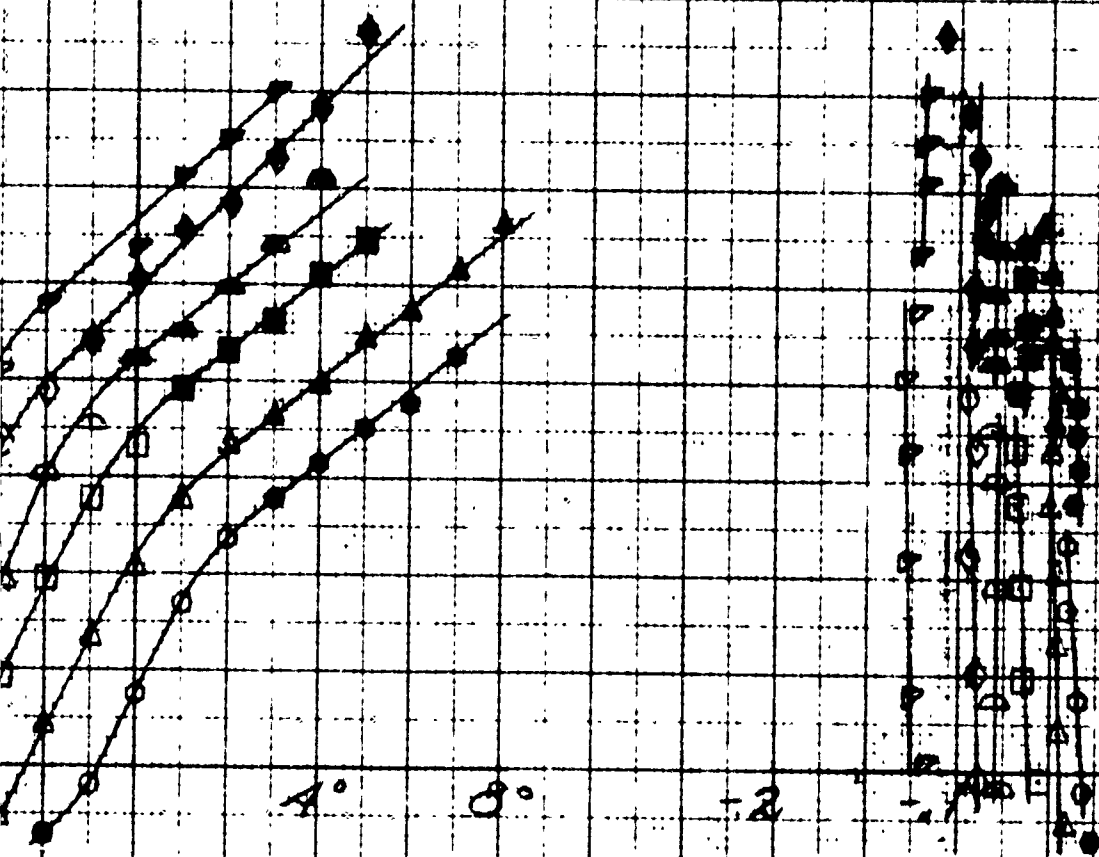
BUSHIPS FOIL NO. 14

M.A.C. = 182 FT. AREA = 8.7"

$\rho/C = 1.00$

$V = 50 \text{ KTS}$

DARKENED SYMBOLS
INDICATE VENTED FLOW



ANGLE OF ATTACK
DEGREES

PITCHING MOMENT
COEFF. - C_m

SYMBOL	FLAP SET	FRONTS NO.	CAV. NO.
○	0°	1	1
□	4°	2	2
△	8°	3	3
◇	12°	4	4
▽	16°	5	5
●	20°	6	6

NOTE:
1. DART KEY
INDICAT

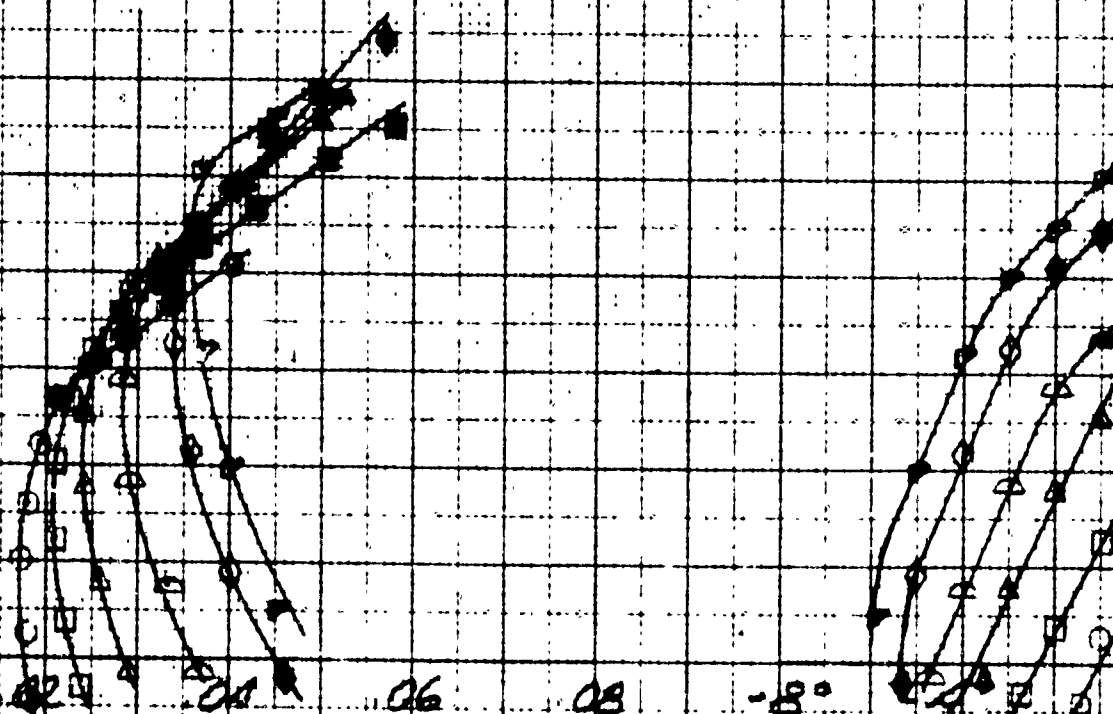
LIFT COEFF ~ C_L

.5
.4
.3
.2
.1
0
-.1

.02 .04 .06 .08 -8°

DRAG COEFF ~ C_D

ANGLE



WHIRLING TANK TEST NO. 43

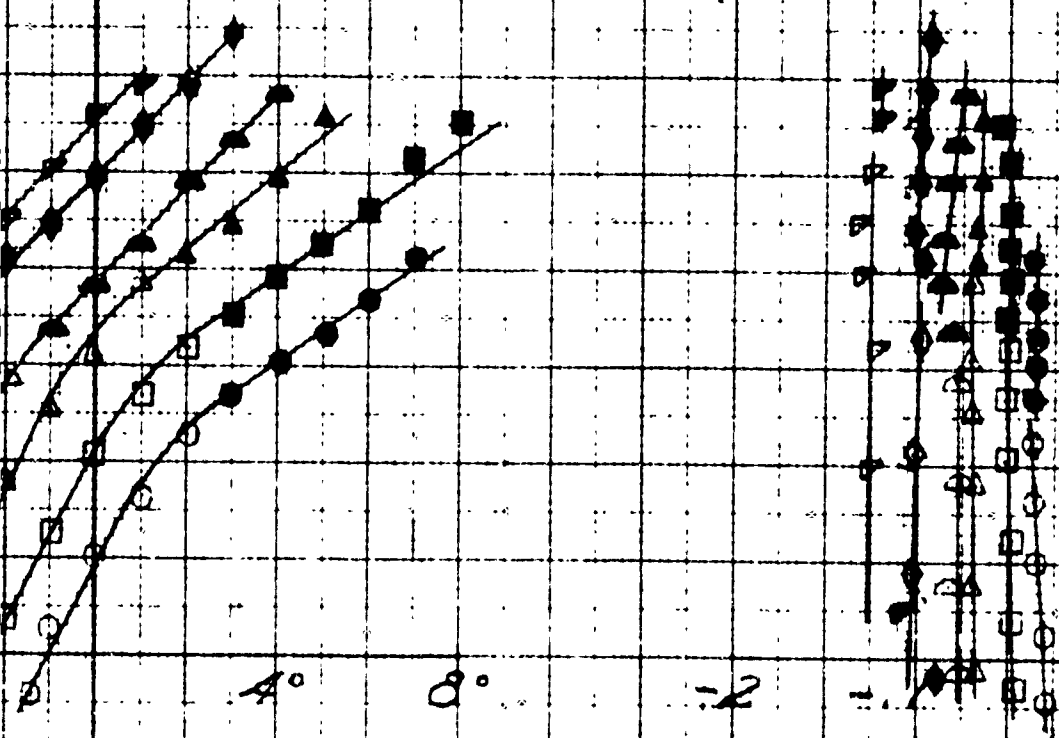
BUSHIPS FOIL NO. 14

M.A.C. = 122 FT AREA = 8.0"

$\gamma_c = 1.00$

$V = 60 KTS$

KEYED SYMBOLS
DATE VENTED FLOW



ANGLE OF ATTACK
DEGREES

PITCHING MOMENT
COEFF - $C_{M_{44}}$

SYMBOL	FLAP SET	FROUDE NO.	CAV. NO.
○	0°		
△	4°	4	2
□	8°	8	2
◇	12°	9	2
○	16°	2	7
◇	20°	4	

NOTE

NO. 11, 12, 13, 14, 15, 16, 17, 18, 19, 20, 21, 22, 23, 24, 25, 26, 27, 28, 29, 30, 31, 32, 33, 34, 35, 36, 37, 38, 39, 40, 41, 42, 43, 44, 45, 46, 47, 48, 49, 50, 51, 52, 53, 54, 55, 56, 57, 58, 59, 60, 61, 62, 63, 64, 65, 66, 67, 68, 69, 70, 71, 72, 73, 74, 75, 76, 77, 78, 79, 80, 81, 82, 83, 84, 85, 86, 87, 88, 89, 90, 91, 92, 93, 94, 95, 96, 97, 98, 99, 100

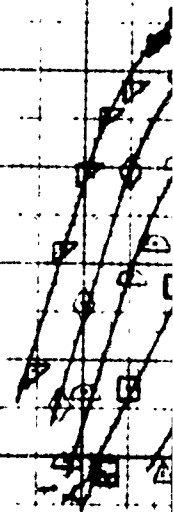
LIFT COEFF ~ C_L

DRAG COEFF ~ C_D

ANGLE OF
DEGREE

5
4
3
2
1
0
-1

.02 .04 .06 .08 -8°



WHIRLING TANK TEST No. 43

BUSHIPS FOIL No. 14

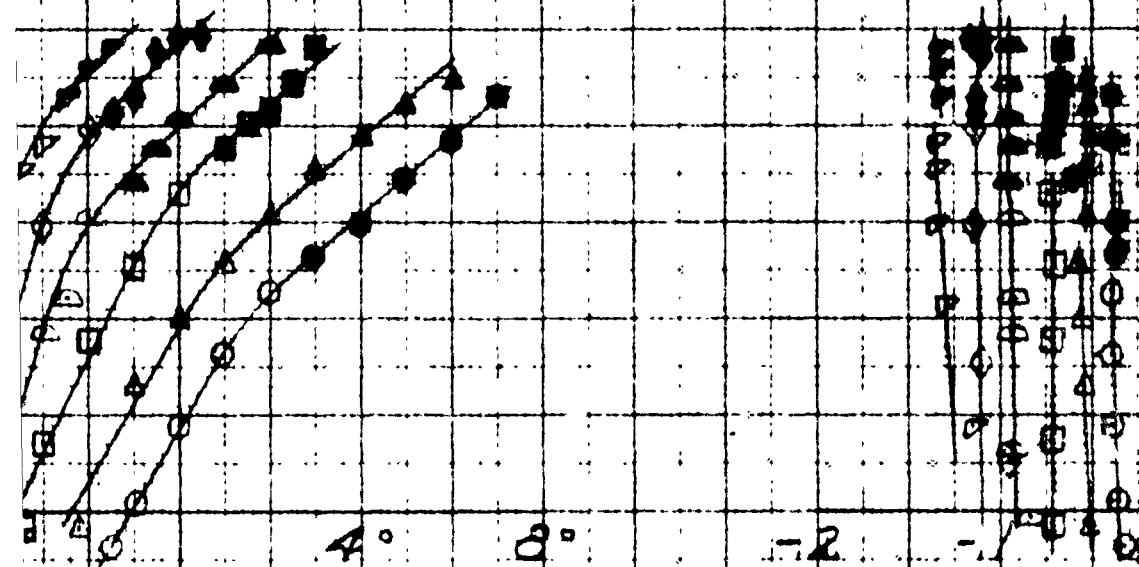
M.A.C. = 1.82 FT AREA = 50"

$$b/c = 1.00$$

$$V = 70 \text{ KTS}$$

NOTE:

1. DARKENED SYMBOLS
INDICATE KENTED FLOW



C_L ATTACK
GREEES

PITCHING MOMENT
COEFF - C_M

SYMBOL	FLAP SET	FROUDE NO.	CAV NO.
○	0°		
△	4°	4.8	.1
□	8°	9	9
◇	12°	2	1
◇	16°	4	
▽	20°		

NOTE:
1. DATA
IND

LIFT COEFF - C_L

.5
.4
.3
.2
.1
0
-.1

.02 .04 .06 .08 8° 4°

DRAW COEFF - C_D

ANGLE OF
DEG

MS 100-1000-1000

WHIRLING TANK TEST NO. 43

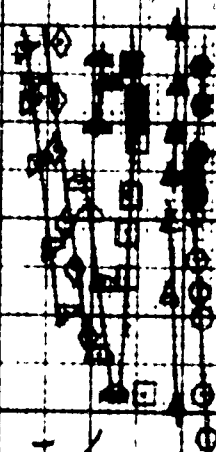
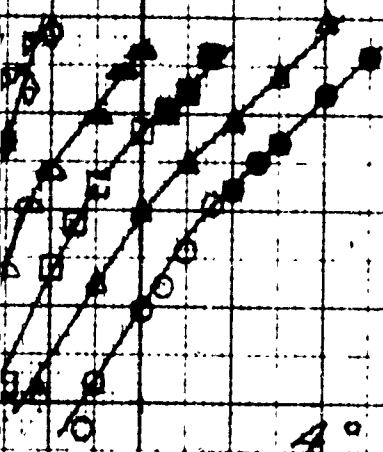
BUSHIPS FOIL NO. 14

M.A.C. = .182 FT AREA = 8.0"

$$\frac{A}{c} = 1.00$$

$$V = 8.0 \text{ KTS}$$

DARKENED SYMBOLS
INDICATE VENTED FLOW.



ANGLE OF ATTACK
DEGREES

PITCHING MOMENT
COEFF - C_m

SYMBOL	FLAP SET	FROUDE NO	CAV. NO
○	0°		
△	4°	5	5
□	8°	3	1
◇	12°	2	
○	16°	2	
▽	20°		

NOTE:
1. DARKENED
INDICATE

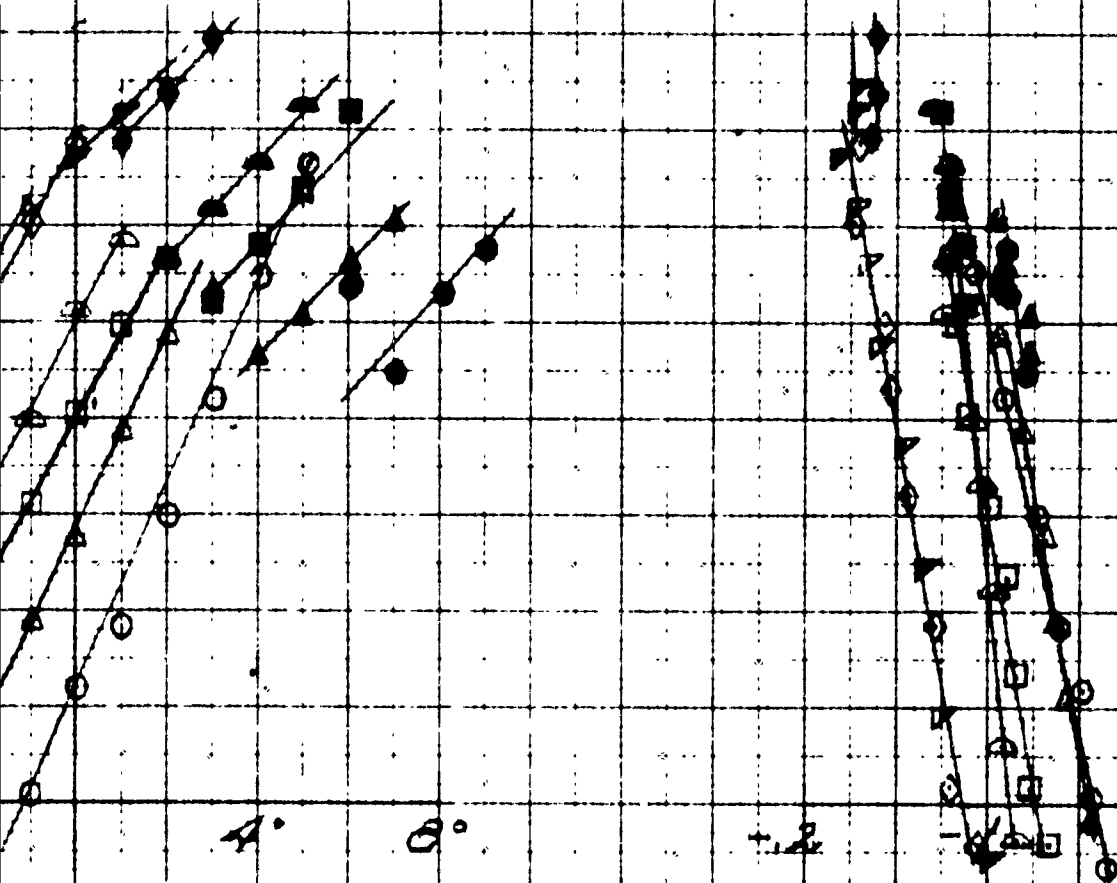


DRAG COEFF - C_D

ANGLE OF
DEGREE

WHIRLING TANK TEST NO. 90
BUSHIPS FOIL NO. 15
M.P.C. = 152 FT. AREA = 8.4"
 $\rho/C = 1.00$ $V = 40$ KTS

OPENED SYMBOLS
ATE VENTED FLOW



ANGLE OF ATTACK
DEGS.

PITCHING MOMENT
COEFF. $\sim C_{M/C}$

WHIRLING TANK TEST NO. 4

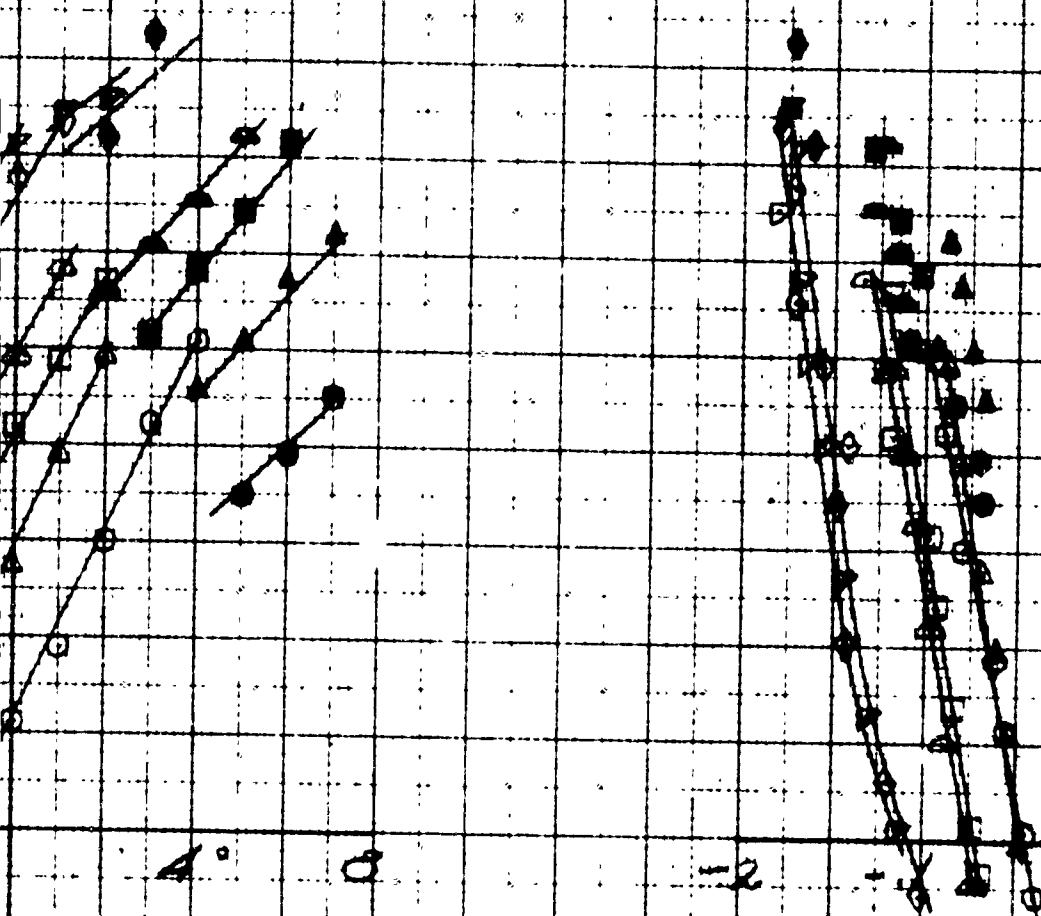
BUSHIPS FOIL NO. 15

M.A.C. = 1.52 FT. $R/R_0 = 0.0$

$\rho/c = 1.00$ $V = 45$ MTS

SYMBOLS

ANTED FLOW.



ATTACK
ANGLES

PITCHING MOMENT
COEFF. ~ $C_{m\alpha}$

SYMBOL	FLAP SET	FROUDE NO.	CAV. NO.
○	0°		
△	4°	5	3
□	8°	3	5
◇	12°	2	5
◊	16°	2	
▽	20°		

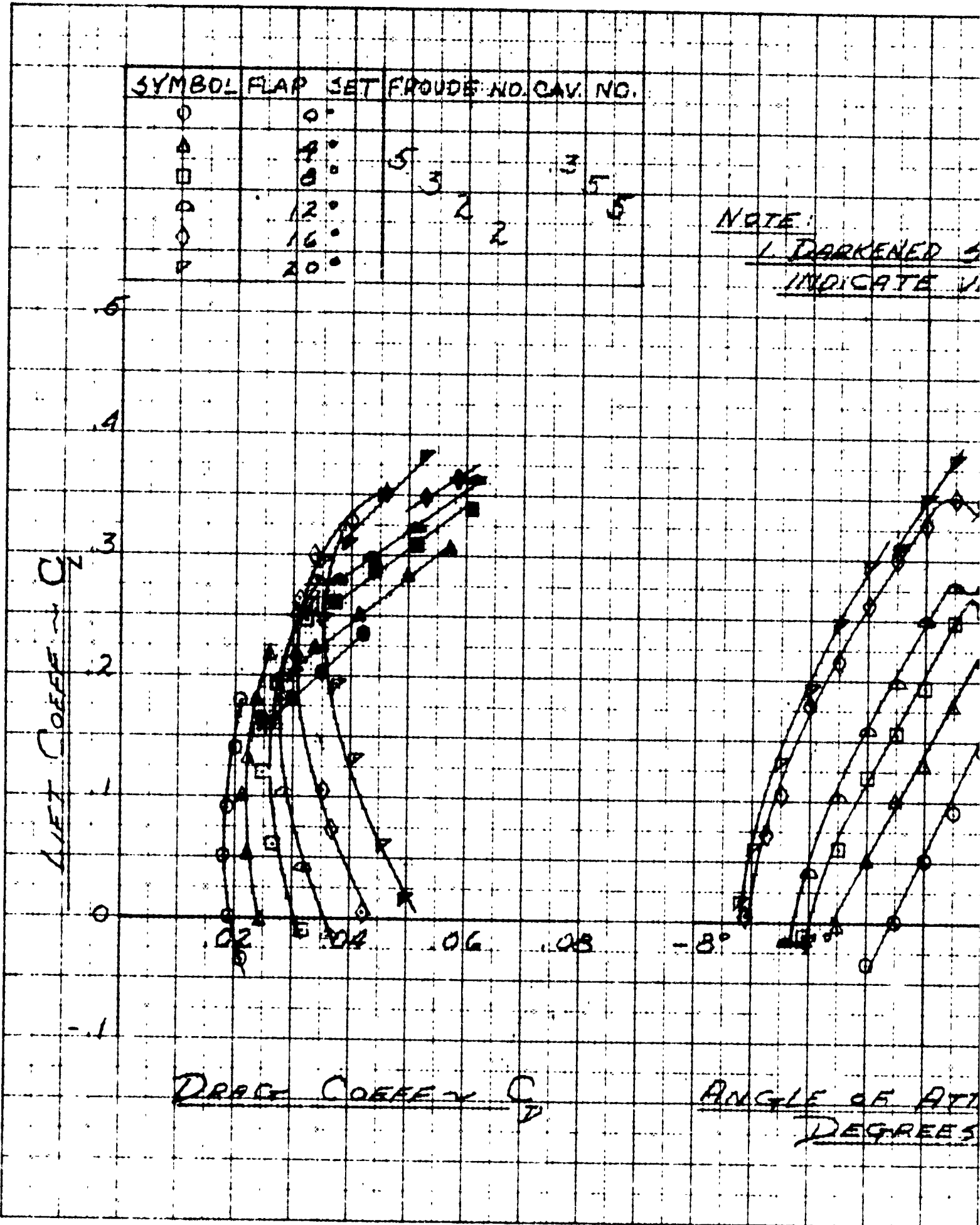
NOTE:

1. DARKENED S
INDICATE V.

LIFT COEFF C_L

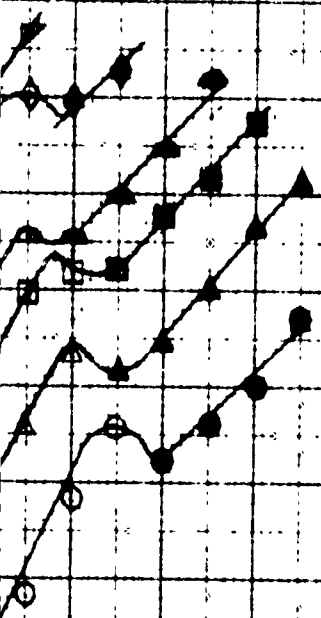
DRAW COEFF C_D

ANGLE OF ATT
DEGREES

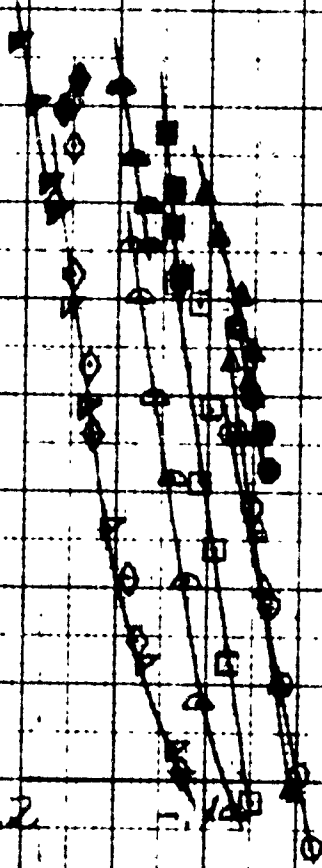


WHIRLING TANK TEST NO. 94
 BUSHIPS FOIL NO 15
 MAC = .152 FT AREA = 8.00 sq in
 $c/c = 1.00$ $V = 50$ KTS

SYMBOLS
 VENTED FLOW



4° 8°



-2 -1.5

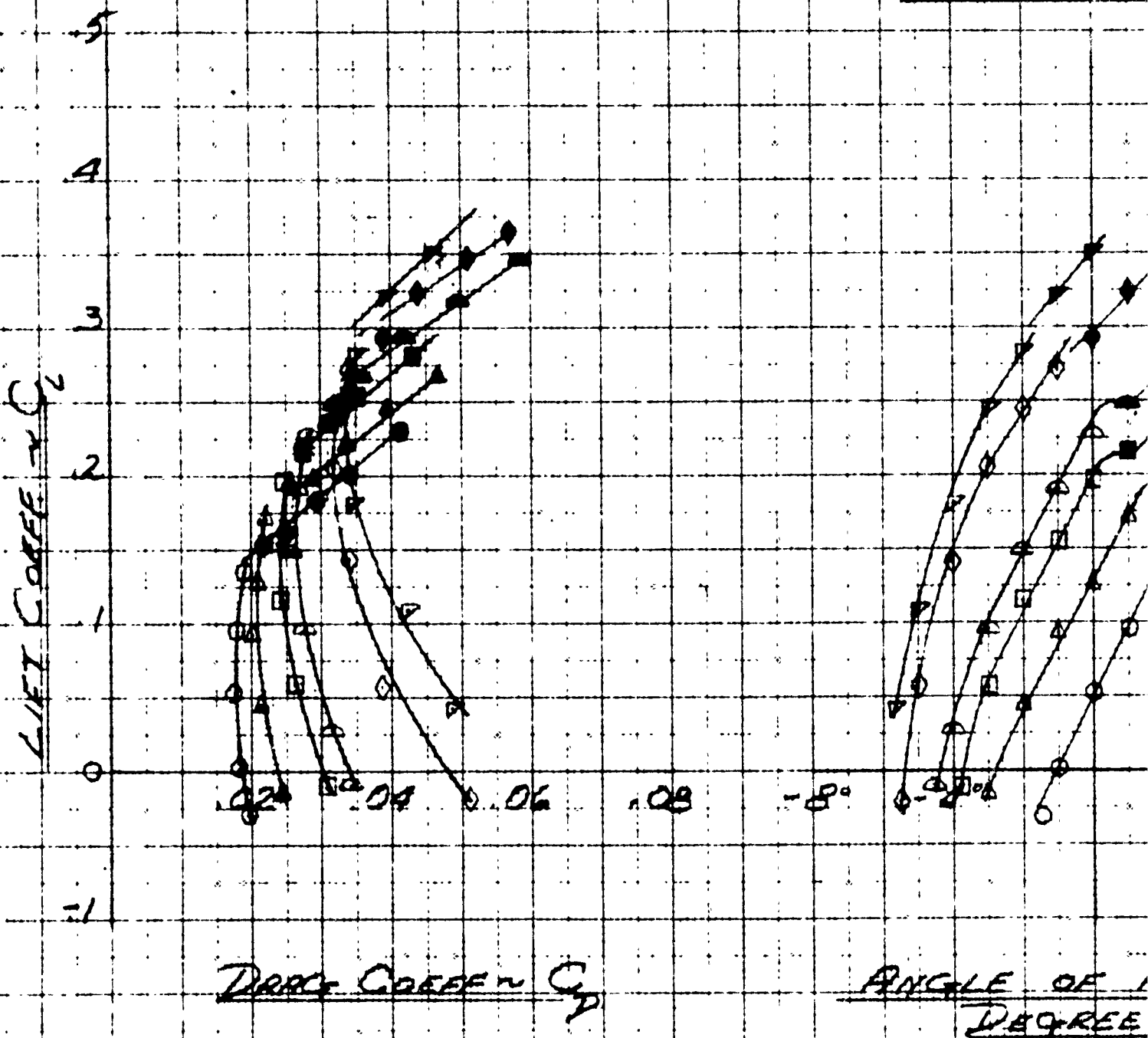
PITCHING
 COEFF.

PITCHING MOMENT
 COEFF. $\sim C_{MCH}$

SYMBOL	FLAP SET	FROUDE NO	CAV. NO.
○	0°		
△	4°	5	2
□	8°	3	6
▽	12°	2	7
◇	16°	2	
⬢	20°		

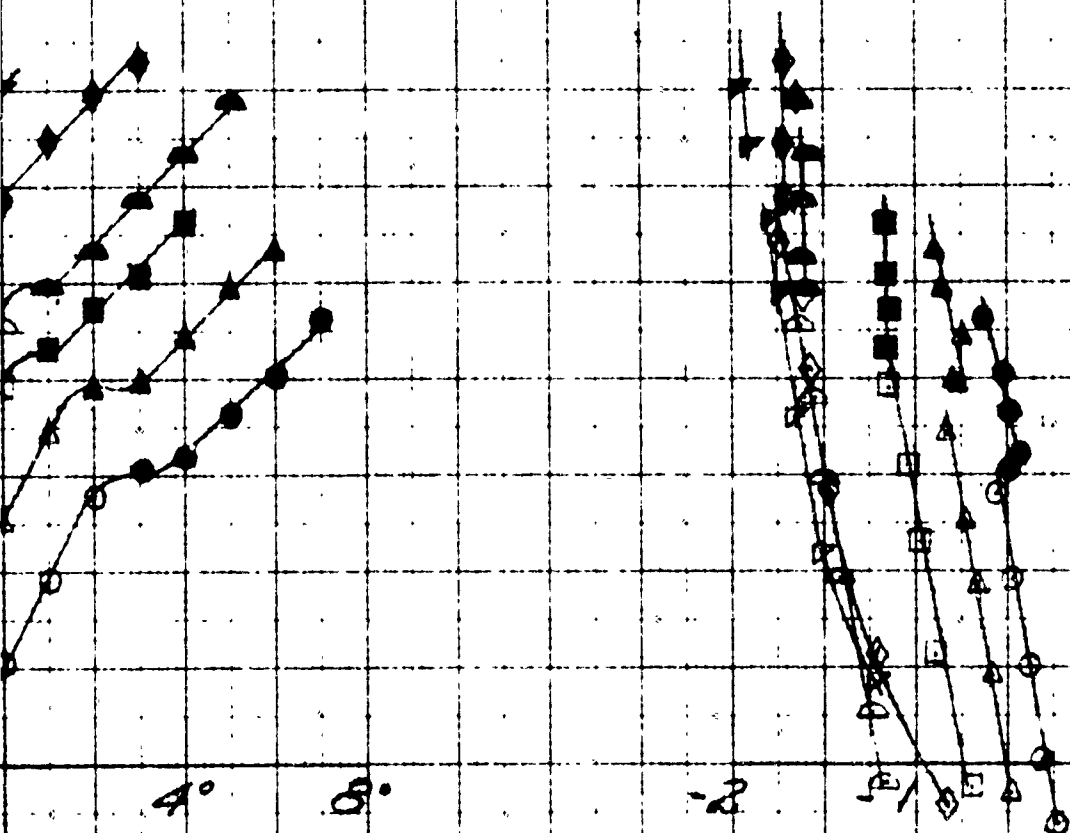
NOTE:

1. DARKENED
INDICATE V



WHIRLING TANK TEST NO. 44
 BUSHIPS FOIL NO. 15
 M.A.C. = .152 FT AREA = 8.9"
 A/C = 1.00 $V = 60$ KTS

D SYMBOLS
 VENTED FLOW



ATTACK
 COEFF

PITCHING MOMENT
 COEFF. $\sim C_{M \frac{1}{2}}$

SYMBOLS	FLAP SET	FROUDE NO	CAK NO
---------	----------	-----------	--------

○	0°	5	2
△	4°	3	
□	8°		
⊙	12°	2	1
◇	16°		
⊕	20°	2	8

NOTE:

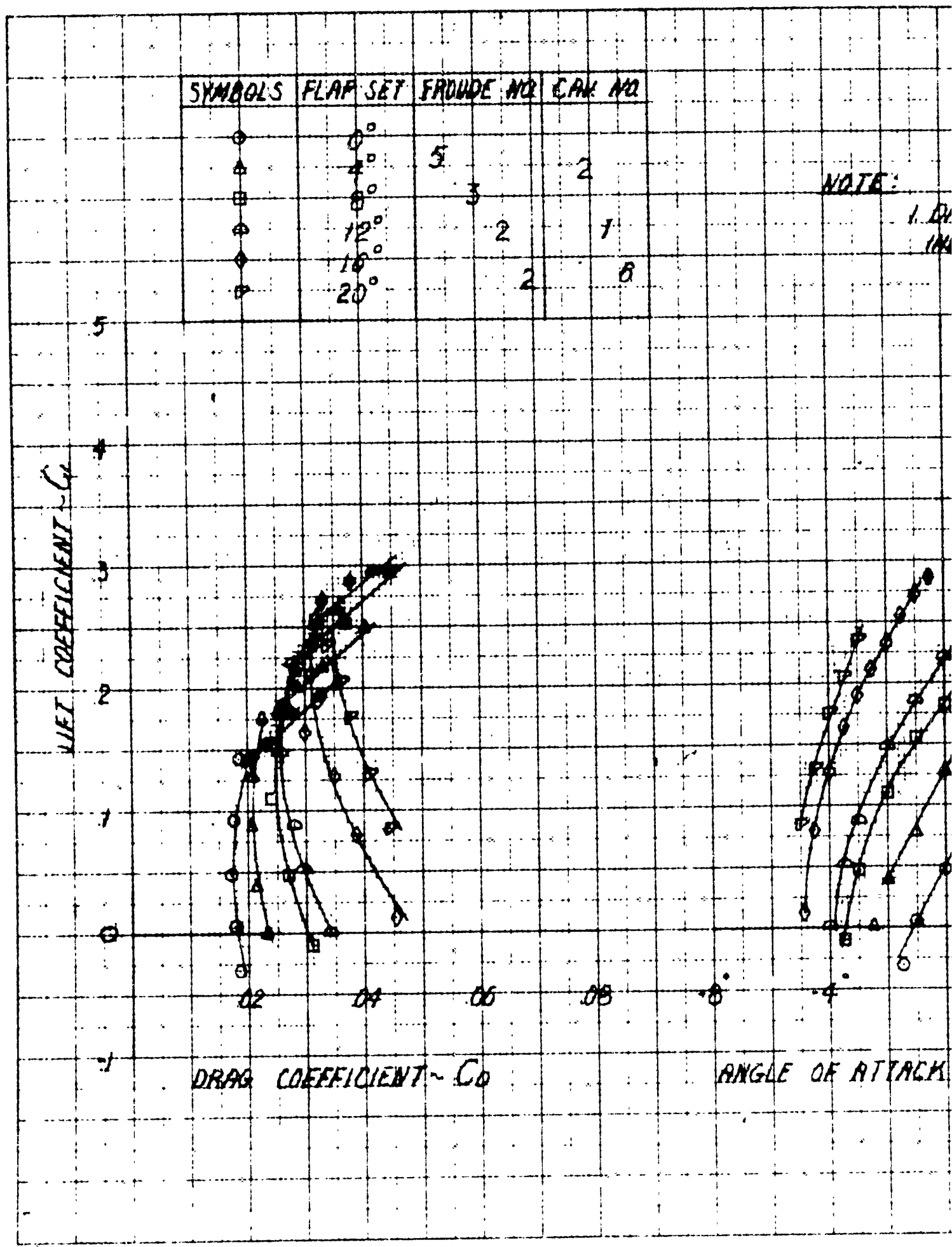
1. DR
144

WEIGHT - 235
LIFT COEFFICIENT - 1.0

LIFT COEFFICIENT - C_L

DRAG COEFFICIENT - C_D

ANGLE OF ATTACK



WHIRLING TANK TEST NO 44

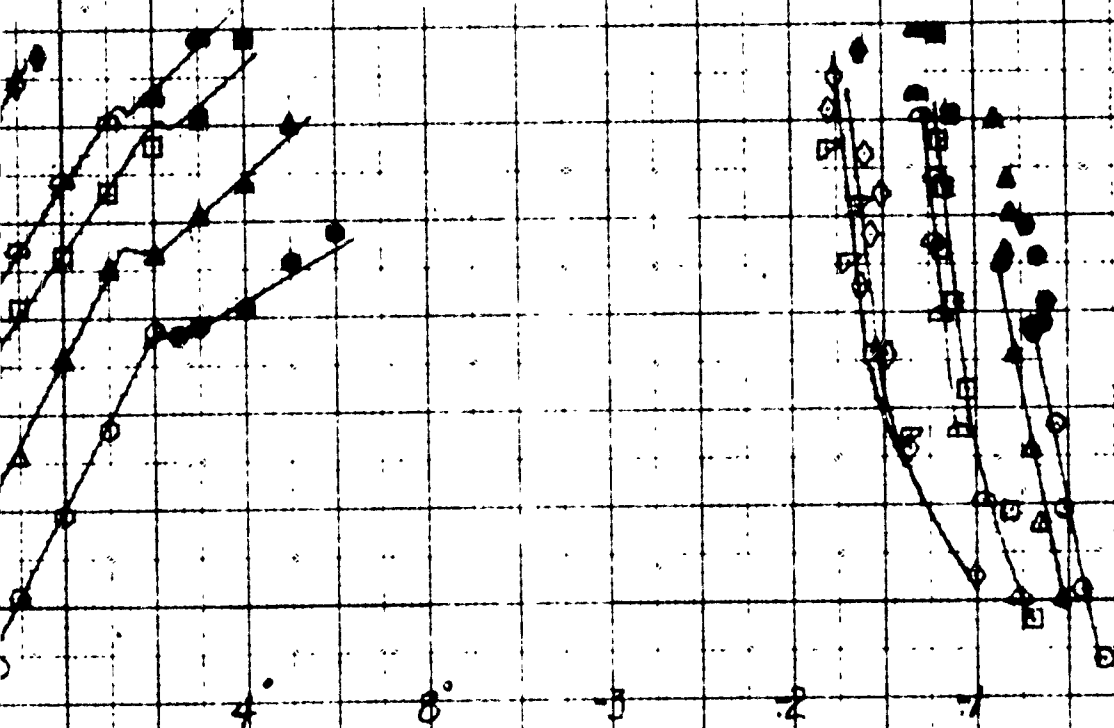
BU SHIPS FOIL NO 15

M.A.C. = 152

S = 800 IN² $d/c = 1.00$

V = 70 KTS.

1. DASHED SYMBOLS
INDICATE REVERSED FLOW.

 α - DEGREESPITCHING MOMENT COEFFICIENT
 C_m

W.E. ALBRECHT, 1907

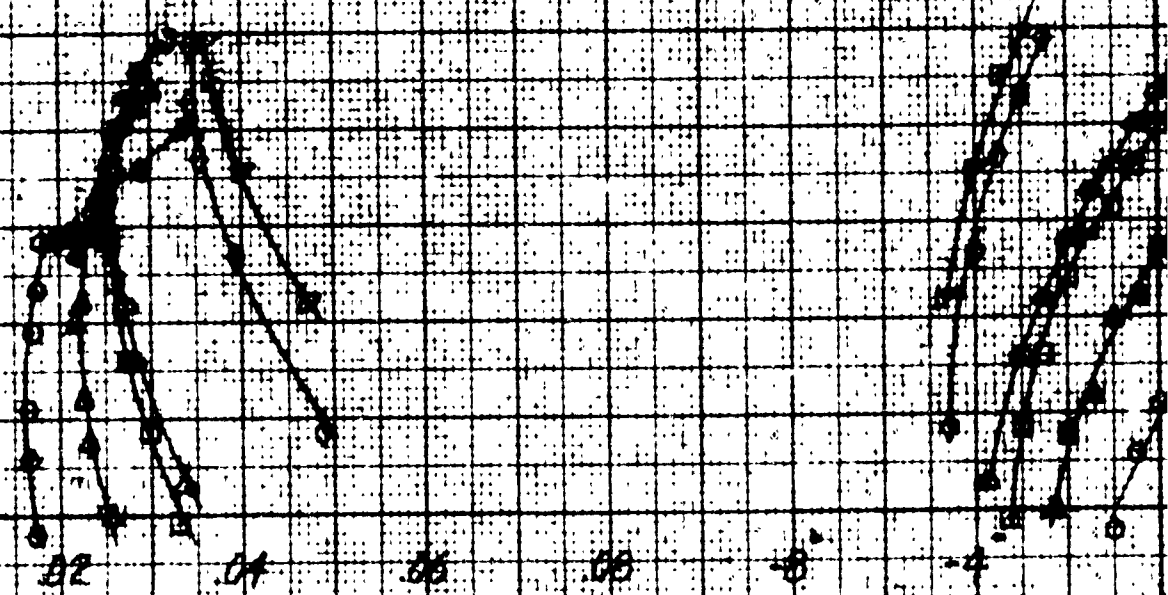
LIFT COEFFICIENT - C_L

SYMBOLS	FLAP SET	FRONT NO.	CHV. NO.
○	0°	5	2
△	4°	3	0
□	8°	2	5
◇	12°	2	5
×	16°	2	5
●	20°	2	5

NOTE:

L.D.A. 14

5
4
3
2
1
0



1

DRAW COEFFICIENT - C_D

ANGLE OF ATTACK

WHIRLING TANM TEST NO. 44

BU SHIPS FOIL NO. 15

MAC = .152 FT $S = 0.00 \text{ in.}^2$ $d/c = 1.00$

V = 80 KTS

DARKENED SYMBOLS
INDICATE REVERSED FLOW.

